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# ACOUSTIC ATTENUATION DESIGN REQUIREMENTS ESTABLISHED THROUGH EPNL PARAMETRIC TRADES

BY  
HENRY F. VELDMAN

1972

THE **BOEING** COMPANY  
WICHITA DIVISION - WICHITA, KANSAS, 67210



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TOPICAL REPORT

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V/STOL AND NOISE DIVISION

## **FOREWORD**

The work described herein was done by The Boeing Company, Wichita Division, under NASA contract NAS 3-14321 with Mr. H. Bloomer, V/STOL and Noise Division, NASA-Lewis Research Center, as Project Manager.

## ABSTRACT

A method is presented to establish acoustic attenuation design requirements through parametric trade studies using the subjective noise unit of effective perceived noise level (EPNL). Results obtained from these studies are then used with an optimization procedure to provide an acoustic lining configuration that is balanced with respect to engine performance losses and lining attenuation characteristics.

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## SUMMARY

An analysis plan was developed for design of acoustically treated nacelles for suppression of fan generated noise from high bypass turbofan engines. The plan was followed in developing two conceptual nacelle designs. The first was a flight nacelle for the NASA/General Electric quiet engine typical of a 707/DC-8 class airplane installation. The quiet engine nacelle design was modified to a design for the NASA Lewis quiet fan. Measured noise data from the NASA quiet fan were used for acoustic design of both conceptual nacelles. This plan included a trade study to select lining attenuation requirements which would provide a 10 EPNdB reduction in flyover noise.

The procedure developed includes a systematic variation of the far field acoustic spectra for each component of the engine noise source and the corresponding change in PNL/EPNL associated with the acoustic spectra variations. Results obtained from these studies are used together with an optimization procedure to minimize the amount of acoustic treatment for a balanced acoustic design that includes engine performance considerations. The design attenuation spectra established in this manner were used to determine the lining characteristics best suited to satisfy the acoustic design objective.

## 1.0 INTRODUCTION

The FAA noise standards set forth in FAR Part 36 have led to the implementation of various designs to reduce the noise being radiated from high bypass turbofan engines. One method of achieving the noise reduction goal is through the design and installation of acoustic liners in the inlet and fan exhaust duct of turbofan engines. This document summarizes one phase of a systematic acoustic lining procedure that has been developed to ensure a reasonable lining design configuration both with respect to acoustic performance and minimum engine performance degradation. This design procedure was developed for and used to establish the attenuation objectives for conceptual nacelle design of the NASA-Lewis QF-2 fan.

The acoustic attenuation design requirements or objectives can be established by a series of PNL/EPNL parametric trade studies. Either test data or estimated values of the engine acoustic signature together with airplane flight path information are used to compute a baseline (untreated) EPNL value for a specific engine configuration. Through the use of a systematic incremental reduction in the various noise components of the engine and computing the associated EPNL value, a matrix of noise reduction values can be established. This matrix will define the various combinations of acoustic attenuations that are required for each noise source to achieve a specific noise reduction goal. A weighting function was used to minimize the total area of acoustic treatment in the engine while meeting the desired EPNL objective. The weighting function related attenuation to the treated area required to obtain that attenuation. By minimizing the sums of these products, the total treated area can be optimized.

The procedure identifies attenuation requirements for inlet and fan duct linings prior to the actual detail design and provides confidence that the acoustic attenuation objective can be obtained. In addition, effects of component noise floors are considered and the effect of the acoustic treatment on engine operational losses are held to a minimum. This procedure allows the engineer flexibility in selecting the attenuated acoustic spectrum. This capability is desirable to respond to geometric and/or manufacturing constraints.

## 2.0 TECHNICAL APPROACH

The basic elements used to provide an acoustic lining configuration that is balanced with respect to engine performance losses and lining attenuation characteristics are outlined in Table I. The steps identified in the flow chart are used to perform a systematic variation of the far field acoustic signature. Each cycle provides one value in the EPNL matrix. Details of the individual analysis elements are outlined in the following paragraphs.

The analysis begins with the identification of the baseline spectra for each component of the noise source (Step 1). In this document, the total noise source is assumed to be a composite of the: (1) fan noise, (2) primary jet noise, (3) secondary jet noise, and (4) turbomachinery noise.

A reduction in Sound Pressure Level (SPL) of an acoustic spectrum is usually referred to as applying attenuation to a baseline component spectra (Step 2). Each component attenuation spectra should be representative of the expected reduction derived from either tests or analysis. The total attenuated noise spectrum of the engine is the power sum of each of the individual attenuated noise components.

The significance of a balanced acoustic design for a particular noise source is closely linked to the "noise floor" concept. Consider two noise components, the fan and jet noise of a turbofan engine. If, at a given frequency, the fan noise has been attenuated to a level below that of the jet noise, then subsequent reduction in the fan noise level will not result in a discernable lower overall noise level. In this case, the noise floor has been established by the jet noise level.

The next step in evaluating the acoustic performance is to extrapolate the attenuated spectra to the requested condition (Step 3). The form at which the input data is available may consist of a set of 1/3 octave data that is either measured or predicted on a polar radius around the noise source or at a sideline distance. The output can be calculated at any one of the above conditions or flight conditions simulating approach, takeoff, or level fly by.

The airplane flight characteristics and atmospheric environment are used to determine the projected SPL spectra by accounting for factors such as the propagation time and distance, the atmospheric attenuation corrections, directivity index and doppler shift.

The atmospheric attenuation corrections are the standard atmospheric absorption as a function of distance, relative humidity, and temperature (Reference 1) and the extra ground attenuation as a function of elevation angle and distance (Reference 2).

The propagation distance is accounted for by the spherical divergence which takes the form of:

$$\Delta \text{dB} = 20 \log_{10} (\text{Distance ratio between input and output condition})$$

The corrections due to the doppler shift effect are taken from Reference 3.

Differences in the number of engines between the input and projected condition are corrected by using the relation

$$\Delta \text{dB} = 10 \log_{10} (\text{number of engines})$$

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The relative jet velocity correction due to the forward airplane speed, outlined in Reference 4, defines a frequency shift as

$$\text{Frequency shift} = \frac{\text{Primary jet velocity} - \text{forward airplane speed}}{\text{Primary jet velocity under static condition}}$$

The change in the sound pressure level (SPL) due to the relative jet velocity is expressed as

$$\Delta \text{dB} = 80 \log_{10} (\text{Frequency shift})$$

The conversion from projected SPL values to subjective units is performed in Step 4. Each 1/3 octave SPL value is converted to the respective NOY unit. The details of this procedure can be found in Reference 5.

The Perceived Noise Level (PNL) is calculated at each angle by using the relation (Reference 6).

$$\text{PNL} = 40 + 33.3 \log_{10} \text{NOY}_{\max} + .15 (\sum \text{NOY} - \text{NOY}_{\max})$$

When spectral irregularities are present, the PNL value is adjusted to account for a tone correction by applying the technique set forth in Reference 7.

The Effective Perceived Noise Level (EPNL) is calculated by integrating the time history of the tone corrected PNL as outlined in Reference 7.

### 3.0 ACOUSTIC TRADE STUDY

The noise abatement system for this program considered the placement of lining material in the inlet and the fan duct. In the analyses, the total fan noise spectrum (Table 2) is treated as being a composite of two independent noise sources. The inlet radiated noise controls the forward quadrant (Table 3), the fan duct radiated noise dominates the aft quadrant (Table 4).

The spectral Power Insertion Loss (PIL) characteristics of tuned single layer acoustic linings tend to have a bell shaped curve about the tuned frequency. The PIL curve shape used for this acoustic trade study, generated from flow duct data, is shown in Figure 1. Segmented acoustic lining configurations were considered for the inlet and fan duct. In one segment, the attenuation was tuned to the fan fundamental frequency and the second segment was tuned to the second fan harmonic. For convenience, the acoustic lining segments in the inlet are referred to as inlet fundamental and inlet harmonic while the lining segments in the fan duct are referred to as fan fundamental and fan harmonic. When the PIL curves of the fundamental and harmonic overlap each other, algebraic addition was used to obtain the resultant PIL.

The attenuated forward and aft fan noise spectrum was combined with jet noise and turbomachinery noise components to form a composite polar spectrum. This spectrum and the appropriate flight parameters were used to obtain an EPNL value.

The EPNL trade studies were conducted for typical takeoff and approach conditions with the observer located at 3.5 n.m. from brake release and 1.0 n.m. from touchdown respectively, as defined in FAR Part 36 (Reference 7). The takeoff and approach were simulated using the flight conditions listed in Table 5. If the value of PNLT at the 10 dB down point is 90 PNdB or less, Reference 7 allows the option of using a time interval between the initial and the final time value for which PNLT equals 90 PNdB. This option was not used because it can give rise to discontinuities when the maximum PNLT value approaches a value equal to or less than 90 PNdB.

The EPNL trade study calculations were accomplished by varying the four lining segment characteristics: (1) inlet fundamental, (2) inlet harmonic, (3) fan fundamental, and (4) fan harmonic, independently at successive values of 0, 5, 10, 20, and 30 dB for the peak value of attenuation. In addition, for the takeoff simulation, primary jet noise attenuations were simulated by applying a uniform attenuation factor to the jet SPL spectrum in increments of 0, 2.5, 5, 7.5, and 10 dB. The approach condition was evaluated with and without the turbomachinery noise component. Table 6 identifies the 8 EPNL trades presented in this document. Runs 1A and 2A represent the baseline engine as simulated for approach and takeoff conditions. Runs 10A and 11A have the primary jet and turbine noise components removed which would be representative of the noise levels measured on a fan test configuration.

The data was plotted in terms of  $\Delta$  EPNL versus inlet fundamental attenuation for constant fan fundamental attenuations. These data reflect combinations of inlet and fan harmonic acoustic treatment. For improved visibility, a cross plotting of the data was accomplished for constant  $\Delta$  EPNL levels of 5 and 10 EPNdB. Each graph depicts the variation of inlet fundamental attenuation versus fan fundamental attenuation for different values of inlet harmonic attenuations (Figures 2 and 3). In each case, the fan harmonic attenuation has a constant preassigned value. Whenever a graph appears blank, this merely indicates that the  $\Delta$  EPNL level cannot be achieved for the attenuation conditions stated. Asterisks appearing within the tabulated data sets indicate that

the attenuation value is outside the range of 0 to 30 dB. The “with tone” notation in the graph heading indicates that the standard tone correction was used as identified in FAR Part 36.

#### 4.0 SELECTION OF DESIGN ATTENUATION SPECTRA

The EPNL trade study provides a data bank of  $\Delta$  EPNL values for combinations of inlet and fan duct attenuation levels tuned to the fundamental and harmonic blade passage frequency of the fan. In view of the large amount of data that can be generated to evaluate various combinations of attenuation levels to satisfy specific design goals, a systematic method was developed to evaluate the EPNL trade study results. This method uses weighting functions that are applied to the individual attenuation levels, such that the summation of these functions are proportional to the total acoustic treatment area.

A four-dimensional interpolation using the  $\Delta$  EPNL data bank was performed to identify each attenuation spectra which gave a 10 EPNdB reduction at takeoff and approach. The results of the interpolation showed that a great number of different spectra satisfied the acoustic design criteria. The selection of the optimum spectra, based on minimum treated area was accomplished by using a lining optimization procedure described in Reference 8, which is based on maximizing the attenuation of a "pink" noise source. This source has a constant SPL for one octave bandwidth and is zero elsewhere. Figures 4 and 5 show results of this optimization procedure for inlet and exhaust Mach numbers of .4 as a function of  $Hf/c$ , where  $H$  is the duct height between the lined walls,  $f$  is the frequency, and  $c$  is the speed of sound. The results presented are attenuation level for a one octave attenuation bandwidth versus the ratio of the lining length,  $L$ , to duct height,  $H$ . The ratio of duct length to height is proportional to the ratio of treated area to flow area in an annular duct with all surfaces lined.

The curves of Figures 4 and 5 were approximated by an equation of the following form:

$$\phi \left( \frac{\text{Treated Area}}{\text{Flow Area}} \right) = C \Delta \text{dB}^{5/3}$$

Figure 6 compares the theoretical and empirical curves for  $Hf/c = 1.0$  when the inlet and exhaust Mach numbers are .4. The constant  $C$  depends on the value of  $Hf/c$  and the flow condition in the duct (i.e., Mach number and direction of flow relative to the noise source). The value of the constant increases with  $Hf/c$ . For example, at an exhaust Mach number of .4, the constant increases by a factor of 1.75 as  $Hf/c$  is doubled. At an inlet Mach number of .4, the constant increases a factor of 1.6 as  $Hf/c$  is doubled. The following table compares relative values of the proportionality constant as a function of  $Hf/c$  and duct flow condition.

<u><math>Hf/c</math></u>	<u>Inlet Mach No. = 0.4</u>	<u>Exhaust Mach No. = 0.4</u>
1	0.455	1.0*
2	0.73	1.75

\*Reference Value

The relative values of the constants listed above are proportional to the theoretical treated area required to obtain a given level of suppression, assuming equivalent flow areas in the inlet and exhaust ducts. The results for  $Hf/c = 1$  show the increased efficiency of inlet linings compared to fan duct linings, which is theoretically predicted assuming no velocity gradients in the ducts. This

magnitude of increased efficiency had not been observed in flow duct tests of linings conducted prior to this study. Consequently, the effectiveness of inlet and fan duct linings were assumed equal for this study. Subsequently, it was discovered that if inlet linings are designed considering velocity gradients, the lining effectiveness approaches the theoretical levels, (Reference 9).

The weighting function used to minimize treated area while achieving a given level of suppression was

$$\phi \left( \frac{\text{Treated Area}}{\text{Flow Area}} \right) = \left( \Delta \text{dB}_{\text{fundamental}}^{5/3} + 1.6 \Delta \text{dB}_{\text{harmonic}}^{5/3} \right)_{\text{inlet}} \\ + \left( \Delta \text{dB}_{\text{fundamental}}^{5/3} + 1.75 \Delta \text{dB}_{\text{harmonic}}^{5/3} \right)_{\text{fan duct.}}$$

The optimum lining configuration determined by the  $\phi$  function satisfied the design objective while minimizing the amount of treated area in the inlet and fan duct. Figure 7 shows the EPNL reduction versus the optimum value of the weighting function  $\phi$ . Each point on the curve represents the inlet and fan duct attenuation spectra for minimum treated area. The results show that the 10 EPNdB reduction goal can be achieved at the approach condition. However, the goal cannot be achieved at takeoff without core jet suppression since the noise floor level is established by the primary jet.

The attenuation requirements for the approach condition were selected as those which gave a 10 EPNdB reduction with the minimum value of the weighting function,  $\phi$ . Figures 8 and 9 show the attenuation requirements at approach for the inlet and fan duct, respectively. Takeoff attenuation requirements were reduced to 7.5 EPNdB due to the presence of the noise floor controlled by the primary jet. The requirements, based on minimum  $\phi$ , are shown in Figures 10 and 11 for the inlet and fan duct, respectively.

## 5.0 RESULTS

The airplane flyover noise prediction computer program (EPNL program) was developed to accept individual noise component spectra. Each spectra defining the 1/3 octave SPL variation with frequency and position. The acoustic characteristic of a turbofan engine was defined by four noise components.

- a. Noise due to the low pressure compressor rotation, usually referred to as "Fan Noise" or simply "Fan."
- b. Primary exhaust jet noise; Jet prim or Jet 1
- c. Secondary exhaust jet noise; Jet sec or Jet 2
- d. Turbo machinery noise; Turb

The array defining the turbomachinery noise includes all other internally generated noise.

The computer program internally separates the fan into forward and aft radiated noise components (fan fwd and fan aft). Equal power distribution is assumed at 90° with a rapidly decreasing influence of the forward fan component in the aft quadrant and vice versa.

At this stage all the noise components are stored individually. Changes in the spectra due to noise suppression techniques or due to different flight characteristics can be accounted for on a per component basis. After all components have been adjusted to match the simulated configuration they are added together for flyover simulation calculations.

The acoustic data for the QF-2 fan was made available in the form of the combined fan and secondary jet spectra at a polar distance of 100 feet. The primary jet noise spectra was obtained at a 250 feet polar distance. Turbomachinery noise was only defined in a frequency region from 1,250 to 10,000 Hz and for angles from 100° to 130°. The above information and the data source are identified in Table 7. The 1/3 octave spectra are listed in Tables 8 through 10.

All the data was subsequently standardized to a 200 feet polar arc distance at 59°F and 70% relative humidity. Furthermore, a polar spectrum defining the secondary jet noise was generated to accomplish the separation of the QF-2 fan and secondary jet noise. The modified data and the type is identified in Table 11. The 1/3 octave spectra are listed in Tables 12 through 15.

The EPNL computer program was modified to include the option allowing either fan or combined fan and secondary jet noise as basic component input.

A sample of the EPNL program output is shown in Tables 16 through 21 for a typical approach condition. The details and explanation of the input code and data format is described in Appendix I (EPNL program description). Table 17 lists the input spectra of the QF-2 fan and secondary jet noise (Fan Jet 2), the primary jet noise (Jet Prim), and the secondary jet noise (Jet Sec). The fan noise spectrum, which is calculated by subtracting the secondary jet noise from the QF-2 noise data, is tabulated in Table 18 (Fan Base). The total jet noise (Jet Tot) is the sum of the primary and secondary jet noise. The turbomachinery noise (Turb Base) was extrapolated to include all angles (10° through 170°) and all frequencies (50 to 10,000 Hz) with an arbitrary lower noise limit of 10 dB.

A sample of the trade study output is listed in Table 19. Each line represents the results of a particular combination of inlet and/or fan duct attenuation. The notation at the head of Table 19 is as follows:

Two columns each under "inlet" and "fan duct" indicate the fundamental and harmonic peak attenuation value in dB. The number under "jet" is an identifier applicable to the jet noise used. The next column contains the EPNL value. Next listed are airplane altitude at overhead, airplane velocity, relative humidity, and temperature. The sideline and offset values are relative to the FAR Part 36 measuring stations. The climb gradient is listed as "Gamma", while "Alpha" is the airplane angle of attack relative to the flight path.

The "duration" is the length of time of the flyover which is used in the EPNL calculation. The "DEPNL" value is the change in EPNdB relative to the condition for no attenuation in the inlet or the fan duct.

Appendix 7 contains the reduced data from the EPNL and PHI computer programs. The results presented are for a 5 and 10 EPNdB noise reduction. The run identification numbers are listed in Table 6.

## **6.0 APPLICATIONS AND LIMITATIONS**

The technique outlined in this document can be applied to any turbofan engine installation whether mounted on a test stand or on an airplane. A sufficiently detailed noise source description on a narrow band or 1/3 octave band basis is desirable to conduct comprehensive EPNL trade studies. If the SPL spectra is only available at a few angles and/or in octave band form, an EPNL trade study should be limited to establishing preliminary design attenuation objectives.

The basic limitations in the computer program are the following:

1. A single airplane configuration can be analyzed during a flyover. This implies a constant power setting, climb gradient, angle of attack, and airplane velocity. Although this may seem like a severe limitation, in actual flight the above parameters do not change very drastically except in a cutback procedure during takeoff.
2. The attenuation is applied in equal amounts at all angles. Angular dependency is neglected because sufficient data were not available to justify the use of a directivity correction. However, inclusion of directivity patterns can easily be incorporated.

## **7.0 CONCLUSIONS AND RECOMMENDATIONS**

A trade study based on EPNL level together with a program to minimize the acoustic treatment required to achieve the attenuation goal is an economic and effective method to estimate lining attenuation requirements for a balanced design with respect to engine and acoustic performance. In addition, the portion of the acoustic spectra where the attenuation goal can be achieved are identified without spending an excessive amount of computer time and/or engineering manhours.

A common method of reducing aircraft noise is through the use of acoustic lining materials and different nozzle configurations; however, there are other techniques which may satisfy a particular requirement. The cutback procedure was introduced to lower the noise level in populated areas around the flight path. A proposed two-segmented approach profile can be examined for the same reason. Changes in airplane configuration, such as different flap settings, have also been suggested as possible candidates for an improved acoustic flyover performance. In view of these considerations, a desirable extension of the present EPNL program would be a modification to account for changes in power setting, engine spool down time, climb gradient, angle of attack, and airplane velocity.

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## **FIGURES**

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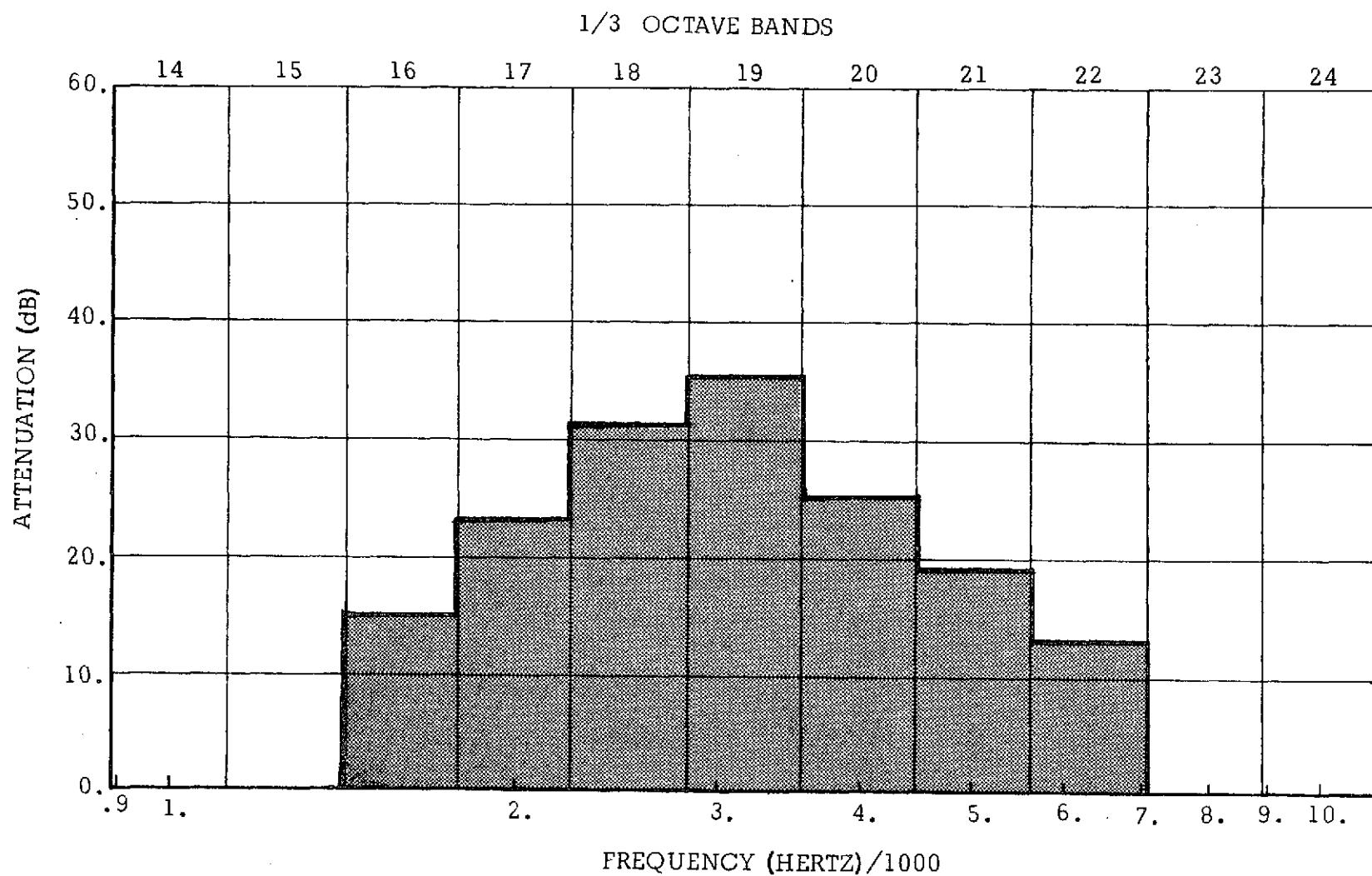
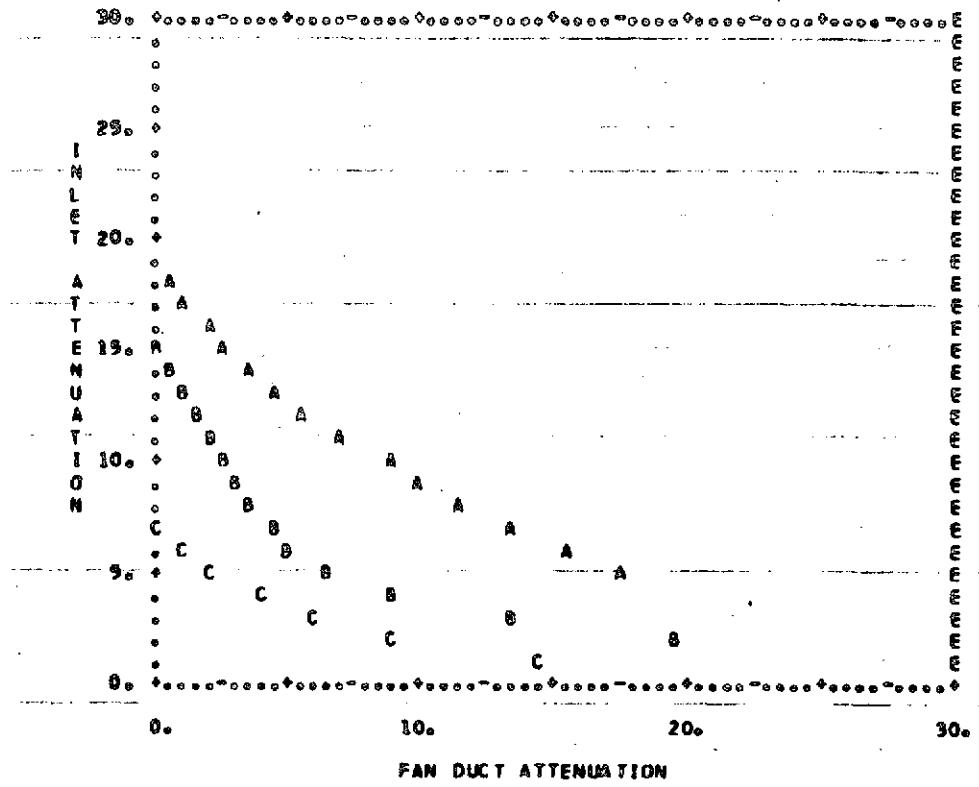


Figure 1. Power Insertion Loss Characteristics

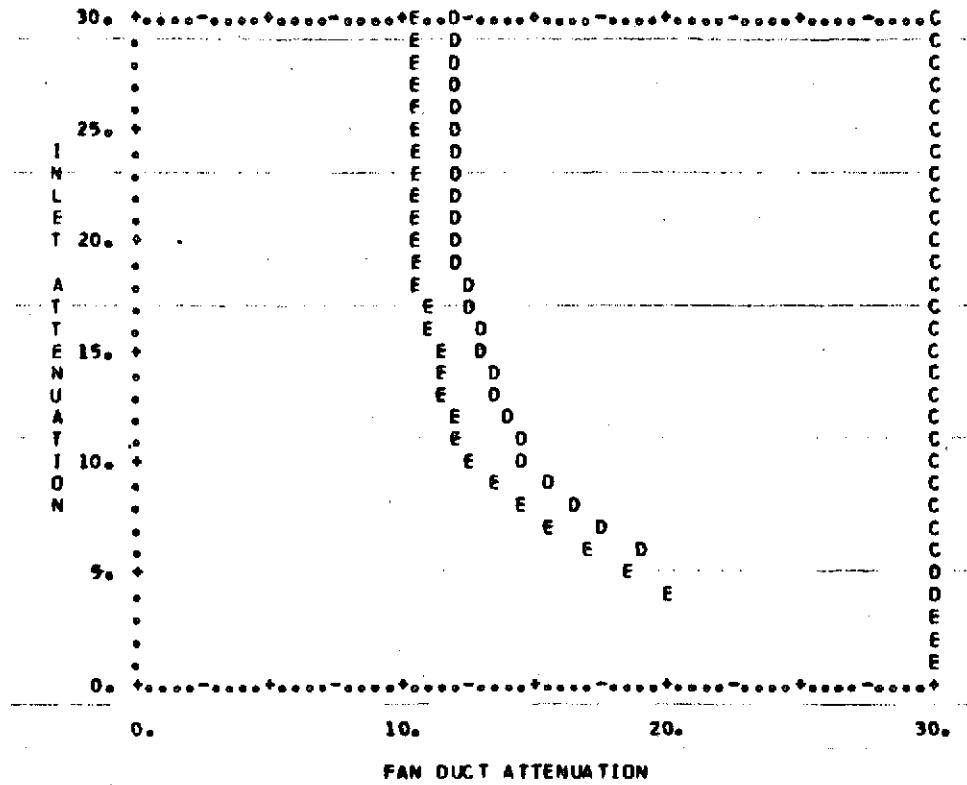
DELTA EPNL = 5.0      FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TOME, 0° JET ATT., WITH TURB. EPNL = 101.00, RUM 1 A



FAN HARMONIC = 20.0		DELTA EPNL = -5.0				
INLET HARMONIC =		A	B	C	D	F
		0	5	10	20	30
1.00	*****	*****	14.53	*****	*****	
2.00	*****	19.33	9.04	*****	*****	
3.00	*****	13.40	6.00	*****	*****	
4.00	*****	8.97	3.79	*****	*****	
5.00	17.65	6.37	1.96	*****	*****	
6.00	15.55	5.21	0.96	*****	*****	
7.00	13.58	4.38	0.06	*****	*****	
8.00	11.72	3.66	*****	*****	*****	
9.00	9.98	2.98	*****	*****	*****	
10.00	8.94	2.32	*****	*****	*****	
11.00	7.20	1.68	*****	*****	*****	
12.00	5.67	1.45	*****	*****	*****	
13.00	4.50	1.02	*****	*****	*****	
14.00	3.57	0.61	*****	*****	*****	
15.00	2.69	0.20	*****	*****	*****	
16.00	1.84	*****	*****	*****	*****	
17.00	1.04	*****	*****	*****	*****	
18.00	0.28	*****	*****	*****	*****	
19.00	*****	*****	*****	*****	*****	
20.00	*****	*****	*****	*****	*****	
21.00	*****	*****	*****	*****	*****	
22.00	*****	*****	*****	*****	*****	
23.00	*****	*****	*****	*****	*****	
24.00	*****	*****	*****	*****	*****	
25.00	*****	*****	*****	*****	*****	
26.00	*****	*****	*****	*****	*****	
27.00	*****	*****	*****	*****	*****	
28.00	*****	*****	*****	*****	*****	
29.00	*****	*****	*****	*****	*****	
30.00	*****	*****	*****	*****	*****	

Figure 2. Inlet Attenuation Versus Fan Duct Attenuation for 5 EPNdB Reduction

DELTA EPNL = 10.0      FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TONE, 0° JET ATT, WITH TURB, EPNL = 101.80, RUN 1 A



FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00		*****	*****	*****	*****	*****
2.00		*****	*****	*****	*****	*****
3.00		*****	*****	*****	*****	*****
4.00		*****	*****	*****	*****	20.02
5.00		*****	*****	*****	*****	18.63
6.00		*****	*****	*****	18.84	18.99
7.00		*****	*****	*****	17.64	15.57
8.00		*****	*****	*****	16.55	14.35
9.00		*****	*****	*****	15.54	13.28
10.00		*****	*****	*****	14.60	12.34
11.00		*****	*****	*****	14.30	12.12
12.00		*****	*****	*****	14.00	11.91
13.00		*****	*****	*****	13.71	11.70
14.00		*****	*****	*****	13.43	11.50
15.00		*****	*****	*****	13.16	11.30
16.00		*****	*****	*****	12.89	11.11
17.00		*****	*****	*****	12.64	10.93
18.00		*****	*****	*****	12.39	10.75
19.00		*****	*****	*****	12.15	10.57
20.00		*****	*****	*****	11.91	10.40
21.00		*****	*****	*****	11.91	10.40
22.00		*****	*****	*****	11.91	10.40
23.00		*****	*****	*****	11.91	10.40
24.00		*****	*****	*****	11.91	10.40
25.00		*****	*****	*****	11.91	10.40
26.00		*****	*****	*****	11.91	10.40
27.00		*****	*****	*****	11.91	10.40
28.00		*****	*****	*****	11.91	10.40
29.00		*****	*****	*****	11.91	10.40
30.00		*****	*****	*****	11.91	10.40

Figure 3. Inlet Attenuation Versus Fan Duct Attenuation for 10 EPNdB Reduction

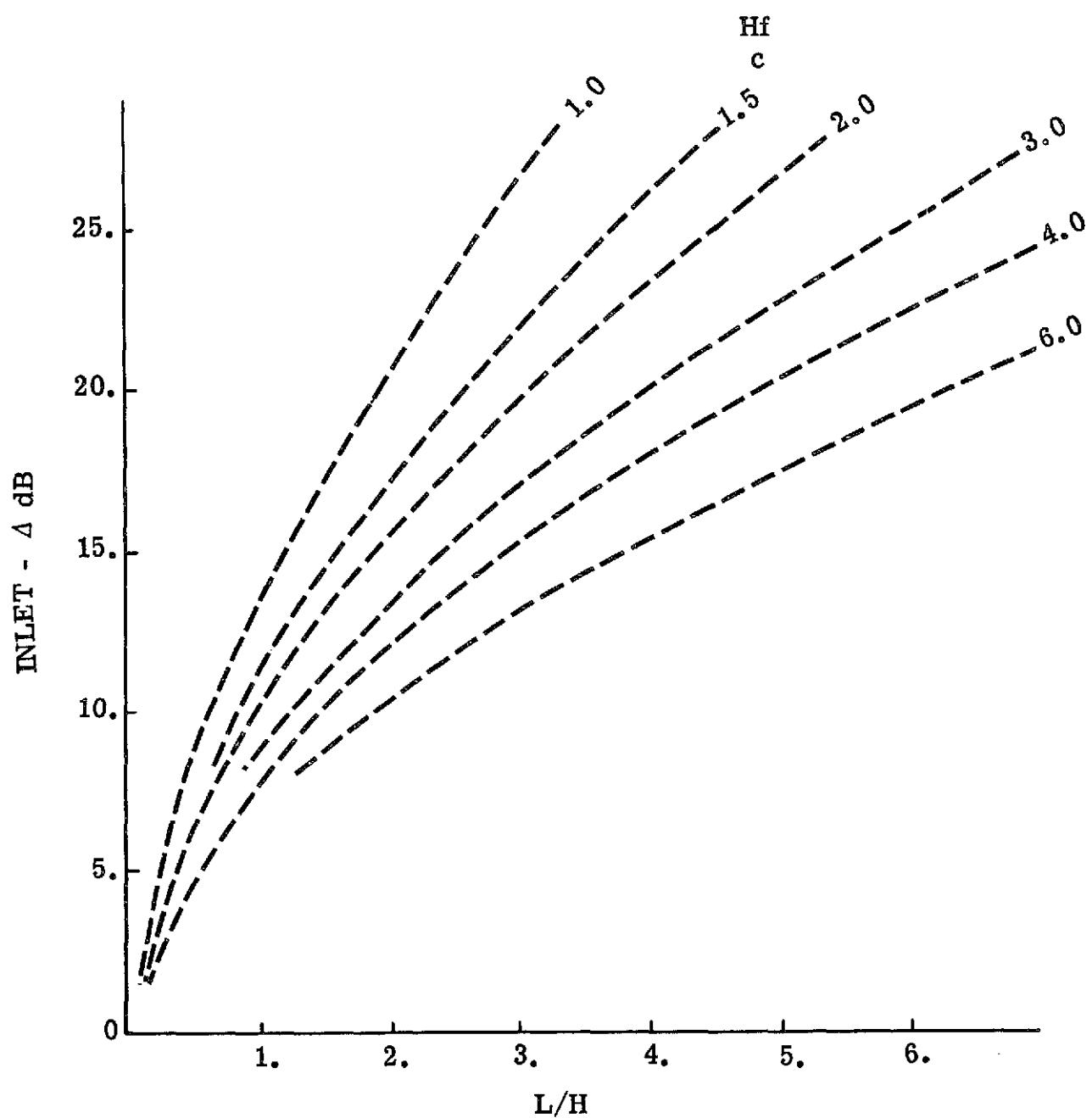


Figure 4. Inlet Mode Attenuation ( $M = .4$ ) over 1 Octave - Band

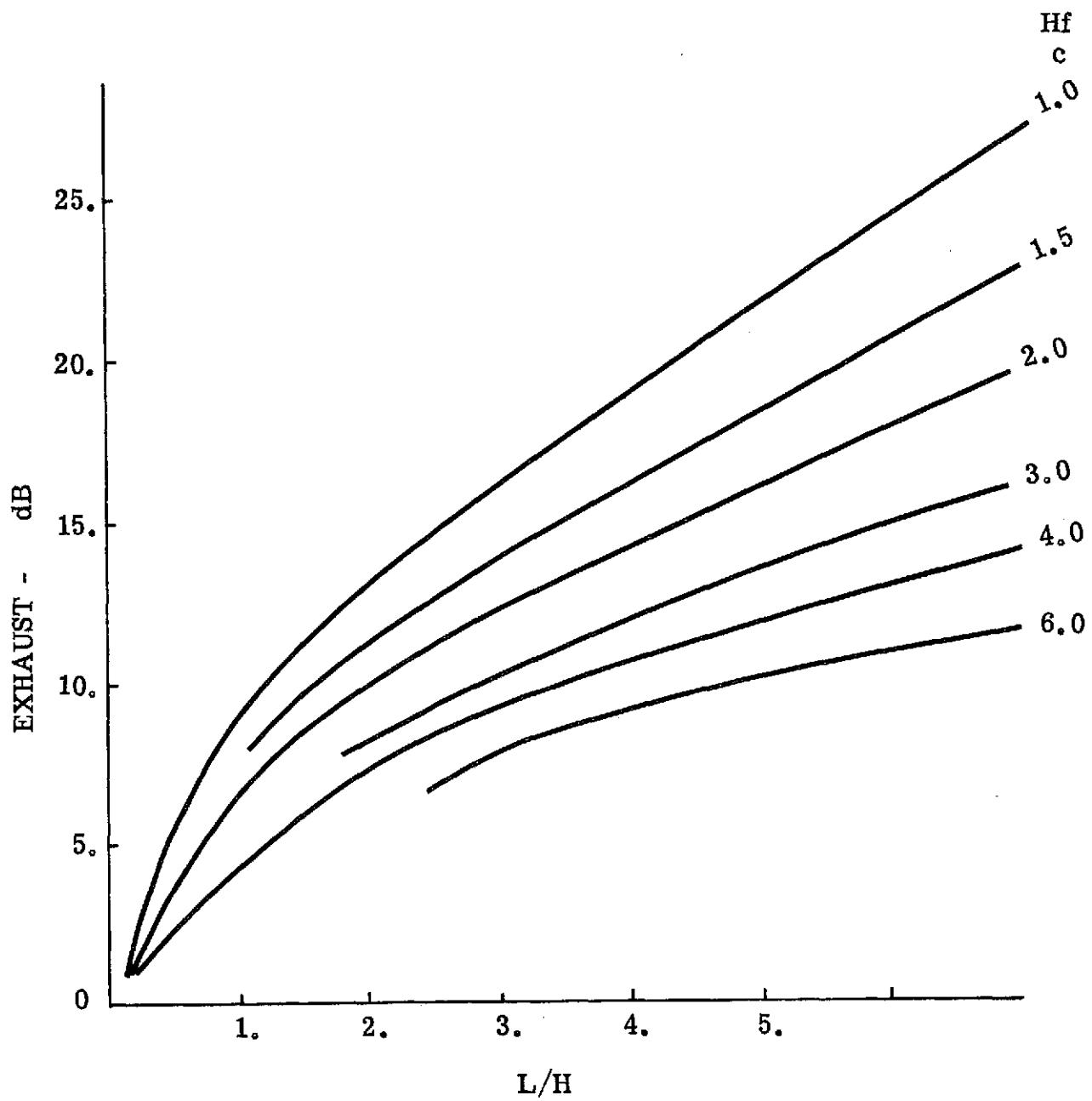


Figure 5. Exhaust Mode Attenuation ( $M = .4$ ) over 1 Octave - Band

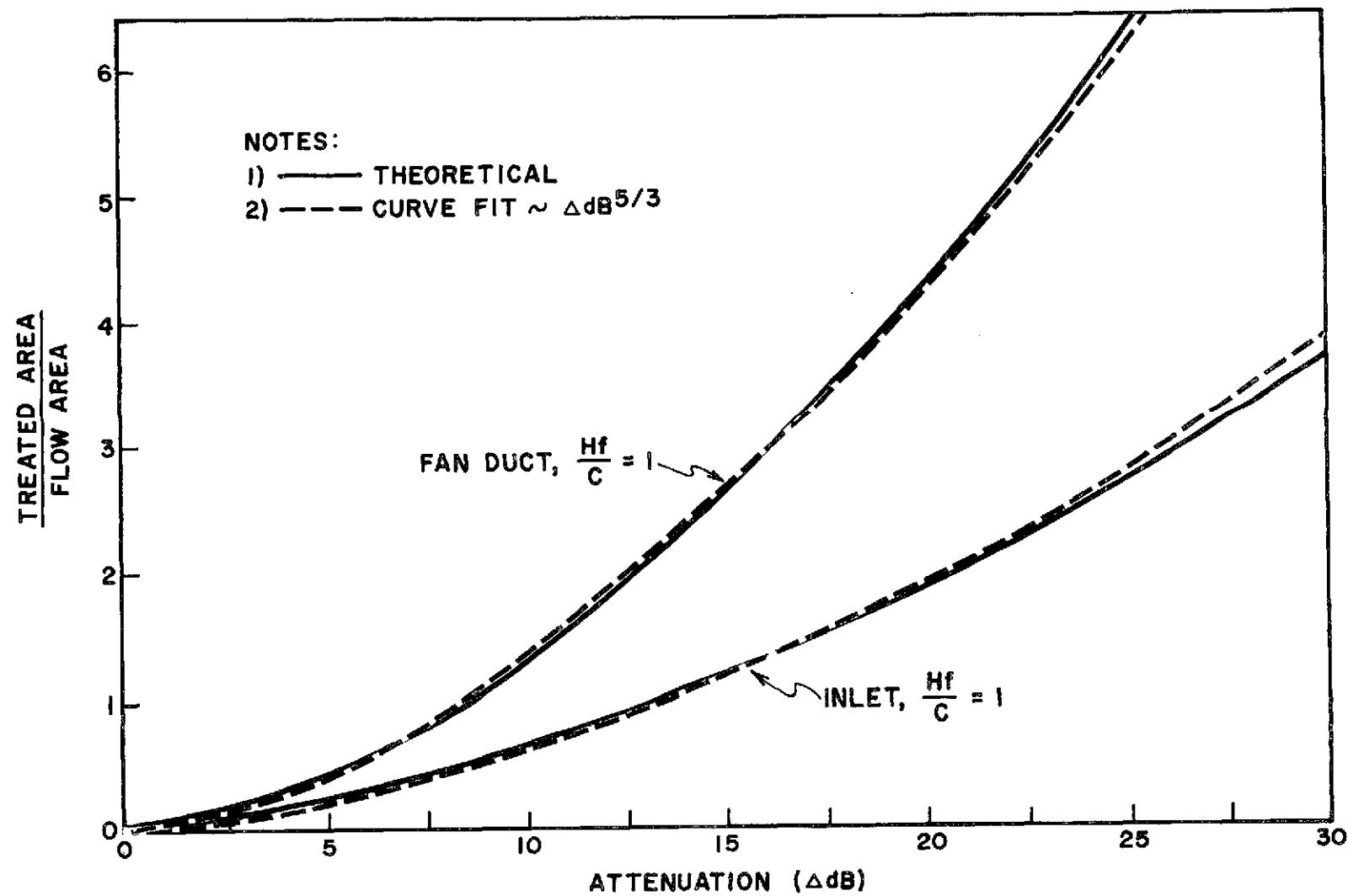


Figure 6. Relation Between Treated Area, Flow Area, and Attenuation

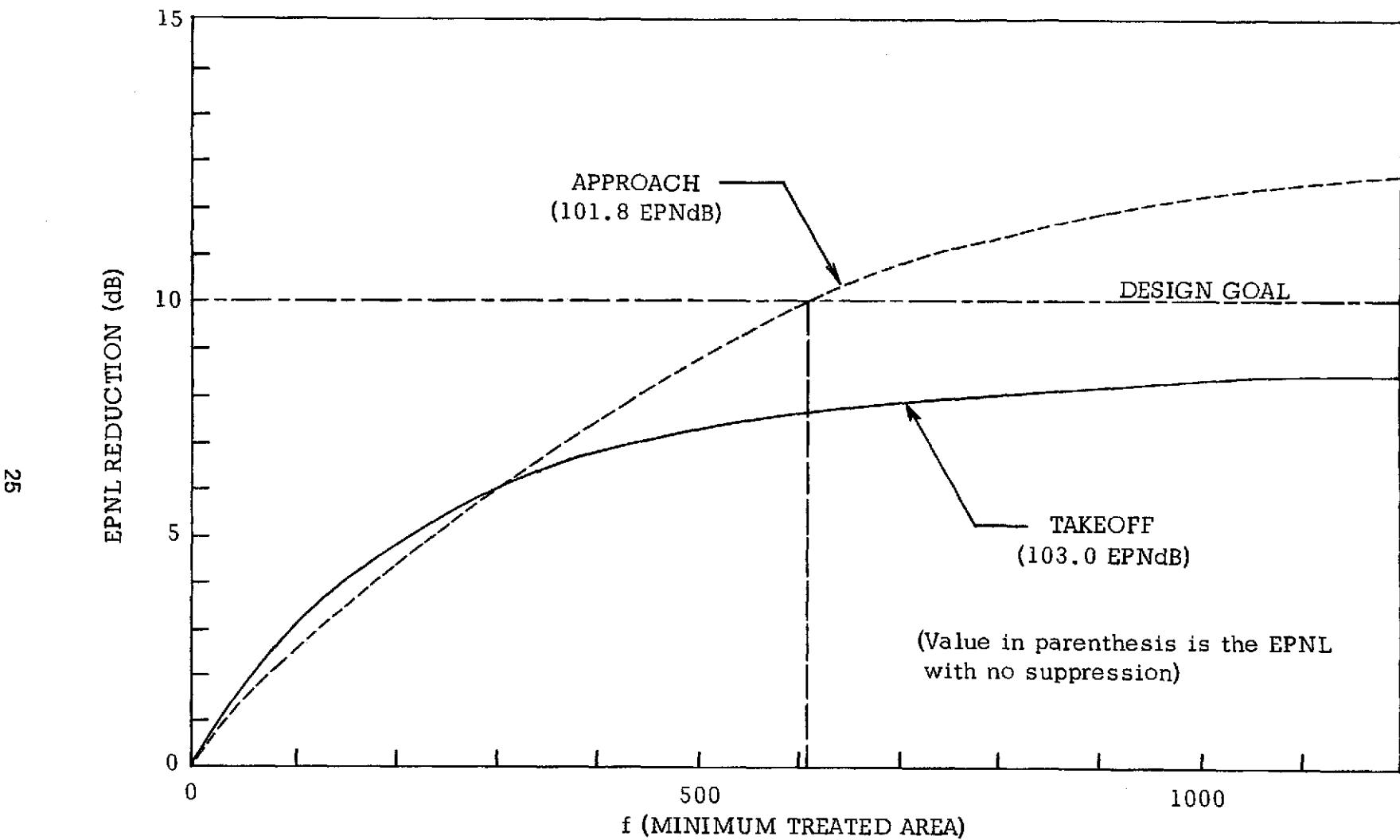


Figure 7. Attenuation Trade Study Results

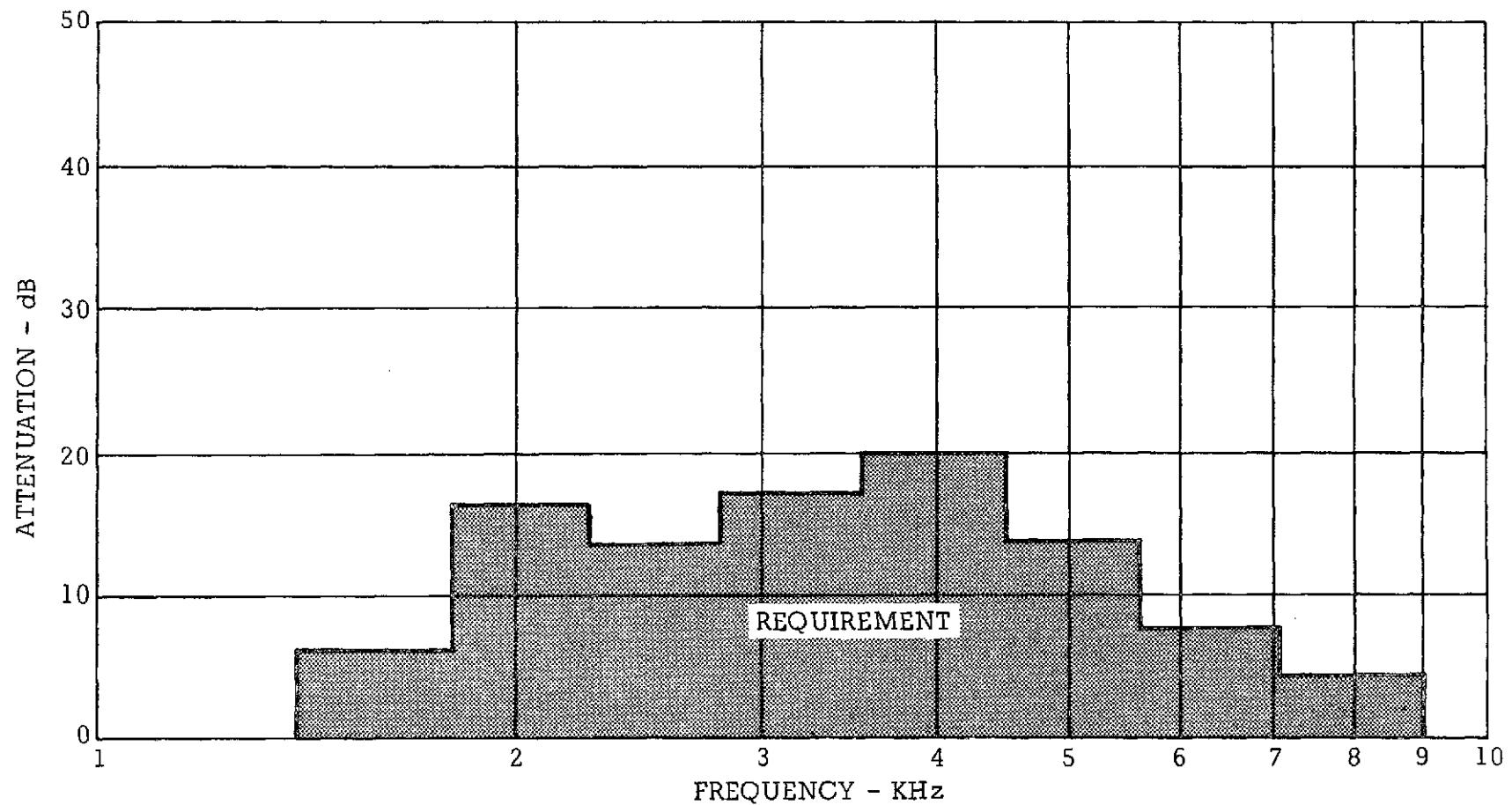


Figure 8. Inlet Lining Attenuation Requirement - Approach

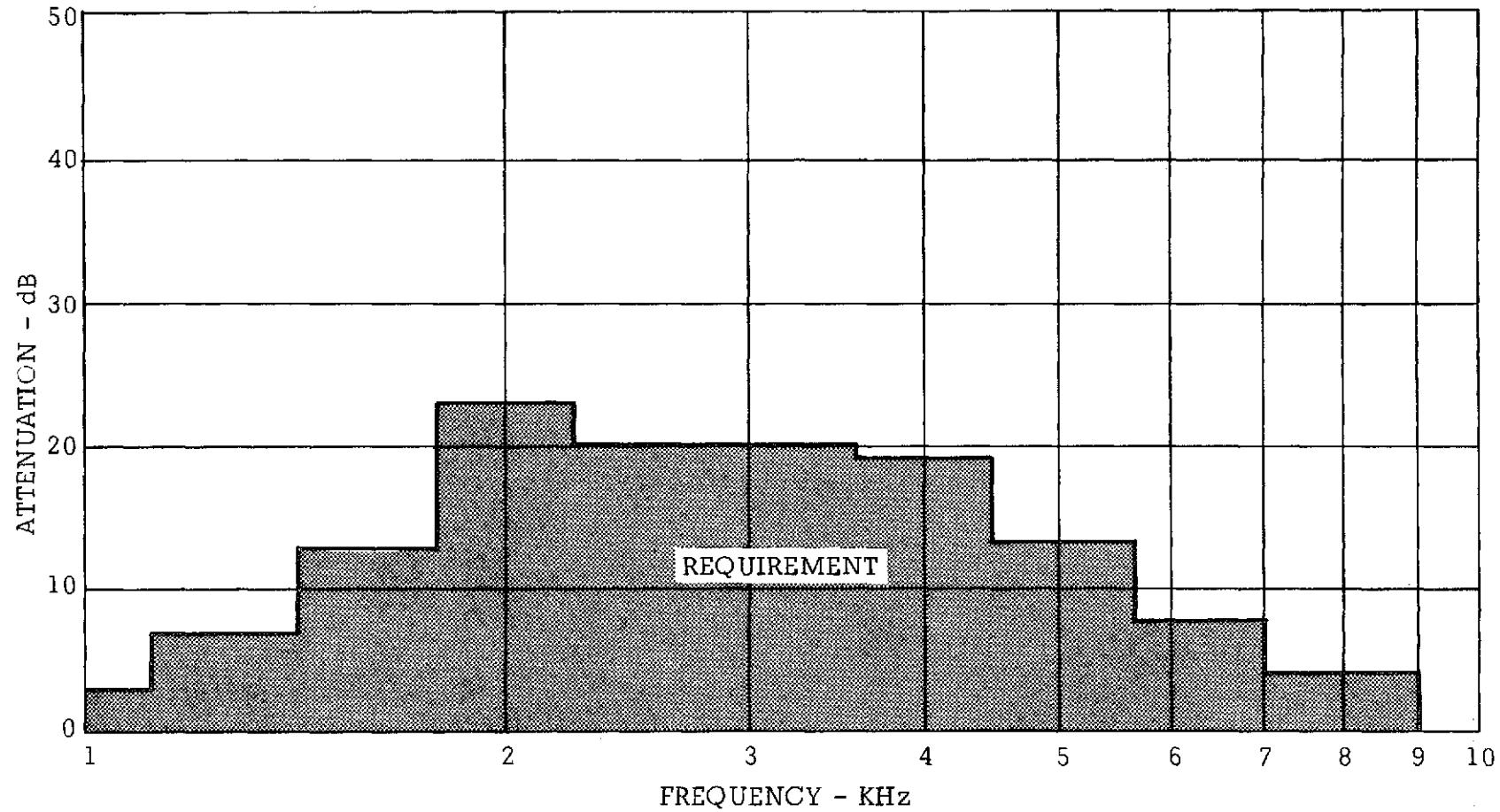


Figure 9. Fan Duct Lining Attenuation Requirement - Approach

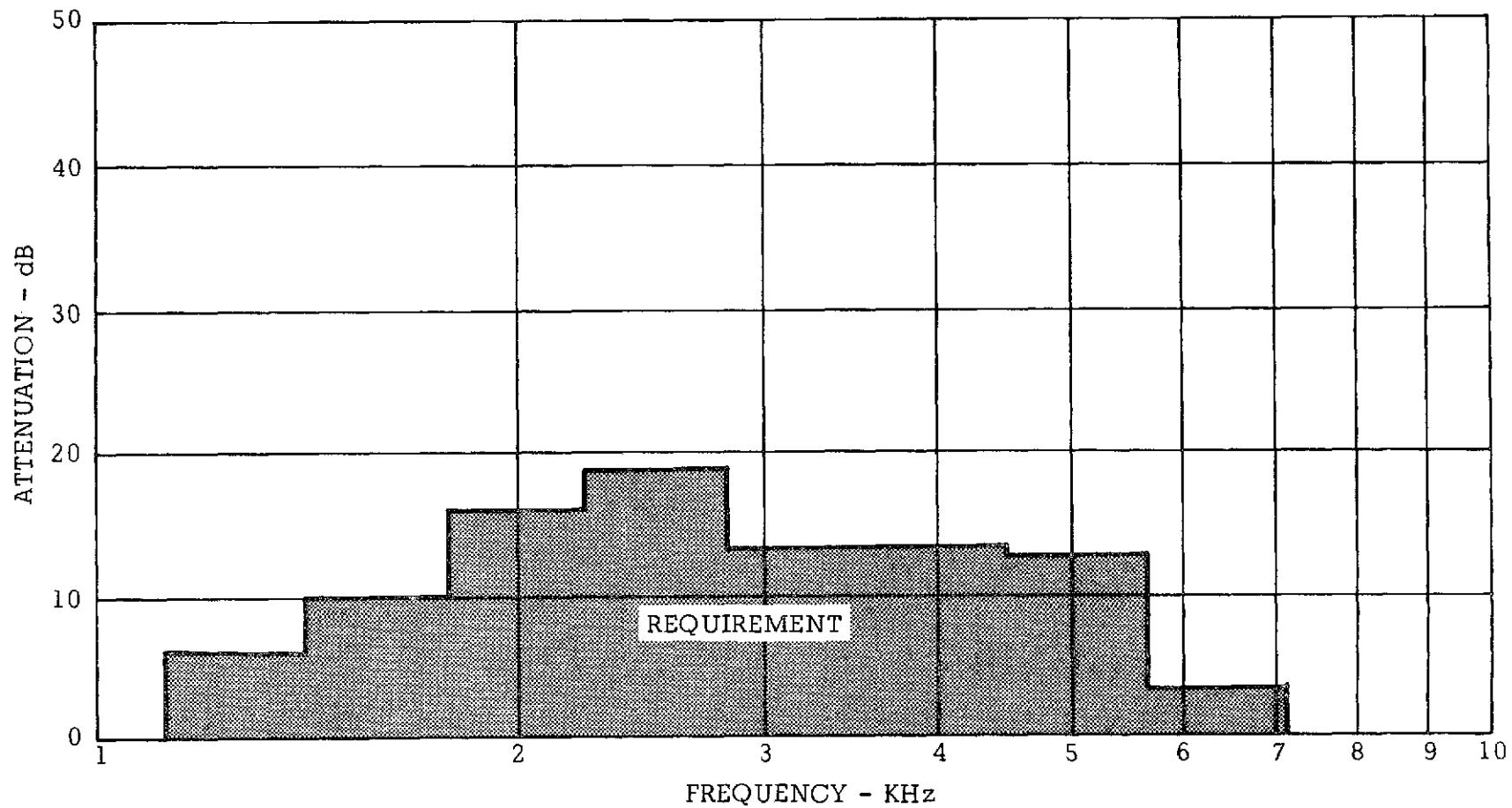


Figure 10. Inlet Lining Attenuation Requirement - Takeoff

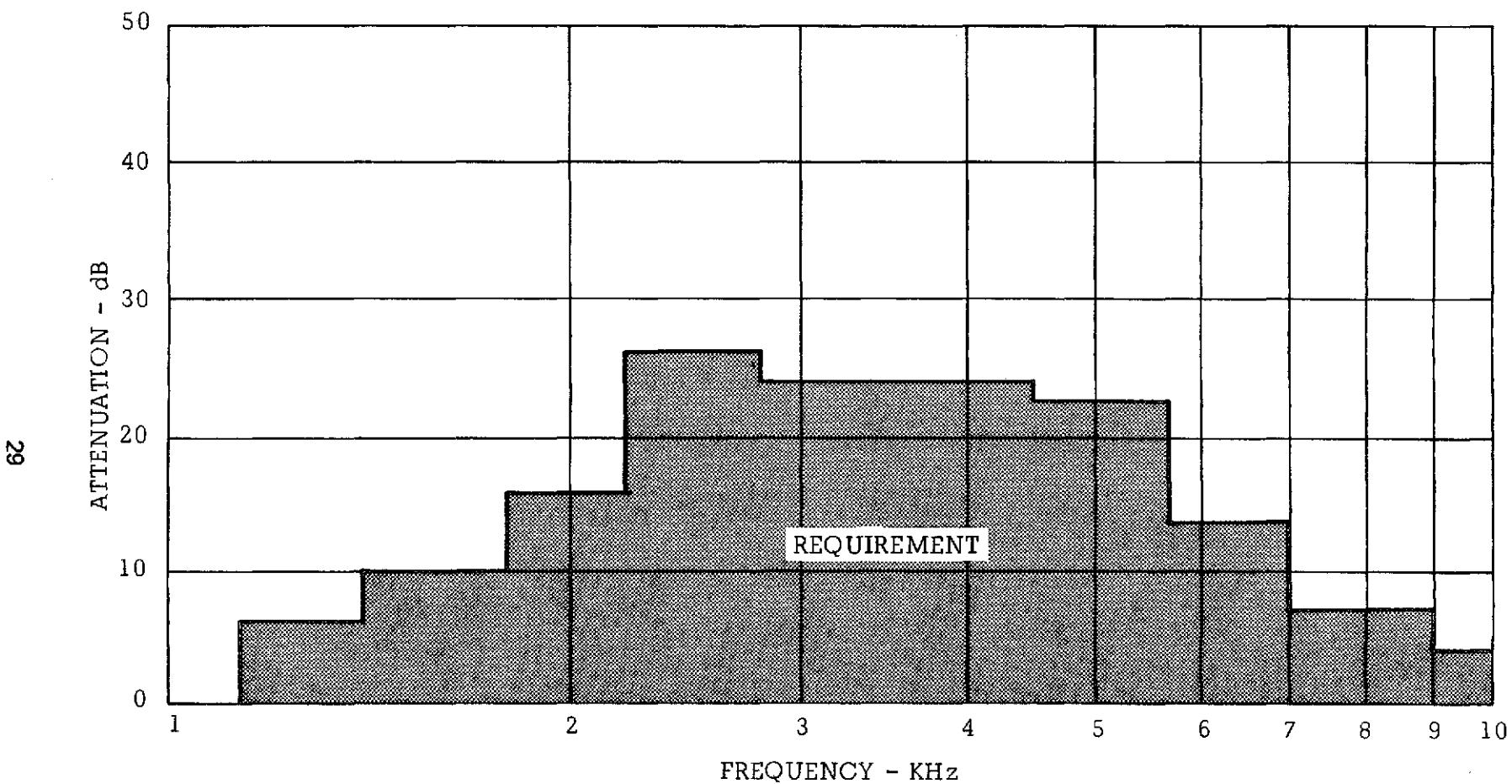


Figure 11. Fan Duct Lining Attenuation Requirement - Takeoff

**TABLES**

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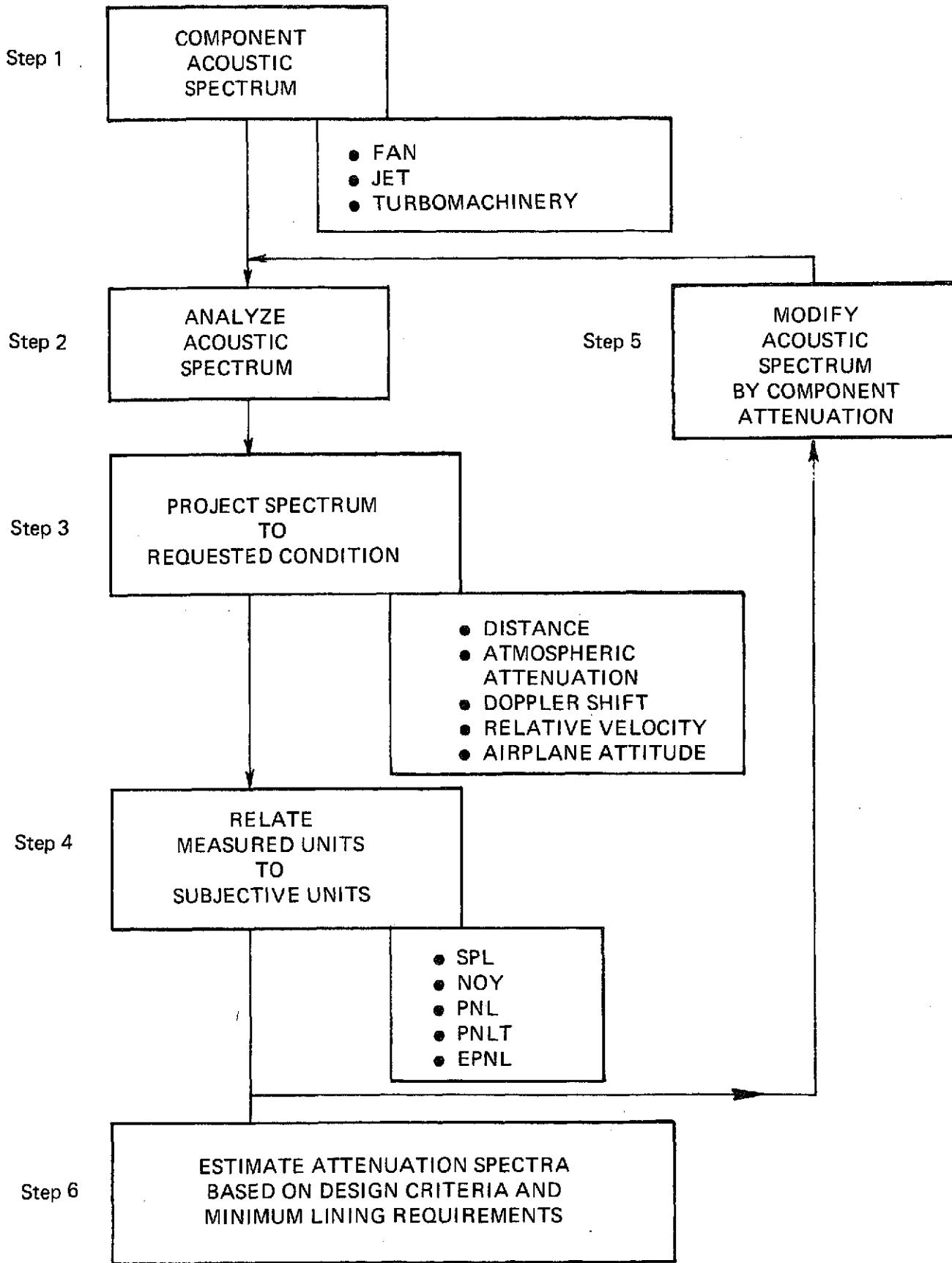


TABLE 1  
SCHEMATIC OF ACOUSTIC TRADE STUDY AND ABSORBER EVALUATION

TABLE 2  
FAN NOISE SPECTRUM

FAN																		
BAND	ANGLE	10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°
	FREQ.	APPROACH																
17	50°	48.6	62.6	50.6	46.6	55.6	46.6	47.6	47.6	49.6	51.6	51.6	53.6	54.6	56.6	58.6	58.6	58.6
18	63°	48.6	62.6	50.6	46.6	55.6	46.6	47.6	47.6	49.6	51.6	51.6	53.6	54.6	56.6	58.6	58.6	58.6
19	80°	48.6	62.6	50.6	46.6	55.6	46.6	47.6	47.6	49.6	51.6	51.6	53.6	54.6	56.6	58.6	58.6	58.6
20	100°	48.6	62.6	50.6	46.6	55.6	46.6	47.6	47.6	49.6	51.6	51.6	53.6	54.6	56.6	58.6	58.6	58.6
21	125°	62.5	68.1	66.5	68.2	65.9	67.5	67.5	65.2	65.5	65.5	67.5	64.9	65.9	64.1	66.1	66.1	66.1
22	160°	66.9	70.5	70.1	66.1	69.1	64.8	65.8	67.1	66.6	65.1	65.1	64.8	62.5	53.6	71.0	71.0	71.0
23	200°	69.5	78.6	71.5	66.6	74.2	65.2	64.9	63.6	62.5	64.5	66.2	66.5	69.2	69.5	66.9	62.1	62.1
24	250°	88.6	74.9	68.3	63.7	72.1	54.3	64.1	64.1	66.1	68.8	69.3	68.8	69.8	68.9	68.8	63.5	63.5
25	315°	64.8	75.6	70.7	68.8	73.5	65.6	65.5	65.5	65.2	65.9	68.4	67.9	68.9	69.6	66.5	59.1	59.1
26	400°	72.9	74.8	70.8	69.7	72.6	65.7	65.6	65.6	67.5	68.6	69.6	70.4	70.1	67.9	63.8	63.8	63.8
27	500°	74.0	77.5	73.0	73.0	75.3	68.9	66.8	66.8	67.4	68.7	70.8	71.0	71.7	71.5	68.6	65.5	65.5
28	630°	75.0	80.8	76.0	74.0	75.0	59.7	63.7	68.9	70.9	71.9	72.9	71.8	73.9	72.7	70.3	67.8	67.8
29	900°	76.0	43.9	77.0	77.0	79.0	72.0	70.0	70.0	72.0	74.0	74.0	73.9	75.9	74.9	70.5	68.2	68.2
30	1000°	79.0	85.0	80.0	78.0	80.0	74.0	73.0	73.0	76.0	77.0	78.0	77.0	79.0	79.0	73.8	71.7	71.7
31	1250°	80.0	85.0	82.0	80.0	81.0	74.0	74.0	75.0	78.0	79.0	80.0	79.0	80.0	80.0	75.9	73.9	73.9
32	1600°	84.0	89.0	86.0	83.0	85.0	78.0	76.0	77.0	80.0	82.0	82.0	82.0	83.0	82.0	75.9	75.9	75.9
33	2000°	89.0	94.0	90.0	88.0	90.0	84.0	81.0	81.0	85.0	86.0	87.0	86.0	86.0	86.0	82.0	79.0	79.0
34	2500°	87.0	88.0	86.0	86.0	85.0	77.0	76.0	77.0	80.0	83.0	83.0	84.0	83.0	83.0	79.0	74.0	74.0
35	3150°	82.0	87.0	86.0	96.0	79.0	76.0	77.0	77.0	81.0	84.0	84.0	84.0	84.0	82.0	77.0	74.0	74.0
36	4000°	83.0	89.0	86.0	88.0	88.0	81.0	77.0	77.0	81.0	82.0	84.0	84.0	84.0	83.0	78.0	75.0	75.0
37	5000°	83.0	88.0	85.0	85.0	87.0	79.0	74.0	75.0	78.0	80.0	81.0	82.0	83.0	83.0	77.0	73.0	73.0
38	6300°	82.0	86.0	84.0	81.0	86.0	76.0	71.0	70.0	74.0	76.0	76.0	79.0	81.0	80.0	75.0	71.0	71.0
39	8000°	79.0	84.0	82.0	81.0	84.0	76.0	71.0	70.0	74.0	76.0	76.0	79.0	81.0	80.0	75.0	71.0	71.0
40	10000°	75.0	80.0	79.0	77.0	80.0	65.0	67.0	68.0	71.0	74.0	74.0	76.0	75.0	70.0	67.0	67.0	67.0

FAN																			
BAND	ANGLE	10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°	
	FREQ.	TAKEOFF																	
17	50°	60.6	65.6	58.6	58.6	64.6	59.6	59.6	59.6	62.6	63.6	65.6	67.6	69.6	72.6	74.6	74.6	74.6	
18	63°	60.6	65.6	58.6	58.6	64.6	59.6	59.6	59.6	62.6	63.6	65.6	67.6	69.6	72.6	74.6	74.6	74.6	
19	80°	60.6	65.6	59.6	58.6	64.6	59.6	59.6	59.6	62.6	63.6	65.6	67.6	69.6	72.6	74.6	74.6	74.6	
20	100°	60.6	65.6	58.6	58.6	64.6	59.6	59.6	59.6	64.6	63.6	65.6	67.6	69.6	72.6	74.6	74.6	74.6	
21	125°	74.5	72.1	76.2	74.5	74.5	73.5	75.5	75.5	78.6	75.9	75.9	81.5	80.9	80.1	82.1	82.1	82.1	
22	160°	77.4	82.4	78.1	78.1	81.4	77.5	79.1	80.3	81.6	78.9	79.1	78.8	77.5	69.6	71.6	71.6	71.6	
23	200°	79.2	82.3	78.4	78.4	81.9	76.9	76.9	76.9	77.5	76.5	78.5	80.5	84.2	85.5	85.5	82.9	82.9	
24	250°	78.3	83.3	77.4	76.3	79.8	76.1	76.1	77.3	78.4	81.3	82.1	82.8	83.4	84.9	85.1	79.5	79.5	
25	315°	80.7	83.5	80.8	79.8	84.7	79.7	78.6	78.6	78.9	80.4	81.2	81.9	93.9	84.2	84.5	75.1	75.1	
26	400°	84.9	85.8	82.9	81.9	83.7	79.8	78.7	79.8	79.3	81.6	82.6	83.4	84.3	84.8	82.1	77.8	79.8	
27	500°	81.9	84.3	90.9	81.9	84.9	79.9	78.9	79.9	79.9	80.7	81.6	82.5	83.4	83.9	81.5	76.9	76.9	
28	630°	83.9	85.5	91.9	81.9	84.9	79.9	78.9	79.9	79.6	81.8	82.8	82.6	83.6	83.1	82.4	79.1	79.1	
29	900°	83.0	88.0	83.0	83.0	85.9	80.9	79.9	79.9	81.8	82.9	83.8	84.8	83.4	81.6	76.5	76.5	76.5	
30	1000°	85.0	89.0	86.0	86.0	90.0	84.0	82.0	83.0	83.9	85.0	85.9	84.9	84.9	84.6	82.1	79.4	79.4	
31	1250°	86.0	90.0	87.0	87.0	90.0	84.0	83.0	84.0	86.0	87.0	87.0	87.0	87.0	85.8	82.4	79.9	79.9	
32	1600°	87.0	91.0	88.0	88.0	91.0	94.0	95.0	95.0	86.0	88.0	89.0	90.0	90.0	89.0	86.9	83.6	81.4	81.4
33	2000°	88.0	93.0	91.0	94.0	92.0	97.0	97.0	97.0	87.0	90.0	92.0	92.0	94.0	94.0	95.0	85.8	82.7	82.7
34	2500°	91.0	98.0	97.0	97.0	100.0	94.0	92.0	91.0	93.0	93.0	95.0	96.0	96.0	91.0	87.9	84.9	84.9	84.9
35	3150°	90.0	96.0	95.0	94.0	98.0	92.0	91.0	90.0	92.0	94.0	94.0	94.0	95.0	95.0	90.0	86.9	83.9	83.9
36	4000°	88.0	93.0	90.0	90.0	93.0	88.0	87.0	88.0	91.0	93.0	93.0	92.0	93.0	88.0	88.0	83.9	81.9	81.9
37	5000°	88.0	94.0	92.0	91.0	94.0	89.0	88.0	89.0	91.0	92.0	93.0	93.0	93.0	90.0	85.0	81.9	81.9	81.9
38	6300°	88.0	92.0	91.0	90.0	93.0	87.0	86.0	88.0	89.0	91.0	92.0	93.0	93.0	90.0	85.0	81.9	81.9	81.9
39	8000°	84.0	89.0	88.0	88.0	90.0	84.0	83.0	84.0	86.0	89.0	90.0	91.0	91.0	88.0	83.0	79.9	79.9	79.9
40	10000°	80.0	86.0	84.0	86.0	80.0	79.0	80.0	82.0	85.0	86.0	87.0	87.0	84.0	84.0	78.9	75.9	75.9	75.9

**TABLE 3**  
**FORWARD FAN NOISE SPECTRUM**

BAND	ANGLE	FAN										FWD						
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°
17	50°	48.6	62.6	50.6	46.6	55.6	46.5	47.2	47.0	46.6	42.7	41.0	32.0	28.2	25.0	22.2	17.0	12.4
18	63°	48.6	62.6	50.6	46.6	55.6	46.5	47.2	47.0	46.6	42.7	41.0	32.0	28.2	25.0	22.2	17.0	12.4
19	80°	48.6	62.6	50.6	46.6	55.6	45.5	47.2	47.0	46.6	42.7	41.0	32.0	28.2	25.0	22.2	17.0	12.4
20	100°	48.6	62.6	50.6	46.6	55.6	46.5	47.2	47.0	46.6	42.7	41.0	32.0	28.2	25.0	22.2	17.0	12.4
21	125°	62.5	68.1	66.5	69.2	66.9	42.5	67.1	64.5	62.5	56.6	57.0	43.3	39.5	32.5	29.8	24.6	20.0
22	160°	66.8	70.5	70.1	66.1	67.1	64.7	65.4	66.5	61.4	56.2	54.5	43.2	35.1	22.0	34.6	24.5	24.8
23	200°	69.5	78.6	71.5	66.4	74.2	65.1	55.5	63.0	59.5	55.6	55.6	44.9	42.8	37.9	30.5	20.6	15.9
24	250°	63.6	74.7	69.3	69.7	72.0	66.2	63.7	63.5	63.1	57.9	58.7	47.2	43.4	37.3	30.5	22.0	17.3
25	315°	69.8	75.0	70.7	68.8	73.5	65.6	45.1	64.9	62.2	57.0	57.8	46.3	42.6	38.0	30.2	17.6	12.9
26	400°	72.9	74.8	70.8	69.9	72.5	65.7	65.0	64.7	59.7	59.1	47.8	44.0	38.5	31.9	22.3	18.0	
27	500°	74.0	77.5	73.9	73.0	75.6	69.9	66.4	66.2	64.8	59.8	60.3	49.4	45.3	39.9	32.3	24.0	19.3
28	630°	73.0	80.5	76.0	74.0	75.9	69.9	68.5	68.3	67.7	63.0	62.4	50.2	47.5	41.1	33.9	26.2	21.6
29	800°	76.0	83.9	77.0	77.0	79.0	71.9	69.6	69.4	69.0	65.1	63.4	52.3	49.6	43.3	34.1	26.6	22.0
30	1000°	79.0	85.0	80.0	78.0	80.0	74.0	72.6	72.4	73.0	68.1	67.4	55.4	52.6	47.4	37.5	30.2	25.6
31	1250°	80.0	85.0	82.0	80.0	81.0	74.0	73.6	74.4	75.0	70.1	69.4	57.4	53.6	48.4	39.5	32.3	28.1
32	1500°	84.0	89.0	86.0	83.0	85.0	78.0	75.6	76.4	77.0	73.1	71.4	60.4	56.6	50.4	42.6	34.4	29.7
33	2000°	89.0	94.0	90.0	88.0	90.0	94.0	89.6	80.4	80.0	76.1	75.4	64.4	60.6	54.4	45.7	37.5	32.8
34	2500°	83.0	88.0	86.0	84.0	85.0	77.0	75.6	76.4	77.0	74.1	72.4	61.4	57.6	51.4	42.6	32.5	27.8
35	3150°	82.0	87.0	86.0	85.0	86.0	79.0	75.6	76.4	78.0	74.1	73.4	62.4	57.6	50.4	40.7	32.4	27.8
36	4000°	83.0	89.0	96.0	85.0	88.0	81.0	76.6	76.4	78.0	73.1	73.4	62.4	57.6	51.4	41.6	33.4	29.2
37	5000°	81.0	86.0	85.0	83.0	87.0	79.0	73.6	74.4	75.0	71.1	70.4	60.6	56.6	50.4	40.7	31.4	26.8
38	6300°	82.0	86.0	84.0	83.0	86.0	79.0	72.6	73.4	74.0	70.1	69.4	59.4	56.6	50.4	40.7	31.5	26.8
39	8000°	79.0	84.0	82.0	81.0	84.0	76.0	70.6	69.4	71.0	67.1	67.4	57.4	54.6	48.4	38.6	29.4	24.9
40	10000°	75.0	80.0	79.0	77.0	80.0	72.0	65.5	66.6	65.0	62.1	63.4	52.4	49.6	43.4	33.7	25.4	20.8

APPROACH

BAND	ANGLE	FAN										FWD						
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°
17	50°	60.6	65.6	58.6	58.6	64.6	59.5	59.2	59.0	59.6	54.7	56.0	66.0	43.2	41.0	38.2	33.0	28.6
18	63°	60.6	65.6	58.6	58.6	64.6	59.5	59.2	59.0	59.6	54.7	55.0	64.0	43.2	41.0	38.2	33.0	28.6
19	80°	60.6	65.6	58.6	58.6	64.6	59.5	59.2	59.0	59.6	54.7	55.0	64.0	43.2	41.0	38.2	33.0	28.6
20	100°	60.6	65.6	58.6	58.6	64.6	59.5	59.2	59.0	61.6	54.7	55.0	66.0	43.2	41.0	38.2	33.0	28.6
21	125°	74.5	72.1	76.1	74.5	78.5	73.5	75.1	78.0	72.9	70.6	69.0	59.9	54.5	48.5	45.8	40.6	36.3
22	160°	77.5	82.5	78.1	81.6	77.7	78.7	79.7	78.6	78.6	70.0	68.5	57.2	51.1	38.0	35.2	30.0	25.8
23	200°	79.2	82.9	78.4	78.4	81.9	76.9	75.5	76.3	74.5	67.6	68.0	58.9	57.8	53.9	49.2	41.4	36.7
24	250°	78.3	83.3	77.4	76.3	79.9	76.0	75.7	76.7	75.4	72.4	71.5	61.2	57.0	53.3	49.7	38.0	33.3
25	315°	80.7	83.5	80.8	78.8	84.7	79.7	71.2	78.0	75.9	71.5	70.6	60.3	57.6	52.5	49.2	33.6	28.9
26	400°	84.9	85.8	82.9	81.9	83.7	79.8	74.3	79.2	76.3	72.8	72.0	61.8	57.9	53.2	45.7	38.3	33.7
27	500°	81.9	84.8	80.9	81.9	85.9	80.8	72.4	78.2	76.5	71.8	71.0	60.9	57.0	52.3	45.1	35.4	30.7
28	630°	83.9	85.7	81.9	81.9	86.7	79.9	79.5	79.3	76.7	72.9	72.2	61.0	57.2	51.5	45.0	37.5	33.3
29	800°	83.0	88.0	83.0	83.0	85.9	40.9	79.5	79.3	78.9	74.0	73.3	62.2	58.4	51.8	45.2	35.0	30.4
30	1000°	85.0	84.9	86.0	86.0	90.0	81.9	82.4	80.9	76.1	75.4	67.3	59.5	53.0	45.7	37.9	33.6	
31	1250°	86.0	90.0	87.0	87.0	90.0	34.0	42.6	43.6	83.0	78.1	76.4	65.3	61.6	54.2	46.0	38.4	33.7
32	1400°	87.0	91.0	88.0	88.0	91.0	34.6	45.4	45.3	80.1	79.4	67.4	64.6	55.3	47.3	39.9	35.2	
33	2000°	88.0	93.0	91.0	90.0	92.0	87.0	86.0	86.4	87.0	83.1	81.4	70.4	67.6	57.3	49.5	41.2	36.5
34	2500°	91.0	98.0	97.0	97.0	100.0	94.3	93.6	90.4	90.4	86.1	85.4	73.4	69.6	59.4	51.6	43.3	38.7
35	3150°	90.0	96.0	95.0	94.0	94.0	92.0	90.6	89.4	89.0	85.1	83.4	72.4	68.6	58.4	50.6	42.3	38.1
36	4000°	88.0	93.0	90.0	90.0	93.0	88.0	96.6	87.4	89.0	84.1	82.4	70.4	66.6	56.4	47.6	40.3	35.7
37	5000°	88.0	94.0	92.0	91.0	94.0	89.0	87.4	88.4	88.0	83.1	82.4	71.4	66.5	58.4	48.6	40.4	35.7
38	6300°	88.0	92.0	91.0	90.0	93.0	87.0	84.6	87.4	86.0	82.1	81.4	71.4	66.6	58.4	48.6	40.4	36.1
39	8000°	84.0	89.0	88.0	88.0	90.0	44.0	82.6	83.4	83.0	80.1	79.4	69.4	64.6	56.4	58.6	38.6	33.8
40	10000°	80.0	86.7	84.0	84.0	86.4	80.0	78.6	79.4	79.0	76.1	75.4	65.4	60.6	57.4	42.6	34.3	27.7

TAKEOFF

**TABLE 4**  
**AFT FAN NOISE SPECTRUM**

BAND	FREQ.	FAN										AFT							APPROACH
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°	
17	50.	2.4	21.0	14.2	15.0	29.2	25.0	37.0	38.7	46.5	51.0	51.2	53.5	54.6	56.6	58.6	58.6	58.6	
18	63.	2.4	21.0	14.2	15.0	29.2	25.0	37.0	38.7	46.5	51.0	51.2	53.5	54.6	56.6	58.6	58.6	58.6	
19	80.	2.4	21.0	14.2	15.0	29.2	25.0	37.0	38.7	46.5	51.0	51.2	53.5	54.6	56.6	58.6	58.6	58.6	
20	100.	2.4	21.0	14.2	15.0	29.2	25.0	37.0	38.7	46.5	51.0	51.2	53.5	54.6	56.6	58.6	58.6	58.6	
21	125.	16.4	26.6	10.1	36.6	40.6	40.7	53.0	56.3	62.5	64.9	67.1	64.9	65.9	64.1	66.1	66.1	66.1	
22	160.	20.6	79.0	33.7	34.5	42.7	43.7	35.2	58.2	67.6	64.5	64.7	64.8	62.5	53.6	71.0	71.0	71.0	
23	200.	23.7	17.1	35.2	34.8	47.8	43.6	54.4	54.7	59.5	63.9	65.8	66.5	69.1	69.5	66.9	62.1	62.1	
24	250.	22.4	33.3	31.9	37.1	45.7	42.7	53.5	55.2	63.0	66.2	68.9	68.7	69.8	68.9	66.8	63.5	63.5	
25	315.	21.0	34.1	34.3	37.2	47.1	44.0	54.9	56.6	62.1	65.3	68.0	67.9	68.9	69.6	66.3	59.1	59.1	
26	400.	26.8	33.2	34.4	38.3	46.2	44.1	55.1	56.8	64.6	68.0	69.2	69.4	70.4	70.1	67.9	63.8	63.8	
27	500.	27.8	36.0	37.6	41.4	49.5	47.3	56.3	57.9	64.7	68.1	70.4	71.0	71.7	71.5	68.6	65.5	65.5	
28	630.	29.2	39.3	39.5	42.4	49.5	48.3	59.6	60.0	67.9	71.3	72.5	71.8	73.9	72.7	70.3	67.8	67.8	
29	800.	29.9	47.4	40.7	45.4	52.6	50.4	59.4	61.1	68.9	73.4	73.6	73.9	75.9	74.9	70.5	68.2	68.2	
30	1000.	32.8	43.5	43.7	46.4	53.6	52.4	62.4	64.1	73.0	76.4	77.6	76.9	79.0	73.8	71.7	71.7	71.7	
31	1250.	31.8	43.4	45.7	48.4	54.6	52.4	63.4	66.1	75.0	78.4	79.6	79.0	80.0	80.0	75.9	73.9	73.9	
32	1600.	37.8	47.5	49.7	51.4	58.6	56.4	65.4	68.1	77.0	81.4	81.6	82.0	83.0	82.0	79.0	75.9	75.9	
33	2000.	42.8	52.5	53.7	50.4	63.6	67.4	70.4	72.1	80.0	84.4	85.6	86.0	87.0	86.0	82.0	79.0	79.0	
34	2500.	36.8	46.5	49.6	52.4	58.6	55.4	65.4	68.1	77.0	82.4	82.6	83.0	84.0	83.0	79.0	74.0	74.0	
35	3150.	35.8	45.4	49.6	53.4	59.6	57.4	65.4	68.1	78.0	82.4	83.6	84.0	84.0	82.0	77.0	74.0	74.0	
36	4000.	37.2	47.5	49.7	53.4	61.6	59.4	65.4	68.1	78.0	81.4	83.6	84.0	84.0	83.0	78.0	75.0	75.0	
37	5000.	36.8	46.5	48.6	51.4	60.6	57.4	63.4	66.1	75.0	79.4	80.6	82.0	83.0	82.0	77.0	73.0	73.0	
38	6300.	35.9	44.5	47.6	51.4	59.6	56.4	62.4	65.1	74.0	78.4	79.6	81.0	83.0	82.0	77.0	73.0	73.0	
39	8000.	32.4	42.5	45.7	49.4	57.6	54.4	60.4	61.1	71.0	75.4	77.6	79.0	81.0	80.0	75.0	71.0	71.0	
40	10000.	28.4	38.5	42.6	45.4	53.6	50.4	55.4	58.1	65.0	70.6	73.6	74.0	76.0	75.0	70.0	67.0	67.0	

BAND	FREQ.	FAN										AFT							TAKEOFF
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°	
17	50.	14.5	24.0	22.2	27.0	38.2	38.0	49.0	50.7	59.5	63.0	65.2	67.5	69.6	72.6	74.6	74.6	74.6	
18	63.	14.5	26.0	22.2	27.0	38.2	38.0	49.0	50.7	59.5	63.0	65.2	67.5	69.6	72.6	74.6	74.6	74.6	
19	80.	14.5	24.0	22.2	27.0	38.2	38.0	49.0	50.7	59.5	63.0	65.2	67.5	69.6	72.6	74.6	74.6	74.6	
20	100.	14.5	24.0	22.2	27.0	38.2	38.0	49.0	50.7	59.5	63.0	65.2	67.5	69.6	72.6	74.6	74.6	74.6	
21	125.	28.4	30.6	39.8	42.9	52.2	51.9	65.0	69.7	72.9	78.9	79.1	81.5	80.9	80.1	82.1	82.1	82.1	
22	160.	31.2	40.9	41.7	46.5	55.0	55.2	58.5	71.4	58.5	78.3	78.7	78.8	77.5	79.6	71.6	71.6	71.6	
23	200.	33.0	41.4	42.0	46.8	55.5	55.1	66.4	68.0	74.5	75.9	78.1	80.5	84.1	85.5	85.5	82.9	82.9	
24	250.	32.1	41.7	41.1	44.7	53.4	54.4	65.5	68.4	75.4	80.7	81.7	82.7	83.4	84.9	85.1	79.5	79.5	
25	315.	34.5	42.0	44.4	48.2	58.3	58.1	68.0	69.7	75.9	79.8	80.8	81.9	83.9	84.1	84.5	75.1	75.1	
26	400.	38.7	46.3	46.5	50.3	57.3	58.2	66.2	70.9	76.2	81.0	82.2	83.4	84.3	84.8	82.1	79.8	79.8	
27	500.	35.7	43.3	44.6	50.3	59.5	59.3	67.1	69.9	76.5	80.1	81.2	82.5	83.3	83.9	81.5	76.9	76.9	
28	630.	37.8	44.3	45.8	50.3	58.5	58.3	68.3	71.0	76.6	81.2	82.4	82.8	83.5	83.1	82.4	79.1	79.1	
29	400.	37.2	46.6	46.6	51.4	59.6	59.3	67.4	71.0	78.8	82.3	83.5	83.8	84.7	83.4	81.6	76.5	76.5	
30	1000.	38.9	47.4	49.6	54.4	63.6	62.4	71.6	74.1	80.9	84.4	85.5	84.9	84.6	92.1	79.4	79.4	79.4	
31	1250.	39.8	48.5	50.7	55.4	63.6	62.4	72.4	75.1	82.9	86.4	86.6	86.9	87.9	85.8	87.4	79.9	79.9	
32	1600.	40.8	49.4	51.6	56.4	64.6	54.4	74.4	77.1	85.0	88.4	89.6	88.9	91.0	86.9	83.6	81.4	81.4	
33	2000.	41.8	51.5	54.6	57.4	65.6	55.4	76.4	78.1	87.0	91.4	91.6	92.0	94.0	88.9	85.8	82.7	82.7	
34	2500.	44.8	56.5	60.6	65.4	73.6	72.4	81.4	82.1	90.0	94.4	95.6	95.0	96.0	91.0	87.9	84.9	84.9	
35	3150.	43.8	54.5	58.7	62.4	71.6	73.4	80.4	81.1	87.0	93.4	93.6	94.0	95.0	90.0	86.9	83.9	83.9	
36	4000.	41.8	51.5	53.7	58.4	66.6	64.4	76.4	79.1	88.0	92.4	92.6	92.0	93.0	88.0	83.3	81.9	81.9	
37	5000.	41.9	52.5	54.6	59.4	67.6	67.4	77.4	80.1	88.0	91.4	92.6	93.0	93.0	90.0	85.0	81.9	81.9	
38	6300.	42.2	50.5	54.7	58.4	66.6	75.4	79.1	86.0	90.4	91.6	93.0	93.0	90.0	85.0	81.9	81.9	81.9	
39	8000.	37.8	47.5	51.7	54.6	63.6	52.4	72.4	75.1	83.0	88.4	89.6	91.0	91.0	88.0	81.0	79.9	79.9	
40	10000.	33.9	44.5	47.7	52.4	59.6	58.6	68.4	71.1	79.0	84.4	85.6	87.0	87.0	84.0	79.9	75.9	75.9	

**TABLE 5**  
**APPROACH AND TAKEOFF FLIGHT CONDITIONS**

<u>CONDITION</u>	<u>ROTOR SPEED (%)</u>	<u>FAA REF. POINT (N. MILE)</u>	<u>VELOCITY (FT/SEC.)</u>	<u>ALTITUDE (FT.)</u>	<u>CLIMB GRADIENT</u>	<u>ANGLE OF ATTACK</u>
Approach	60	1	254	370	-3.0°	3.7°
Takeoff	90	3.5	317.44	1000	4.57°	7.6°

**TABLE 6**  
**EPNL TRADE STUDY RUN IDENTIFICATION**

<u>RUN</u>	<u>CONDITION</u>	<u>TURBINE NOISE</u>	<u>PRIMARY JET ATTENUATION (dB)</u>	<u>EPN LEVEL WITHOUT LININGS (EPNdB)</u>
1A	Approach	Yes	0	101.80
2A	Takeoff	No	0	103.02
5A	Takeoff	No	10	101.83
10A	Takeoff	No		101.70
11A	Approach	No		101.66
12A	Takeoff	No	2.5	102.56
13A	Takeoff	No	5.0	102.22
14A	Takeoff	No	7.5	101.98

**TABLE 7**  
**INPUT DATA IDENTIFICATION**

<u>DATA</u>	<u>TYPE</u>	<u>ORIGIN</u>
Fan + Jet 2	100-ft. Polar	NASA-Lewis
Jet	250-ft. Polar	General Electric
Turbomachinery	370-ft. Sideline	General Electric

TABLE 8  
FAN AND SECONDARY JET NOISE SPECTRUM (100-FT. POLAR)

BAND	ANGLE	FAN										JET2						APPROACH
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.
FREQ.																		
17	50.	72.0	86.0	74.0	70.0	74.0	70.0	71.0	71.0	73.0	75.0	75.0	77.0	78.0	80.0	82.0	82.0	82.0
18	63.	72.0	86.0	74.0	70.0	79.0	73.0	71.0	71.0	73.0	75.0	75.0	77.0	78.0	80.0	82.0	82.0	82.0
19	80.	72.0	86.0	74.0	70.0	79.0	70.0	71.0	71.0	73.0	75.0	75.0	77.0	78.0	80.0	82.0	82.0	82.0
20	100.	72.0	86.0	74.0	70.0	79.0	70.0	71.0	71.0	73.0	75.0	75.0	77.0	78.0	80.0	82.0	82.0	82.0
21	125.	73.0	82.0	76.0	76.0	79.0	72.0	73.0	74.0	75.0	76.0	77.0	77.0	78.0	80.0	80.0	80.0	80.0
22	160.	75.0	84.0	78.0	74.0	79.0	73.0	74.0	75.0	75.0	75.0	76.0	76.0	77.0	78.0	78.0	78.0	78.0
23	200.	77.0	87.0	79.0	74.0	82.0	73.0	75.0	72.0	72.0	74.0	75.0	76.0	78.0	79.0	79.0	79.0	79.0
24	250.	76.0	84.0	76.0	80.0	72.0	72.0	72.0	74.0	75.0	77.0	77.0	78.0	78.0	78.0	77.0	77.0	77.0
25	315.	77.0	84.0	78.0	76.0	81.0	73.0	73.0	73.0	73.0	74.0	76.0	76.0	77.0	78.0	77.0	75.0	75.0
26	400.	80.0	83.0	78.0	77.0	80.0	73.0	73.0	73.0	73.0	75.0	76.0	77.0	77.0	78.0	77.0	75.0	75.0
27	500.	81.0	85.0	81.0	80.0	83.0	76.0	74.0	76.0	76.0	78.0	78.0	78.0	79.0	79.0	78.0	76.0	76.0
28	630.	82.0	88.0	83.0	81.0	83.0	77.0	76.0	76.0	78.0	79.0	80.0	79.0	81.0	80.0	78.0	76.0	76.0
29	800.	83.0	91.0	84.0	84.0	86.0	79.0	77.0	77.0	79.0	81.0	81.0	81.0	83.0	82.0	78.0	76.0	76.0
30	1000.	86.0	92.0	87.0	85.0	87.0	81.0	80.0	80.0	83.0	84.0	85.0	84.0	86.0	86.0	87.0	87.0	87.0
31	1250.	87.0	92.0	89.0	87.0	88.0	81.0	81.0	82.0	85.0	86.0	87.0	86.0	87.0	87.0	87.0	87.0	87.0
32	1600.	91.0	96.0	93.0	90.0	92.0	85.0	83.0	84.0	87.0	89.0	89.0	89.0	90.0	90.0	90.0	90.0	90.0
33	2000.	96.0	101.0	97.0	95.0	97.0	91.0	88.0	88.0	90.0	92.0	93.0	93.0	94.0	94.0	94.0	94.0	94.0
34	2500.	98.0	95.0	93.0	91.0	92.0	84.0	83.0	84.0	87.0	90.0	90.0	91.0	90.0	90.0	89.0	89.0	89.0
35	3150.	90.0	95.0	94.0	93.0	94.0	87.0	86.0	85.0	89.0	91.0	92.0	92.0	92.0	90.0	91.0	91.0	91.0
36	4000.	91.0	97.0	94.0	93.0	96.0	95.0	94.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0
37	5000.	91.0	96.0	93.0	91.0	90.0	93.0	92.0	92.0	92.0	92.0	92.0	92.0	92.0	91.0	91.0	91.0	91.0
38	6300.	91.0	94.0	92.0	91.0	94.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0	88.0
39	8000.	88.0	93.0	91.0	90.0	93.0	92.0	92.0	92.0	93.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0	95.0
40	10000.	85.0	90.0	89.0	87.0	90.0	82.0	82.0	76.0	77.0	78.0	81.0	84.0	86.0	85.0	85.0	85.0	85.0

BAND	ANGLE	FAN										JET2						TAKEOFF	
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.	
FREQ.																			
17	50.	84.0	89.0	82.0	82.0	88.0	83.0	83.0	83.0	86.0	87.0	89.0	91.0	93.0	96.0	98.0	98.0	98.0	
18	63.	84.0	89.0	82.0	82.0	88.0	83.0	83.0	83.0	86.0	87.0	89.0	91.0	93.0	96.0	98.0	98.0	98.0	
19	80.	84.0	89.0	82.0	82.0	88.0	83.0	83.0	83.0	86.0	87.0	89.0	91.0	93.0	96.0	98.0	98.0	98.0	
20	100.	84.0	89.0	82.0	82.0	88.0	83.0	83.0	83.0	86.0	87.0	89.0	91.0	93.0	96.0	98.0	98.0	98.0	
21	125.	85.0	88.0	85.0	84.0	89.0	84.0	85.0	85.0	87.0	88.0	89.0	90.0	92.0	94.0	96.0	96.0	96.0	
22	160.	86.0	91.0	86.0	86.0	90.0	86.0	87.0	87.0	88.0	88.0	89.0	90.0	92.0	94.0	95.0	95.0	95.0	
23	200.	87.0	91.0	86.0	86.0	90.0	83.0	85.0	85.0	85.0	86.0	86.0	86.0	87.0	89.0	90.0	91.0	91.0	
24	250.	86.0	91.0	85.0	84.0	88.0	84.0	84.0	85.0	85.0	87.0	87.0	88.0	90.0	92.0	94.0	95.0	95.0	
25	315.	88.0	91.0	88.0	87.0	92.0	87.0	86.0	86.0	87.0	87.0	89.0	90.0	91.0	92.0	93.0	94.0	95.0	
26	400.	92.0	93.0	90.0	89.0	91.0	87.0	87.0	87.0	87.0	89.0	89.0	90.0	91.0	92.0	93.0	94.0	95.0	
27	500.	89.0	92.0	83.0	89.0	93.0	88.0	87.0	87.0	88.0	87.0	88.0	89.0	90.0	91.0	92.0	93.0	94.0	
28	630.	91.0	93.0	89.0	89.0	92.0	92.0	87.0	86.0	86.0	87.0	87.0	89.0	90.0	90.0	91.0	91.0	91.0	
29	800.	90.0	95.0	90.0	90.0	93.0	93.0	97.0	87.0	87.0	89.0	90.0	91.0	91.0	92.0	92.0	92.0	92.0	
30	1000.	92.0	96.0	93.0	93.0	97.0	91.0	91.0	90.0	91.0	92.0	92.0	93.0	93.0	94.0	94.0	94.0	94.0	
31	1250.	93.0	97.0	96.0	96.0	97.0	91.0	91.0	90.0	91.0	93.0	94.0	94.0	95.0	95.0	96.0	96.0	96.0	
32	1600.	94.0	98.0	95.0	95.0	98.0	93.0	92.0	93.0	95.0	96.0	97.0	98.0	98.0	99.0	99.0	99.0	99.0	
33	2000.	95.0	100.0	94.0	96.0	99.0	94.0	94.0	94.0	97.0	97.0	98.0	99.0	101.0	101.0	96.0	96.0	96.0	96.0
34	2500.	98.0	105.0	104.0	104.0	107.0	101.0	101.0	99.0	98.0	100.0	102.0	102.0	102.0	102.0	103.0	98.0	95.0	92.0
35	3150.	98.0	106.0	104.0	102.0	106.0	101.0	99.0	98.0	100.0	102.0	102.0	102.0	102.0	102.0	103.0	98.0	95.0	92.0
36	4000.	95.0	101.0	96.0	95.0	101.0	98.0	95.0	96.0	99.0	101.0	101.0	100.0	101.0	101.0	98.0	93.0	90.0	90.0
37	5000.	96.0	102.0	101.0	99.0	102.0	97.0	96.0	97.0	99.0	100.0	101.0	101.0	101.0	101.0	101.0	98.0	93.0	90.0
38	6300.	96.0	100.0	99.0	98.0	101.0	95.0	94.0	96.0	97.0	99.0	100.0	101.0	101.0	101.0	101.0	98.0	93.0	90.0
39	8000.	93.0	98.0	91.0	91.0	99.0	93.0	92.0	93.0	95.0	98.0	99.0	100.0	100.0	100.0	97.0	92.0	89.0	89.0
40	10000.	90.0	98.0	94.0	94.0	95.0	99.0	99.0	99.0	99.0	92.0	95.0	96.0	97.0	97.0	94.0	89.0	86.0	86.0

TABLE 9  
PRIMARY JET NOISE SPECTRUM (250 FT. POLAR)

BAND FREQ.	ANGLE	JET										PRIM									APPROACH
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°			
17	50.	53.2	52.1	52.9	53.6	54.6	55.2	55.9	57.2	58.9	60.8	63.0	65.9	69.2	73.2	77.9	75.6	75.6			
18	63.	54.8	53.5	53.8	54.4	54.9	56.0	56.3	58.1	59.5	61.5	63.5	66.4	69.4	73.1	76.7	74.8	74.8			
19	80.	56.0	54.5	54.7	55.3	55.9	56.6	57.5	58.7	60.2	61.9	64.0	66.7	69.4	72.7	75.3	73.4	73.4			
20	100.	56.4	54.7	55.0	55.4	56.2	57.0	57.8	58.9	60.4	62.1	64.2	66.6	69.1	72.1	73.9	71.7	71.7			
21	125.	56.9	55.0	54.9	55.5	56.2	57.0	57.8	58.9	60.3	62.0	63.8	66.2	68.5	71.2	72.0	70.1	70.1			
22	160.	56.8	54.7	54.7	55.3	56.0	56.4	57.4	58.6	60.0	61.6	63.3	65.5	67.7	69.9	70.1	68.3	68.3			
23	200.	56.5	54.3	54.4	54.8	55.5	56.3	57.1	58.1	59.3	60.9	62.6	64.6	66.6	68.5	68.2	66.4	66.4			
24	250.	56.2	53.8	53.7	54.2	54.7	55.3	56.4	57.3	58.6	60.1	61.6	63.4	65.3	66.8	66.0	64.3	64.3			
25	315.	55.7	53.0	52.9	53.3	53.8	54.6	55.5	56.4	57.5	59.1	60.5	62.2	63.9	64.9	63.9	62.2	62.2			
26	400.	55.0	52.2	52.0	52.2	52.4	53.6	54.3	55.4	56.4	57.8	59.1	60.9	62.4	61.8	60.3	60.3				
27	500.	54.2	51.2	50.8	51.1	51.8	52.5	53.3	54.3	55.3	56.5	57.8	59.5	60.8	60.8	59.5	58.3	58.3			
28	630.	53.3	50.1	49.8	50.0	50.6	51.3	52.1	53.0	54.0	55.1	56.5	57.9	59.2	58.8	57.4	56.3	56.3			
29	800.	52.6	49.1	48.5	48.8	49.4	50.0	50.8	51.7	52.7	53.6	55.0	56.4	57.4	58.8	55.3	54.4	54.4			
30	1000.	51.5	46.0	47.3	47.5	48.1	48.7	49.5	50.4	51.2	52.3	53.5	54.8	55.7	56.8	53.2	52.5	52.5			
31	1250.	50.5	46.7	46.0	46.2	46.7	47.3	48.1	49.0	49.8	51.0	52.1	53.3	53.9	53.0	51.2	50.6	50.6			
32	1600.	49.4	45.4	44.7	44.9	45.2	46.1	46.8	47.6	48.4	49.5	50.8	51.9	52.0	49.2	48.9	48.9				
33	2000.	48.2	44.1	43.4	43.6	43.9	44.7	45.5	46.2	47.0	48.1	49.3	50.3	49.4	47.4	47.2	47.2				
34	2500.	46.8	42.7	42.0	41.9	42.5	43.3	44.1	44.8	45.7	46.7	47.9	49.0	48.6	47.7	45.6	45.4	45.4			
35	3150.	45.4	41.3	40.6	40.7	41.2	41.9	42.8	43.5	44.4	45.4	46.7	47.7	47.0	46.1	44.0	43.8	43.8			
36	4000.	44.0	39.8	39.4	39.6	40.0	40.6	41.5	42.2	43.2	44.2	45.4	46.5	46.6	46.6	42.5	42.2	42.2			
37	5000.	42.7	36.5	36.1	38.1	38.7	39.3	40.2	41.0	42.0	42.9	44.1	45.1	43.1	41.0	40.7	40.7				
38	6300.	41.5	37.3	36.9	36.4	37.5	38.2	39.0	39.8	40.8	41.7	42.9	43.9	42.7	41.7	39.6	39.3	39.3			
39	8000.	40.4	36.2	35.7	35.7	36.2	37.0	37.7	38.6	39.5	40.6	41.6	42.7	41.3	40.3	38.3	38.0	38.0			
40	10000.	39.2	35.1	34.4	34.6	35.1	35.8	36.5	37.3	38.3	39.2	40.4	41.5	40.0	39.0	36.9	36.7	36.7			

BAND FREQ.	ANGLE	JET										PRIM									TAKEOFF
		10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°			
17	50.	77.7	76.6	77.3	78.1	79.1	79.6	80.3	81.6	83.3	85.2	87.4	90.3	93.7	97.7	102.3	100.1	100.1			
18	63.	80.3	79.0	79.3	79.9	80.5	81.6	82.3	83.6	85.3	87.0	89.0	92.0	95.0	98.6	102.2	100.3	100.3			
19	80.	82.6	81.1	81.3	81.9	82.5	83.2	84.1	85.3	86.8	88.5	90.6	93.3	96.0	99.3	101.9	100.0	100.0			
20	100.	86.0	82.3	82.6	83.0	83.8	84.6	85.4	86.6	88.1	89.8	91.8	94.2	96.7	99.7	101.5	93.3	93.3			
21	125.	85.5	83.6	83.5	84.1	84.8	85.5	86.4	87.5	88.9	90.6	92.4	94.7	97.1	99.8	100.6	98.7	98.7			
22	160.	86.2	84.1	84.2	84.7	85.4	86.2	86.9	88.1	89.5	91.1	92.8	94.9	97.1	99.3	99.6	97.7	97.7			
23	200.	86.8	84.6	84.6	85.0	85.8	86.5	87.4	88.3	89.5	91.1	92.9	94.8	96.9	98.7	94.4	96.7	96.7			
24	250.	87.2	84.6	84.7	85.1	85.7	86.5	87.4	88.3	89.5	91.1	92.6	94.6	96.3	97.7	97.0	95.3	95.3			
25	315.	87.2	84.6	84.4	85.4	86.2	87.1	87.9	89.1	90.6	92.1	93.8	95.5	96.5	95.4	91.8	91.8				
26	400.	87.1	84.2	84.1	84.3	85.0	85.7	86.5	87.5	88.5	89.9	91.1	93.0	94.5	95.0	93.4	92.3	92.3			
27	500.	86.7	83.7	83.3	83.6	84.3	85.0	85.8	86.4	87.8	89.0	90.3	92.0	93.3	93.3	92.0	90.8	90.8			
28	630.	86.1	82.9	82.6	82.8	83.3	84.1	84.9	85.8	86.6	87.9	89.3	90.7	91.9	91.6	90.2	89.1	89.1			
29	800.	85.4	82.0	81.5	81.8	82.4	83.0	83.8	84.7	85.6	86.4	88.0	89.4	90.3	89.4	88.3	87.3	87.3			
30	1000.	84.6	81.0	80.7	81.1	81.8	82.6	83.4	84.3	85.3	86.0	87.0	88.7	89.7	86.5	85.5	85.5				
31	1250.	83.6	79.3	79.1	79.3	79.8	80.4	81.2	82.0	82.9	84.0	85.2	86.4	87.0	86.1	84.3	83.7	83.7			
32	1600.	82.6	76.4	77.0	77.9	78.3	78.1	78.8	80.0	81.4	82.5	83.8	84.9	85.1	84.2	82.3	81.9	81.9			
33	2000.	81.1	77.0	76.3	76.3	76.8	76.6	76.4	77.1	77.9	81.0	82.2	83.2	82.3	82.3	80.3	80.1	80.1			
34	2500.	79.6	75.4	74.6	74.6	75.2	76.0	76.4	77.6	78.4	79.5	80.6	81.7	81.4	80.4	78.4	78.1	78.1			
35	3150.	78.3	73.8	73.2	73.2	73.7	74.4	75.3	76.0	76.9	77.9	79.2	80.2	79.6	78.6	76.5	76.3	76.3			
36	4000.	76.3	72.2	71.7	71.7	72.3	73.4	74.5	75.5	76.5	77.7	78.9	79.9	79.9	78.6	76.6	74.6	74.6			
37	5000.	74.8	70.7	70.2	70.2	70.8	71.4	72.3	73.1	74.1	75.0	76.2	77.2	76.2	75.2	73.1	72.9	72.9			
38	6300.	73.5	69.4	69.4	69.5	69.5	70.2	71.0	71.8	72.8	73.7	74.9	75.9	74.7	73.7	71.6	71.4	71.4			
39	8000.	72.3	68.2	67.6	67.7	68.2	69.0	69.7	70.5	71.5	72.5	73.6	74.7	73.3	72.3	70.2	70.0	70.0			
40	10000.	71.2	67.1	66.4	66.6	67.1	67.8	68.5	69.3	70.3	71.2	72.4	73.5	72.0	71.0	69.9	68.7	68.7			

## APPROACH

BAND FREQ.	ANGLE	TURB										BASE						
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.
17	50.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
18	61.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
19	80.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
20	100.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
21	125.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
22	160.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
23	200.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
24	250.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
25	315.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
26	400.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
27	500.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
28	640.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
29	800.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
30	1000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
31	1250.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
32	1600.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
33	2000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
34	2500.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
35	3150.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
36	4000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
37	5000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
38	6300.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
39	8000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0
40	10000.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.0	10.0	10.0	10.0	0.0	0.0	0.0	0.0	0.0

TABLE 10  
TURBOMACHINERY NOISE SPECTRUM (370 FT. SIDELINE)

**TABLE 11**  
**MODIFIED DATA IDENTIFICATION**

<u>DATA</u>	<u>TYPE</u>
Fan + Jet 2	200-ft. Polar
Jet 1	200-ft. Polar
Jet 2	200-ft. Polar
Turbomachinery	200-ft. Polar

TABLE 12  
FAN AND SECONDARY JET NOISE SPECTRUM (200-FT. POLAR)

BAND	ANGLE	FAN										JET2							APPROACH
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.	
17	50.	65.0	79.0	67.0	63.0	72.0	63.0	64.0	64.0	66.0	68.0	68.0	70.0	71.0	73.0	75.0	75.0	75.0	
18	63.	65.0	79.0	67.0	63.0	72.0	63.0	64.0	64.0	66.0	68.0	68.0	70.0	71.0	73.0	75.0	75.0	75.0	
19	80.	65.0	79.0	67.0	63.0	72.0	64.0	64.0	64.0	66.0	68.0	68.0	70.0	71.0	73.0	75.0	75.0	75.0	
20	100.	65.0	79.0	67.0	63.0	72.0	63.0	64.0	64.0	66.0	68.0	68.0	70.0	71.0	73.0	75.0	75.0	75.0	
21	125.	66.0	75.0	69.0	69.0	72.0	65.0	66.0	67.0	68.0	69.0	70.0	70.0	71.0	71.0	73.0	73.0	73.0	
22	800.	48.0	47.0	46.0	47.0	46.0	46.0	47.0	46.0	46.0	46.0	46.0	46.0	46.0	46.0	46.0	46.0	46.0	
23	200.	70.0	80.0	72.0	67.0	75.0	66.0	66.0	65.0	65.0	67.0	68.0	69.0	71.0	72.0	71.0	71.0	71.0	
24	300.	69.0	77.0	69.0	69.0	73.0	65.0	65.0	65.0	67.0	68.0	70.0	70.0	71.0	71.0	70.0	70.0	70.0	
25	315.	70.0	77.0	71.0	69.0	74.0	66.0	66.0	66.0	66.0	67.0	69.0	69.0	70.0	71.0	70.0	68.0	68.0	
26	400.	73.0	76.0	71.0	70.0	73.0	66.0	66.0	66.0	68.0	69.0	70.0	70.0	71.0	70.0	68.0	68.0	68.0	
27	500.	74.0	76.0	74.0	73.0	76.0	69.0	67.0	67.0	68.0	69.0	71.0	71.0	72.0	72.0	70.0	68.0	68.0	
28	610.	75.0	81.0	76.0	76.0	76.0	70.0	69.0	69.0	71.0	72.0	73.0	72.0	74.0	73.0	71.0	69.0	69.0	
29	800.	76.0	84.0	77.0	77.0	79.0	72.0	73.0	70.0	72.0	74.0	76.0	76.0	75.0	71.0	69.0	69.0	69.0	
30	1000.	79.0	85.0	80.0	78.0	80.0	74.0	74.0	71.0	73.0	76.0	77.0	78.0	77.0	79.0	79.0	74.0	72.0	
31	1250.	80.0	85.0	82.0	80.0	81.0	74.0	74.0	75.0	78.0	79.0	80.0	79.0	80.0	80.0	76.0	74.0	74.0	
32	1600.	84.0	89.0	86.0	83.0	85.0	78.0	78.0	77.0	80.0	82.0	82.0	82.0	83.0	82.0	78.0	76.0	76.0	
33	2000.	89.0	94.0	90.0	88.0	90.0	84.0	81.0	81.0	83.0	85.0	86.0	86.0	87.0	86.0	82.0	79.0	79.0	
34	2500.	83.0	86.0	86.0	84.0	84.0	77.0	76.0	77.0	80.0	83.0	83.0	83.0	84.0	83.0	79.0	74.0	74.0	
35	3150.	82.0	87.0	86.0	85.0	86.0	79.0	76.0	77.0	81.0	83.0	84.0	84.0	84.0	82.0	77.0	74.0	74.0	
36	4000.	83.0	89.0	86.0	85.0	88.0	81.0	77.0	77.0	81.0	84.0	84.0	84.0	84.0	83.0	78.0	75.0	75.0	
37	5000.	83.0	88.0	85.0	83.0	87.0	79.0	74.0	75.0	78.0	80.0	81.0	82.0	83.0	82.0	77.0	71.0	73.0	
38	6300.	82.0	86.0	84.0	83.0	86.0	74.0	73.0	74.0	77.0	79.0	80.0	81.0	83.0	82.0	77.0	73.0	73.0	
39	8000.	79.0	84.0	82.0	81.0	84.0	76.0	71.0	70.0	74.0	76.0	78.0	79.0	81.0	80.0	75.0	71.0	71.0	
40	10000.	75.0	80.0	74.0	77.0	80.0	72.0	66.0	67.0	68.0	71.0	74.0	76.0	76.0	75.0	70.0	67.0	67.0	

BAND	ANGLE	FAN										JET2							TAKEOFF
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.	
17	50.	77.0	82.0	79.0	75.0	81.0	76.0	75.0	76.0	79.0	80.0	82.0	84.0	86.0	89.0	91.0	91.0	91.0	
18	63.	77.0	82.0	79.0	75.0	81.0	76.0	76.0	76.0	79.0	80.0	82.0	84.0	86.0	89.0	91.0	91.0	91.0	
19	80.	77.0	82.0	79.0	75.0	81.0	76.0	76.0	76.0	79.0	80.0	82.0	84.0	86.0	89.0	91.0	91.0	91.0	
20	100.	77.0	82.0	79.0	75.0	81.0	76.0	76.0	76.0	79.0	80.0	82.0	84.0	86.0	89.0	91.0	91.0	91.0	
21	125.	78.0	81.0	78.0	77.0	82.0	77.0	76.0	80.0	81.0	82.0	83.0	85.0	86.0	87.0	89.0	89.0	89.0	
22	100.	79.0	84.0	79.0	79.0	83.0	79.0	79.0	80.0	81.0	82.0	83.0	84.0	85.0	86.0	88.0	88.0	88.0	
23	200.	80.0	84.0	79.0	79.0	83.0	78.0	78.0	79.0	80.0	81.0	83.0	85.0	86.0	88.0	86.0	86.0	86.0	
24	250.	79.0	84.0	78.0	77.0	81.0	77.0	77.0	78.0	80.0	82.0	83.0	84.0	85.0	87.0	88.0	86.0	86.0	
25	315.	81.0	84.0	81.0	80.0	85.0	80.0	79.0	80.0	80.0	81.0	82.0	84.0	85.0	87.0	87.0	84.0	84.0	
26	400.	85.0	86.0	43.0	82.0	84.0	80.0	79.0	80.0	80.0	82.0	84.0	85.0	86.0	85.0	85.0	82.0	82.0	
27	500.	82.0	85.0	81.0	82.0	86.0	81.0	80.0	80.0	80.0	81.0	82.0	83.0	84.0	85.0	85.0	82.0	82.0	
28	630.	84.0	86.0	82.0	82.0	85.0	83.0	83.0	84.0	85.0	86.0	87.0	88.0	89.0	89.0	89.0	87.0	87.0	
29	800.	83.0	84.0	83.0	83.0	86.0	81.0	80.0	80.0	82.0	83.0	84.0	85.0	86.0	86.0	86.0	86.0	86.0	
30	1000.	85.0	89.0	86.0	86.0	90.0	84.0	84.0	84.0	86.0	87.0	87.0	88.0	89.0	89.0	89.0	81.0	81.0	
31	1250.	86.0	90.0	87.0	87.0	90.0	87.0	87.0	87.0	88.0	88.0	88.0	89.0	90.0	91.0	91.0	81.0	81.0	
32	1600.	87.0	91.0	88.0	88.0	91.0	86.0	86.0	86.0	87.0	87.0	87.0	87.0	87.0	87.0	87.0	82.0	82.0	
33	2000.	88.0	93.0	91.0	91.0	92.0	87.0	87.0	87.0	90.0	92.0	92.0	94.0	94.0	94.0	94.0	86.0	83.0	
34	2500.	91.0	94.0	97.0	97.0	100.0	94.0	92.0	91.0	92.0	93.0	93.0	94.0	95.0	95.0	95.0	95.0	95.0	
35	3150.	90.0	96.0	95.0	94.0	98.0	92.0	91.0	90.0	92.0	94.0	94.0	94.0	95.0	95.0	95.0	95.0	94.0	
36	4000.	86.0	93.0	91.0	90.0	93.0	89.0	87.0	86.0	88.0	91.0	91.0	92.0	93.0	93.0	93.0	92.0	92.0	
37	5000.	88.0	94.0	92.0	91.0	94.0	89.0	87.0	86.0	89.0	91.0	92.0	93.0	93.0	93.0	93.0	92.0	92.0	
38	6300.	88.0	92.0	91.0	90.0	93.0	87.0	85.0	84.0	89.0	91.0	92.0	93.0	93.0	93.0	93.0	92.0	92.0	
39	8000.	86.0	89.0	88.0	88.0	90.0	86.0	84.0	83.0	84.0	86.0	87.0	89.0	91.0	91.0	88.0	87.0	89.0	
40	10000.	86.0	86.0	85.0	85.0	86.0	84.0	84.0	84.0	86.0	86.0	86.0	86.0	87.0	87.0	87.0	84.0	84.0	

TABLE 13  
PRIMARY JET NOISE SPECTRUM (200-FT. POLAR)

BAND	ANGLE	JET										PRIM							APPROACH
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.	
17	50°	55.0	54.0	55.0	56.0	57.0	57.0	58.0	59.0	61.0	63.0	65.0	68.0	71.0	75.0	80.0	85.0	88.0	
18	63°	57.0	55.0	56.0	56.0	57.0	58.0	59.0	60.0	61.0	63.0	65.0	68.0	71.0	75.0	79.0	77.0	77.0	
19	80°	58.0	57.0	57.0	57.0	58.0	59.0	60.0	61.0	62.0	64.0	66.0	69.0	71.0	75.0	77.0	75.0	75.0	
20	100°	58.0	57.0	57.0	57.0	58.0	59.0	60.0	61.0	62.0	64.0	66.0	69.0	71.0	74.0	76.0	74.0	74.0	
21	125°	59.0	57.0	57.0	58.0	58.0	59.0	60.0	61.0	62.0	64.0	66.0	68.0	71.0	73.0	74.0	72.0	72.0	
22	140°	59.0	57.0	57.0	57.0	58.0	59.0	60.0	61.0	62.0	64.0	65.0	68.0	70.0	72.0	72.0	70.0	70.0	
23	200°	59.0	56.0	56.0	57.0	58.0	58.0	59.0	60.0	61.0	63.0	65.0	67.0	69.0	71.0	70.0	68.0	68.0	
24	250°	58.0	56.0	56.0	56.0	57.0	58.0	59.0	59.0	60.0	62.0	64.0	65.0	67.0	69.0	66.0	66.0	66.0	
25	315°	58.0	55.0	55.0	55.0	56.0	57.0	58.0	58.0	60.0	61.0	63.0	64.0	66.0	67.0	66.0	64.0	64.0	
26	400°	57.0	54.0	54.0	54.0	55.0	55.0	57.0	57.0	58.0	60.0	61.0	63.0	64.0	65.0	64.0	62.0	62.0	
27	500°	56.0	53.0	53.0	53.0	54.0	55.0	55.0	56.0	57.0	59.0	60.0	62.0	63.0	63.0	62.0	63.0	60.0	
28	630°	55.0	54.0	52.0	52.0	53.0	53.0	54.0	55.0	56.0	57.0	59.0	60.0	61.0	61.0	60.0	54.0	58.0	
29	800°	55.0	51.0	51.0	51.0	52.0	52.0	53.0	54.0	55.0	56.0	57.0	59.0	60.0	59.0	57.0	57.0	57.0	
30	1000°	54.0	50.0	49.0	50.0	51.0	52.0	53.0	53.0	54.0	56.0	57.0	58.0	57.0	55.0	55.0	55.0	55.0	
31	1250°	53.0	49.0	48.0	48.0	49.0	50.0	50.0	51.0	52.0	53.0	54.0	56.0	56.0	55.0	53.0	53.0	53.0	
32	1600°	52.0	48.0	47.0	47.0	48.0	49.0	50.0	51.0	52.0	53.0	54.0	55.0	55.0	51.0	51.0	51.0		
33	2000°	51.0	46.0	46.0	46.0	46.0	47.0	48.0	49.0	50.0	52.0	53.0	53.0	52.0	50.0	50.0	50.0		
34	2500°	49.0	45.0	44.0	44.0	45.0	46.0	46.0	47.0	48.0	49.0	49.0	50.0	51.0	50.0	48.0	48.0		
35	3150°	48.0	44.0	43.0	43.0	44.0	44.0	45.0	45.0	46.0	46.0	47.0	48.0	49.0	49.0	46.0	46.0		
36	4000°	47.0	42.0	42.0	42.0	43.0	43.0	44.0	44.0	45.0	46.0	47.0	48.0	48.0	47.0	45.0	45.0		
37	5000°	45.0	41.0	41.0	41.0	41.0	42.0	43.0	44.0	45.0	46.0	47.0	48.0	47.0	46.0	44.0	43.0		
38	6000°	44.0	40.0	40.0	40.0	40.0	41.0	42.0	43.0	44.0	45.0	46.0	47.0	46.0	45.0	42.0	42.0		
39	8000°	44.0	39.0	39.0	39.0	39.0	40.0	41.0	42.0	43.0	44.0	45.0	46.0	44.0	43.0	41.0	41.0		
40	10000°	43.0	39.0	38.0	38.0	39.0	39.0	40.0	41.0	42.0	43.0	44.0	45.0	44.0	43.0	40.0	40.0		

BAND	ANGLE	JET										PRIM							TAKEOFF
		10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	170.	
17	50°	80.0	79.0	79.0	80.0	81.0	82.0	82.0	84.0	85.0	87.0	89.0	92.0	96.0	100.0	104.0	102.0	102.0	
18	63°	82.0	81.0	81.0	82.0	84.0	84.0	86.0	87.0	89.0	91.0	94.0	97.0	101.0	104.0	102.0	102.0		
19	80°	85.0	83.0	83.0	84.0	85.0	86.0	87.0	87.0	89.0	90.0	92.0	94.0	96.0	99.0	101.0	104.0	102.0	
20	100°	86.0	84.0	85.0	85.0	86.0	87.0	87.0	89.0	90.0	92.0	94.0	96.0	99.0	102.0	104.0	95.0	95.0	
21	125°	88.0	86.0	86.0	86.0	87.0	87.0	88.0	89.0	90.0	91.0	93.0	96.0	97.0	99.0	102.0	103.0	101.0	
22	140°	86.0	86.0	86.0	87.0	87.0	88.0	88.0	89.0	90.0	92.0	93.0	95.0	97.0	99.0	101.0	103.0	100.0	
23	200°	89.0	87.0	87.0	87.0	88.0	89.0	89.0	90.0	91.0	92.0	95.0	97.0	99.0	101.0	103.0	99.0	99.0	
24	250°	84.0	87.0	87.0	87.0	88.0	89.0	89.0	90.0	92.0	93.0	95.0	96.0	98.0	100.0	99.0	97.0	97.0	
25	315°	84.0	87.0	86.0	87.0	87.0	88.0	88.0	89.0	90.0	91.0	93.0	94.0	96.0	98.0	99.0	97.0	96.0	
26	400°	86.0	86.0	86.0	86.0	86.0	87.0	88.0	89.0	90.0	91.0	92.0	93.0	95.0	97.0	97.0	96.0	96.0	
27	500°	84.0	86.0	85.0	86.0	86.0	87.0	87.0	88.0	89.0	90.0	91.0	92.0	94.0	95.0	95.0	94.0	93.0	
28	630°	84.0	85.0	85.0	85.0	85.0	86.0	86.0	87.0	88.0	89.0	90.0	91.0	94.0	94.0	92.0	91.0	91.0	
29	800°	85.0	84.0	84.0	84.0	84.0	85.0	85.0	86.0	87.0	88.0	89.0	90.0	92.0	92.0	90.0	89.0	89.0	
30	1000°	87.0	87.0	83.0	83.0	83.0	84.0	84.0	85.0	86.0	87.0	89.0	90.0	91.0	90.0	88.0	88.0	88.0	
31	1250°	86.0	82.0	81.0	82.0	82.0	83.0	84.0	85.0	86.0	87.0	89.0	89.0	88.0	87.0	86.0	86.0	86.0	
32	1600°	85.0	81.0	80.0	80.0	81.0	81.0	82.0	83.0	84.0	85.0	86.0	87.0	87.0	86.0	85.0	84.0	84.0	
33	2000°	83.0	79.0	79.0	79.0	80.0	81.0	81.0	82.0	83.0	84.0	85.0	86.0	86.0	85.0	83.0	82.0	82.0	
34	2400°	82.0	78.0	77.0	77.0	78.0	79.0	79.0	80.0	81.0	82.0	83.0	84.0	84.0	83.0	81.0	80.0	80.0	
35	3150°	80.0	76.0	76.0	76.0	77.0	77.0	78.0	79.0	79.0	80.0	82.0	83.0	82.0	81.0	79.0	79.0	79.0	
36	4000°	79.0	75.0	74.0	74.0	75.0	76.0	77.0	78.0	79.0	80.0	80.0	81.0	80.0	79.0	77.0	77.0	77.0	
37	5000°	73.0	73.0	73.0	73.0	73.0	74.0	75.0	76.0	77.0	78.0	79.0	80.0	80.0	79.0	76.0	76.0	76.0	
38	6000°	76.0	72.0	72.0	72.0	72.0	73.0	74.0	75.0	76.0	77.0	78.0	79.0	78.0	77.0	74.0	74.0	74.0	
39	8000°	75.0	71.0	71.0	71.0	71.0	72.0	73.0	74.0	75.0	76.0	77.0	78.0	76.0	75.0	73.0	73.0	73.0	
40	10000°	74.0	71.0	70.0	71.0	71.0	72.0	72.0	73.0	74.0	75.0	76.0	77.0	76.0	75.0	72.0	72.0	72.0	

TABLE 14  
SECONDARY JET NOISE SPECTRUM (200-FT. POLAR)

		JET								SEC								APPROACH	
BAND	ANGLE	10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°	
	FREQ.																		
17	50°	64.9	78.9	66.4	62.9	71.9	62.9	63.9	65.9	67.9	67.9	69.9	70.9	72.9	74.9	74.9	76.9		
18	63°	64.9	78.9	66.4	62.9	71.9	62.9	63.9	65.9	67.9	67.9	69.9	70.9	72.9	74.9	74.9	76.9		
19	80°	64.9	78.9	66.4	62.9	71.9	62.9	63.9	65.9	67.9	67.9	69.9	70.9	72.9	74.9	74.9	76.9		
20	100°	64.9	78.9	66.4	62.9	71.9	62.9	63.9	65.9	67.9	67.9	69.9	70.9	72.9	74.9	74.9	76.9		
21	125°	63.4	77.4	65.4	61.4	70.4	61.4	62.4	64.4	66.4	66.4	68.4	69.4	71.4	73.4	73.4	75.4		
22	160°	61.9	75.9	63.9	59.4	68.4	59.4	60.4	60.9	62.9	64.4	64.9	66.4	67.9	69.9	69.9	69.9		
23	200°	60.4	74.4	62.4	56.4	67.4	58.4	59.4	59.4	61.4	63.4	65.4	66.4	68.4	70.4	70.4	70.4		
24	250°	58.9	72.9	60.9	56.9	65.9	56.9	57.9	57.9	59.4	61.4	61.9	63.9	64.9	66.9	68.9	68.9		
25	315°	57.4	71.4	59.4	55.4	64.4	55.5	56.4	56.4	58.4	60.4	60.4	62.4	63.4	65.4	67.4	67.4		
26	400°	55.9	69.9	57.9	53.9	62.9	53.9	54.9	54.9	56.4	58.4	58.9	61.4	63.9	65.9	65.9	65.9		
27	500°	54.4	68.4	56.4	52.4	61.4	52.4	53.4	53.4	55.4	57.4	57.4	59.4	60.4	62.4	64.4	64.4		
28	630°	52.9	66.9	54.9	50.9	59.9	50.9	51.9	51.9	53.9	55.9	55.9	57.9	58.9	60.9	62.9	62.9		
29	800°	51.4	65.4	53.4	49.4	58.4	49.4	50.4	50.4	52.4	54.4	54.4	56.4	57.4	59.4	61.4	61.4		
30	1010°	49.9	63.9	51.9	47.9	56.9	47.9	48.9	48.9	50.9	52.9	52.9	55.9	57.9	59.9	59.9	59.9		
31	1250°	48.4	62.4	50.4	46.4	46.4	47.4	47.4	49.4	51.4	51.4	53.4	54.4	56.4	58.4	58.4	58.4		
32	1600°	46.9	60.9	48.4	44.9	53.9	44.9	45.9	45.9	47.9	49.9	49.9	51.9	52.9	54.9	56.9	56.9		
33	2000°	45.4	59.4	47.4	43.4	52.4	43.4	44.4	44.4	46.4	48.4	50.4	51.4	53.4	55.4	55.4	55.4		
34	2500°	43.9	57.9	45.9	41.9	50.9	41.9	42.9	42.9	44.9	46.9	48.9	51.9	53.9	53.9	53.9			
35	3150°	42.4	56.4	44.4	40.4	49.4	40.4	41.4	41.4	43.4	45.4	47.4	48.4	50.4	52.4	52.4			
36	4000°	40.9	54.9	42.9	38.9	47.9	38.9	39.9	39.9	41.9	43.9	45.9	46.9	48.9	50.9	50.9			
37	5000°	39.4	53.4	41.4	37.4	46.4	37.4	38.4	38.4	40.4	42.4	42.4	44.4	47.4	49.4	49.4			
38	6300°	37.9	51.9	39.9	35.9	44.9	35.9	36.9	36.9	38.4	40.4	40.4	42.9	43.9	45.9	47.9	47.9		
39	8000°	36.4	50.4	34.4	32.4	43.4	34.4	35.4	35.4	37.4	39.4	39.4	41.4	42.4	44.4	46.4	46.4		
40	10000°	34.9	48.4	36.4	32.9	41.9	32.9	33.9	33.9	35.9	37.9	37.9	39.9	40.9	42.9	44.9	44.9		

		JET								SEC								TAKEOFF	
BAND	ANGLE	10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	120°	130°	140°	150°	160°	170°	
	FREQ.																		
17	50°	76.9	81.9	74.9	76.9	80.4	75.9	75.9	78.9	79.9	81.9	83.9	85.9	88.9	90.9	90.9	90.9		
18	63°	76.9	81.9	74.9	76.9	80.9	75.9	75.9	78.9	79.9	81.9	83.9	85.9	88.9	90.9	90.9	90.9		
19	80°	76.9	81.9	74.9	74.9	80.9	75.9	75.9	78.9	79.9	81.9	83.9	85.9	88.9	90.9	90.9	90.9		
20	100°	76.9	81.9	74.9	74.9	80.9	75.9	75.9	78.9	79.9	81.9	83.9	85.9	88.9	90.9	90.9	90.9		
21	125°	75.4	80.4	73.4	73.4	79.4	74.4	74.4	79.4	79.4	82.4	84.4	87.4	89.4	90.4	90.4	90.4		
22	160°	73.9	78.4	71.9	71.9	77.9	72.9	72.9	72.9	72.9	76.9	78.9	80.9	82.9	85.9	87.9	87.9		
23	200°	72.4	77.4	70.4	70.4	76.4	71.4	71.4	71.4	76.4	75.4	77.4	79.4	81.4	84.4	86.4	86.4		
24	250°	70.9	75.9	68.9	68.9	74.9	63.9	63.9	64.9	64.9	73.9	75.9	77.9	79.9	82.9	84.9	86.9		
25	315°	69.4	74.4	67.4	67.4	73.4	68.4	68.4	69.4	69.4	73.4	74.4	76.4	78.4	81.4	83.4	83.4		
26	400°	67.9	72.9	65.9	65.9	71.9	66.9	66.9	66.9	71.9	70.9	72.9	74.9	76.9	79.9	81.9	81.9		
27	500°	66.4	71.4	64.4	64.4	70.4	65.4	65.4	65.4	70.4	69.4	71.4	73.4	75.4	78.4	80.4	80.4		
28	630°	64.9	69.9	62.4	62.4	68.4	63.4	63.4	63.4	68.4	67.4	69.4	71.4	73.4	76.4	78.4	78.4		
29	800°	63.4	68.4	61.4	61.4	67.4	62.4	62.4	62.4	67.4	66.4	68.4	70.4	72.4	75.4	77.4	77.4		
30	1000°	61.9	66.9	59.4	59.4	65.9	60.9	60.9	60.9	65.9	64.4	65.9	68.9	70.9	73.9	75.9	75.9		
31	1250°	60.4	65.4	58.4	58.4	64.4	57.4	57.4	59.4	64.4	63.4	65.4	67.4	69.4	72.4	74.4	74.4		
32	1600°	59.9	63.9	56.9	56.9	62.9	57.9	57.9	57.9	62.9	61.9	63.9	65.9	67.9	70.9	72.9	72.9		
33	2000°	57.4	62.4	55.4	55.4	61.4	56.4	56.4	56.4	61.4	60.4	62.4	64.4	66.4	69.4	71.4	71.4		
34	2500°	55.9	60.4	53.9	53.9	59.4	54.9	54.9	54.9	59.4	58.4	60.4	62.9	64.9	67.4	69.4	69.4		
35	3150°	54.4	59.4	52.4	52.4	58.4	53.4	53.4	53.4	58.4	57.4	59.4	61.4	64.4	66.4	68.4	68.4		
36	4000°	52.9	57.9	50.9	50.9	56.9	51.9	51.9	51.9	56.9	55.9	57.9	59.9	61.9	65.9	66.9	66.9		
37	5000°	51.4	56.4	49.4	49.4	55.4	50.4	50.4	50.4	55.4	54.4	56.4	58.4	60.4	63.4	65.4	65.4		
38	6300°	49.9	54.9	47.9	47.9	51.9	45.9	45.9	45.9	51.9	50.9	52.9	54.9	57.9	61.9	64.9	64.9		
39	8000°	48.4	53.4	46.4	46.4	52.4	47.4	47.4	47.4	52.4	51.4	53.4	56.4	59.4	62.4	65.4	65.4		
40	10000°	46.9	51.9	44.9	44.9	50.9	45.9	45.9	45.9	50.9	49.4	51.4	53.4	56.4	60.4	63.4	63.4		

TABLE 15  
TURBOMACHINERY NOISE SPECTRUM (200-FT. POLAR)

S/N#	FREQ.	ANGLE	TURB															APPROACH	
			10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	
17	50.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
18	63.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
19	80.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
20	100.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
21	125.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
22	160.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
23	200.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
24	250.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
25	312.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
26	400.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
27	500.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
28	630.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
29	800.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
30	1000.	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0	10.0
31	1250.	10.0	10.0	10.0	10.0	11.0	14.0	18.0	23.0	28.0	34.0	40.0	48.0	50.0	42.0	36.0	24.0	12.0	12.0
32	1600.	10.0	10.0	10.0	10.0	12.0	16.0	20.0	26.0	31.0	38.0	50.0	51.0	53.0	50.0	34.0	21.0	12.0	12.0
33	2000.	10.0	10.0	10.0	13.0	18.0	23.0	27.0	32.0	39.0	42.0	53.0	54.0	56.0	54.0	42.0	32.0	23.0	23.0
34	2500.	10.0	10.0	10.0	14.0	20.0	25.0	32.0	38.0	40.0	55.0	57.0	60.0	53.0	46.0	38.0	32.0	25.0	25.0
35	3120.	10.0	10.0	10.0	15.0	21.0	28.0	35.0	42.0	50.0	59.0	61.0	63.0	61.0	50.0	42.0	35.0	28.0	28.0
36	4000.	10.0	10.0	10.0	16.0	23.0	30.0	38.0	45.0	54.0	60.0	64.0	69.0	67.0	54.0	45.0	38.0	30.0	30.0
37	5000.	10.0	10.0	10.0	17.0	25.0	33.0	41.0	49.0	58.0	70.0	69.0	74.0	72.0	58.0	49.0	41.0	33.0	33.0
38	6300.	10.0	10.0	10.0	18.0	27.0	35.0	44.0	53.0	62.0	76.0	75.0	80.0	78.0	62.0	53.0	44.0	35.0	35.0
39	8000.	10.0	10.0	10.0	19.0	28.0	37.0	47.0	56.0	66.0	78.0	81.0	81.0	80.0	66.0	56.0	47.0	37.0	37.0
40	10000.	10.0	10.0	10.0	20.0	30.0	40.0	50.0	60.0	70.0	80.0	80.0	83.0	77.0	70.0	60.0	50.0	40.0	40.0

BREAKDOWN OF CONTROL CARD

1 --- 1 = TIME HEADER CARD	1
2 --- 1 = STANDARD HEADER CARD	1
3 --- 0 = FIXED NOISE SOURCE	0
4 --- 0 = STANDARD UNIN POLAR ANGLE INPUT	0
5 --- NOT USED	0
6 --- 0 = STOELINE INPUT = 1 POLAR INPUT	1
7 --- 4 = STANDARD FORMAT TTT SPL INPUT	4
8 --- NOT USED	0
9 --- 0 = JET TIME CORRECTION	0
10 --- NOT USED	0
11 --- 1 = PUNCHED OUTPUT	1
12-15 ---- NOT USED	0101
16 --- 1 = INVERTED ATT. TERMS INTERNALLY	1
17 --- NOT USED	0
18 --- 1 = READ OUTFROM JET - SUPP. CARD	1
19 --- 1 = SHIFT FREQ OUTPUT	1
20 --- 1 = STANDARD (LONG FORM) OUTPUT	0
21 --- 1 = JET CORE INPUT	1
22 --- 1 = DEVELOP FORTPRINT	0
23 --- 1 = TIME PRINT (GROM SUR)	0
24 --- 1 = DEVELOPING DOPPLER EFFECTS	1
25 --- 1 = FAN NOISE INPUT	1
26 --- 1 = JET INLET SHIFT	1
27 --- 1 = TURBINE NOISE INPUT	1
28 --- 0 = USE GROUND ATTENATION	0
29 --- 1 = USE JET, INLET, FWD DUCT ATTEN.	1
30 --- 1 = GRID GEN. FOR DEVELOPING EPNL CNTRS	0
31-32 --- NO OF INPUT LASERS	1
33-34 --- NO OF POLAR ANGLES	17
35-36 --- 24 = 1/3 OCTAVE DATA	24
37-38 ---	0
39-40 --- NO OF FLIGHT PROFILE SECTIONS	0
43-44 --- NO OF PARAM. INPUTS	0
43-44 ---	0
45-46 --- NO OF MODES SURFED	0
47-48 --- NO OF GRID POINTS	0
49-50 --- NO OF ATTENUATIONS	1
51 --- NO OF ENGINES	4.0

CONTROL CARD AND DATA INPUT

11000140001010110110011110101 11724 0 0 0 0 0 0 1 4.

SPL24=SPL+FAN+JET 2 SPLJ1=SPL JET 1 SPLREF=SPL JET 2  
CONVERTED FROM 100 F 0 200 FT APP 271977

ALITUDE	VELOCITY	HUMIDITY	TEMP	STOELINE	OFFSET	POWER	OVTIM	NJ
200.00	0.0	70.00	59.00	0.0	0.0	0.0	0.0	17
310.00	754.00	70.00	59.00	0.0	0.0			

TABLE 16  
EPNL PROGRAM INPUT (CONTROL CARDS)

ANGLE	FAN												JETZ												
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	
10.00	65.	65.	65.	65.	66.	68.	70.	69.	70.	73.	74.	75.	76.	79.	80.	84.	86.	83.	82.	83.	83.	82.	79.	75.	
20.00	75.	74.	74.	74.	75.	75.	77.	76.	77.	77.	78.	78.	81.	82.	85.	85.	88.	84.	86.	87.	87.	88.	86.	84.	80.
30.00	67.	67.	67.	67.	67.	69.	71.	72.	69.	71.	71.	74.	76.	77.	80.	82.	86.	86.	86.	86.	85.	84.	82.	79.	
40.00	63.	63.	63.	63.	67.	66.	67.	67.	69.	69.	70.	73.	74.	77.	78.	80.	83.	85.	84.	85.	85.	83.	81.	77.	
50.00	72.	72.	72.	72.	72.	72.	73.	72.	75.	73.	74.	73.	76.	76.	79.	80.	81.	85.	90.	85.	86.	84.	84.	80.	
60.00	67.	63.	63.	63.	63.	66.	66.	66.	66.	66.	66.	66.	66.	66.	66.	66.	66.	66.	77.	79.	81.	75.	76.	72.	
70.00	64.	64.	64.	64.	64.	66.	67.	66.	65.	66.	66.	66.	67.	69.	70.	73.	74.	76.	81.	76.	77.	74.	73.	71.	66.
80.00	64.	64.	64.	64.	64.	64.	68.	68.	65.	66.	66.	66.	67.	69.	70.	73.	73.	77.	81.	77.	77.	75.	75.	70.	67.
90.00	66.	66.	66.	66.	66.	67.	67.	67.	66.	68.	68.	71.	72.	76.	78.	80.	83.	86.	81.	81.	78.	77.	74.	68.	
100.00	68.	68.	68.	68.	68.	69.	68.	68.	67.	67.	68.	67.	69.	72.	74.	77.	79.	87.	85.	83.	83.	82.	80.	76.	71.
110.00	68.	68.	68.	68.	68.	70.	68.	68.	70.	69.	70.	71.	73.	74.	78.	80.	82.	86.	83.	84.	84.	81.	80.	78.	74.
120.00	70.	70.	70.	70.	70.	70.	70.	70.	70.	70.	70.	71.	72.	74.	77.	79.	87.	86.	84.	84.	82.	81.	74.	74.	
130.00	71.	71.	71.	71.	71.	71.	71.	71.	71.	70.	71.	72.	74.	76.	79.	80.	83.	87.	84.	84.	84.	83.	81.	76.	
140.00	73.	73.	73.	73.	73.	73.	73.	73.	73.	71.	71.	72.	73.	73.	75.	79.	80.	82.	86.	83.	82.	82.	80.	75.	
150.00	75.	75.	75.	75.	75.	75.	75.	75.	75.	71.	70.	70.	71.	71.	74.	76.	79.	82.	76.	77.	78.	77.	75.	70.	
160.00	75.	75.	75.	75.	75.	75.	75.	75.	75.	71.	71.	70.	68.	69.	69.	69.	72.	74.	74.	75.	73.	73.	71.	67.	
170.00	75.	75.	75.	75.	75.	75.	75.	75.	75.	71.	71.	70.	68.	68.	68.	69.	72.	74.	76.	79.	74.	75.	73.	71.	67.

ANGLE	JET												PRIM												
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	
10.00	55.	57.	58.	59.	59.	58.	58.	58.	58.	58.	57.	56.	55.	55.	54.	53.	52.	51.	49.	48.	47.	45.	44.	44.	43.
20.00	54.	55.	57.	57.	57.	57.	57.	57.	57.	57.	56.	55.	54.	53.	52.	51.	50.	49.	48.	48.	45.	44.	42.	41.	40.
30.00	55.	56.	57.	57.	57.	57.	56.	56.	56.	56.	55.	54.	53.	52.	51.	49.	48.	47.	46.	46.	43.	42.	41.	40.	39.
40.00	56.	56.	57.	57.	58.	57.	57.	57.	56.	55.	54.	53.	52.	51.	50.	49.	47.	46.	46.	44.	43.	42.	41.	40.	39.
50.00	57.	57.	58.	58.	58.	58.	58.	58.	58.	57.	56.	55.	54.	53.	52.	50.	49.	47.	46.	45.	44.	43.	41.	40.	39.
60.00	57.	58.	58.	58.	58.	58.	58.	58.	58.	58.	57.	56.	55.	54.	53.	52.	51.	50.	49.	48.	47.	46.	45.	43.	42.
70.00	58.	59.	60.	60.	60.	59.	58.	58.	58.	57.	56.	55.	54.	53.	52.	50.	49.	48.	46.	45.	44.	43.	42.	41.	40.
80.00	55.	60.	61.	61.	61.	61.	60.	59.	58.	57.	57.	57.	57.	57.	57.	57.	57.	57.	56.	56.	56.	55.	54.	53.	52.
90.00	61.	61.	62.	62.	62.	62.	62.	61.	61.	60.	58.	57.	56.	55.	53.	52.	51.	49.	48.	47.	46.	45.	44.	43.	42.
100.00	63.	63.	63.	64.	64.	64.	64.	63.	62.	61.	60.	58.	57.	56.	54.	53.	52.	50.	49.	48.	47.	46.	45.	44.	43.
110.00	65.	65.	66.	66.	66.	66.	65.	65.	64.	63.	61.	60.	59.	57.	56.	54.	53.	52.	50.	49.	48.	47.	46.	45.	44.
120.00	68.	68.	68.	69.	69.	69.	69.	67.	65.	64.	63.	62.	60.	59.	57.	56.	54.	53.	51.	50.	49.	48.	47.	46.	45.
130.00	71.	71.	71.	71.	71.	70.	69.	67.	66.	64.	63.	61.	60.	58.	56.	54.	53.	51.	49.	48.	47.	46.	44.	44.	44.
140.00	75.	75.	75.	74.	73.	73.	72.	71.	69.	67.	67.	63.	61.	59.	57.	55.	53.	52.	50.	49.	47.	46.	45.	43.	43.
150.00	80.	79.	77.	76.	74.	72.	70.	68.	66.	64.	62.	60.	57.	55.	53.	51.	50.	48.	46.	45.	44.	42.	41.	40.	40.
160.00	75.	75.	75.	74.	72.	70.	68.	66.	64.	62.	60.	58.	57.	55.	53.	51.	50.	48.	46.	45.	43.	42.	41.	40.	40.
170.00	78.	77.	75.	74.	72.	70.	68.	66.	64.	62.	60.	58.	57.	55.	53.	51.	50.	48.	46.	45.	43.	42.	41.	40.	40.

ANGLE	JET												SEC												
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	
10.00	65.	65.	55.	65.	63.	62.	60.	54.	57.	55.	54.	53.	51.	50.	48.	47.	45.	44.	42.	41.	35.	36.	35.		
20.00	74.	74.	74.	74.	77.	76.	74.	73.	71.	70.	68.	67.	65.	64.	62.	61.	59.	58.	56.	55.	53.	52.	50.	49.	
30.00	67.	67.	67.	67.	67.	65.	64.	62.	61.	59.	58.	56.	55.	53.	52.	50.	49.	47.	46.	44.	43.	41.	40.	38.	
40.00	63.	53.	53.	63.	61.	50.	54.	57.	55.	54.	52.	51.	49.	48.	46.	45.	43.	42.	40.	39.	37.	36.	34.	33.	
50.00	72.	72.	72.	72.	70.	69.	67.	66.	64.	63.	61.	60.	58.	57.	55.	54.	52.	51.	49.	48.	46.	45.	43.	42.	
60.00	63.	63.	63.	63.	61.	60.	58.	57.	55.	54.	52.	51.	49.	48.	46.	45.	43.	42.	40.	39.	37.	36.	34.	33.	
70.00	64.	64.	64.	64.	62.	61.	59.	58.	56.	55.	54.	52.	50.	49.	47.	46.	44.	43.	41.	40.	38.	37.	35.	34.	
80.00	64.	64.	64.	64.	62.	61.	59.	58.	56.	55.	53.	52.	50.	49.	47.	46.	44.	43.	41.	40.	38.	37.	35.	34.	
90.00	66.	66.	66.	66.	66.	65.	63.	61.	60.	58.	57.	56.	54.	52.	51.	49.	48.	46.	45.	43.	42.	40.	39.	37.	36.
100.00	68.	68.	68.	68.	66.	66.	63.	63.	62.	60.	59.	57.	56.	54.	53.	51.	50.	48.	47.	45.	44.	42.	41.	39.	38.
110.00	68.	68.	68.	68.	67.	67.	65.	63.	62.	60.	59.	57.	56.	54.	53.	51.	50.	48.	47.	45.	44.	42.	41.	39.	38.
120.00	70.	70.	70.	70.	68.	67.	65.	64.	62.	61.	59.	58.	56.	55.	53.	52.	50.	49.	47.	46.	44.	43.	41.	40.	
130.00	71.	71.	71.	71.	71.	69.	68.	66.	65.	63.	62.	60.	57.	56.	54.	53.	51.	50.	48.	47.	45.	44.	42.	41.	
140.00	73.	73.	73.	73.	71.	70.	69.	67.	65.	64.	62.	61.	59.	58.	56.	55.	53.	52.	50.	49.	47.	46.	44.	43.	
150.00	75.	75.	75.	75.	73.	72.	70.	69.	67.	66.	64.	63.	61.	60.	58.	57.	55.	54.	52.	51.	49.	48.	46.	45.	
160.00	75.	75.	75.	75.	75.	73.	72.	70.	69.	67.	65.	63.	61.	60.	58.	57.	55.	54.	52.	51.	49.	48.	46.	45.	
170.00	75.	75.	75.	75.	73.	72.	70.	69.	67.	66.	64.	63.	61.	60.	58.	57.	55.	54.	52.	51.	49.	48.	46.	45.	

**TABLE 17**  
**EPNL PROGRAM INPUT**  
**(FAN, PRIMARY AND SECONDARY JET NOISE SPECTRUM)**



INLET	FANDET	JET	TEMP	ALTITUDE	VEL	TREL HUM	TEMP	SIDELINE	OFFSET	GAMMA	ALPHA	DURATION	DEPNL	
1.0	0.0	0.0	20.0	4.465	95.13	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.73	2.67
5.0	0.0	0.0	20.0	4.465	95.23	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.53	3.57
10.0	0.0	0.0	20.0	4.465	95.38	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.58	3.92
20.0	0.0	0.0	20.0	4.465	95.58	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.26	5.21
30.0	0.0	0.0	20.0	4.465	95.80	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.81	5.30
0.0	0.0	5.0	20.0	4.465	95.87	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.43	3.18
5.0	0.0	5.0	20.0	4.465	95.97	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.17	4.21
10.0	0.0	5.0	20.0	4.465	95.17	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.20	4.63
20.0	0.0	5.0	20.0	4.465	95.58	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.39	6.12
30.0	0.0	5.0	20.0	4.465	95.59	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.65	6.21
0.0	0.0	10.0	20.0	4.465	95.78	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.16	3.52
5.0	0.0	10.0	20.0	4.465	95.19	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.94	4.61
10.0	0.0	10.0	20.0	4.465	95.70	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.96	5.10
20.0	0.0	10.0	20.0	4.465	95.98	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.28	6.92
30.0	0.0	10.0	20.0	4.465	95.77	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.56	7.03
0.0	0.0	20.0	20.0	4.465	95.93	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.81	3.87
5.0	0.0	20.0	20.0	4.465	95.68	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.48	5.12
10.0	0.0	20.0	20.0	4.465	95.11	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.47	5.69
20.0	0.0	20.0	20.0	4.465	95.89	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.68	7.91
30.0	0.0	20.0	20.0	4.465	95.48	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.93	8.12
0.0	0.0	30.0	20.0	4.465	95.00	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.82	3.80
5.0	0.0	30.0	20.0	4.465	95.76	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.47	5.04
10.0	0.0	30.0	20.0	4.465	95.17	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.45	5.63
20.0	0.0	30.0	20.0	4.465	95.11	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.58	7.69
30.0	0.0	30.0	20.0	4.465	95.71	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.86	7.89
0.0	5.0	0.0	20.0	4.465	95.73	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.32	3.07
5.0	0.0	0.0	20.0	4.465	95.68	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.10	4.12
10.0	0.0	0.0	20.0	4.465	95.14	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.88	4.61
20.0	0.0	0.0	20.0	4.465	95.47	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.72	5.32
30.0	0.0	0.0	20.0	4.465	95.53	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.77	5.27
0.0	5.0	5.0	20.0	4.465	95.14	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.97	3.66
5.0	5.0	5.0	20.0	4.465	95.44	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.70	4.86
10.0	5.0	5.0	20.0	4.465	95.35	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.44	5.45
20.0	5.0	5.0	20.0	4.465	95.56	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.75	6.24
30.0	5.0	5.0	20.0	4.465	95.62	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.80	6.18
0.0	5.0	10.0	20.0	4.465	95.79	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.73	4.01
5.0	5.0	10.0	20.0	4.465	95.43	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.45	5.37
10.0	5.0	10.0	20.0	4.465	95.77	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.19	6.03
20.0	5.0	10.0	20.0	4.465	95.35	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.02	6.96
30.0	5.0	10.0	20.0	4.465	95.83	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.81	6.97
0.0	5.0	20.0	20.0	4.465	95.38	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.29	4.42
5.0	5.0	20.0	20.0	4.465	95.85	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.94	5.95
10.0	5.0	20.0	20.0	4.465	95.24	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.65	6.76
20.0	5.0	20.0	20.0	4.465	95.81	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.38	7.99
30.0	5.0	20.0	20.0	4.465	95.74	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.18	6.06
0.0	5.0	30.0	20.0	4.465	95.45	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.29	4.35
5.0	5.0	30.0	20.0	4.465	95.93	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.91	5.87
10.0	5.0	30.0	20.0	4.465	95.11	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.61	6.69
20.0	5.0	30.0	20.0	4.465	95.00	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.29	7.80
30.0	5.0	30.0	20.0	4.465	95.93	270.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.10	7.87
0.0	10.0	0.0	20.0	4.465	95.37	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.58	3.43
5.0	10.0	0.0	20.0	4.465	95.09	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.42	4.71
10.0	10.0	0.0	20.0	4.465	95.14	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.73	5.41
20.0	10.0	0.0	20.0	4.465	95.05	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.05	5.85
30.0	10.0	0.0	20.0	4.465	95.71	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.66	5.89
0.0	10.0	5.0	20.0	4.465	95.73	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.21	4.07
5.0	10.0	5.0	20.0	4.465	95.35	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.32	5.45
10.0	10.0	5.0	20.0	4.465	95.45	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.79	6.35
20.0	10.0	5.0	20.0	4.465	95.75	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.96	6.95
30.0	10.0	5.0	20.0	4.465	95.92	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	4.99	6.98
0.0	10.0	10.0	20.0	4.465	95.35	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.97	4.45
5.0	10.0	10.0	20.0	4.465	95.75	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	6.06	6.05
10.0	10.0	10.0	20.0	4.465	95.73	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.57	7.07
20.0	10.0	10.0	20.0	4.465	95.99	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.27	7.81
30.0	10.0	10.0	20.0	4.465	95.96	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.37	7.86
0.0	10.0	20.0	20.0	4.465	95.89	370.00	254.00	70.00	59.00	0.0	0. -3.00	3.70	5.49	4.91

TABLE 19a.  
EPNL PROGRAM OUTPUT





**APPENDIX 1  
EPNL PROGRAM DESCRIPTION**

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## PROGRAM DESCRIPTION

The object of this computer program is to use aircraft sound data to estimate the time pattern at other hypothetical observer locations and convert the projected sound into subjective units. The time history of an aircraft flyby is simulated using either stationary ground runup data, flyover data measured at other than requested conditions, or predicted data. Acceptable input consists of one-third octave band sound pressure levels (SPL) as a function of: (1) directivity angle at a constant polar radius, or (2) directivity angle at a constant sideline distance. Since these input spectra may be derived by other prediction methods, or may be actual test measurements, this computer program will be useful in all phases of aircraft development from preliminary design to final test. Output consists of SPL spectra for both the input and projected conditions with the results presented in terms of perceived noise level (PNL), tone corrected PNL (TCPNL), peak TCPNL, effective perceived noise level (EPNL, as defined by the FAA), and overall sound pressure level (OASPL).

The simulation technique consists of applying various corrections to the input SPL spectra to obtain the projected SPL spectra. Corrections due to the following effects are determined by this program: (1) spherical divergence, (2) atmospheric absorption, (3) extra ground attenuation, and (4) number of engines. Also, the program has the options to determine: (1) doppler shift effects and (2) tone corrected PNL and effective perceived noise level.

State-of-the-art techniques of sound calculation are incorporated in such a way that if the calculation procedures change and become standardized, this program can be easily modified. The basic idea is to control the correction effects by selection of the calculation techniques used. A program user controls the selection of computation by means of an input control card which selects the sequence of calculation subroutines. The subroutines are in a general form such that the conditions supplied by input determine the value of the corrections to be applied. Variations in noise characteristics might be achieved for example by considering doppler shift effects and extra ground attenuation for one simulation and neglecting these effects for another simulation.

### Input Data

The input data can consist of three basic types of noise spectra: (1) SPL measurements on a polar arc around a stationary engine, (2) SPL measurements on a sideline parallel to the stationary engine, and (3) SPL measurements at a given location during an aircraft flyby. All data inputting is handled by the main program.

In discussing the input data, a set consists of the SPL measurements for one operating condition, and a case consists of one to four sets of data. A computer run consists of an unlimited number of cases and is terminated by an end-of-file card. This section of the report describes the data input cards in the sequence necessary for proper program execution. Unless specified in the writeup, the cards are needed only when indicated on the control card. A description of each storage array, where the numbers in parenthesis denote the dimensions of the array, is presented in the section on the main program.

### Control Card

The first of each case must be a control card. Columns (1-30) are 30 one-digit indicators which

control the sequence of program execution and are stored in (MF(30)) in order. Columns (31–50) are 10 two-digit indicators which also control the sequence of execution and are stored in (MN(10)) in order. Column (51) is an indicator for the number of engines to be considered. In the following description of the indicators in Columns (1–50), blank values are considered as zeros and indicate no special action is to be taken.

<u>Column</u>	<u>Description</u>
(3)	(MF(3)) – For a value less than 0, the program expects the SPL input as a function of time. A value of 1 indicates one-quarter second input and a value of 2 indicates one-half second input.
(4)	(MF(4)) – For a value of 4, the program expects polar angle input using a format (F10., 12F5./10X, 12F5.).
(6) IN	(MF(6)) – For any nonzero value, the program expects the SPL input as a function of a constant polar distance. For a zero value and also a zero value in (MF(3)), the program expects the SPL input as a function of a constant sideline distance.
(9)	(MF(9)) – For a zero value determine the pure tone correction.
(10)	(MF(10)) – For a value of 1, apply engine shielding effect.
(11)	(MF(11)) – For a value of 1, generate card output for EPNL trade study.
(16)	(MF(16)) – Generate the attenuation trades internally.
(17)	(MF(17)) – For a value of 1, omit the relative jet velocity correction.
(18)	(MF(18)) – For a value of 1, read a card which indicates the uniform jet suppression value.
(19)	(MF(19)) – For a value of 1, use the short form printout.
(20)	(MF(20)) – For a value of 1, use the long form printout.
(21)	(MF(21)) – For a value of 1, the SPLJE array is interpreted as the primary jet noise spectrum.
(22)	(MF(22)) – For a value of 1, develop a footprint output.
(24)	(MF(24)) – For a value of 1, determine the doppler shift effect.
(25)	(MF(25)) – For a value of 1, the SPL24 array is interpreted as the fan noise and secondary jet noise spectrum.
(26)	(MF(26)) – For a value of 1, omit any jet noise correction for irregularities.
(27)	(MF(27)) – For a value of 1, input the turbine noise spectrum.

- (28) (MF(28)) – For any nonzero value, the program suppresses the correction calculation for extra ground attenuation. A value of 1 suppresses the correction for the predicted data. A value of 2 suppresses the correction for both input and predicted data.
- (29) (MF(29)) – For a value of 1, the program expects component noise spectrum. It also expects data cards specifying the peak attenuation which is applied to an attenuation spectrum.
- (30) (MF(30)) – For a value of 1, the program expects a card describing a new condition (such as velocity, altitude, etc.).
- (1) (MN(1)) – Number of input cases.
- (2) (MN(2)) – Number of polar angles.
- (3) (MN(3)) – Number of 1/3 octave bands (24).
- (5) (MN(5)) – Number of flight profile sections.
- (6) (MN(6)) – Number of R.P.M. inputs.
- (8) (MN(8)) – Number of modes summed.
- (9) (MN(9)) – Number of grid line points.
- (10) (MN(10)) – Number of attenuations.

#### Input Condition Card

This card is automatically read (FORMAT 9F8.0) and is the first card of a series of data sets (number of sets indicated by (MN(1))). The numbers in each field are stored in sequence in (H(3,4); V(3,4); HREL (3,4); TEMP (3,4); X(3,4); Y (3,4); GAMM (2); ALP 2; VMJET). Field one is the polar distance (field two = 0) or altitude of the aircraft in feet; field two is the aircraft speed in feet per second; field three is the atmospheric relative humidity in percent; field four is the atmospheric temperature in °F; field five is the sideline distance in feet; field six is the parallel offset distance in feet; field seven is the climb gradient in degrees; field eight is the angle of attack in degrees; field nine is the jet velocity at the aircraft speed in feet per second.

#### Prediction Condition Card

This card is automatically read (FORMAT 9F8.0) and must follow the input condition card. The information on this card (corresponding field for field with the first nine fields of the input condition card) describes the conditions of the projected data.

## Spectral Data Cards

These cards are read according to the format (F10.3, 12F5.2/10X, 12F5.2). Information on these cards is stored in SPL24, SPLJE, SPLREF, and SPLTU. Each spectral set is preceded by a data card containing an identifier, the jet velocity at zero airplane velocity (if applicable), and a title descriptor (FORMAT 2F10., 13A4). After the array SPLJE is read in, the card describing the attenuation spectra are to be read in where CAT is the maximum value in the array ATEN.

## Output

The output from this program consists of printed, plotted, and/or punched parameters. All printing, plotting, and punching of the output data are controlled by the program user by control card indicators.

### Printed Output

At the start of a case, the title, control indicators, and the conditions describing the input and predicted spectra are printed. The input SPL data and the predicted SPL, PNL, OASPL, and EPNL values are automatically printed. IF MF(19) = 1 and MF(20) = 0, a special short form printout is used showing the essential flight and environmental parameters, the peak attenuation values, the time duration factor, the relative change in EPNL, and a counter indicating the number of runs made.

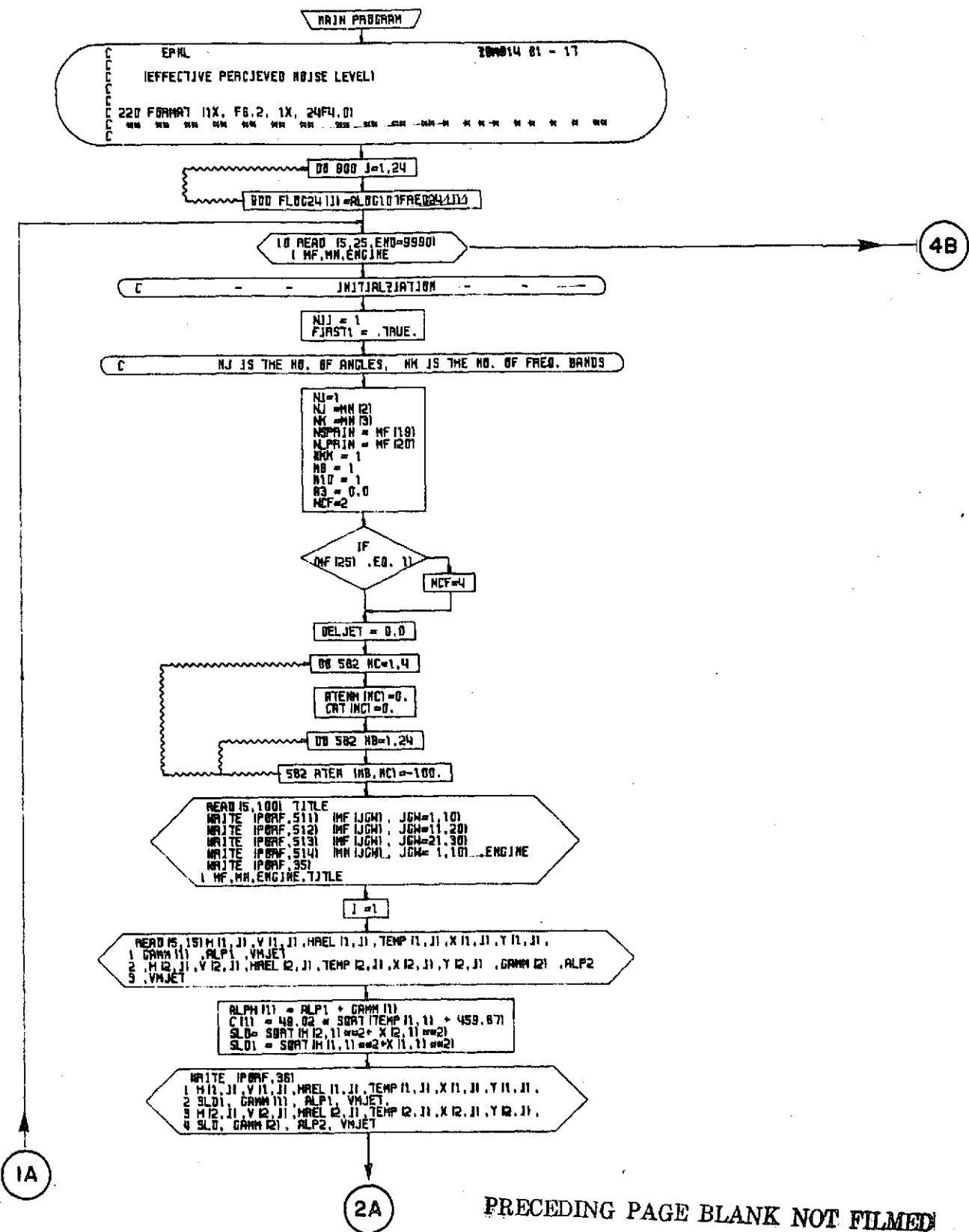
### Punched Output

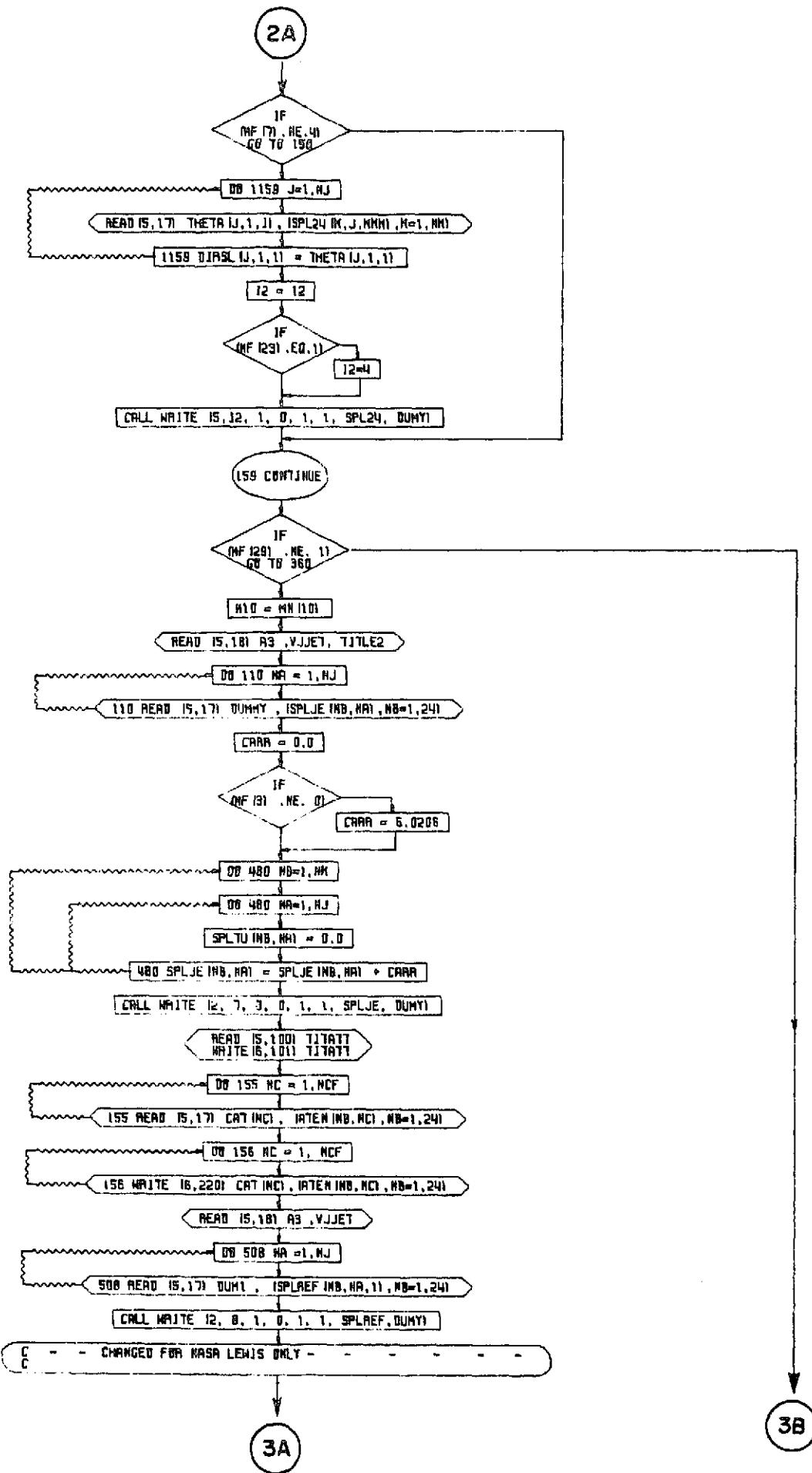
The punched output consists of one card for each run (MF(11) = 1). The card contains the peak attenuations, the jet noise identifier, the EPNL level, the change in EPNL, and a counter indicating the number of runs made.

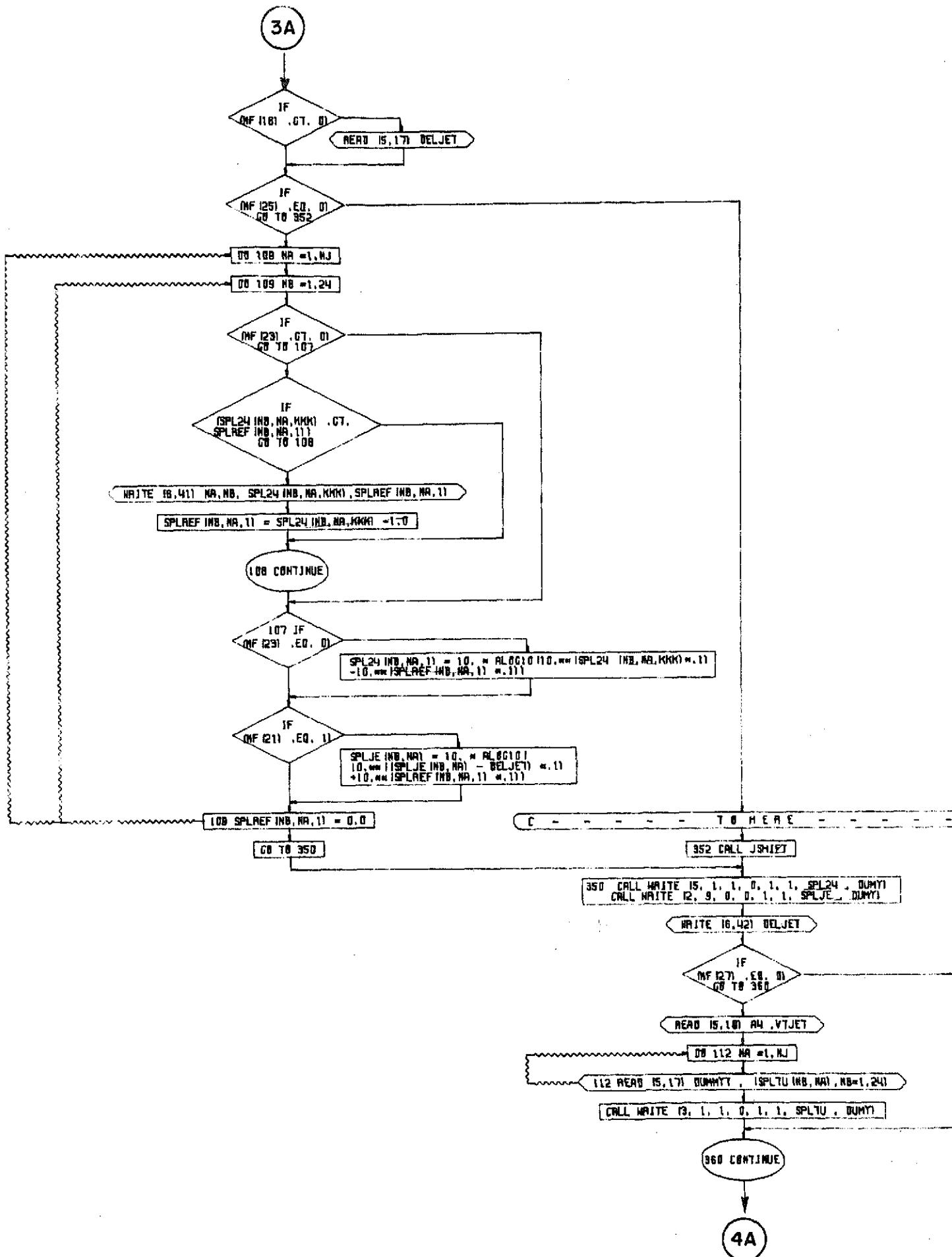
## Main Control Program

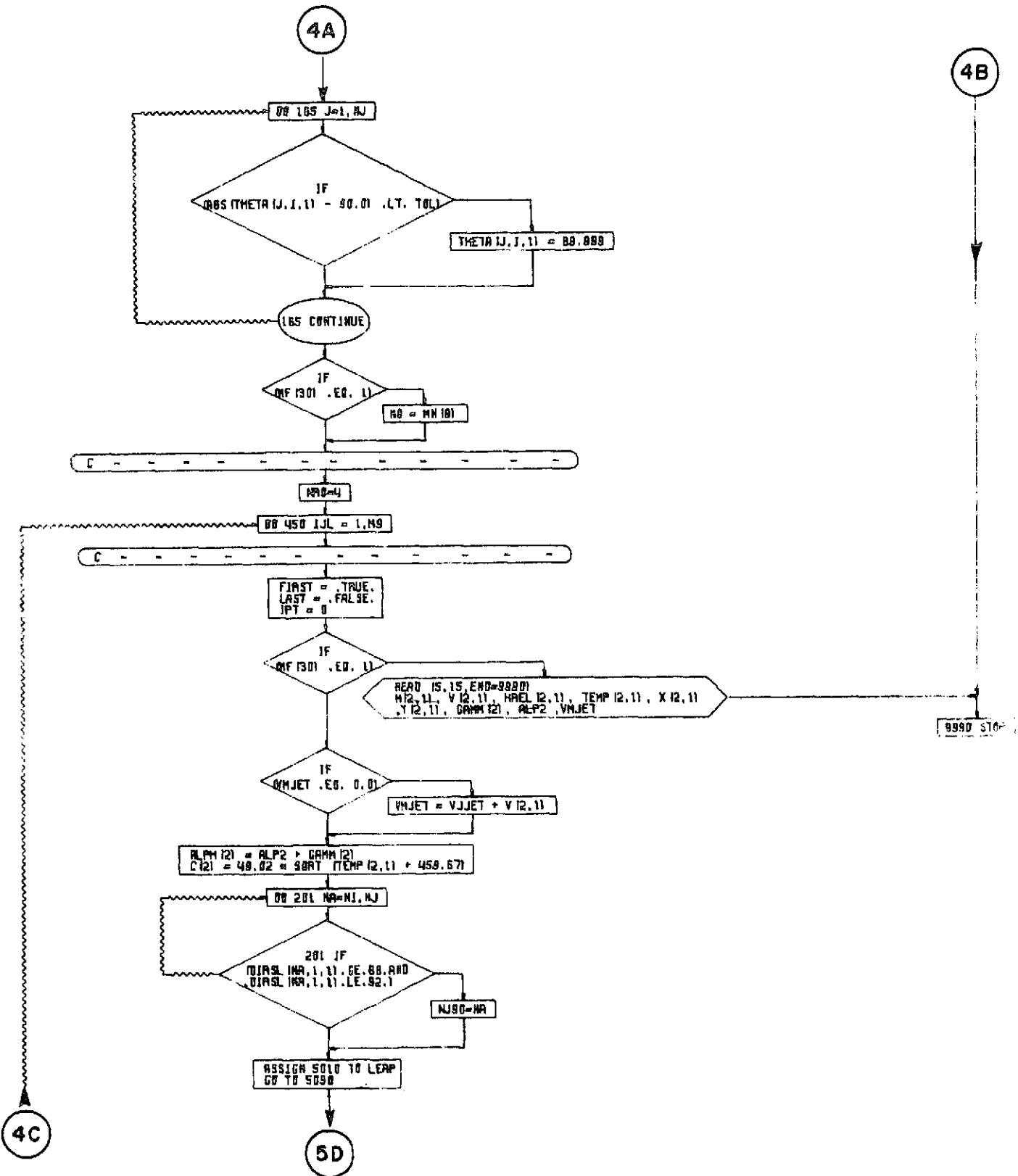
The main control program has general control over input, order of solution, and output throughout the computer run. This is done by calling into operation specialized subroutines in a controlled logical order.

**APPENDIX 2**  
**EPNL PROGRAM FLOW CHART**

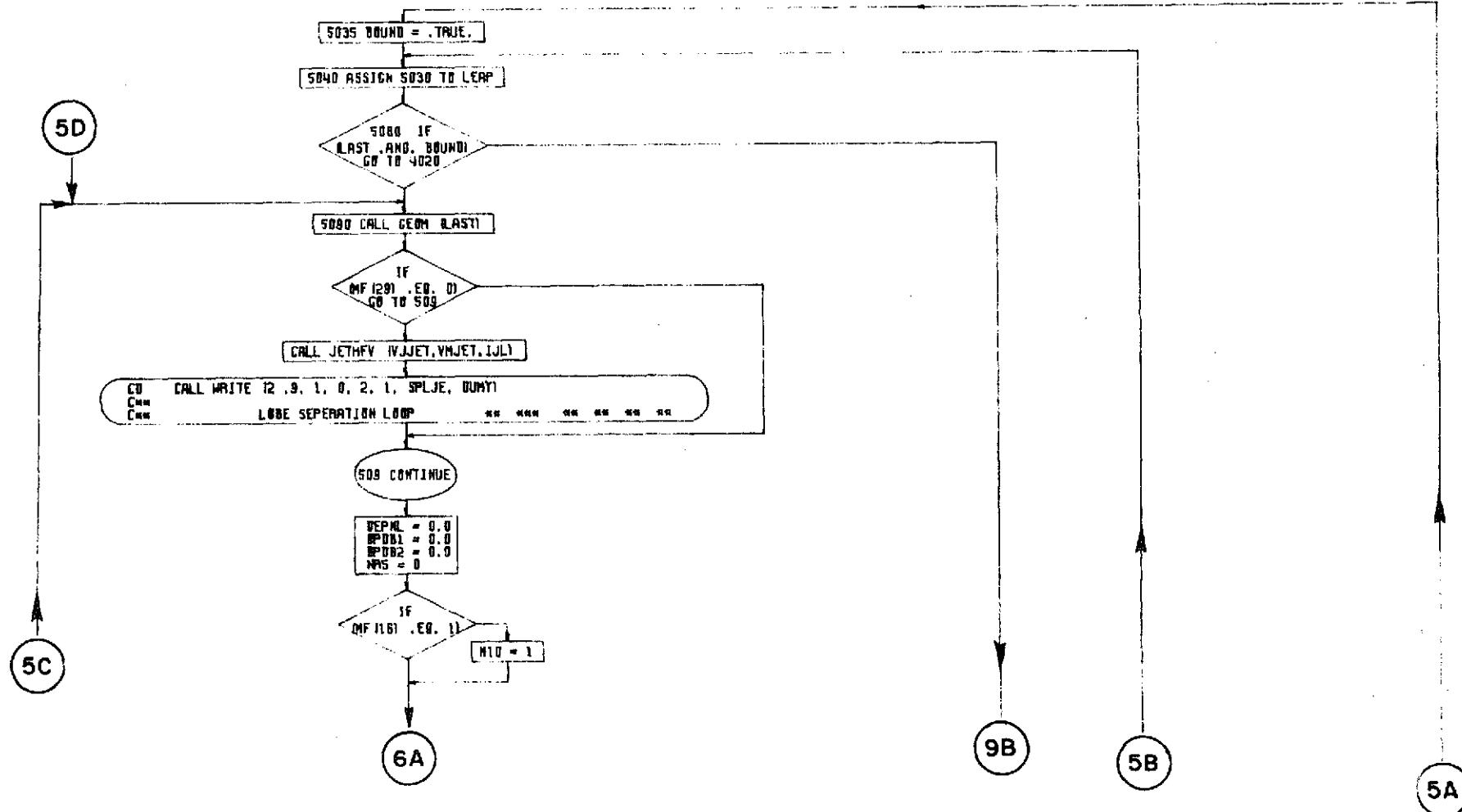


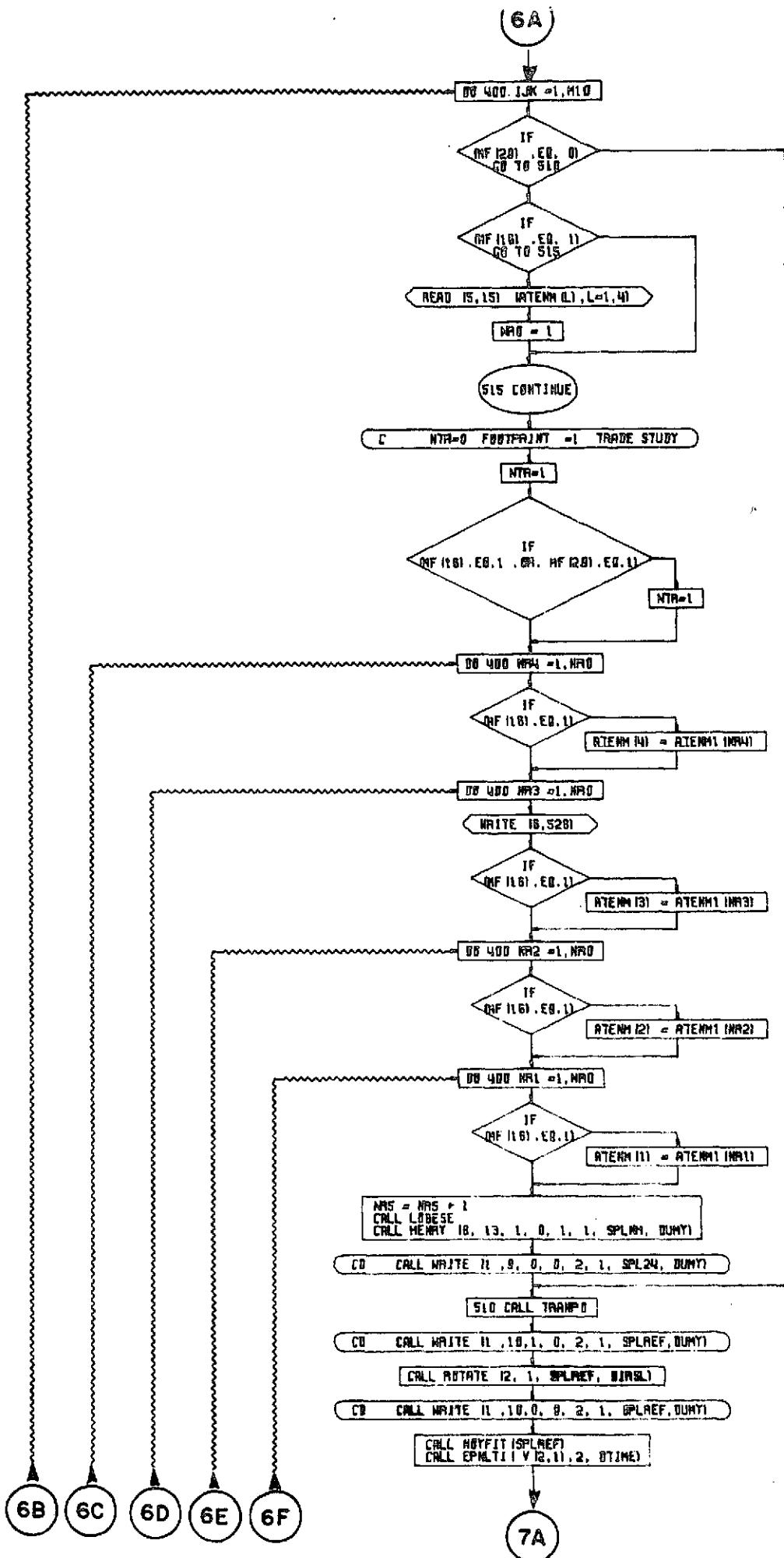


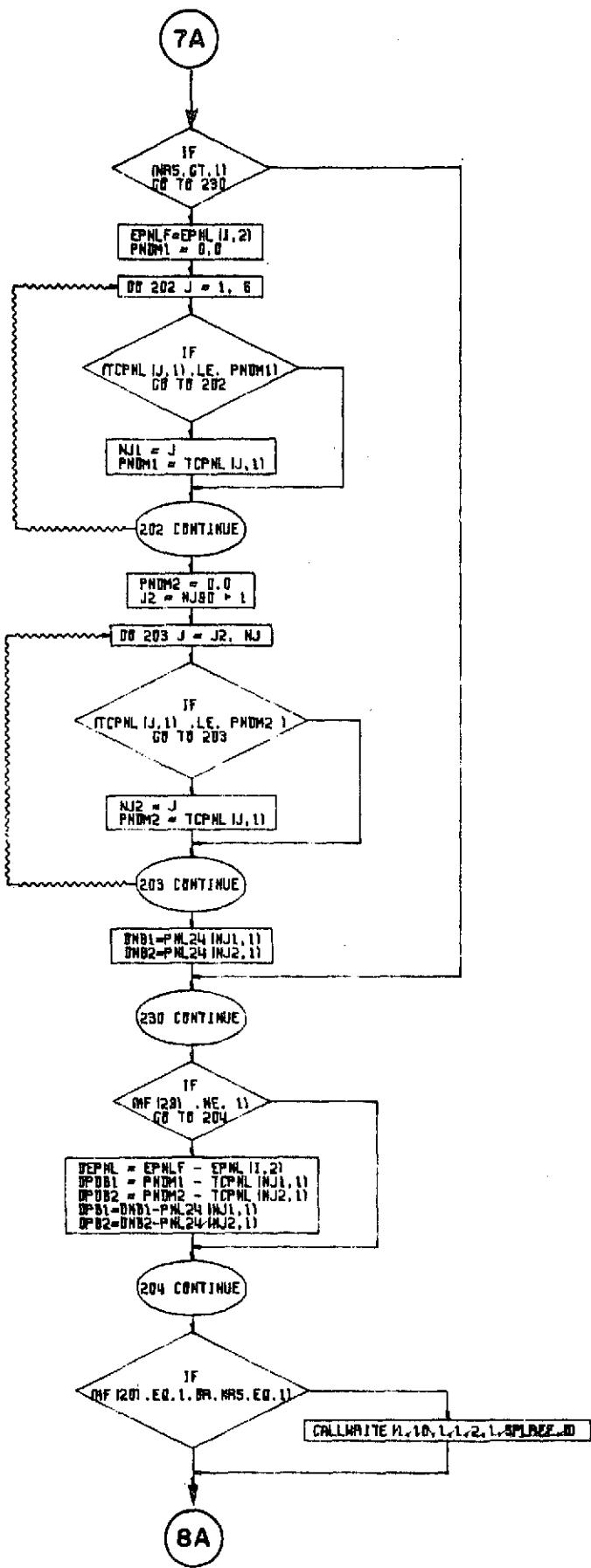


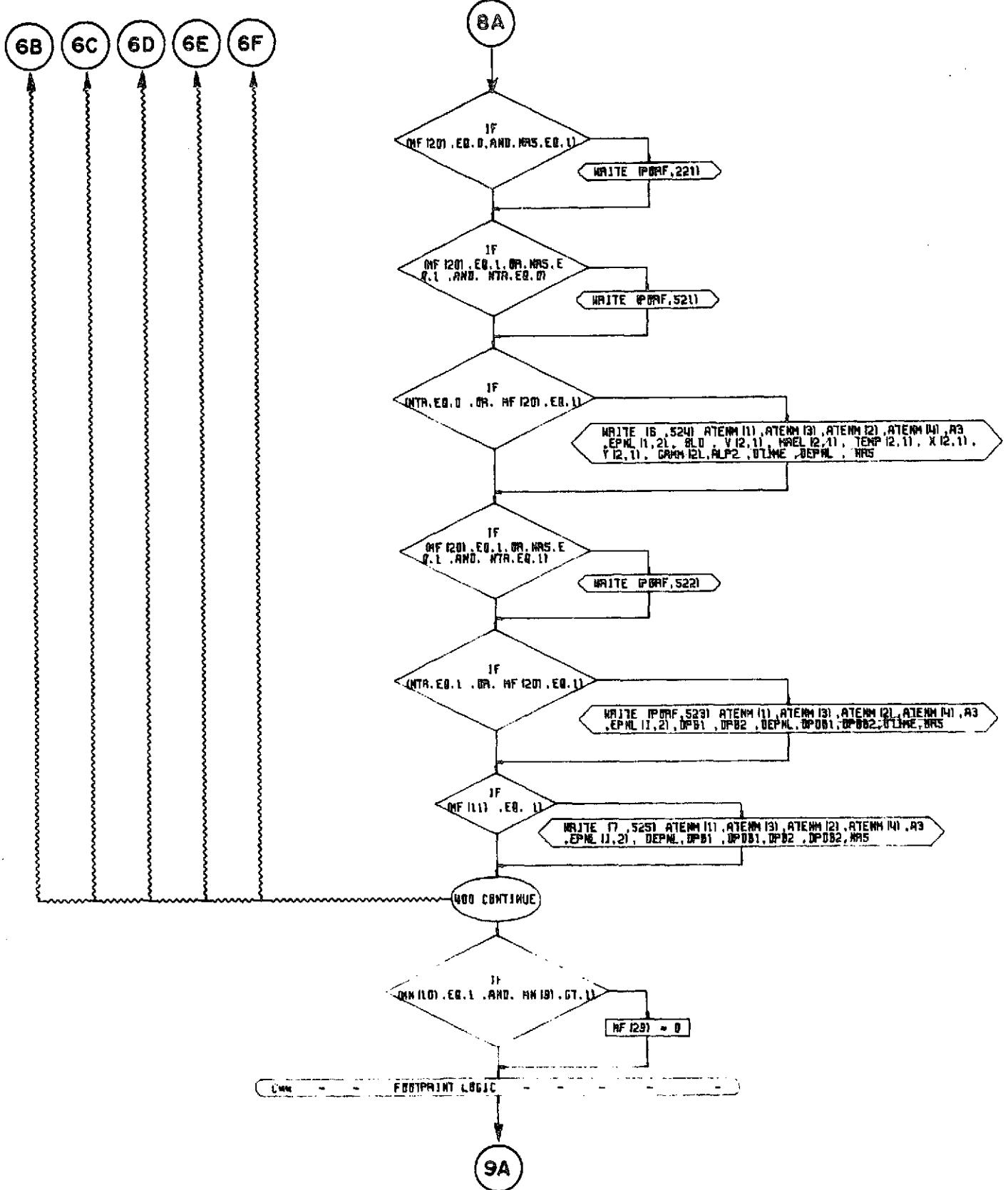


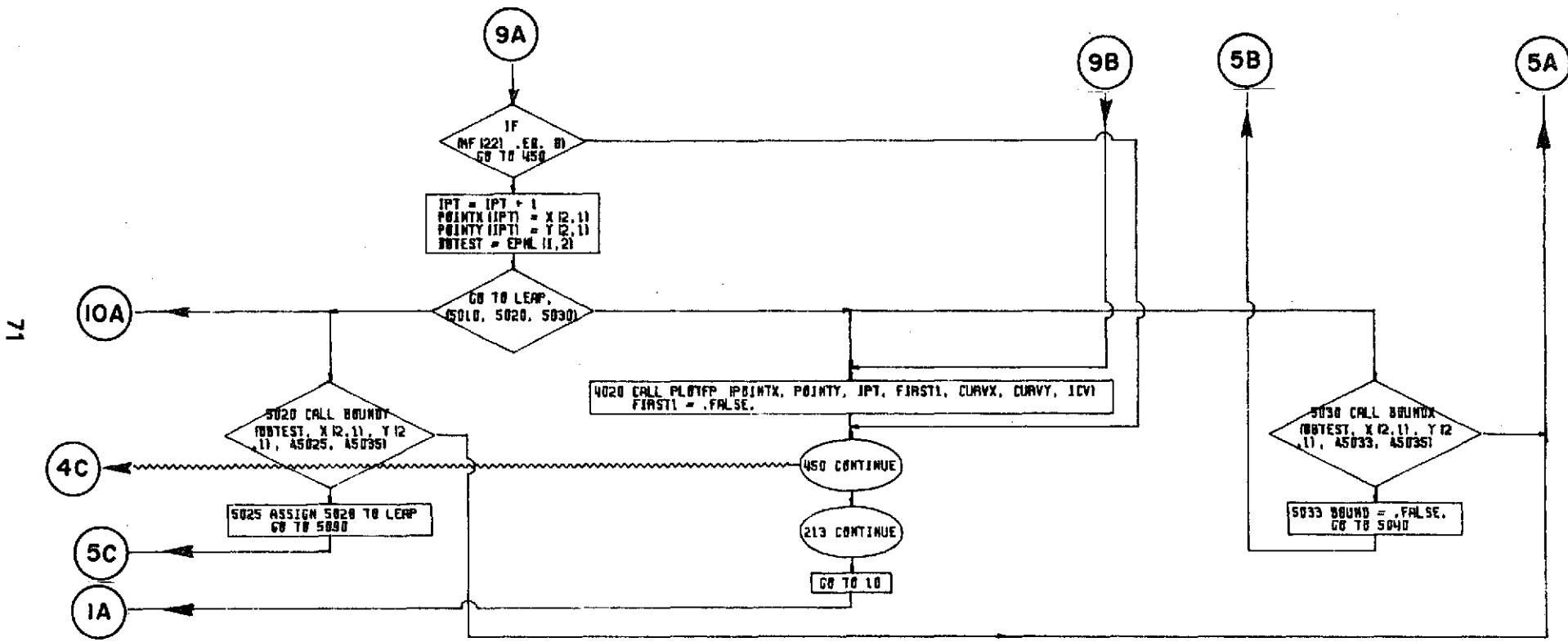
67

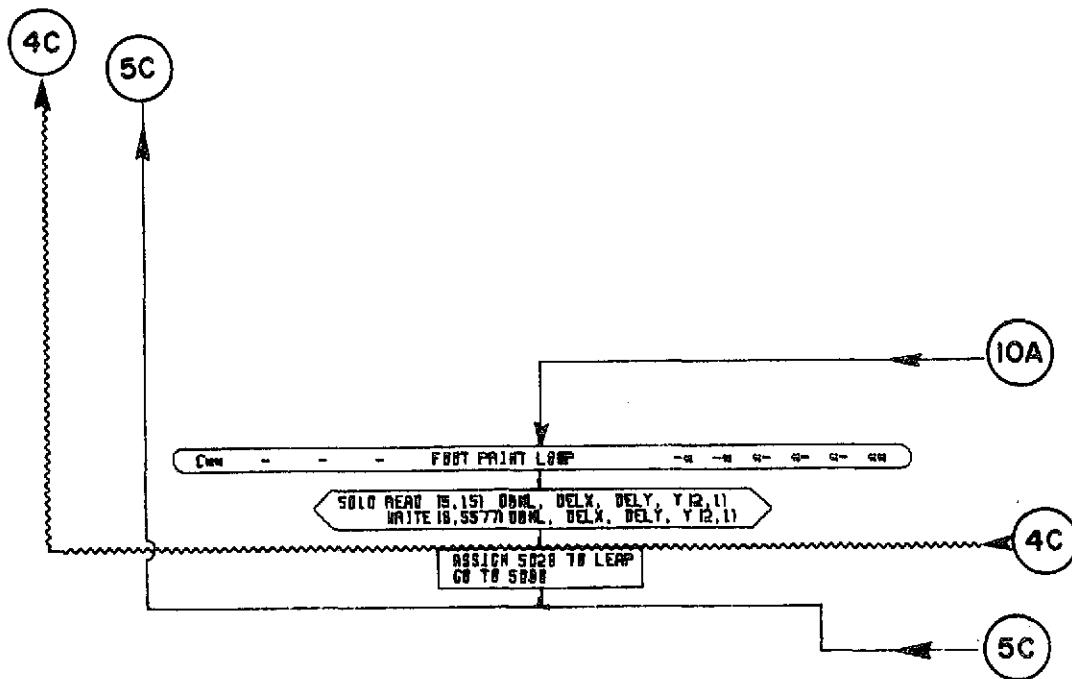


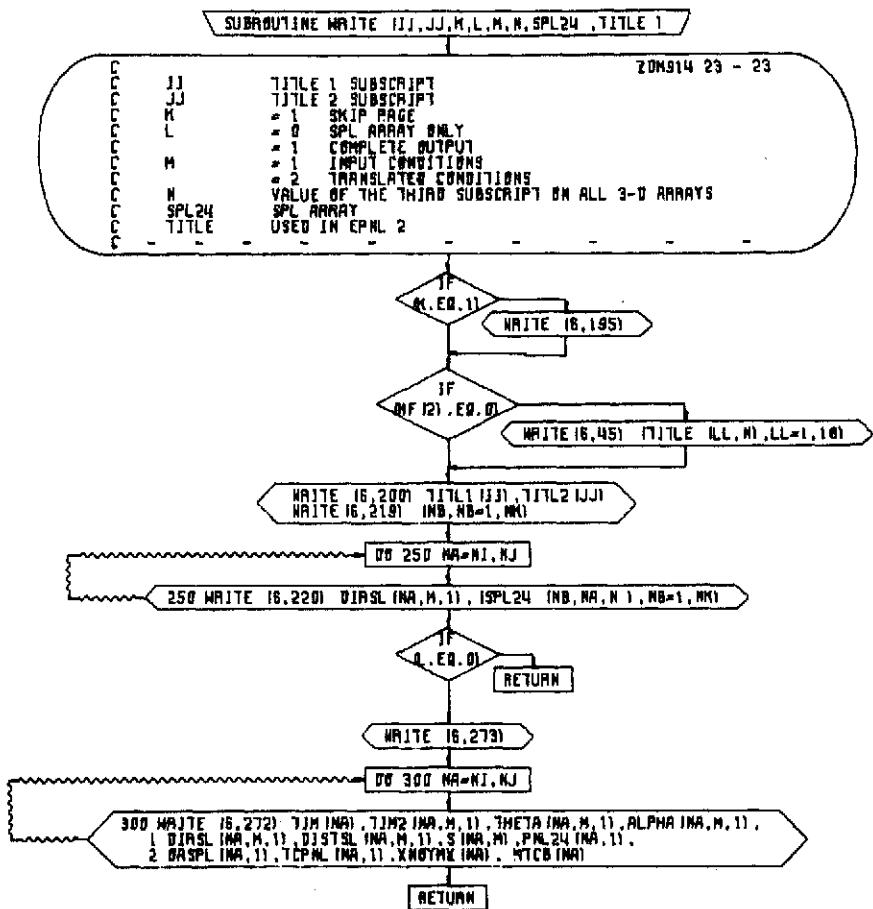


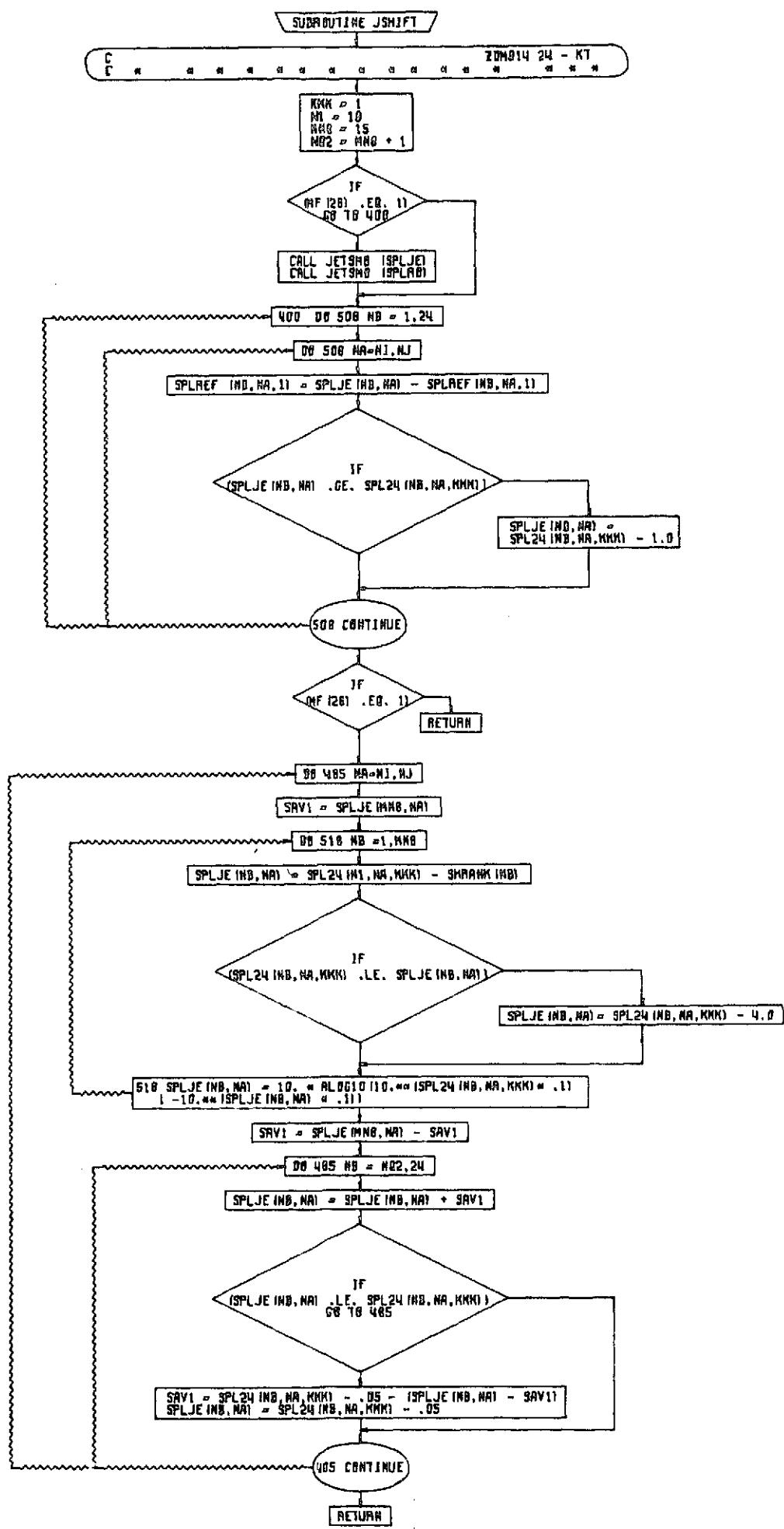


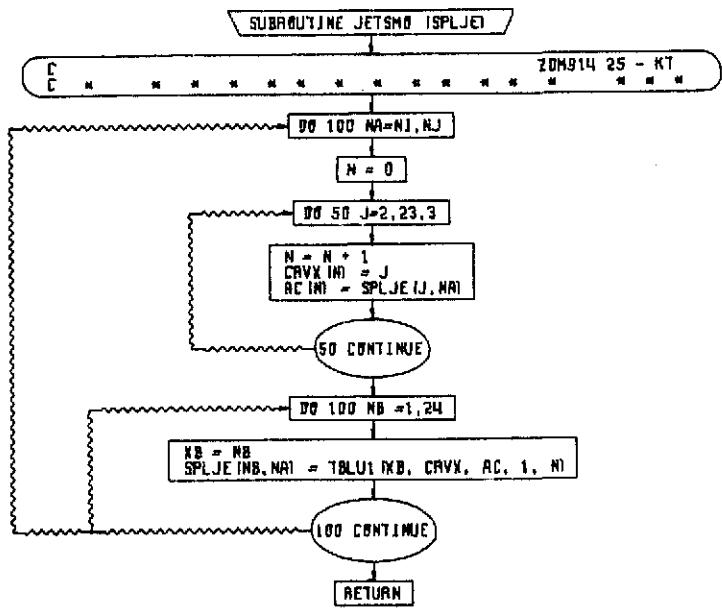


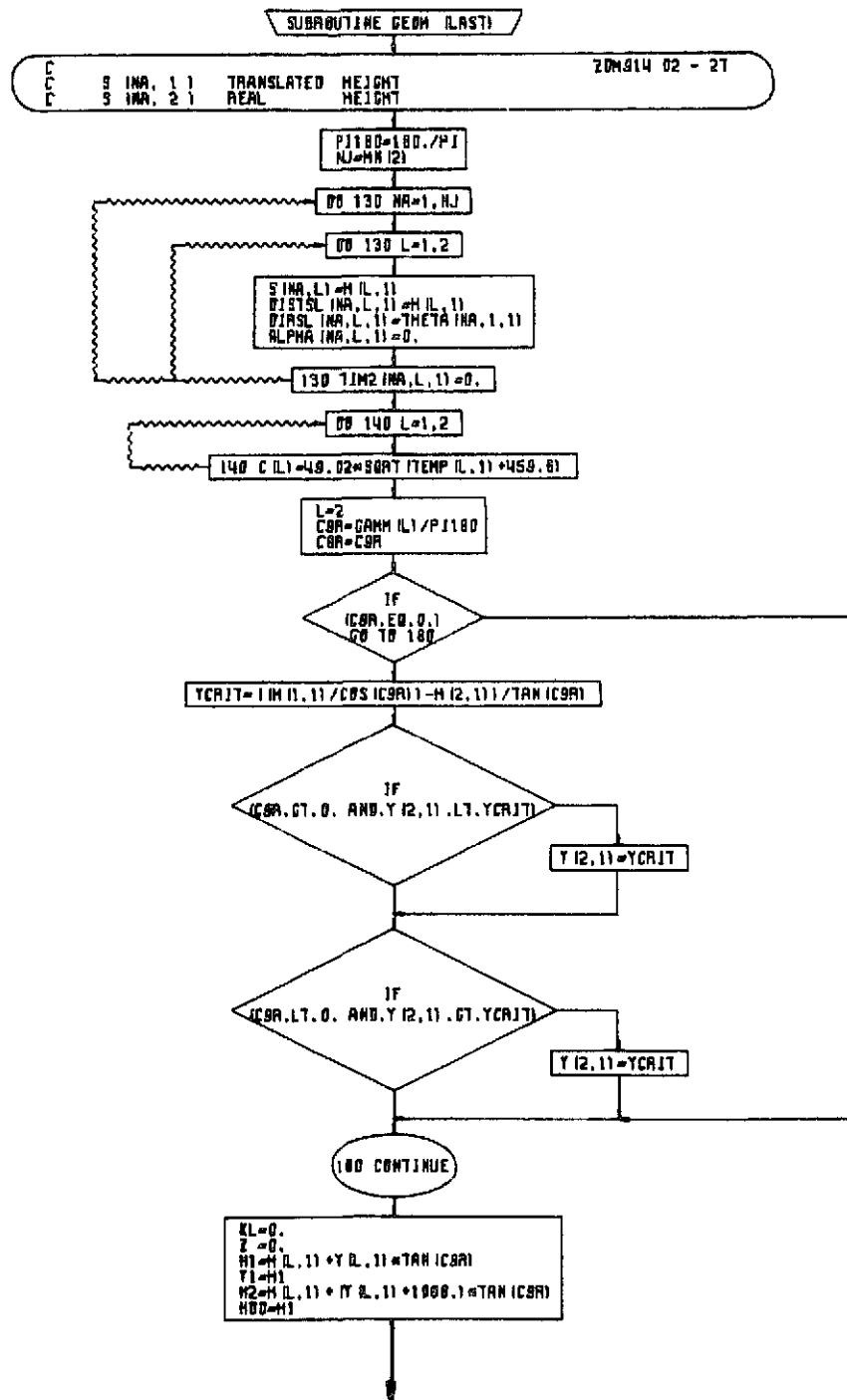


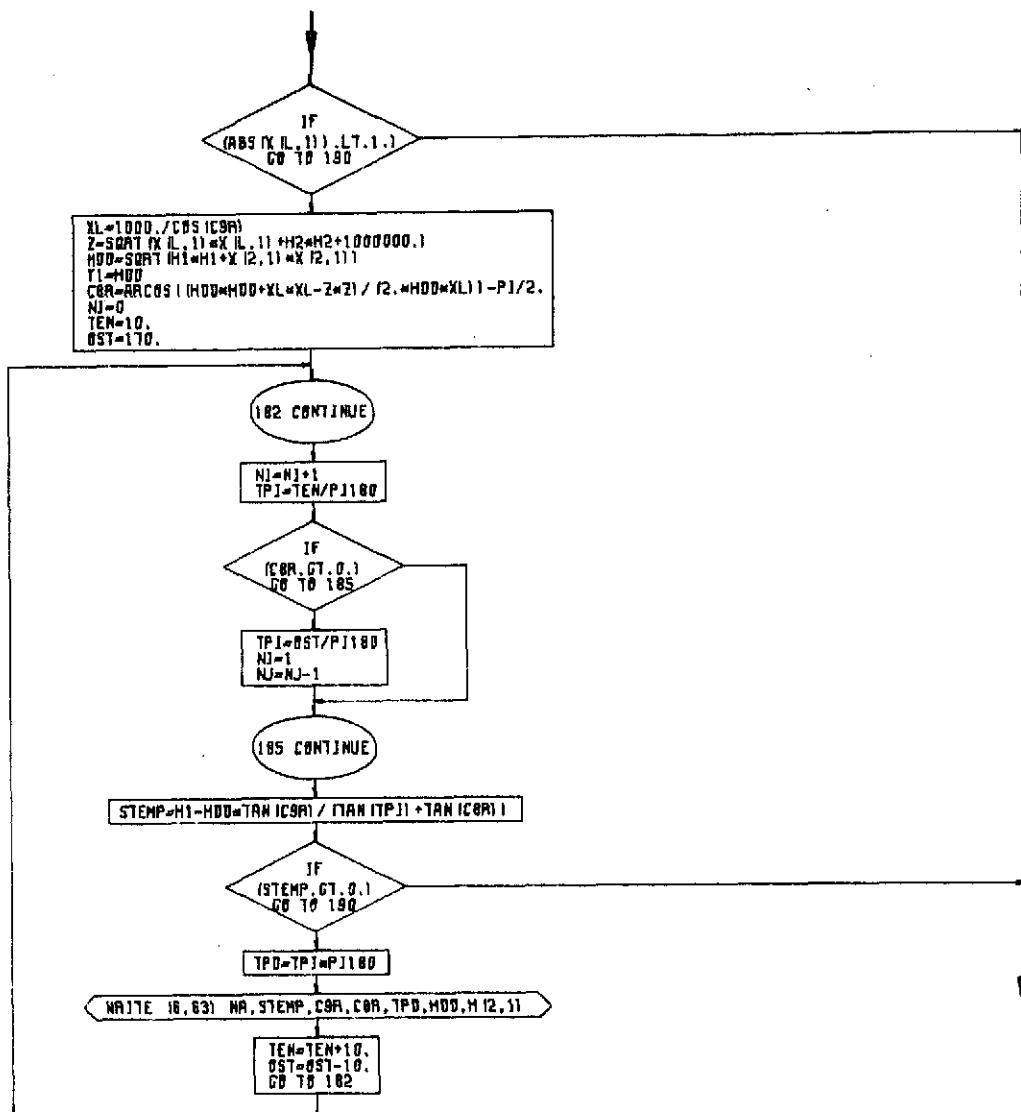


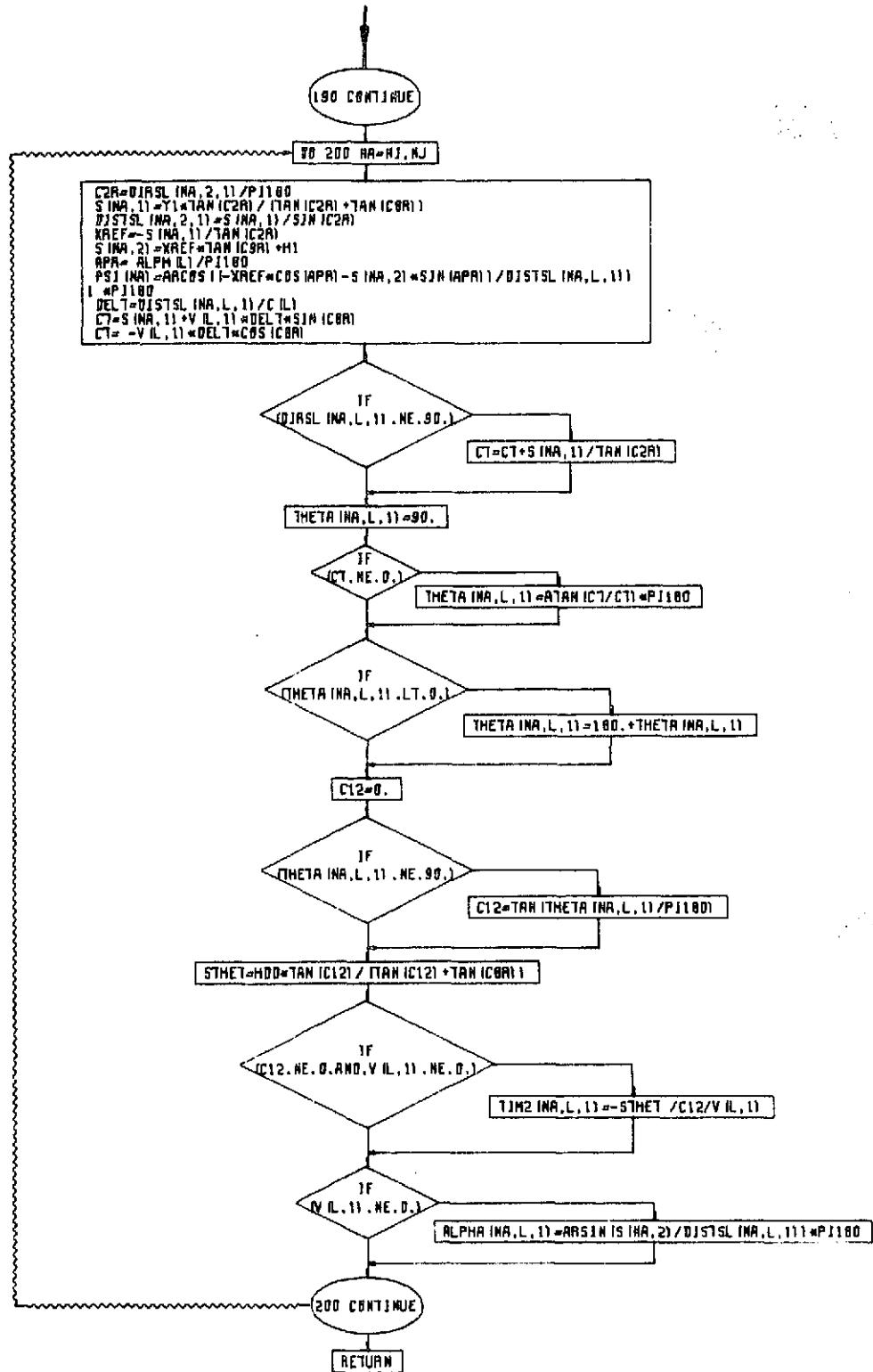






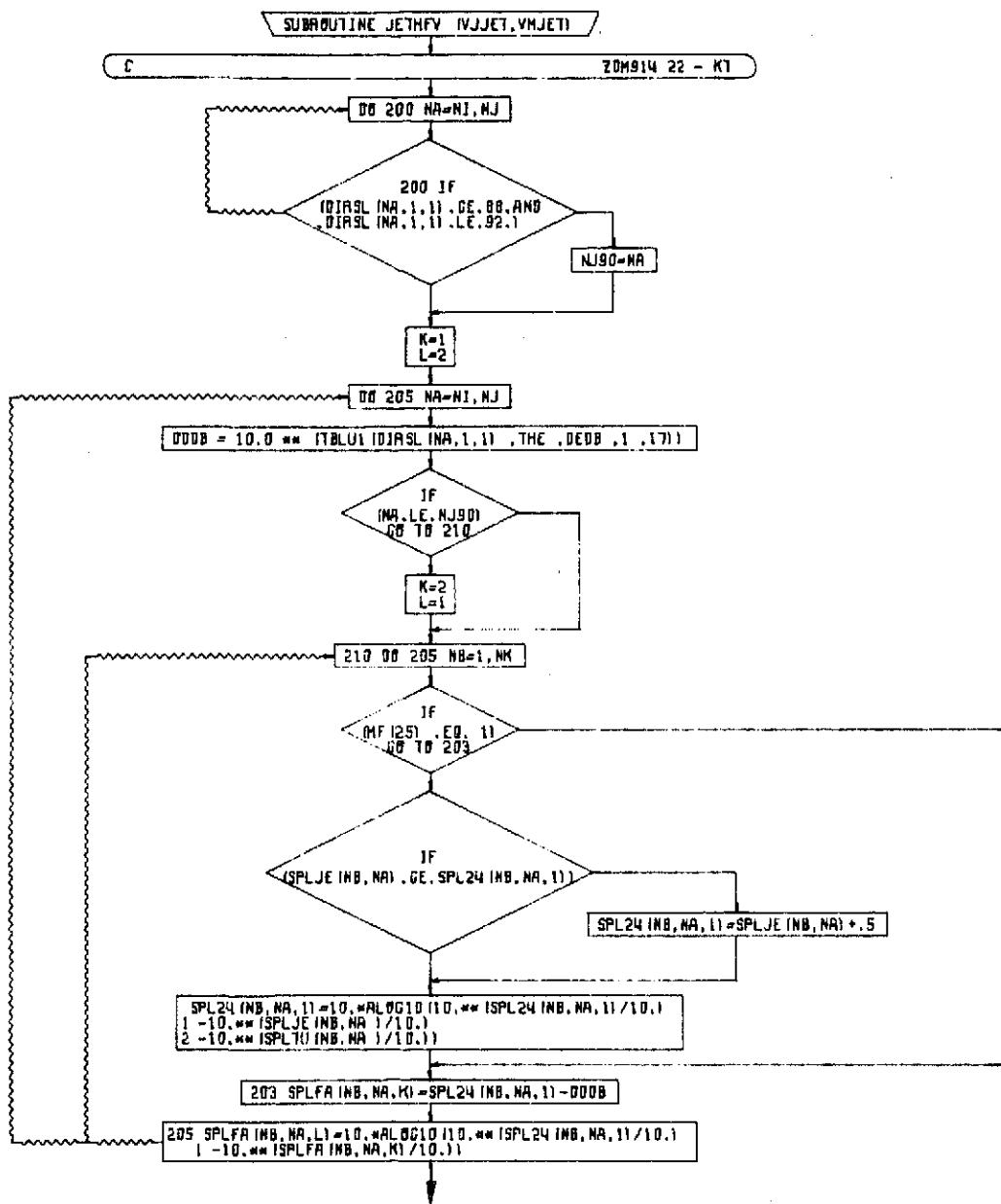


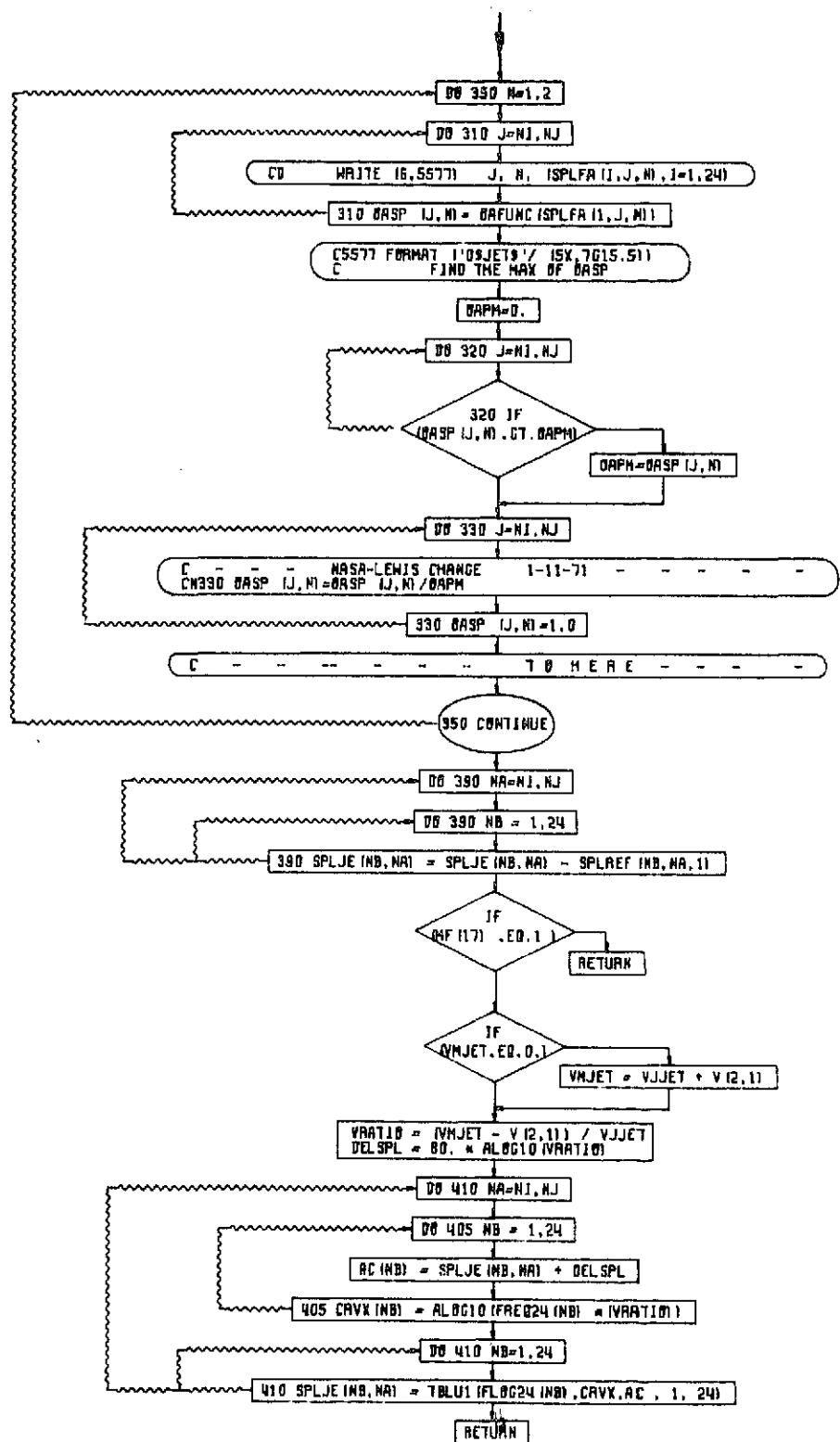


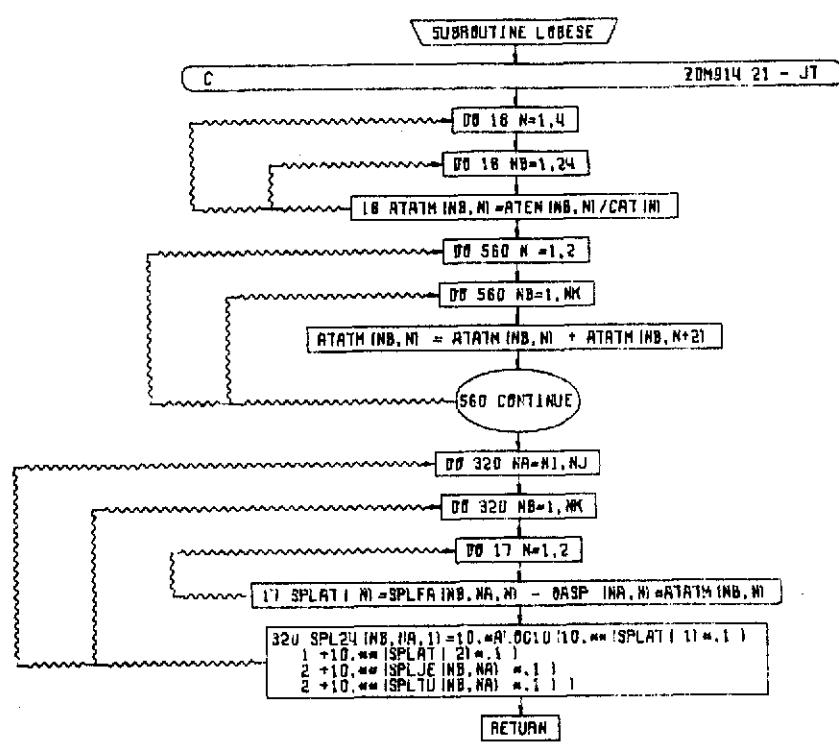


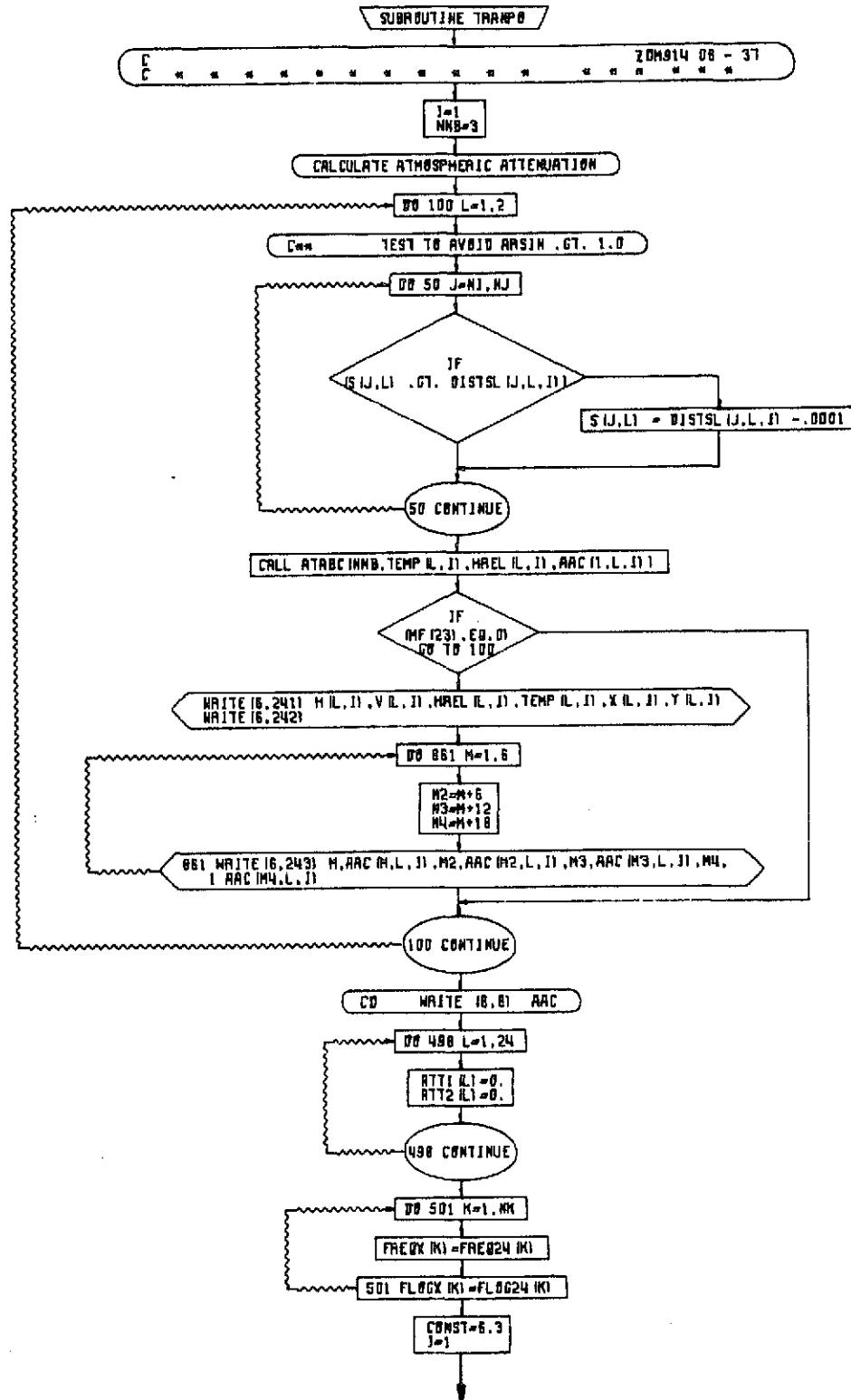
200 CONTINUE

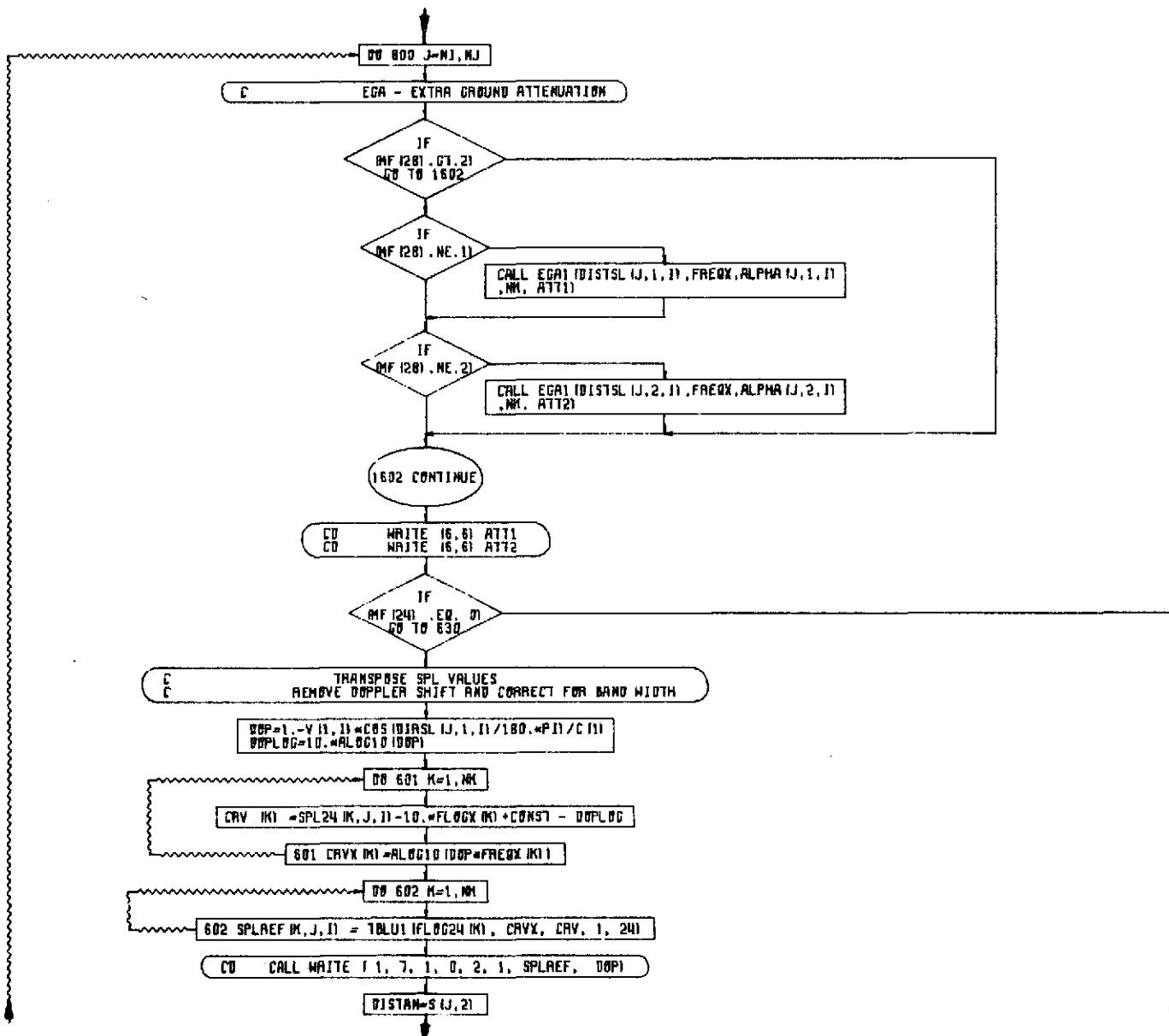
RETURN

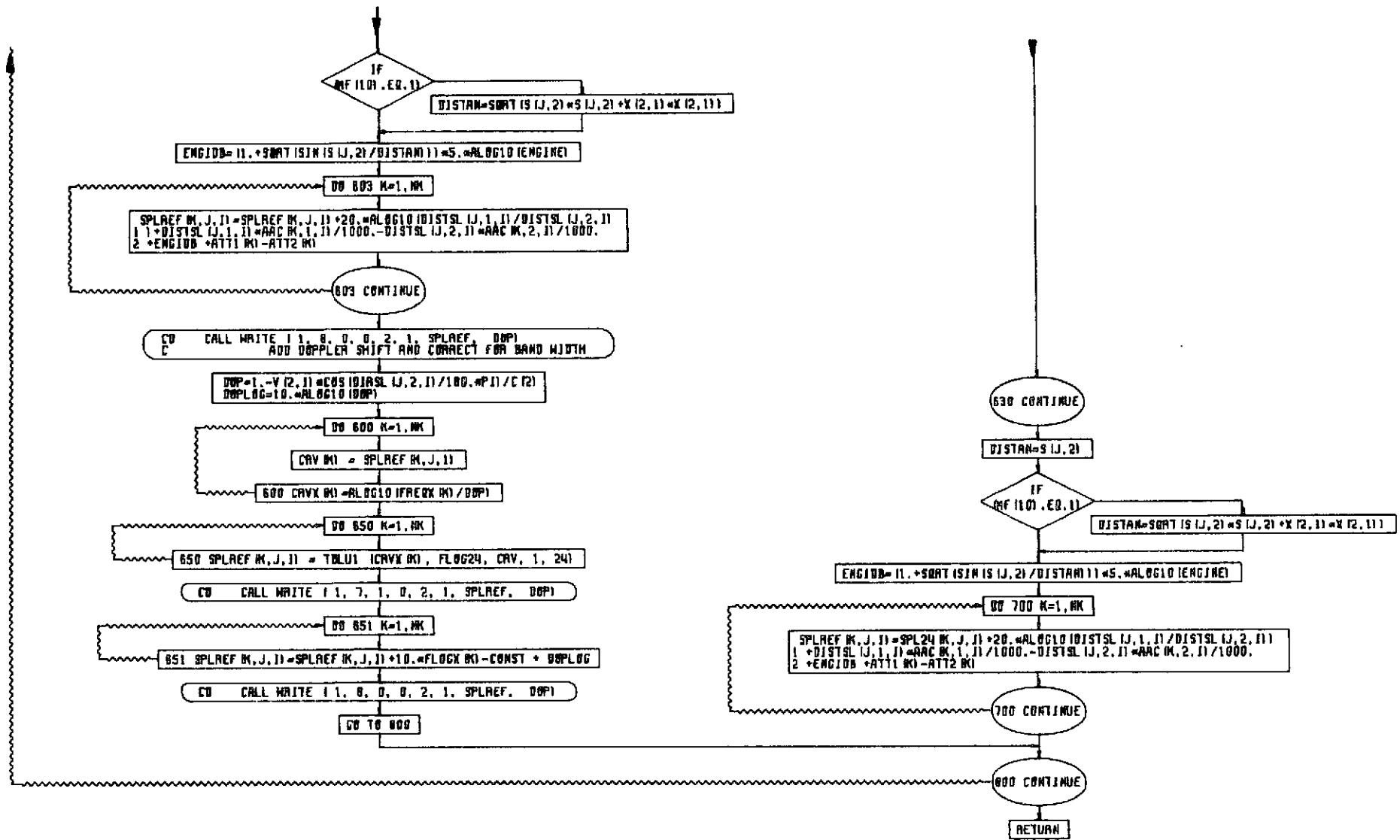


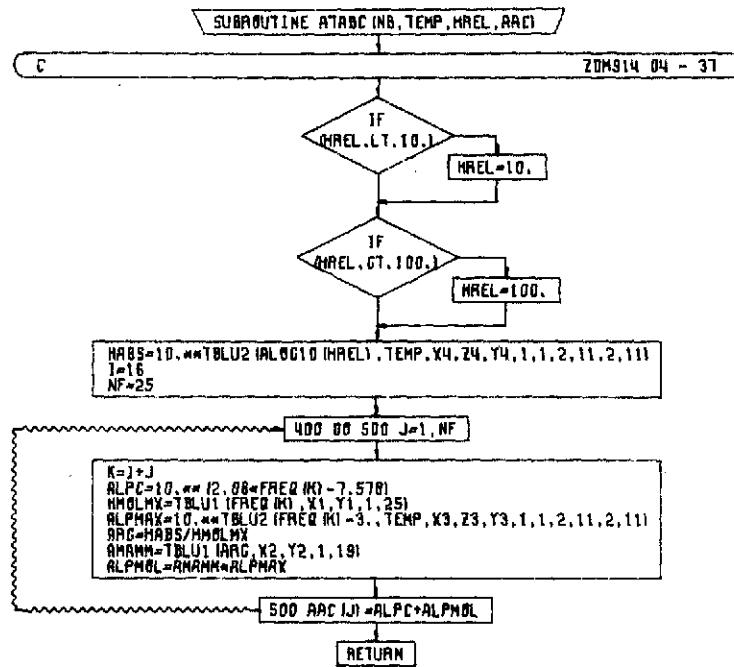


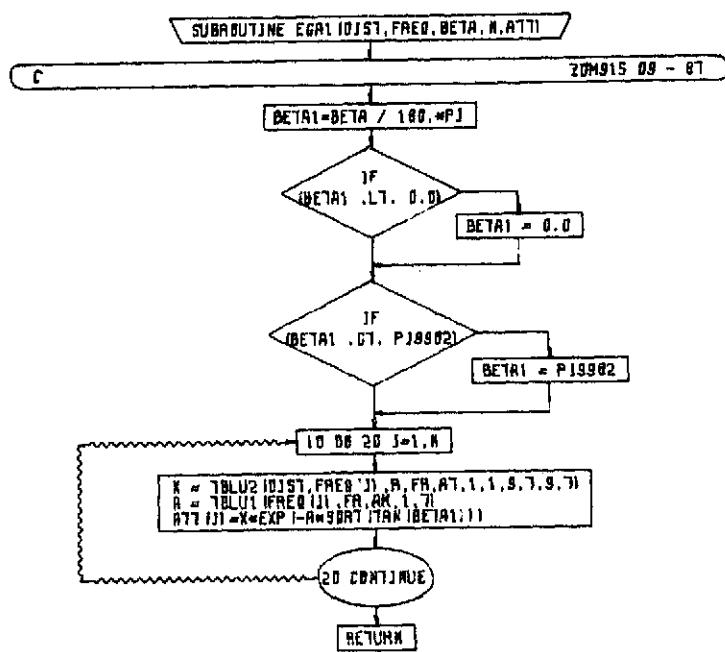


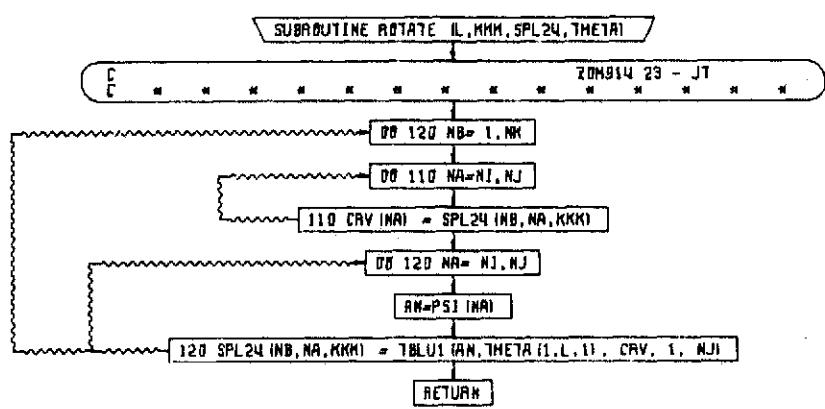


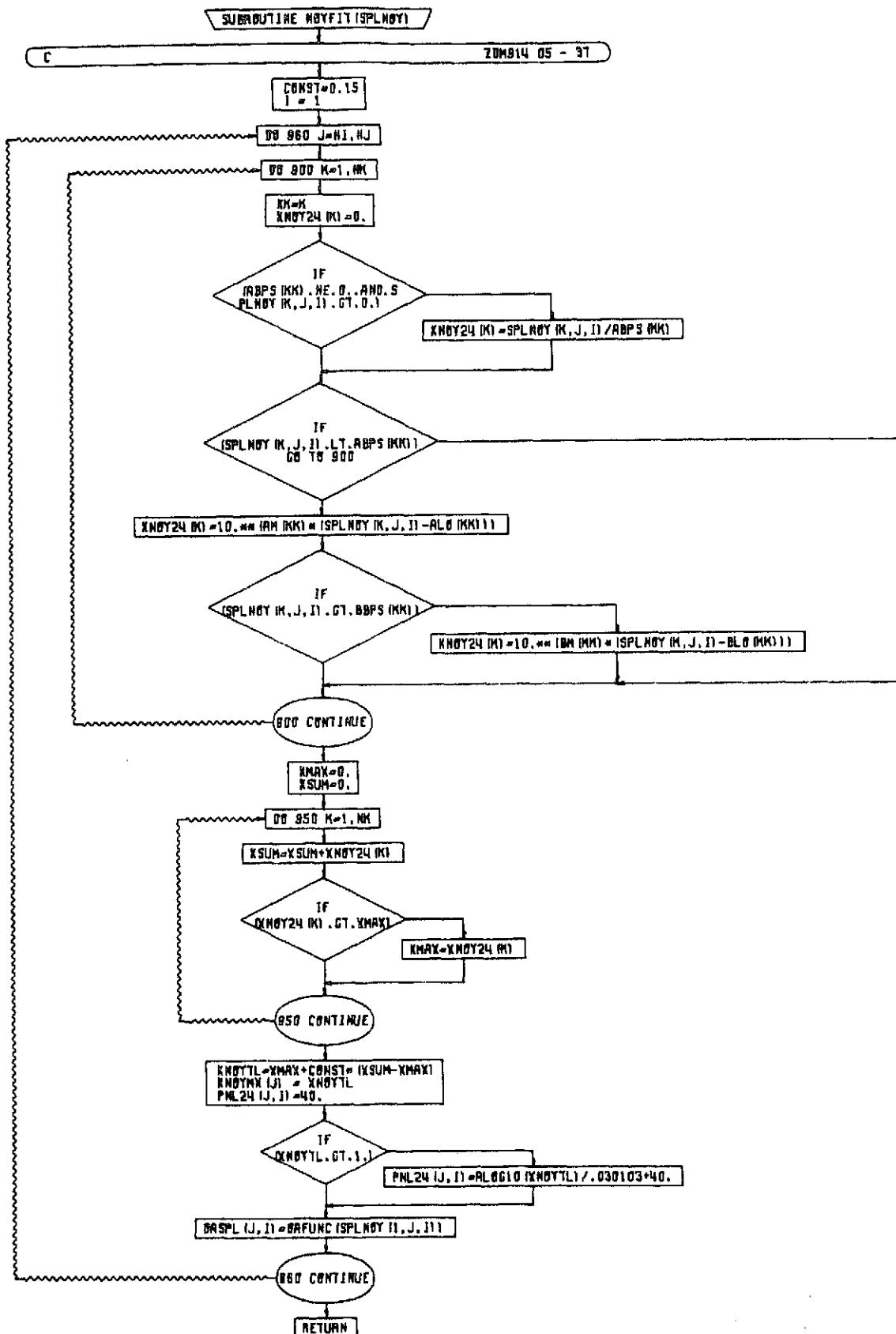


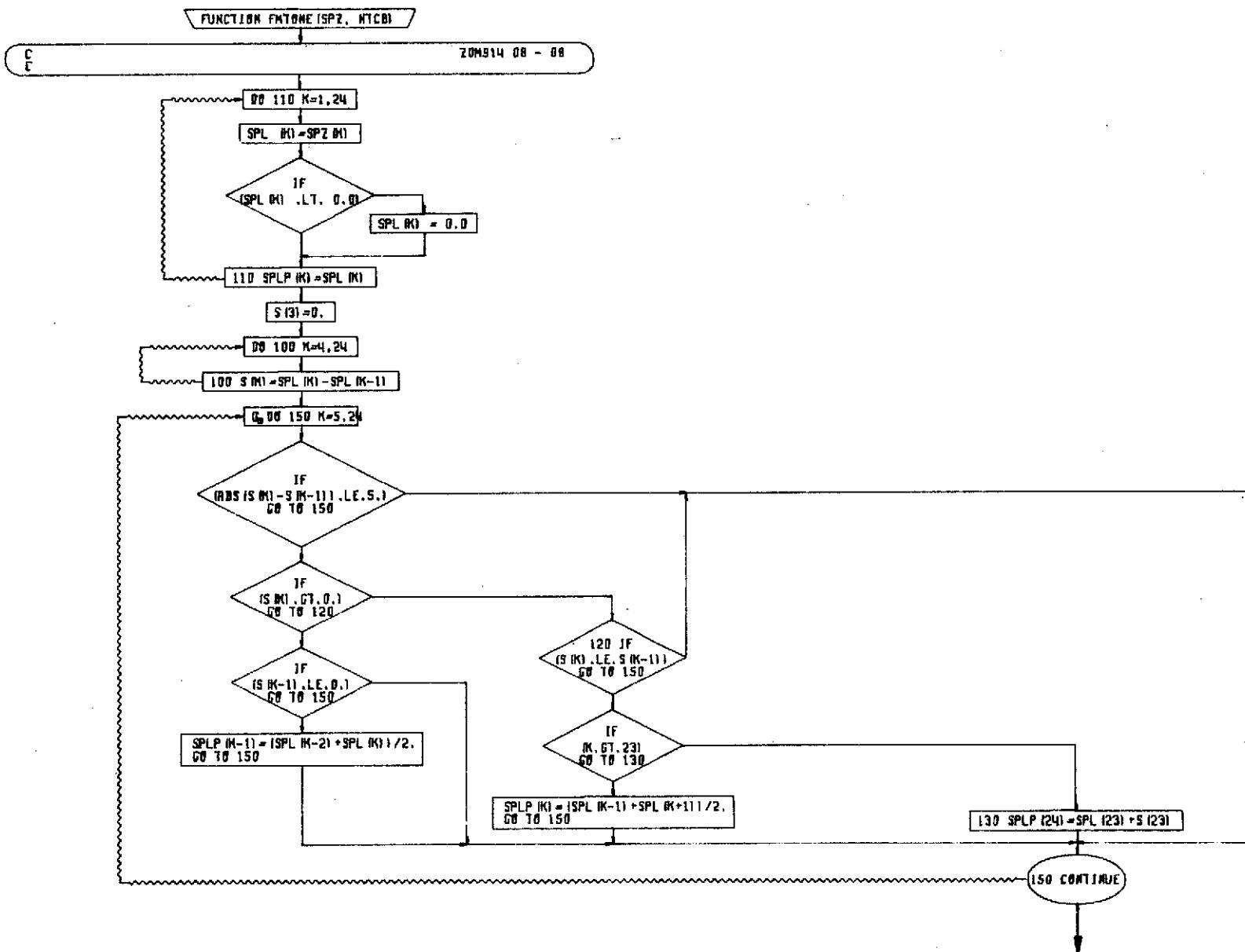


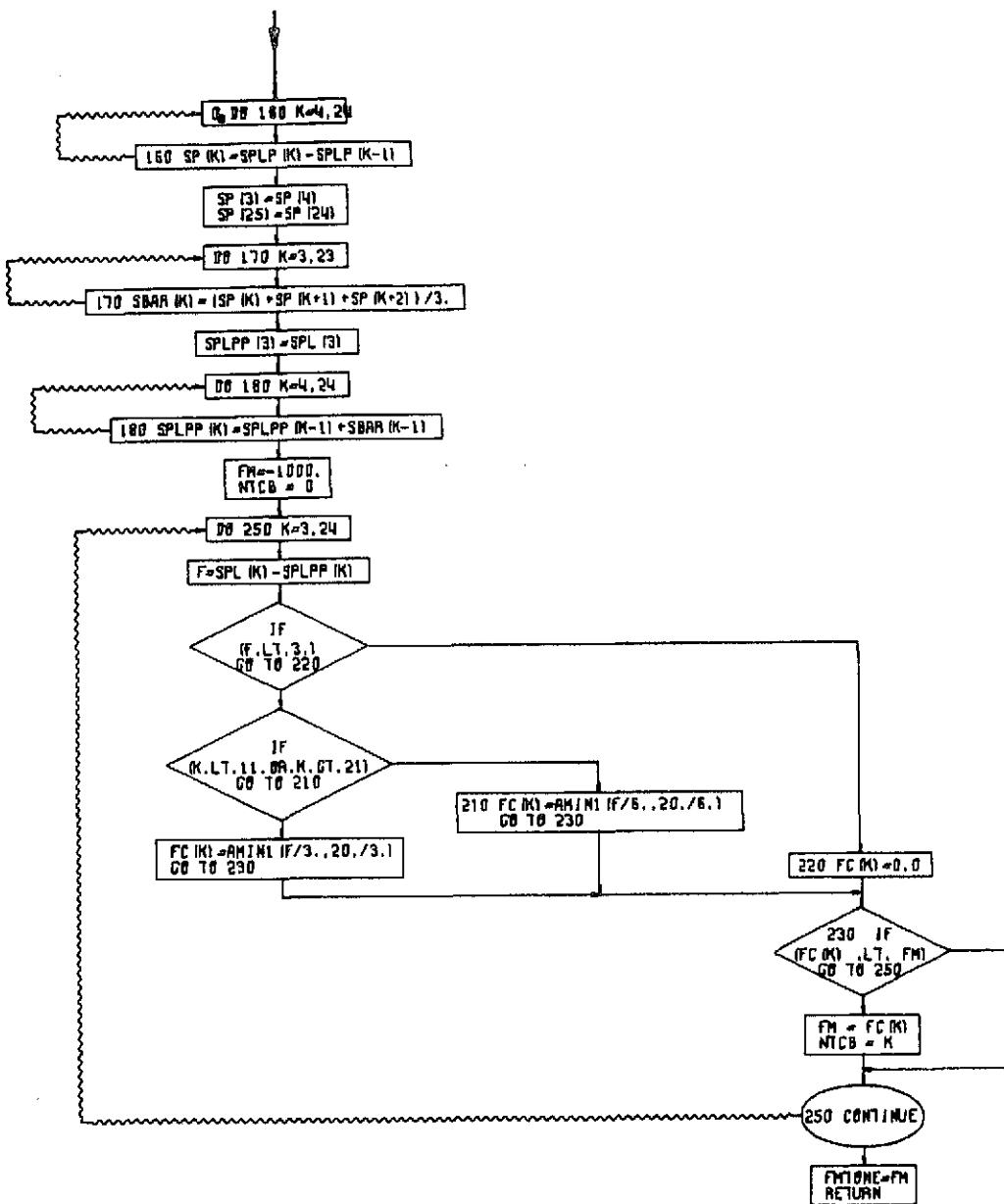


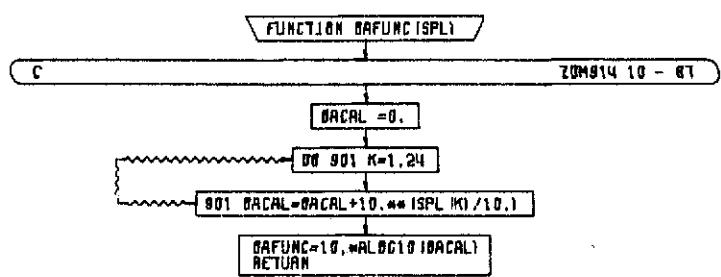


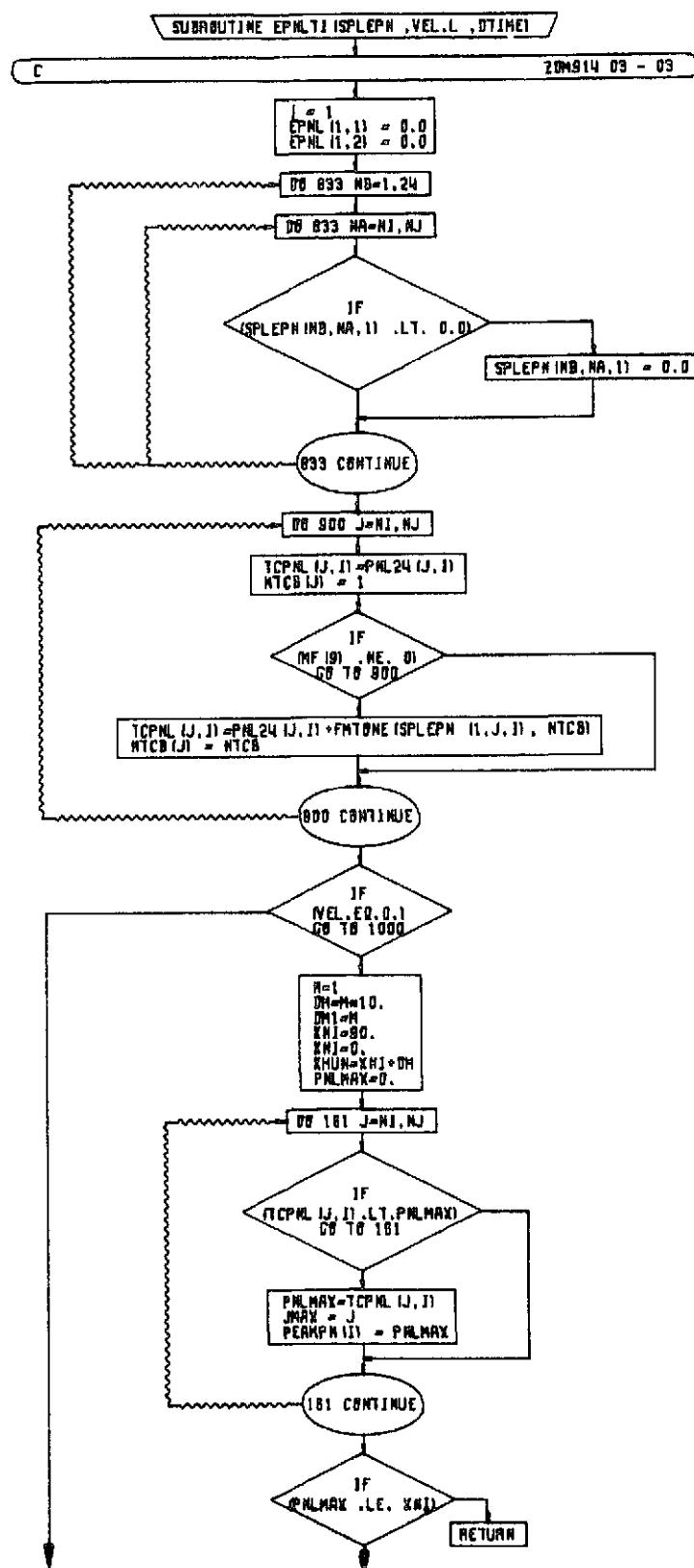


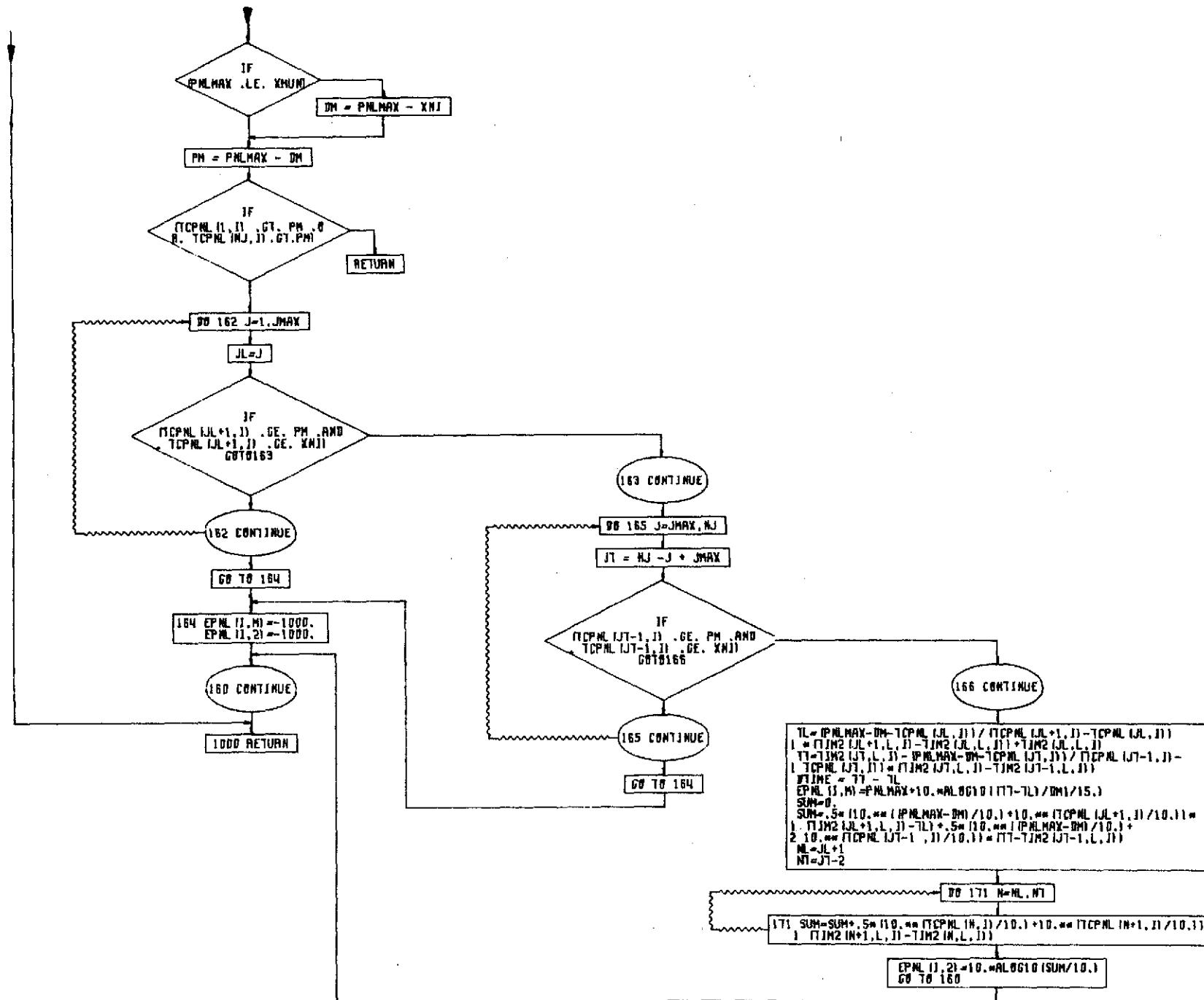


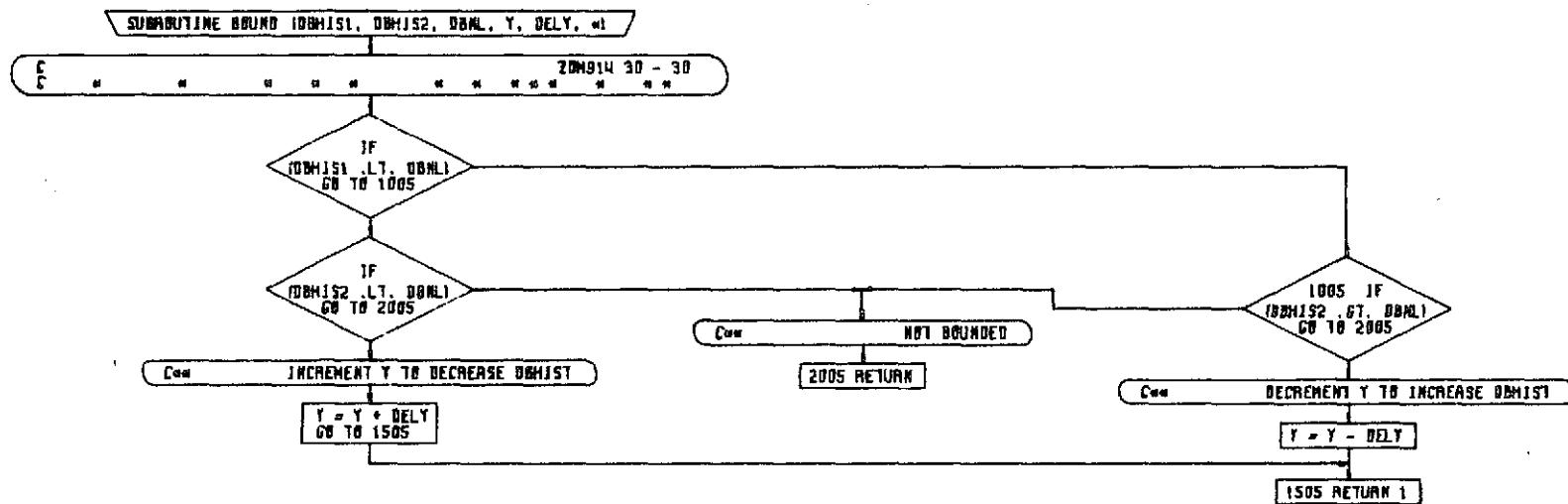


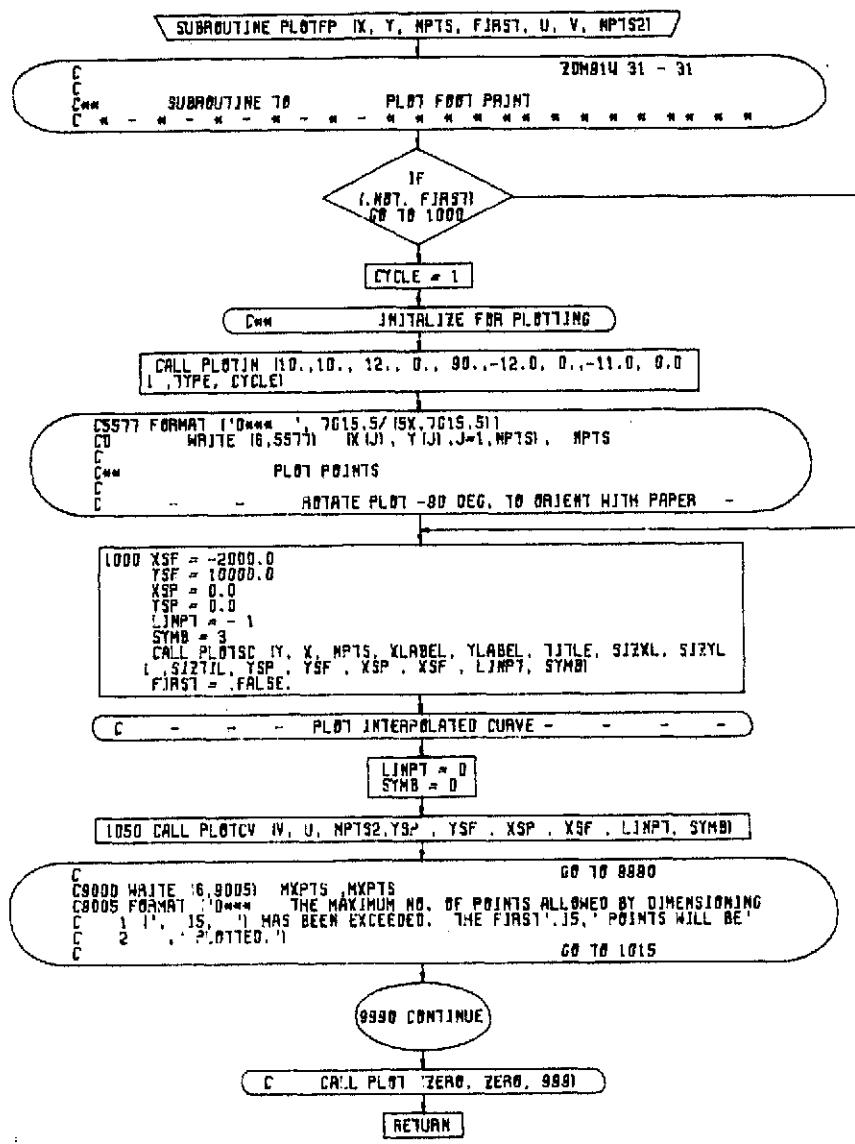


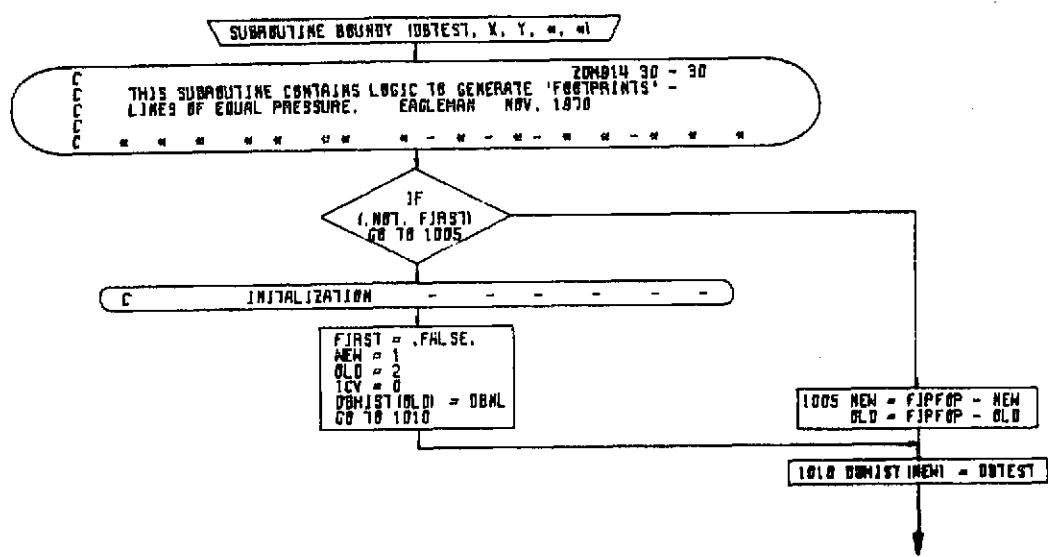


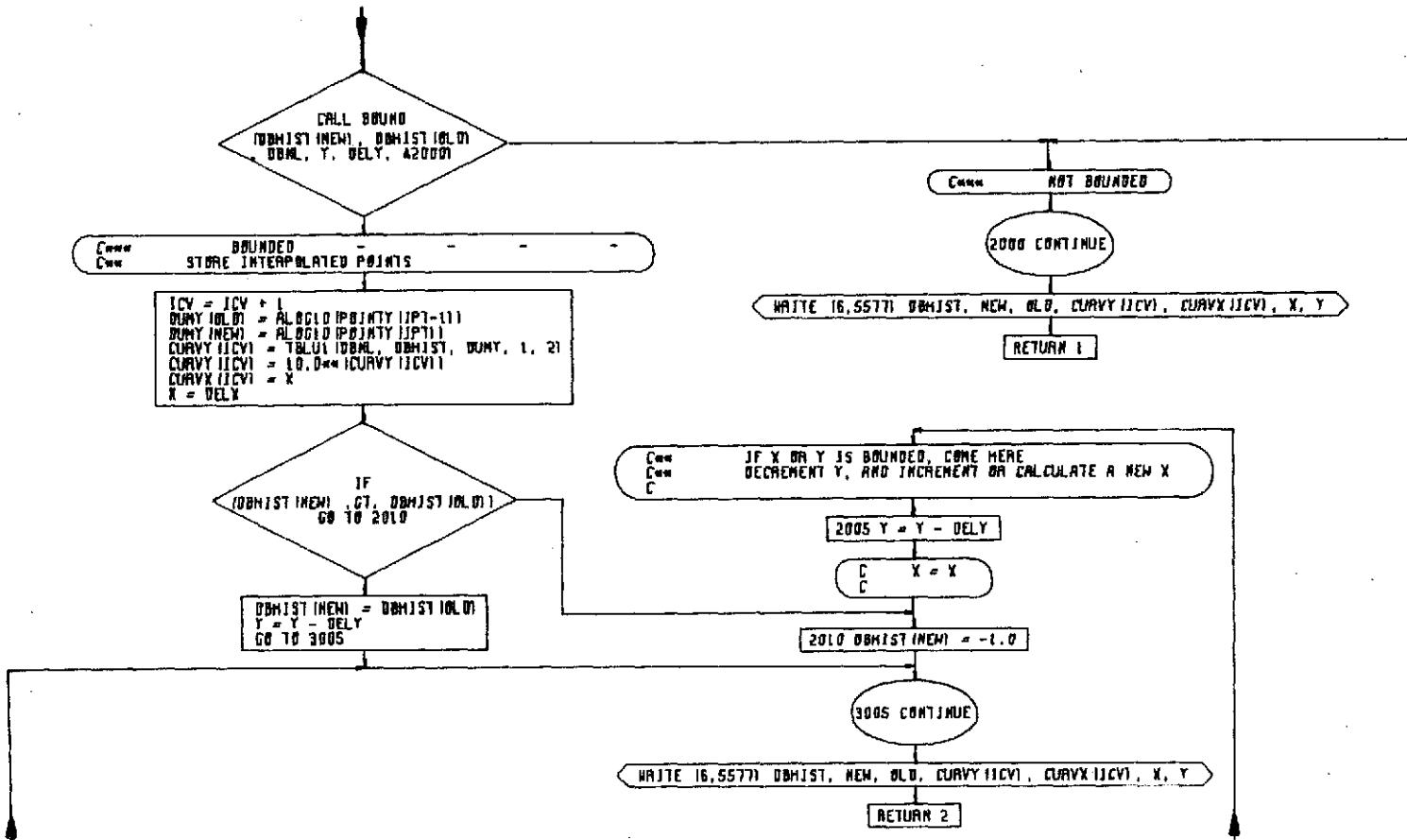


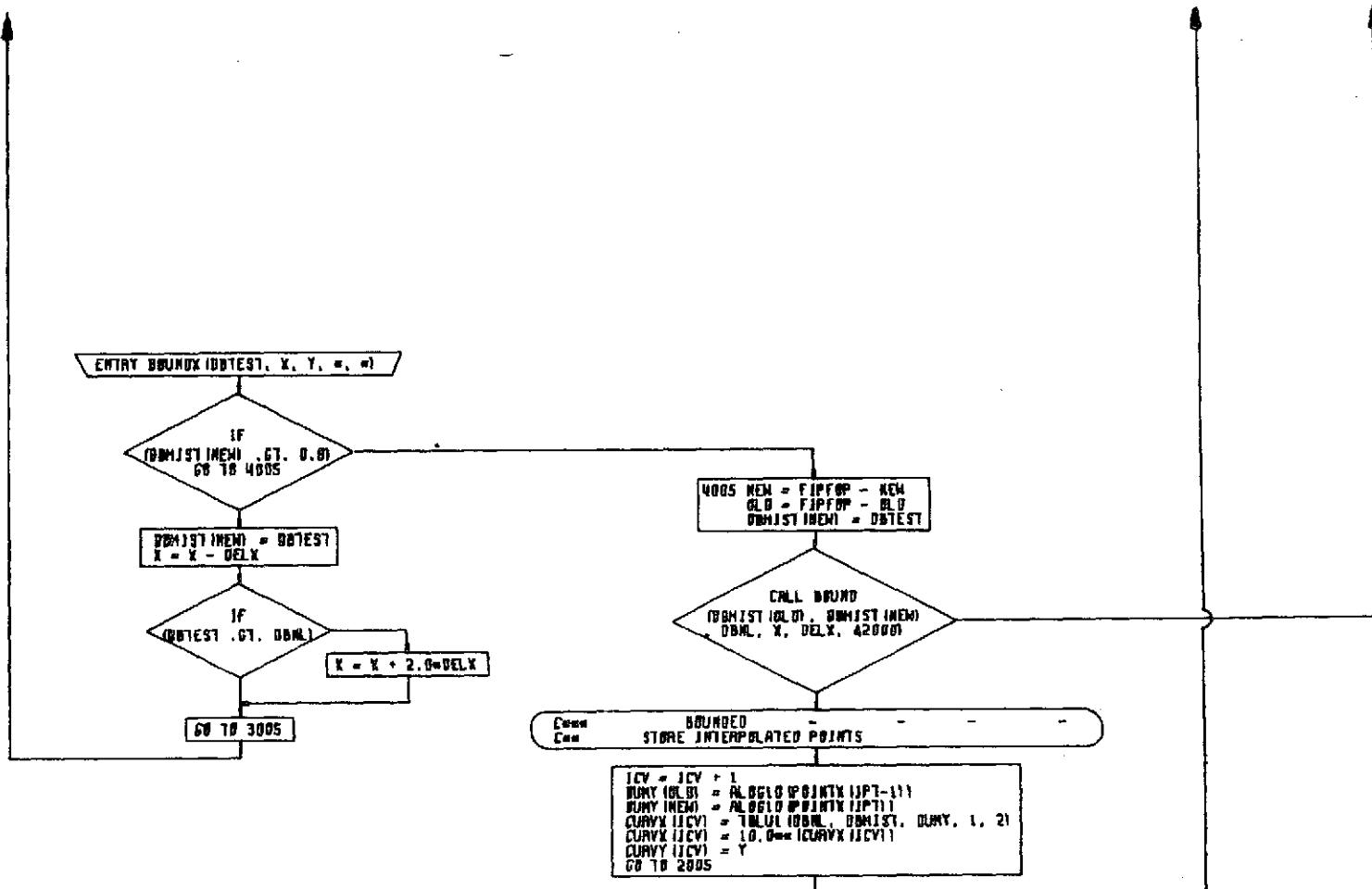












**APPENDIX 3**  
**EPNL PROGRAM LISTING**

*C-2*

C EPNL

ZDM914 01 - 1T

C (EFFECTIVE PERCIEVED NOISE LEVEL)

LOGICAL \*4 FIRST, LAST ,FIRST1 ,BOUND  
INTEGER \*4 PAPER/06/, PORF/06/  
REAL \*4 TOL/1.0E-04/  
1 ,BLANK//' 1 / ,TITLE2(13)  
2 ,BAST(3)/\*1 BASELINE //, INPT(5)/\*INPUT) //  
3 ,ATTT(3)/\*0 ATTENUATED //, CALT(5)/\*CALCULATED) //  
4 ,DELT(3)/\*0 DELTA//,CALST(5)/\*CALCULATED - SHRANK) //  
5 , TURB(5)/\*TURBINE) //  
6 ,ATENM1(5)/0.0 ,5.0 ,10.0 ,20.0, 30.0/  
COMMON / AREA1 / SPL24(24,70,2),SPLRFF(24,70,1)  
COMMON/AREA2/TITLE(40)  
COMMON / AREA3 / FREQ24(24),FLOG24(24),FREQ8(8),FLOG8(8),BN(24)  
COMMON / AREA4 / MF(30),MN(10),ENGINE ,NI,NJ,NK,NII,CRV(400),MKT  
COMMON / AREAS / PNL24(70,1),XNOY24(24),EPNL(4,2),PEAKPN (4),  
1 OASPL(70,1),TCPNL(70,1) ,XNOYMX(70), MTCB(70)  
COMMON/AREA6/THRUST(4),THRUSI (6),OVHTIM(4),TIM(70),TIM2(70,3,1)  
COMMON /FOOPTPT/ ICV, CURVX(100), CURVY(100), IPT, PCINTX(250)  
1 ,POINTY(250), DBNL, DELX, DELY  
COMMON/GEOM1/H(3,4),V(3,4),HREL(3,4),TEMP(3,4),X(3,4),Y(3,4), PI  
1 ,GAMM(2) ,ALPH(2), S(70,2), W(70,2) ,C(2)  
COMMON/GEOM2/THETA(70,3,1),ALPHA(70,3,1),DIRSL(70,3,1),  
1 DISTSL(70,3,1),AAC(25,2,4)  
COMMON/CCNNOY/ABPS(24),BBPS(24),ALD(24),BLO(24),AM(24),BM(24)  
COMMON /HFVJF/ SPLFA(24,70,2) ,SPLJE(24,70), DAPS(70,2)  
1 ,SPLTU(24,70)  
COMMON /HFVJT/ CAT(4) ,ATEN(24,4) ,ATATM(24,4) ,ATENM(4)  
  
C  
C DIMENSION CRVX(50),HH(32),XX1(32),FREQP(8),FREQC(8),NBAND(24)  
C  
5 FORMAT(7F10.4/F8.4,I2)  
15 FORMAT(9F8.0)  
16 FORMAT (SF7.0)  
17 FORMAT( F10.3, 12F5.2 / 10X,12F5.2)  
18 FORMAT(2F10.0, 13A4)  
20 FORMAT (18A4)  
25 FORMAT(3CI1,10I2,F5.0 +2F8.0)  
31 FORMAT(2X,18F4.0)  
35 FORMAT(1H1, 28H CONTROL CARD AND DATA INPUT //  
1 2X,30I1,10I2,F5.0 // 2X,18A4 / 2X,18A4 /  
2 2X, 7A4,F4.1, 444, F4.1, 3A4, F4.1, A4//)  
36 FORMAT(/// 96H ALTITUDE VELOCITY HUMIDITY TEMP SID  
1EL INE OFFSET POWER OVHTIM NJ /  
2 7X,3F10.2,I10 / 7X,6F10.2 )  
41 FORMAT (2X,2I5, 3F10.2)  
42 FORMAT (// 15X, 'DELTA JET PRIMARY =', F4.1)  
100 FORMAT (18A4)  
101 FORMAT (1H1,36A4)

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105 FORMAT (7A4,F4.1, 4A4, F4.1, 3A4, F4.1, A4)
199 FORMAT (5X,18A4)
200 FORMAT (1H1,55X,22H S P L      I N P U T /)
2000 FORMAT(1H1,55X,36H S P L      T R A N S L A T E D /)
210 FORMAT(1H , 3X,15H POWER LEVEL = ,F10.2, 2H ,6X,11HALITUDE = ,
1   F10.2,6X,11HSIDELINE = ,F10.2 )
219 FORMAT (1X,5HANGLE,24(1X,I3 ) )
220 FORMAT (3X,27F4.0)
221 FORMAT(1H1)
272 FORMAT (1X, 5F8.2 ,2F10.2, 4F8.2, 14)
273 FORMAT (1X, / * TIME     TIME     THETA    ELEVA   DIRSL   DIST*
1   ,T52, *HEIGHT    PNL     OASPL   TCPNL  XNOYMX  TCBAND*)
280 FORMAT ( /1X,16H MAXIMUM TCPNL = , F10.2,
1 1X,19HAPPROXIMATE EPNL = , F10.2,
2 1X,19HINTEGRATION EPNL = , F10.2)
511 FORMAT (*1BREAKDOWN OF CONTROL CARD E P N L 1*,//),
1   *   1 --- 1 = ONE HEADER CARD ----- *, 11, / ,
2   *   2 --- 1 = SECOND HEADER CARD ----- *, 11, / ,
3   *   3 --- 0 = FIXED NOISE SOURCE ----- *, 11, / ,
4   *   4 --- 0 = STANDARD (NON POLAR ANGLE) INPUT - *, 11, / ,
5   *   5 --- NOT USED ----- *, 11, / ,
6   *   6 --- 0 = SIDELINE INPUT = 1 POLAR INPUT --- *, 11, / ,
7   *   7 --- 4 = STANDARD (FORMAT 17) SPL INPUT --- *, 11, / ,
8   *   8 --- NOT USED ----- *, 11, / ,
9   *   9 --- 0 = USE TONE CORRECTION ----- *, 11, / ,
X   * 10 --- 1= APPLY SHIELDING EFFECT ----- *, 11, / /)
512 FORMAT (
1   *   11 --- 1 = PUNCHED OUTPUT ----- *, 11, / ,
5   *   12-15 ---- NOT USED ----- *, 411, / ,
6   *   16 --- 1 = GENERATE ATT. TRADES INTERNALLY --- *, 11, / ,
7   *   17 --- 1 = OMIT REL. JET VEL. CORRECTION --- *, 11, / ,
8   *   18 --- 1 = READ UNIFORM JET - SUPP. CARD --- *, 11, / ,
9   *   19 --- 1 = SHORT FORM OUTPUT ----- *, 11, / ,
X   * 20 --- 1 = STANDARD (LONG FORM) OUTPUT ----- *, 11, / /)
513 FORMAT (
1   *   21 --- 1 = JET CORE INPUT ----- *, 11, / ,
2   *   22 --- 1 = DEVELOP FOOTPRINT ----- *, 11, / ,
3   *   23 --- 1 = TIME PRINT (GEOM SUB) ----- *, 11, / ,
4   *   24 --- 1 = DETERMINE DOPPLER EFFECTS --- *, 11, / ,
5   *   25 --- 1 = FAN NOISE INPUT ----- *, 11, / ,
6   *   26 --- 1= OMIT JET SHIFT ----- *, 11, / ,
7   *   27 --- 1 = TURBINE NOISE INPUT ----- *, 11, / ,
8   *   28 --- 0 = USE GROUND ATENUATION ----- *, 11, / ,
9   *   29 --- 1 = USE JET, INLET, FWD DUCT ATTN.--- *, 11, / ,
A   * 30 --- 1 = GRID GEN. FOR DEVELOPING EPNL CNTRS *, 11, / /)
514 FORMAT (* 31-32 --- NO OF INPUT CASES ----- *, 12, / ,
1   * 33-34 --- NO OF POLAR ANGLES ----- *, 12, / ,
2   * 35-36 --- 24 = 1/3 OCTAVE DATA ----- *, 12, / ,
3   * 37-38 --- ----- *, 12, / ,
4   * 39-40 --- NO OF FLIGHT PROFILE SECTIONS ----- *, 12, / ,
5   * 41-42 --- NO OF R.P.M. INPUTS ----- *, 12, / )

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6  * 43-44 --- -----
7  * 45-46 --- NO OF MODES SUMMED ----- *, I2, /
8  * 47-48 --- NO OF GRIDPOINTS ----- *, I2, /
9  * 49-50 --- NO OF ATTENUATIONS ----- *, I2, /
A  * 51 --- NO OF ENGINES ----- *, F5.1)
522 FORMAT (// ' INLET FANDUCT JET FPNL SLANT D VEL'
 1  ' REL HUM TEMP SIDLINE OFFSET GAMMA ALPHA DURATION',
 2  ' D EPNL SEQ//)
523 FORMAT (1X, 4F5.1, F8.3, 5F8.2, 2X, F8.1, F8.0, F6.2, F7.2
 1 ,2F9.2, I4)
524 FORMAT (3A4, ' JET DATA (*, 5A4, 5X, 13A4//' ANGLE', 24(1X,I3)/
 1 (3X,25F4.0))
525 FORMAT (1X, 4F5.1, F8.3, 2F8.2, 30X, 15)
526 FORMAT (*0*)
C*** PREFERRED FULL OCTAVE
  DATA  FREQP
  1 / 63., 125., 250., 500., 1000., 2000., 4000., 8000. /
C*** COMMON FULL OCTAVE
  DATA  FREQC
  1 / 53., 106., 212., 424., 848., 1696., 3392., 6784. /
C
C  **  **  **  **  **  **  **  **  **  **  *  *  *  *
C
C      DO 900 I=1,24
900 FLOG24(I)= ALOGIC(FREQ24(I))
C
 10 READ (5,25,END=5990)
 1           MF,MN,ENGINE ,THRUST(1) ,OVHTIM(1)
C
C      -          INITIALZIATION -          -
C      FIRST1 = .TRUE.
C      NJ IS THE NO. OF ANGLES, NK IS THE NO. OF FREQ. BANDS
NI=1
NJ =MN(2)
NK =MN(2)
NSPRIN = MF(19)
NLPRIN = MF(20)
KKK = 1
M9 = 1
M10 = 1
A3 = 0.0
NCF=2
IF (MF(25).EQ. 1) NCF=4
DELJET = 0.0
DO 582 NC=1,4
ATENM(NC)=C.
CAT(NC)=1.
DO 582 NB=1,24
582 ATEN (NB,NC)=0.0
C
      READ(5,100) (TITLE(L),L=1,36)
      WRITE (PORF,511) (MF(JGW), JGW=1,10)

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      WRITE (PORF,512)      (MF(JGW), JGW=11,20)
      WRITE (PORF,513)      (MF(JGW), JGW=21,30)
      WRITE (PORF,514)      (MN(JGW), JGW= 1,10) ,ENGINE
      WRITE (PORF,35)
      1                   MF,MN,ENGINE,(TITLE(L),L=1,25)
      2   ,ATENM(1),(TITLE(L),L=27,30), ATENM(2), (TITLE(L),L=32,34)
      3   ,A3, TITLE(36)
      I =1
      READ(5,15)H(1,I),V(1,I),HREL(1,I),TEMP(1,I),X(1,I),Y(1,I),
      1   GAMM(1) ,ALP1 ,VMJET
      2   ,H(2,I),V(2,I),HREL(2,I),TEMP(2,I),X(2,I),Y(2,I) ,GAMM(2) ,ALP2
      3   ,VMJET
      ALPH(1) = ALP1 + GAMM(1)
      C(1) = 49.02 * SQRT(TEMP(1,1) + 459.67)
      WRITE (PORF,36)
      1   H(1,I),V(1,I),HREL(1,I),TEMP(1,I),X(1,I),Y(1,I),
      2   THRUST(I),OVHTIM(I),NJ
      3   ,H(2,I),V(2,I),HREL(2,I),TEMP(2,I),X(2,I),Y(2,I)
      IF(MF(7).NE.4) GO TO 159
      DO 1159 J=1,NJ
      READ(5,17) THETA(J,1,I),(SPL24(K,J,KKK),K=1,NK)
1159 DIRSL(J,1,I) = THETA(J,1,I)
      CALL WRITE (5,12, 1, 0, 1, 1, SPL24, DUMMY)
159 CONTINUE
      IF (MF(29) .NE. 1)                               GO TO 360
      M10 = MN(1C)
      READ (5,18) A3 ,VJJET, TITLE2
      DO 110 NA = 1,NJ
110 READ (5,17) DUMMY           ,(SPLJE(NB,NA),NB=1,24)
      CARR = 0.0
      IF (MF(3) .NE. 0)                               CARR = 6.0206
      DO 480 NB=1,NK
      DO 480 NA=1,NJ
      SPLTU(NB,NA) = 0.0
480 SPLJE(NB,NA) = SPLJE(NB,NA) + CARR
      CALL WRITE (2, 7, 0, 0, 1, 1, SPLJE, DUMMY)
      DO 155 NC = 1,NCF
155 READ (5,17) CAT(NC), (ATEN(NB,NC),NB=1,24)
      READ (5,18) A3 ,VJJET
      DO 508 NA =1,NJ
508 READ (5,17) DUM1 , (SPLREF(NB,NA,1),NB=1,24)
      CALL WRITE (2, 8, 1, 0, 1, 1, SPLREF,DUMMY)

C
C - - - CHANGED FOR NASA LEWIS ONLY - - - - -
C
      IF (MF(18) .GT. 0)      READ (5,17) DELJET
      IF (MF(25) .EQ. 0)                               GO TO 352
      DO 109 NA =1,NJ
      DO 109 NB =1,24
      IF (SPL24(NB,NA,KKK) .GT. SPLREF(NB,NA,1))      GO TO 108
      WRITE (6,41) NA,NB, SPL24(NB,NA,KKK),SPLREF(NB,NA,1)

```

```

SPLREF(NB,NA,1) = SPL24(NB,NA,KKK) -1.0
108 CONTINUE
    SPL24(NB,NA,1) = 10. * ALOG10(10.**(SPL24(NB,NA,KKK)*.1)
                                -10.**(SPLREF(NB,NA,1) *.1))
    1      IF (MF(21) .EQ. 1)
    1SPLJE(NB,NA) = 10. * ALOG10(
    2          10.**( (SPLJE(NB,NA) - DELJET) *.1)
    3          +10.**(SPLREF(NB,NA,1) *.1))
109 SPLREF(NB,NA,1) = 0.0
                                         GO TO 350
C - - - - - TO HERE - - - - -
352 CALL JSHIFT
350 CALL WRITE (5, 1, 1, 0, 1, 1, SPL24 , DUMY)
    CALL WRITE (2, 9, 0, 0, 1, 1, SPLJE , DUMY)
    WRITE (6,42) DELJET
    IF (MF(27) .EQ. 0)                               GO TO 360
    READ (5,181) A4 ,VTJET
    DO 112 NA =1,NJ
112 READ (5,17) THETA(NA,1,I), (SPLTU(NB,NA),NB=1,24)
    CALL WRITE (3, 1, 1, 0, 1, 1, SPLTU , DUMY)
360 CONTINUE
    DO 165 J=1,NJ
    IF (ABS(THETA(J,I,1) - 90.0) .LT. TOL) THETA(J,I,1) = 89.999
165 CONTINUE
    IF (MF(30) .EQ. 1)      M9 = MN(9)
    IF (MF(20) .EQ. 1)                               GO TO 166
    WRITE (6,221)
    WRITE (6,522)
166 CONTINUE
C - - - - - NR0=5
    DO 450 IJL = 1,M9
C - - - - -
    FIRST = .TRUE.
    LAST = .FALSE.
    IPT = 0
    IF (MF(30) .EQ. 1)      READ (5,15,END=9990)
    1          H(2,1), V(2,1), HREL(2,1), TEMP(2,1), X(2,1)
    2          ,Y(2,1), GAMM(2), ALP2 ,VMJFT
    IF (VMJET .EQ. 0.0)      VMJET = VJJET + V(2,1)
    ALPH(2) = ALP2 + GAMM(2)
    C(2) = 49.02 * SQRT (TEMP(2,1) + 459.67)
    ASSIGN 5010 TO LEAP
                                         GO TO 5090
C** - - - - - FOOT PRINT LOOP      ** - * - * - * - **
5010 READ (5,15) DBNL, DELX, DELY, Y(2,1)
5577 FORMAT ('$ $$M$$ ', BG15.7)
    WRITE(6,5577)DBNL, DELX, DELY, Y(2,1)
    ASSIGN 5020 TO LEAP
                                         GO TO 5090
5020 CALL BOUNDY (DBTEST, X(2,1), Y(2,1), +5025, +5035)

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5025 ASSIGN SC2C TO LEAP
5030 CALL BOUNDX (DBTEST, X(2,1), Y(2,1), +5033, +5035) GO TO 5090
5033 BOUND = .FALSE.
5035 BOUND = .TRUE.
5040 ASSIGN SC3C TO LEAP
5080 IF (LAST .AND. BOUND) GO TO 4020
5090 CALL GEOM (LAST)
    IF (MF(29) .EQ. 0) GO TO 509
    CALL JETHFV (VJJET, VMJET)
CD    CALL WRITE (2,9, 1, 0, 2, 1, SPLJE, DUMY)
C**
C**          LOBE SEPERATION LOOP      ** *** ** ** **
509 CONTINUE
NR5 = 0
    IF (MF(16) .EQ. 1) M10 = 1
    DO 400 IJK = 1,M10
        IF (MF(29) .EQ. 0) GO TO 510
        IF (MF(16) .EQ. 1) GO TO 515
        READ (5,15) (ATENM(L),L=1,4)
        NR0 = 1
515 CONTINUE
        DO 400 NR4 = 1,NR0
        ATENM(4) = ATENM1(NR4)
        DO 400 NR3 = 1,NR0
        WRITE (6,526)
        ATENM(3) = ATENM1(NR3)
        DO 400 NR2 = 1,NR0
        ATENM(2) = ATENM1(NR2)
        DO 400 NR1 = 1,NR0
        ATENM(1) = ATENM1(NR1)
        NR5 = NR5 + 1
        CALL LOBES
CD    CALL WRITE (1,9, 0, 0, 2, 1, SPL24, DUMY)
510 CALL TRANPO
CD    CALL WRITE (1,10, 1, 0, 2, 1, SPLREF,DUMY)
    CALL ROTATE(2,1,SPLREF,DIRSL)
CD    CALL WRITE (1,10, 0, 0, 2, 1, SPLREF,DUMY)
        CALL NOYFIT(SPLREF)
        CALL EPNLTI(SPLREF,V(2,1),2, DTIME)
        IF (NR5 .EQ. 1) EPNLF = EPNL(1,2)
        DEPNL = C.0
        IF (MF(29) .EQ. 1) DEPNL = EPNLF - EPNL(1,2)
        IF (MF(20) .EQ. 1) CALL WRITE (1, 10, 1, 1, 2, 1, SPLREF,D)
        IF (NLPRIN .EQ. 1) WRITE (PORF,522)
        SLD= SQRT(H(2,1)**2+ X(2,1)**2)
        WRITE (PORF,523) ATENM(1),ATENM(3),ATENM(2),ATENM(4),A3
1    ,EPNL(1,2), SLD , V(2,1), HREL(2,1), TEMP(2,1), X(2,1),
2    Y(2,1), GAMM(2), ALP2 ,DTIME ,DEPNL , NR5
        IF (MF(11) .EQ. 1)

```

```

1      WRITE (7,525) ATENM(1),ATENM(3),ATENM(2),ATENM(4),A3
2      ,EPNL(I,2), DEPNL ,NR5
CF          WRITE (PAPER,523) ATENM(1), ATENM(2), A3,
CF 1      EPNL(I,2), H(2,1), V(2,1), HREL(2,1), TEMP(2,1), X(2,1),
CF 2      Y(2,1), GAMM(2), ALP2 ,DTIME ,DEPNL
400 CONTINUE
    IF (MN(10).EQ.1 .AND. MN(9).GT.1)      MF(29) = 0
C** - - - FOOTPRINT LOGIC - - - - -
    IF (MF(22) .EQ. 0)                      GO TO 450
    IPT = IPT + 1
    POINTX(IPT) = X(2,1)
    POINTY(IPT) = Y(2,1)
    DBTEST = EPNL(I,2)
                                GO TO LEAP,(5010, 5020, 5030)

C
4020 CONTINUE
C4020 CALL PLCTFP (POINTX, POINTY, IPT, FIRST1, CURVX, CURVY, [CV])
    FIRST1 = .FALSE.

450 CONTINUE
213 CONTINUE
    GO TO 1C
9990 STOP
    END

```

SUBROUTINE WRITE ((I,JJ,K,L,M,N,SPL24 ,TITLE )

ZDM914 23 - 23

C  
C     II           TITLE 1 SUBSCRIPT  
C     JJ           TITLE 2 SUBSCRIPT  
C     K           = 1 SKIP PAGE  
C     L           = 0 SPL ARRAY ONLY  
C     M           = 1 COMPLETE OUTPUT  
C     M           = 1 INPUT CONDITIONS  
C     M           = 2 TRANSLATED CONDITIONS  
C     N           VALUE OF THE THIRD SUBSCRIPT ON ALL 3-D ARRAYS  
C     SPL24       SPL ARRAY  
C     TITLE       USED IN EPNL 2  
COMMON / AREA4 / MF(30),MN(10),ENGINE ,NT,NJ,NK,NII,CRV(400),MKT  
COMMON / AREA5 / PNL24(70,1),XNOY24(24),EPNL(4,21),PEAKPN (4),  
1           OASPL(70,1),TCPNL(70,1),XNOYMX(70), MTCB(70)  
COMMON/AREA6/THRUST(4),THRUSI (6),OVHTIM(4),TIM(70),TIM2(70,3,1)  
COMMON/GEOM1/H(3,4),V(3,4),HREL(3,4),TEMP(3,4),X(3,4),Y(3,4), PI  
1           ,GAMM(2) ,ALPH(2), S(70,21), W(70,2) ,C(2)  
COMMON/GEOM2/THETA(70,3,1),ALPHA(70,3,1),DIRSL(70,3,1),  
1   DISTSL(70,3,1),AAC(25,2,4)  
COMMON /HFVJE/  SPLFA(24,70,2) ,SPLJE(24,70), OAPS(70,2)  
1           ,SPLTU(24,70)  
DIMENSION      SPL24(24,17,1)  
DIMENSION      TITL1(8),TITL2(13), TITLE(18,3)  
DATA    TITL1    /\*ENG\*, 'JET ', 'TURB', ' ', 'FAN ', 'JET2',  
1        ' ', ' ', ' ', '/'  
2        ,TITL2    /\*BASE\*, 'ATT.', 'CORR', ' ', 'FWD ', 'AFT ', 'PRIM'  
3        ,'SEC ', 'TOT ', 'TRAN', '(DA)', 'JET2', ' '  
45 FORMAT ( 1X,18A4)  
195 FORMAT ('1')  
200 FORMAT (/55X,A4,10X,A4/)  
219 FORMAT (5X"ANGLE"2X,(24(1X,12,1X) ) /)  
220 FORMAT (5X,F6.2,1X,24F4.0)  
272 FORMAT (1X,5F8.2,2F10.2,4F8.2,14)  
273 FORMAT ( /5X   TIME   TIME   THETA   FLEVA   DIRSL   DIST'  
1 4X"HEIGHT    PNL    OASPL   TCPNL    NOYMAX   NB'/)  
C    - - - - -  
IF (K.EQ.1) WRITE (6,195)  
IF (MF(2).EQ.0) WRITE(6,45) (TITLE (LL,N),LL=1,18)  
WRITE (6,200) TITL1(JJ),TITL2(JJ)  
WRITE(6,219) (NB,NB=1,NK)  
DO 250 NA=NT,NJ  
250 WRITE (6,220) DIRSL(NA,M,1),(SPL24 (NB,NA,N ),NB=1,NK)  
IF (L.EQ.0) RETURN  
WRITE (6,273)  
DO 300 NA=NT,NJ  
300 WRITE (6,272) TIM(NA),TIM2(NA,M,1),THETA(NA,M,1),ALPHA(NA,M,1),  
1  DIRSL(NA,M,1),DISTSL(NA,M,1),S(NA,M),PNL24(NA,1),  
2  OASPL(NA,1),TCPNL(NA,1),XNOYMX(NA), MTCB(NA)  
RETURN  
END

## SUBROUTINE GEOM (LAST)

ZDM914 02 - 2T

```

C LOGICAL *4 LAST
REAL * 4 BIG/1.0E+50/
COMMON / AREA4 / MF(30), MN(10), ENGINE , NI, NJ, NK, NII, CRV(400), MKT
COMMON/AREA6/THRUST(4), THRUSI (6), OVHTIM(4), TIM(70), TIM2(70,3,1)
COMMON/GEOM1/H(3,4), V(3,4), HREL(3,4), TEMP(3,4), X(3,4), Y(3,4), PI
1 , GAMM(2), ALPH(2), S(70,2), W(70,2) , C(2)
COMMON/GEOM2/THETA(70,3,1), ALPHA(70,3,1), DIRSL(70,3,1),
1 DISTSL(70,3,1), AAC(25,2,4)
COMMON / GEOM7 / PSI(70)
DIMENSION CRVX(50) ,P(3,4)
C S (NA, 1 ) TRANSLATED HEIGHT
C S (NA, 2 ) REAL HEIGHT
63 FORMAT (//10X*** X(2,1) TOO LARGE S(I3,1) IS NEG*,2X,F15.5
1 // 10X,E12.5///)
PI180=180./PI
NJ=MN(2)
DO 130 NA=1,NJ
DO 130 L=1,2
S(NA,L)=H(L,1)
DISTSL(NA,L,1)=H(L,1)
DIRSL(NA,L,1)=THETA(NA,1,1)
ALPHA(NA,L,1)=0.
130 TIM2(NA,L,1)=0.
DO 140 L=1,2
140 C(L)=49.02*SQRT(TEMP(L,1)+459.6)
L=2
C9R=GAMM(L)/PI180
C8R=C9R
IF (C9R.EQ.0.) GO TO 180
YCRIT=((H(1,1)/COS(C9R))-H(2,1))/TAN(C9R)
IF (C9R.GT.0. AND.Y(2,1).LT.YCRIT) Y(2,1)=YCRIT
IF (C9R.LT.0. AND.Y(2,1).GT.YCRIT) Y(2,1)=YCRIT
180 CONTINUE
XL=0.
Z =0.
H1=H(L,1)+Y(L,1)*TAN(C9R)
Y1=H1
H2=H(L,1)+(Y(L,1)+1000.)*TAN(C9R)
HDD=H1
IF(ABS(X(L,1)).LT.1.) GO TO 190
XL=1000./COS(C9R)
Z=SQRT(X(L,1)*X(L,1)+H2*H2+1000000.)
HDD=SQRT(H1*H1+X(2,1)*X(2,1))
Y1=HDD
C8R=ARCOS((HDD*HDD+XL*XL-Z*Z)/(2.*HDD*XL))-PI/2.
NI=0
TEN=10.
OST=170.
182 CONTINUE

```

```

NI=N I+1
TP I=TEN/PI180
IF(C8R.GT.C0.) GO TO 185
TP I=OST/P1180
NI=1
NJ=NJ-1
185 CONTINUE
STEMP=H1-HDD*TAN(C9R)/(TAN(TPI)+TAN(C8R))
IF(STEMP.GT.0.) GO TO 190
TPD=TP I*P1180
WRITE(6,63) NA,STEMP,C9R,C8R,TPD,HDD,H(2,1)
TEN=TEN+10.
OST=OST-10.
GO TO 182
190 CONTINUE
DO 200 NA=NI,NJ
C2R=DIRSL(NA,2,1)/PI180
S(NA,1)=Y1*TAN(C2R)/(TAN(C2R)+TAN(C8R))
DISTSL(NA,2,1)=S(NA,1)/SIN(C2R)
XREF=-S(NA,1)/TAN(C2R)
S(NA,2)=XREF*TAN(C9R)+H1
APR=ALPH(L1)/PI180
PSI(NA)=ARCCOS((-XREF*COS(APR)-S(NA,2)*SIN(APR))/DISTSL(NA,L,1))
1 *P1180
DELT=DISTSL(NA,L,1)/C(L)
CT=S(NA,1)+V(L,1)*DELT*SIN(C8R)
CT=-V(L,1)*DELT*COS(C8R)
IF(DIRSL(NA,L,1).NE.90.) CT=CT+S(NA,1)/TAN(C2R)
THETA(NA,L,1)=90.
IF(CT.NE.0.) THETA(NA,L,1)=ATAN(CT/CT)*PI180
IF(THETA(NA,L,1).LT.0.) THETA(NA,L,1)=180.+THETA(NA,L,1)
C12=0.
IF(THETA(NA,L,1).NE.90.) C12=TAN(THETA(NA,L,1)/PI180)
STHET=HDD*TAN(C12)/(TAN(C12)+TAN(C8R))
IF(C12.NE.0.AND.V(L,1).NE.0.) TIM2(NA,L,1)=-STHET/C12/V(L,1)
IF(V(L,1).NE.0.)ALPHA(NA,L,1)=ARSIN(S(NA,2)/DISTSL(NA,L,1))*PI180
200 CONTINUE
RETURN
END

```

## SUBROUTINE TRANPO

ZDM914 06 - 3T

```

C COMMON / AREA1 / SPL24(24,70,2),SPLREF(24,70,1)
COMMON / AREA3 / FREQ24(24),FLOG24(24),FREQ8(8),FLOG8(8),BWN(24)
COMMON / AREA4 / MF(30),MN(10),ENGINE ,NI,NJ,NK,NIT,CRV(400),MKT
COMMON/GEOM1/H(3,4),V(3,4),HREL(3,4),TEMP(3,4),X(3,4),Y(3,4), PI
1      ,GAMM(2), ALPH(2), S(70,2), W(70,2) ,C(2)
COMMON/GEOM2/THETA(70,3,1),ALPHA(70,3,1),DIRSL(70,3,1),
1 DISTSL(70,3,1),AAC(25,2,4)
DIMENSION CRVX(50),FREQX(24),FLOGX(24)
DIMENSION ATT1(24),ATT2(24)
241 FORMAT(1, 3X,4HH = ,F12.2,4HV = ,F12.2,7HHREL = ,F12.2,7HTEMP = ,
1   F12.2,4HX = ,F12.2,4HY = ,F12.2)
242 FORMAT(3X,2IHABSORPTION CORRECTORS )
243 FORMAT(3X,4(I3,1H=,G15.6,3X))
C * * * * * * * * * * * * * * * * * * * * * * *
I=1
NNB=3
CALCULATE ATMOSPHERIC ATTENUATION
DO 100 L=1,2
C** TEST TO AVOID ARSIN .GT. 1.0
DO 50 J=NI,NJ
  IF (S(J,L) .GT. DISTSL(J,L,I))    S(J,L) = DISTSL(J,L,I) -.0001
50 CONTINUE
CALL ATABC(NNB,TEMP(L,I),HREL(L,I),AAC(1,L,I))
IF(MF(23).EQ.0) GO TO 100
WRITE(6,241) H(L,I),V(L,I),HREL(L,I),TEMP(L,I),X(L,I),Y(L,I)
WRITE(6,242)
DO 861 M=1,6
M2=M+6
M3=M+12
M4=M+18
861 WRITE(6,243) M,AAC(M,L,I),M2,AAC(M2,L,I),M3,AAC(M3,L,I),M4,
1 AAC(M4,L,I)
100 CONTINUE
6 FORMAT (/12F9.3)
CD     WRITE (6,6) AAC
DO 498 L=1,24
ATT1(L)=C.
ATT2(L)=C.
498 CONTINUE
DO 501 K=1,NK
FREQX(K)=FREQ24(K)
501 FLOGX(K)=FLOG24(K)
CONST=6.2
I=1
DO 800 J=NI,NJ
      EGA - EXTRA GROUND ATTENUATION
IF(MF(28).GT.2) GO TO 1602
IF(MF(28).NE.1) CALL EGA1(DISTSL(J,1,I),FREQX,ALPHA(J,1,I)
1   ,NK, ATT1)

```

```

    IF(MF(28).NE.2) CALL EGA1(DISTSL(J,2,I),FREQX,ALPHA(J,2,I)
1      ,NK, ATT2)
1602 CONTINUE
CD      WRITE (6,6) ATT1
CD      WRITE (6,61 ATT2
CD      IF (MF(24).EQ.0)                               GO TO 630
C          TRANPOSE SPL VALUES
C          REMOVE DOPPLER SHIFT AND CORRECT FOR BAND WIDTH
DOP=1.-V(1,1)*COS(DIRSL(J,1,1)/180.*P1)/C(1)
DOPLOG=10.* ALOG10(DOP)
DO 601 K=1,NK
CRV(K)=SPL24(K,J,1)-10.*FLOGX(K)+CONST - DOPLOG
601 CRVX(K)=ALOG10(DOP*FREQX(K))
DO 602 K=1,NK
CRV(K)=SPLREF(K,J,1) = TBLU1(FLOG24(K), CRVX, CRV, 1, 24)
CD      CALL WRITE (1, 7, 1, 0, 2, 1, SPLREF, DOP)
DISTAN=S(J,2)
IF (MF(10).EQ.1) DISTAN=SQRT(S(J,2)*S(J,2)+X(2,1)*X(2,1))
ENGICB=(1.+SQRT(SIN(S(J,2)/DISTAN)))*5.* ALOG10(ENGINE)
DO 603 K=1,NK
SPLREF(K,J,1)=SPLREF(K,J,1)+20.* ALOG10(DISTSL(J,1,1)/DISTSL(J,2,1)
1  )+DISTSL(J,1,1)*AAC(K,1,1)/1000.-DISTSL(J,2,1)*AAC(K,2,1)/1000.
2  +ENGICB +ATT1(K)-ATT2(K)
603 CONTINUE
CD      CALL WRITE (1, 8, 0, 0, 2, 1, SPLREF, DOP)
C          ADD DOPPLER SHIFT AND CORRECT FOR BAND WIDTH
DOP=1.-V(2,1)*COS(DIRSL(J,2,1)/180.*P1)/C(2)
DOPLOG=10.* ALOG10(DOP)
DO 600 K=1,NK
CRV(K)=SPLREF(K,J,1)
600 CRVX(K)=ALOG10(FREQX(K)/DOP)
DO 650 K=1,NK
SPLREF(K,J,1)=TBLU1(CRVX(K), FLOG24, CRV, 1, 24)
CD      CALL WRITE (1, 7, 1, 0, 2, 1, SPLREF, DOP)
DO 651 K=1,NK
651 SPLREF(K,J,1)=SPLREF(K,J,1)+10.*FLOGX(K)-CONST + DOPLOG
CD      CALL WRITE (1, 8, 0, 0, 2, 1, SPLREF, DOP)
GO TO 800
630 CONTINUE
DISTAN=S(J,2)
IF (MF(10).EQ.1) DISTAN=SQRT(S(J,2)*S(J,2)+X(2,1)*X(2,1))
ENGIDB=(1.+SQRT(SIN(S(J,2)/DISTAN)))*5.* ALOG10(ENGINE)
DO 700 K=1,NK
SPLREF(K,J,1)=SPL24(K,J,1)+20.* ALOG10(DISTSL(J,1,1)/DISTSL(J,2,1))
1  +DISTSL(J,1,1)*AAC(K,1,1)/1000.-DISTSL(J,2,1)*AAC(K,2,1)/1000.
2  +ENGICB +ATT1(K)-ATT2(K)
700 CONTINUE
800 CONTINUE
RETURN
END

```

SUBROUTINE ROTATE (L,KKK,SPL24,THETA)

ZDM914 23 - JT

```
C      REAL *4 SPL24(24,70,2), THETA(70,3,1)
C      COMMON / AREA4 / MF(30), MN(10), ENGINE, NT, NJ, NK, NII, CRV(400), MKT
C      COMMON / GEOM1 / H(3,4), V(3,4), HREL(3,4), TEMP(3,4), X(3,4),
1      Y(3,4), PI, GAMM(2), ALPH(2), S(70,2), W(70,2), C(2)
C      COMMON / GEOM7 / PSI(70)
C      *      *      *      *      *      *      *      *      *      *      *
C      DO 120 NB= 1,NK
C      DO 110 NA=NI,NJ
110 CRV(NA) = SPL24(NB,NA,KKK)
      DO 120 NA= NI,NJ
      AN=PSI(NA)
120 SPL24(NB,NA,KKK) = TBLU1(AN,THETA(1,L,1), CRV, 1, NJ)
      RETURN
      END
```

## SUBROUTINE NOVFT(SPLNOY)

ZDM914 05 - 3T

```

COMMON / AREA4 / MF(30), MN(10), ENGINE , NI, NJ, NK, NII, CRV(400), MKT
COMMON / AREA5 / PNL24(70,1), XNOY24(24), EPNL(4,2), PEAKPN (4),
1          DA SPLIT(70,1), TCPNL(70,1), XNOYMX(70), MTCB(70)
COMMON/CCNNOY/ABPS(24), BBPS(24), AL0(24), BL0(24), AM(24), BM(24)
DIMENSION SPLNOY(24,70,1)
CONST=0.15
I = 1
DO 960 J=NI,NJ
DO 900 K=1,NK
KK=K
XNOY24(K)=0.
IF(ABPS(KK).NE.0.0.AND.SPLNOY(K,J,I).GT.0.)
1  XNOY24(K)=SPLNOY(K,J,I)/ABPS(KK)
IF(SPLNOY(K,J,I).LT.ABPS(KK)) GO TO 900
XNOY24(K)=10.**(AM(KK)*(SPLNOY(K,J,I)-AL0(KK)))
IF(SPLNOY(K,J,I).GT.BBPS(KK))
1  XNOY24(K)=10.**(BM(KK)*(SPLNOY(K,J,I)-BL0(KK)))
900 CONTINUE
XMAX=0.
XSUM=0.
DO 950 K=1,NK
XSUM=XSUM+XNOY24(K)
IF(XNOY24(K).GT.XMAX) XMAX=XNOY24(K)
950 CONTINUE
XNOYTL=XMAX+CONST*(XSUM-XMAX)
XNOYMX(J)=XNOYTL
PNL24(J,I)=40.
IF(XNOYTL.GT.1.) PNL24(J,I)= ALOG10(XNOYTL)/.030103+40.
DASPL(J,I)=DAFUNC(SPLNOY(1,J,I))
960 CONTINUE
RETURN
END

```

## SUBROUTINE EPNLTI(SPLEPN ,VEL,L ,DTIME)

ZDM914 03 - 03

```

C
COMMON / AREA1 / SPL24(24,70,2),SPLREF(24,70,1)
COMMON / AREA3 / FREQ24(24),FLOG24(24),FREQ8(8),FLOG8(8),BWN(24)
COMMON / AREA4 / MF(30),MN(10),ENGINE ,NI,NJ,NK,NII,CRV(400),MKT
COMMON / AREA5 / PNL24(70,1),XNDY24(24),EPNL(4,2),PEAKPN (4),
1          OASPL(70,1),TCPNL(70,1) ,XNOYMX(70), MTCB(70)
COMMON/AREA6/THRUST(4),THRUSI (6),OVHTIM(4),TIM(70),TIM2(70,3,1)
COMMON/GEOM1/H(3,4),V(3,4),HREL(3,4),TEMP(3,4),X(3,4),Y(3,4), PI
1          ,GAMM(2) ,ALPH(2) ,S(70,2) ,W(70,2) ,C(2)
COMMON/GEOM2/THETA(70,3,1),ALPHA(70,3,1),DIRSL(70,3,1),
1  DISTSL(70,3,1),AAC(25,2,4)
DIMENSION SPLEPN (24,60,1)
I = 1
EPNL(1,1) = 0.0
EPNL(1,2) = 0.0
DO E33 NB=1,24
DO 833 NA=NI,NJ
IF(SPLEPN(NB,NA,1) .LT. 0.0) SPLEPN(NB,NA,1) = 0.0
833 CONTINUE
DO 900 J=NI,NJ
TCPNL(J,I)=PNL24(J,I)
MTCB(J) = 1
IF (MF(91) .NE. 0)                               GO TO 900
TCPNL(J,I)=PNL24(J,I)+FMTONE(SPLEPN (1,J,I), NTCB)
MTCB(J) = NTCB
900 CONTINUE
IF(VEL.EQ.0.) GO TO 1000
M=1
DM=M*10.
DM1=M
XNI=90.
XNI=0.
XHUN=XNI+DM
PNLMAX=0.
DO 161 J=NI,NJ
IF(TCPNL(J,I).LT.PNLMAX) GO TO 161
PNLMAX=TCPNL(J,I)
JMAX = J
PEAKPN(I) = PNLMAX
161 CONTINUE
IF (PNLMAX .LE. XNI)                               RETURN
IF (PNLMAX .LE. XHUN)      DM = PNLMAX - XNI
PM = PNLMAX - DM
IF (TCPNL(1,I) .GT. PM .OR. TCPNL(NJ,I).GT.PM)   RETURN
DO 162 J=1,JMAX
JL=J
IF(TCPNL(JL+1,I) .GE. PM .AND. TCPNL(JL+1,I) .GE. XNI) GOTO163
162 CONTINUE
GO TO 164
163 CONTINUE

```

```

DO 165 J=JMAX,NJ
JT = NJ - J + JMAX
IF(TCPNL(JT-1,I) .GE. PM .AND. TCPNL(JT-1,I) .GE. XNII) GOT0166
165 CONTINUE
GO TO 164
166 CONTINUE
TL=(PNLMAX-DM-TCPNL(JL,I))/(TCPNL(JL+1,I)-TCPNL(JL,I))
1 *(TIM2(JL+1,L,I)-TIM2(JL,L,I))+TIM2(JL,L,I)
TT=TIM2(JT,L,I)-(PNLMAX-DM-TCPNL(JT,I))/(TCPNL(JT-1,I)-
1 TCPNL(JT,I))* (TIM2(JT,L,I)-TIM2(JT-1,L,I))
DTIME = TT - TL
EPNL(I,M)=PNLMAX*10.* ALOG10((TT-TL)/DM1/15.)
SUM=0.
SUM=.5*((10.*((PNLMAX-DM)/10.))+10.*((TCPNL(JL+1,I)/10.)*
1 (TIM2(JL+1,L,I)-TL))+.5*(10.*((PNLMAX-DM)/10.)*
2 10.*((TCPNL(JT-1,I)/10.))* (TT-TIM2(JT-1,L,I)))
NL=JL+1
NT=JT-2
DO 171 N=NL,NT
171 SUM=SUM+.5*(10.*((TCPNL(N,I)/10.))+10.*((TCPNL(N+1,I)/10.))*(
1 (TIM2(N+1,L,I)-TIM2(N,L,I)))
EPNL(I,2)=10.*ALOG10(SUM/10.)
GO TO 160
164 EPNL(I,M)=-1000.
EPNL(I,2)=-1000.
160 CONTINUE
1000 RETURN
END

```

## SUBROUTINE JSHIFT

ZDM914 24 - KT

```

COMMON / AREA1 / SPL24(24,70,2), SPLREF(24,70,1)
COMMON / AREA4 / MF(30), MN(10), ENGINE, NI, NJ, NK, NII, CRV(400), MKT
COMMON / HFVJE / SPLFA(24,70,2), SPLJE(24,70), OASP(70,2)
1     , SPLTU(24,70)
REAL *4 ACC(25), CRVX(25), SPLRO(24,70)
1   , SHRANK(24) / 17., 16.8, 16.5, 16., 15.0, 14.5, 14.0, 13.5, 12.5,
2   12., 11.5, 11., 10., 9., 8., 6., 4., 1., 0., .5, .1.5, 3., 5.,
3   7./
EQUIVALENCE (SPLREF(1,1,1), SPLRO(1,1))
C *      *      *      *      *      *      *      *      *      *      *      *      *
KKK = 1
N1 = 19
MN8 = 15
NQ2 = MN8 + 1
IF (MF(26) .EQ. 1)                               GO TO 400
CALL JETSMO (SPLJE)
CALL JETSMO (SPLRO)
400    DO 508 NB = 1,24
DO 508 NA=NI,NJ
SPLREF (NB,NA,1) = SPLJE(NB,NA) - SPLREF(NB,NA,1)
IF (SPLJE(NB,NA) .GE. SPL24(NB,NA,KKK)) SPLJE(NB,NA) =
1           SPL24(NB,NA,KKK) - 1.0
508 CONTINUE
IF (MF(26) .EQ. 1) RETURN
DO 485 NA=NI,NJ
SAV1 = SPLJE(MN8,NA)
DO 518 NB = 1,MN8
SPLJE(NB,NA) = SPL24(NI,NA,KKK) - SHRANK(NB)
IF (SPL24(NB,NA,KKK) .LE. SPLJE(NB,NA))
1   SPLJE(NB,NA) = SPL24(NB,NA,KKK) - 4.0
518 SPLJE(NB,NA) = 10. * ALOG10(10.**(SPL24(NB,NA,KKK)* .1)
1           -10.**(SPLJE(NB,NA) * .1))
SAV1 = SPLJE(MN8,NA) - SAV1
DO 485 NB = NQ2,24
SPLJE(NB,NA) = SPLJE(NB,NA) + SAV1
IF (SPLJE(NB,NA) .LE. SPL24(NB,NA,KKK)) GO TO 485
SAV1 = SPL24(NB,NA,KKK) - .05 - (SPLJE(NB,NA) - SAV1)
SPLJE(NB,NA) = SPL24(NB,NA,KKK) - .05
485 CONTINUE
RETURN
END

```

## SUBROUTINE JETSMD (SPLJE)

ZDM914 25 - KT

```
C      COMMON / AREA4 / MF(30), MN(10), ENGINE, NI, NJJ(1), NK, NII, CRV(400)
1, MKT
REAL *4 AC(25), CRVX(25), SPLJE(24,70)
EQUIVALENCE (NJ, NJJ(1))
C   *      *      *      *      *      *      *      *      *      *      *
DO 100 NA=NI,NJ
N = 0
DO 50 J=2,23,3
N = N + 1
CRVX(N) = J
AC(N) = SPLJE(J,NA)
50 CONTINUE
DO 100 NB = 1,24
XB = NB
SPLJE(NB,NA) = TBLU1(XB, CRVX, AC, 1, N)
100 CONTINUE
RETURN
END
```

## SUBROUTINE JETHFV (VJJET, VMJET)

ZDM914 22 - KT

```

C
      REAL *4 AC(25), CRVX(25)
      COMMON / AREA1 / SPL24(24,70,2), SPLREF(24,70,1)
      COMMON / AREA3 / FREQ24(24), FLOG24(24), FREQ8(8), FLOG8(8), BWN(24)
      COMMON / AREA4 / MF(30), MN(10), ENGINE, NI, NJ, NK, NII, CRV(400), MKT
      COMMON / GEOM1 / H(3,4), V(3,4), HREL(3,4), TEMP(3,4), X(3,4)
      1 , Y(3,4), PI, GAMM(2), ALPH(2), S(70,2), W(70,2), C(2)
      COMMON / GEOM2 / THETA(70,3,1), ALPHA(70,3,1), DIRSL(70,3,1),
      1 DISTSL(70,3,1), AAC(25,2,4)
      COMMON / HFVJE / SPLFA(24,70,2), SPLJE(24,70), OASP(70,2)
      1 , SPLTU(24,70)
      DIMENSION DE DB(17) , THE(17)
      DATA DE DB/-4., -3.52288, -3. , -2.52288, -2. , -1.52288, -0.39794,
      1 -0.22185, -0.47712, -0.22185, -0.39794, -1.52288, -2.0, -2.52288
      2 , -3. , -3.52288, -4. /
      3 , THE/10., 20., 30., 40., 50., 60., 70., 80., 90., 100.,
      4 110., 120., 130., 140., 150., 160., 170. /
      DO 200 NA=NI,NJ
 200 IF (DIRSL(NA,1,1).GE.88.AND.DIRSL(NA,1,1).LE.92.) NJ90=NA
      K=1
      L=2
      DO 205 NA=NI,NJ
      DDDB = 10.0 ** (TBLU1(DIRSL(NA,1,1) ,THE ,DEDB ,1 ,17))
      IF (NA.LE.NJ90) GO TO 210
      K=2
      L=1
 210 DO 205 NB=1,NK
      IF (MF(25) .EQ. 1) GO TO 203
      IF (SPLJE(NB,NA).GE.SPL24(NB,NA,1)) SPL24(NB,NA,1)=SPLJE(NB,NA)+.5
      SPL24(NB,NA,1)=10.* ALOG10(10.**(SPL24(NB,NA,1)/10.))
      1 -10.**((SPLJE(NB,NA )/10.))
      2 -10.**((SPLTU(NB,NA )/10.))
 203 SPLFA(NB,NA,K)=SPL24(NB,NA,1)-DDDB
 205 SPLFA(NB,NA,L)=10.* ALOG10(10.**(SPL24(NB,NA,1)/10.))
      1 -10.**((SPLFA(NB,NA,K)/10.))
      DO 350 N=1,2
      DO 310 J=NI,NJ
  CC      WRITE (6,5577) J, N, (SPLFA(I,J,N),I=1,24)
  310 OASP (J,N)= OAFUNC(SPLFA(1,J,N))
  C5577 FORMAT ('0$JETS$/ (5X,7G15.5)')
  C      FIND THE MAX OF OASP
      OAPM=0.
      DO 320 J=NI,NJ
  320 IF (OASP(J,N).GT.OAPM) OAPM=OASP(J,N)
      DO 330 J=NI,NJ
  C      - - - NASA-LEWIS CHANGE      1-11-71 - - - - -
  CN330 OASP (J,N)=OASP (J,N)/OAPM
  330 OASP (J,N)=1.0
  C      - - - -- - - - - - - - - - - - - - - - - - - -
  350 CONTINUE

```

```
DO 390 NA=NI,NJ
    DO 390 NB = 1,24
390 SPLJE(NB,NA) = SPLJE(NB,NA) - SPLREF(NB,NA,1)
    IF (MF(17) .EQ. 1) RETURN
    IF (VMJET.EQ.0.1 VMJET = VJJET + V(2,1)
    VRATIO = (VMJET - V(2,1)) / VJJET
    DELSPL = 80. * ALOG10(VRATIO)
    DO 410 NA=NI,NJ
        DO 405 NB = 1,24
        AC(NB) = SPLJE(NB,NA) + DELSPL
405 CRVX(NB) = ALOG10(FREQ24(NB) *(VRATIO))
        DO 410 NB=1,24
410 SPLJE(NB,NA) = TBLU1(FLOG24(NB),CRVX,AC + 1, 24)
    RETURN
END
```

## SUBROUTINE LOBESE

ZDM914 21 - JT

C

```
COMMON / AREA1 / SPL24(24,70,2), SPLREF(24,70,1)
COMMON / AREA4 / MF(30), MN(10), ENGINE , NI,NJ,NK,NEI,CRV(400),MKT
COMMON / HFVJE / SPLFA(24,70,2), SPLJE(24,70), OASP(70,2)
1     ,SPLTU(24,70)
COMMON / HFVJT / CAT(4), ATEN(24,4), ATATM(24,4), ATENM(4)
DIMENSION SPLAT(2)
DO 18 N=1,4
DO 18 NB=1,24
18 ATATM(NB,N)=ATEN(NB,N)/CAT(N)
DO 560 N =1,2
DO 560 NB=1,NK
      ATATM(NB,N) = ATATM(NB,N) + ATATM(NB,N+2)
560 CONTINUE
DO 320 NA=NI,NJ
DO 320 NB=1,NK
DO 17 N=1,2
17 SPLAT(   N)=SPLFA(NB,NA,N) - OASP (NA,N)*ATATM(NB,N)
320 SPL24(NB,NA,1)=10.* ALOG10(10.**(SPLAT(   1)*.1 )
1                               +10.** (SPLAT(   2)*.1 )
2                               +10.** (SPLJE(NB,NA) *.1 )
2                               +10.** (SPLTU(NB,NA) *.1 ) )
RETURN
END
```

```

FUNCTION FMTONE(SPZ, NTCB) ZDM914 08 - 08
C
      DIMENSION SPL(25),S(25),SPLP(25),SP(25),SBAR(25),SPLPP(25),FC(25)
1      ,SPZ(1)
      EQUIVALENCE (SP,S,SPLPP) + (SPLP, SBAR,FC)
C
      DO 110 K=1,24
      SPL(K)=SPZ(K)
      IF (SPL(K) .LT. 0.0)    SPL(K) = 0.0
110 SPLP(K)=SPL(K)
      S(3)=0.
      DO 100 K=4,24
100 S(K)=SPL(K)-SPL(K-1)

      DO 150 K=5,24
      IF(ABS(S(K)-S(K-1)).LE.5.) GO TO 150
      IF(S(K).GT.0.) GO TO 120
      IF(S(K-1).LE.0.) GO TO 150
      SPLP(K-1)=(SPL(K-2)+SPL(K))/2.
      GO TO 150
120 IF(S(K).LE.S(K-1)) GO TO 150
      IF(K.GT.23) GO TO 130
      SPLP(K)=(SPL(K-1)+SPL(K+1))/2.
      GO TO 150
130 SPLP(24)=SPL(23)+S(23)
150 CONTINUE

      DO 160 K=4,24
160 SP(K)=SPLP(K)-SPLP(K-1)
      SP(3)=SP(4)
      SP(25)=SP(24)
      DO 170 K=3,23
170 SBAR(K)=(SP(K)+SP(K+1)+SP(K+2))/3.
      SPLPP(3)=SPL(3)
      DO 180 K=4,24
180 SPLPP(K)=SPLPP(K-1)+SBAR(K-1)
      FM=-1000.
      NTCB = C
      DO 250 K=3,24
      F=SPL(K)-SPLPP(K)
      IF(F.LT.3.) GO TO 220
      IF(K.LT.11.OR.K.GT.21) GO TO 210
      FC(K)=AMIN1(F/3.,20./3.)
      GO TO 230
210 FC(K)=AMIN1(F/6.,20./6.)
      GO TO 220
220 FC(K)=0.C
230 IF (FC(K) .LT. FM)                               GO TO 250
      FM = FC(K)
      NTCB = K
250 CONTINUE

```

FM TONE=FM  
RETURN  
END

SUBROUTINE ATABC(NB,TEMP,HRLF,AAC)

ZDM914 04 - 3T

C  
1        DIMENSION AAC(1),PFREQ(8),CFREQ(8),TFREQ(25),FREQ(41),X1(25),  
2                Y1(25),X2(19),Y2(19),X3(2),Y3(2,11),Z3(11),X4(2),  
2                Y4(2,11),Z4(11)  
1        EQUIVALENCE (FREQ(1),PFREQ(1)),(FREQ(9),CFREQ(1)),  
1                (FREQ(17),TFREQ(1)),X1(1),(Z3(1),Z4(1))  
1        DATA PFREQ/1.799341,2.09691,2.39794,2.69897,3.,3.30103,3.60206,  
1                3.773786/  
1        DATA CFREQ/1.724276,2.025306,2.326336,2.628389,2.929419,3.230449,  
1                3.531479,3.702431/  
1        DATA TFREQ/1.69897,1.799341,1.90309,2.,2.09691,2.20412,2.30103,  
1                2.39794,2.498311,2.60206,2.69897,2.799341,2.90309,3.,  
2                3.09691,3.20412,3.30103,3.39794,3.498311,3.60206,  
3                3.650308,3.748963,3.851258,3.951338,4.049218/  
1        DATA Y1/.22,.25,.28,.32,.36,.41,.45,.51,.56,.63,.7,.79,.89,.1.01,  
1                1.14,1.27,1.41,1.57,1.75,1.98,2.10,2.36,2.67,2.98,3.26/  
1        DATA X2/0.,.5,.65,.75,.9,1.,1.12,1.25,1.5,2.,2.5,3.,3.5,4.,4.5,5.,  
1                6.,6.5,8./  
1        DATA Y2/0.,.69,.9,.96,1.,1.,.95,.865,.75,.57,.45,.37,.31,.27,.24,  
1                .225,.21,.2,.2/  
1        DATA X3/-0.522879,.954243/  
1        DATA Y3/.0086,1.498311,.079181,1.556303,.136721,1.618048,.198657,  
1                1.672098,.247973,1.724276,.292256,1.770852,.342423,  
2                1.812913,.380211,1.869232,.431364,1.908485,.477121,  
3                1.954243,.525045,1.995635/  
1        DATA Z3/0.,10.,20.,30.,40.,50.,60.,70.,80.,90.,100./  
1        DATA X4/1.,2./  
1        DATA Y4/-0.958607,.021189,-.744727,.230449,-.545155,.431364,  
1                -.3467E7,.633468,-.180454,.80618,-.017729,.963788,.113943,  
2                1.10721,.255273,1.255273,.406540,1.389166,.531479,  
3                1.518514,.662758,1.643453/  
1        IF(HREL.LT.10.) HREL=10.  
1        IF(HREL.GT.100.) HREL=100.  
1        HABS=10.\*TBLU2 ALOG10(HREL),TEMP,X4,Z4,Y4,1,1,2,11,2,11)  
1        I=16  
1        NF=25  
400      DO 500 J=1,NF  
1        K=I+J  
1        ALPC=10.\*\*(2.06\*FREQ(K)-7.578)  
1        HMOLMX=TBLU1(FREQ(K),X1,Y1,1,25)  
1        ALPMAX=10.\*TBLU2(FREQ(K)-3.,TEMP,X3,Z3,Y3,1,1,2,11,2,11)  
1        ARG=FABS/HMOLMX  
1        AMAMM=TBLU1(ARG,X2,Y2,1,19)  
1        ALPMOL=AMAMM\*ALPMAX  
500      AAC(J)=ALPC+ALPMOL  
1        RETURN  
1        END

## SUBROUTINE EGAI(DIST,FREQ,BETA,N,ATT)

ZDM915 09 - 8T

```

C
DIMENSION FREQ(N),ATT(N),AK(7),AT(63),FR(7),R(9)
DATA           AK / 2.7, 2.27, 2.19, 1.93, 1.63, 2.11, 2.11/
DATA           AT/
1   .5  ,1.  ,1.  ,1.  ,1.5 ,2.  ,4.  ,4.5 ,5.  ,
2   1.  ,1.5 ,2.  ,2.  ,2.5 ,3.  ,6.  ,7.  ,7.  ,
3   1.  ,1.5 ,2.  ,2.5 ,3.  ,4.  ,7.5 ,9.  ,9.  ,
4   1.  ,1.5 ,2.  ,3.  ,4.  ,5.  ,10. ,11. ,11. ,
5   1.  ,1.5 ,2.  ,3.5 ,5.  ,6.  ,11. ,13. ,14. ,
6   1.  ,2.  ,3.  ,4.5 ,6.  ,7.  ,13. ,15. ,15. ,
7   1.  ,2.  ,3.  ,4.5 ,6.  ,7.  ,13. ,15. ,15. /
DATA           FR /53., 106., 212., 425., 850., 1700., 3400./
DATA           R /100., 200., 400., 600., 800., 1000., 2000., 3000.,
1  4000./
DATA PI/3.14159265/
DATA PI9902/1.55508836/
BETA1=BETA / 180.*PI
IF (BETA1 .LT. 0.0) BETA1 = 0.0
IF (BETA1 .GT. PI9902) BETA1 = PI9902
10 DO 20 I=1,N
X = TBLU2(DIST,FREQ(I),R,FR,AT,1,1,9,7,9,7)
A = TBLU1(FREQ(I),FR,AK,1,7)
ATT(I)=X*EXP(-A*SQRT(TAN(BETA1)))
20 CONTINUE
RETURN
END

```

FUNCTION OAFUNC(SPL)

ZDM914 10 - 8T

C  
DIMENSION SPL(24)  
OACAL =0.  
DO 901 K=1,24  
901 OACAL=OACAL+10.\*\*(SPL(K)/10.)  
OAFUNC=10.\* ALOG10(OACAL)  
RETURN  
END

## BLOCK DATA

ZDM914 13 - 8T

```

COMMON / AREA1 / SPL24(24,70,2), SPLREF(24,70,1)
COMMON / AREA2 / TITLE(40)
COMMON / AREA3 / FREQ24(24), FLOG24(24), FREQ8(8), FLOG8(8), BWN(24)
COMMON / AREA4 / MF(30), MN(10), ENGINE , NI, NJ, NK, NIT, CRV(400), MKT
COMMON / AREA5 / PNL24(70), XNOY24(24), EPNL(8), PEAKPN(4),
1          OASPL(70,1), TCPNL(70,1) , XNOYMX(70), MTCB(70)
COMMON / AREA6 / THRUST(4), THRUSI (6), OVHTIM(4), TIM(70), TIM2(70,3,1)
COMMON / GEOM1 / H(3,4), V(3,4), HREL(3,4), TEMP(3,4), X(3,4), Y(3,4), PI
1          , GAMM(2) , ALPH(2) , S(70,2) , W(70,2) , C(2)
COMMON / GEOM2 / THETA(14), THETA2(196), ALPHA(210), DIRSL(210),
1  DISTSL(210), AAC(200)
COMMON / CONNNOY / ABPS(24), BBPS(24), ALO(24), BLO(24), AM(24), BM(24)
COMMON / HFVJE / SPLFA(24,70,2), SPLJE(24,70), OASP(70,2)
1          , SPLTU(24,70)
DATA PI / 3.1415926536   / , SPLTU/1680*0.0/
2          , AAC/ 200*0.0/, TITLE/40*0.0/
4          , PEAKPN/4*0.0/
5          , EPNL /8*0.0/
6          , TIM /70*0.0/
8          , TCPNL /70*0.0/
DATA THETA/30., 40., 50., 60., 70., 80., 90., 100., 110., 120., 130., 140.,
1          150., 160./
DATA BWN /1.0, 2.0, 3.0, 4.0, 5.0, 6.0, 7.0, 8.0, 9.0, 10.0, 11.0, 12.0,
1 13.0, 14.0, 15.0, 16.0, 17.0, 18.0, 19.0, 20.0, 21.0, 22.0, 23.0, 24.0/
DATA FRE024
1 / 50., 63., 80., 100., 125., 160., 200., 250., 315., 400.,
2 500., 630., 800., 1000., 1250., 1600., 2000., 2500., 3150., 4000.,
3 5000., 6300., 8000., 10000. /
DATA ABPS/
1          64. , 60. , 56. , 53. , 51. , 48. , 46. , 44. , 42. , 40. ,
2          40. , 40. , 40. , 40. , 38. , 34. , 32. , 30. , 29. , 29. ,
3          30. , 31. , 37. , 41. , / 
DATA BBPS/
1          91. , 85.9 , 87.3 , 79.9 , 79.8 , 76. , 74. , 74.9 , 94.6 , 1000.,
2          1000. , 1000. , 1000. , 1000. , 1000. , 1000. , 1000. , 1000. , 1000. ,
3          1000. , 1000. , 44.3 , 50.7 / 
DATA ALO /
1          64. , 60. , 56. , 53. , 51. , 48. , 46. , 44. , 42. , 40. ,
2          40. , 40. , 40. , 40. , 38. , 34. , 32. , 30. , 29. , 29. ,
3          30. , 31. , 37. , 41. , / 
DATA BLO /
1          52. , 51. , 49. , 47. , 46. , 45. , 43. , 42. , 41. , 40. ,
2          40. , 40. , 40. , 40. , 38. , 34. , 32. , 30. , 29. , 29. ,
3          30. , 31. , 34. , 37. , / 
DATA AM /
1          .043478  ..04057    ..036831  ..036831  ..035336  ,
2          ..033233  ..033333  ..032051  ..030675  ..030103  ,
3          ..030103  ..030103  ..030103  ..030103  ..030103  ,
4          ..02996   ..02996   ..02996   ..02996   ..02996   ,

```

5 .02996 ,.02996 ,.042285 ,.042285/  
DATA BM/15\*.030103,9\*.02996/  
END

FUNCTION TBLU1(XX,X,Y,MDX,NLX)

ZDM914 14 - 14

C  
DIMENSION X(1),Y(1)  
NO X = MDX-1  
CALL SEARCH(X,XX,NO X,NLX,I,DELTA 1,A 1)  
TBLU1 = Y(I) + (Y(I+1)-Y(I))\*DELTA 1/A 1  
RETURN  
END

```
FUNCTION TBLU2(XX,YY,X,Y,F2,MDX,MDY,NLX,NLY,NFX,NFY)
DIMENSION F2(NFX,NFY),X(1),Y(1)
NO X = MDX-1
NO Y = MDY-1
CALL SEARCH(X,XX,NO X,NLX,I,DELTA 1,A 1)
CALL SEARCH(Y,YY,NO Y,NLY,J,DELTA 2,A 2)
DFX1 = (F2(I+1,J)-F2(I,J))*DELTA 1/A1
DFX2 = (F2(I+1,J+1)-F2(I,J+1))*DELTA 1/A1
TBLU2 = F2(I,J) + DFX1 + (F2(I,J+1)-F2(I,J)+DFX2-DFX1)*DELTA 2/A 2
RETURN
END
```

```
SUBROUTINE SEARCH(X,XX,NOX,NLX,I,DELTA,A)
DIMENSION X(1)
1 NO X = NO X + 1
IF(XX.LT.X(NO X)) GO TO 5
IF(NO X.GE.NLX)GO TO 5
GO TO 1
5 DELTA = XX - X(NO X)
I = NO X
IF(NO X.NE.1) GO TO 10
A = X(NO X + 1) - X(NO X)
RETURN
10 IF(XX.LT.X(NLX)) GO TO 20
DELTA = XX - X(NO X-1)
A = X(NO X) - X(NO X-1)
I = NO X - 1
RETURN
20 DELTA = XX - X(NO X - 1)
A = X(NO X) - X(NO X-1)
I = (NO X-1)
RETURN
END
```

```

SUBROUTINE BOUNDY (DBTEST, X, Y, *, *)
C
C                                     ZDM914 30 - 30
C THIS SUBROUTINE CONTAINS LOGIC TO GENERATE 'FOOTPRINTS' -
C LINES OF EQUAL PRESSURE. EAGLEMAN NOV. 1970
C
LOGICAL *4 FIRST
INTEGER *4 OLD ,FIFPOP/03/
REAL *4 DBHIST(2), DUMY(2)
COMMON /FOOTPT/ ICV, CURVX(100), CURVY(100), IPT, POINTX(250)
1 ,POINTY(250), DBNL, DELX, DELY, FIRST
C * * * * * * * * - * - * - * - * - * - *
      IF (.NOT. FIRST)                                GO TO 1005
C           INITIALIZATION   - - - - -
FIRST = .FALSE.
NEW = 1
OLD = 2
ICV = 0
DBHIST(OLD) = DBNL                               GO TO 1010
1005 NEW = FIFPOP - NEW
      OLD = FIFPOP - OLD
1010 DBHIST(NEW) = DBTEST
      CALL BOUND (DBHIST(NEW), DBHIST(OLD), DBNL, Y, DELY, +2000)
C***          BOUNDED   - - - - -
C**          STORE INTERPOLATED POINTS
ICV = ICV + 1
DUMY(OLD) = ALOG10(POINTY(IPT-1))
DUMY(NEW) = ALOG10(POINTY(IPT))
CURVY(ICV) = TBLU1(DBNL, DBHIST, DUMY, 1, 2)
CURVY(ICV) = 10.0**[CURVY(ICV)]
CURVX(ICV) = X
X = DELX
      IF (DBHIST(NEW) .GT. DBHIST(OLD))              GO TO 2010
DBHIST(NEW) = DBHIST(OLD)
Y = Y - DELY                                      GO TO 3005
C***          NOT BOUNDED
2000 CONTINUE
5577 FORMAT ('0$$B$$ ', 8G15.7)
      WRITE (6,5577) DBHIST, NEW, OLD, CURVY(ICV), CURVX(ICV), X, Y
RETURN 1
C
C**          IF X OR Y IS BOUNDED, COME HERE
C**          DECREMENT Y, AND INCREMENT OR CALCULATE A NEW X
C
2005 Y = Y - DELY
C
C          X = X
C
2010 DBHIST(NEW) = -1.0

```

```

C
3005 CONTINUE
    WRITE(6,5577) DBHIST, NEW, OLD, CURVY(ICV), CURVX(ICV), X, Y
    RETURN 2

C
C
    ENTRY BOUNDX(DBTEST, X, Y, *, *)
C
        IF (DBHIST(NEW) .GT. 0.0)                      GO TO 4005
        DBHIST(NEW) = DBTEST
        X = X - DELX
        IF (DBTEST .GT. DBNL)      X = X + 2.0*DELX
                                    GO TO 3005

4005 NEW = FIPFOP - NEW
OLD = FIPFOP - OLD
DBHIST(NEW) = DBTEST
CALL BOUND (DBHIST(OLD), DBHIST(NEW), DBNL, X, DELX, +2000)
        BOUNDED
C**
        STORE INTERPOLATED POINTS
ICV = ICV + 1
DUMY(OLD) = ALOG10(POINTX(IPT-1))
DUMY(NEW) = ALOG10(POINTX(IPT))
CURVX(ICV) = TBLU1(DBNL, DBHIST, DUMY, 1, 2)
CURVX(ICV) = 10.0**CURVX(ICV)
CURVY(ICV) = Y
                                    GO TO 2005
END

```

C SUBROUTINE BOUND (DBHIS1, DBHIS2, DBNL, Y, DELY, \*)  
C \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \*  
C \* IF (DBHIS1 .LT. DBNL) GO TO 1005  
C \* IF (DBHIS2 .LT. DBNL) GO TO 2005  
C\*\* INCREMENT Y TO DECREASE DBHIST  
C \* Y = Y + DELY GO TO 1505  
1005 IF (DBHIS2 .GT. DBNL) GO TO 2005  
C\*\* DECREMENT Y TO INCREASE DBHIST  
C \* Y = Y - DELY  
1505 RETURN 1  
C\*\* NOT BOUNDED  
2005 RETURN BOUNDED  
C\*\* END

```

SUBROUTINE PLOTFP (X, Y, NPTS, FIRST, U, V, NPTS2)
C                                     ZDM914 31 - 31
C
C**      SUBROUTINE TO          PLOT FOOT PRINT
LOGICAL *4 FIRST
INTEGER *4 MXPTS/1000/, XLABEL(1)/*  '/, SIZXL/0/ ,CYCLE
1           ,YLABEL(1)/*  '/
2           ,TITLE(4)/'EPNL TEST DATA  '/, SIZTIL/14/ ,TYPE/'REC'/
3   ,SYMB
REAL *4 X(1)  , Y(1), U(1),      ZERO/0.0/ ,V(1)
C * - * - * - * - * - * * * * * * * * * * * *
IF (.NOT. FIRST)                               GO TO 1000
CYCLE = 1
C
C**      INITIALIZE FOR PLOTTING
CALL PLCTIN (10.,10., 12., 0., 90.,-12.0, 0.,-11.0, 0.0
1   ,TYPE, CYCLE)
C
C5577 FORMAT ('0*** ', 7G15.5/15X,7G15.5)
CD      WRITE (6,5577) (X(J), Y(J),J=1,NPTS), NPTS
C
C**      PLOT POINTS
C
C      - - - ROTATE PLOT -90 DEG. TO ORIENT WITH PAPER - -
1000 XSF = -2000.0
YSF = 10000.0
XSP = 0.0
YSP = 0.0
LINPT = - 1
SYMB = 3
CALL PLOTCSC (Y, X, NPTS, XLABEL, YLABEL, TITLE, SIZXL, SIZYL
1           ,SIZTIL, YSP, YSF, XSP, XSF, LINPT, SYMB)
FIRST = .FALSE.
C      - - - PLOT INTERPOLATED CURVE - - - - -
LINPT = 0
SYMB = 0
1050 CALL PLOTCV (V, U, NPTS2,YSP, YSF, XSP, XSF, LINPT, SYMB)
C                                     GO TO 9990
C9000 WRITE (6,9005) MXPTS,MXPTS
C9005 FORMAT ('0*** THE MAXIMUM NO. OF POINTS ALLOWED BY DIMENSIONING
C   1 (', IS, ') HAS BEEN EXCEEDED. THE FIRST',IS,' POINTS WILL BE'
C   2 , ' PLOTTED.')
C                                     GO TO 1015
C
9990 CONTINUE
C     CALL PLOT (ZERO, ZERO, 999)
RETURN
END

```

**APPENDIX 4**  
**PHI PROGRAM DESCRIPTION**

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## PHI PROGRAM

This computer program evaluates the relative effectiveness of the inlet and fan duct lining and the fundamental and harmonic attenuation.

The main program accepts the punched output from the Aircraft Noise Time History Prediction Program (EPNL program).

### Program Input

1. Title card columns (1-72)

2. K1, K10 (2I5)

K1 indicates the number of attenuations used with values as indicated by DATA statement ATENM2. If K1 = 4, then the data statement is rearranged by replacing ATENM2(4) = ATENM2(5).

K10 = 0 at all times.

3. LON1, LON2, LON3, LON4, MTH, NST, NIN, IPHI (8I5)

LON1 = 1 Write input

LON2 = 1 Write revised input

LON3 = 1 Plot attenuation

LON4 = Plot phi study

IPHI = 1 Long printout for design information

= 0 Short printout for documentation

NST = Start of phi calculation (integer between initial and final peak attenuation value)

NTH = Final value of NST

NIN = Increment to be used in going from NST to NTH

4. C1, C2, C3, C4, C5 (5F10.1)

Constants used in the elevation of the phi function.

5. ATT2 (40X, F6.0)

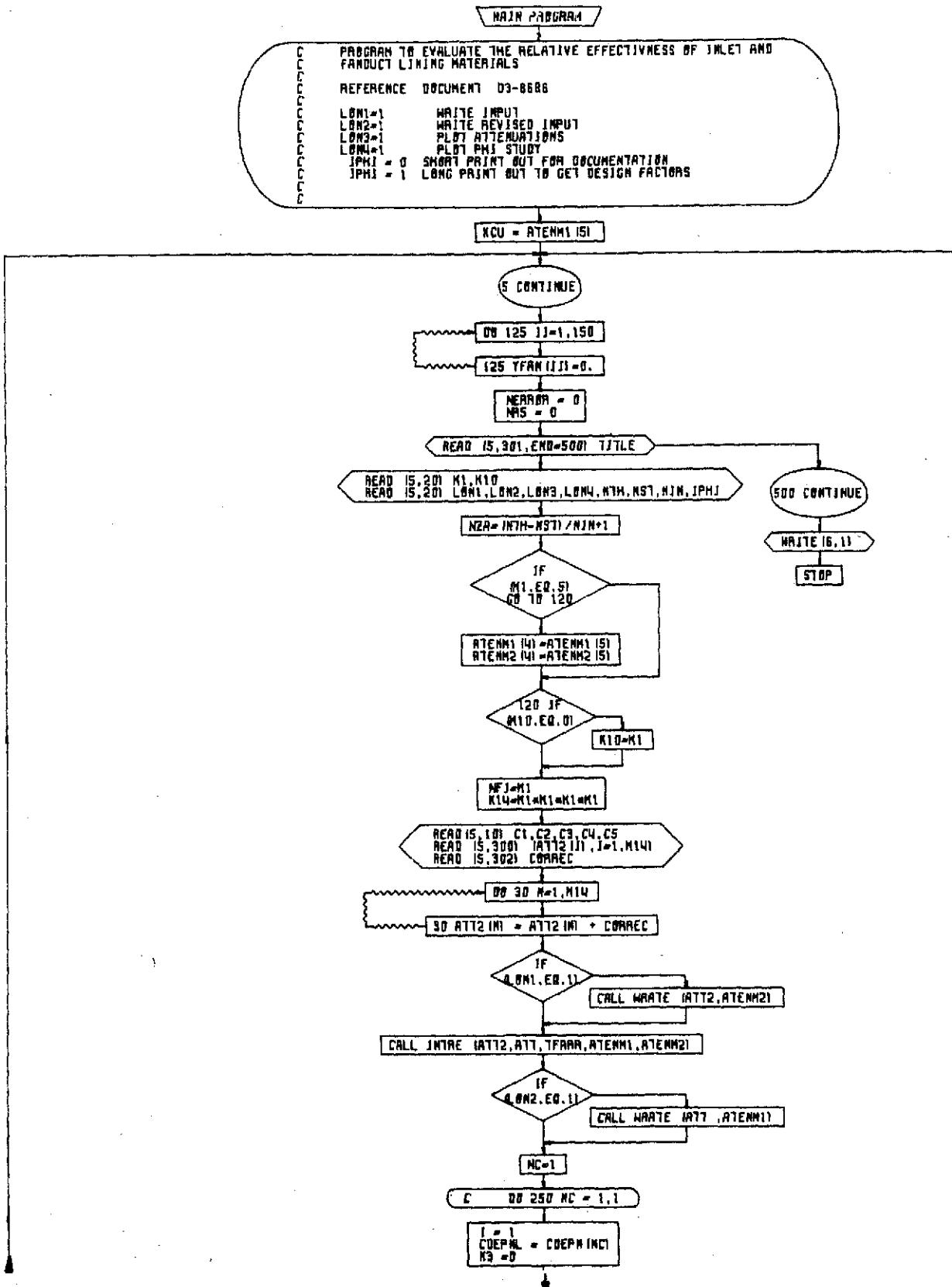
The change in EPNL as effected by the attenuation.

## 6. CORREC (F10.2)

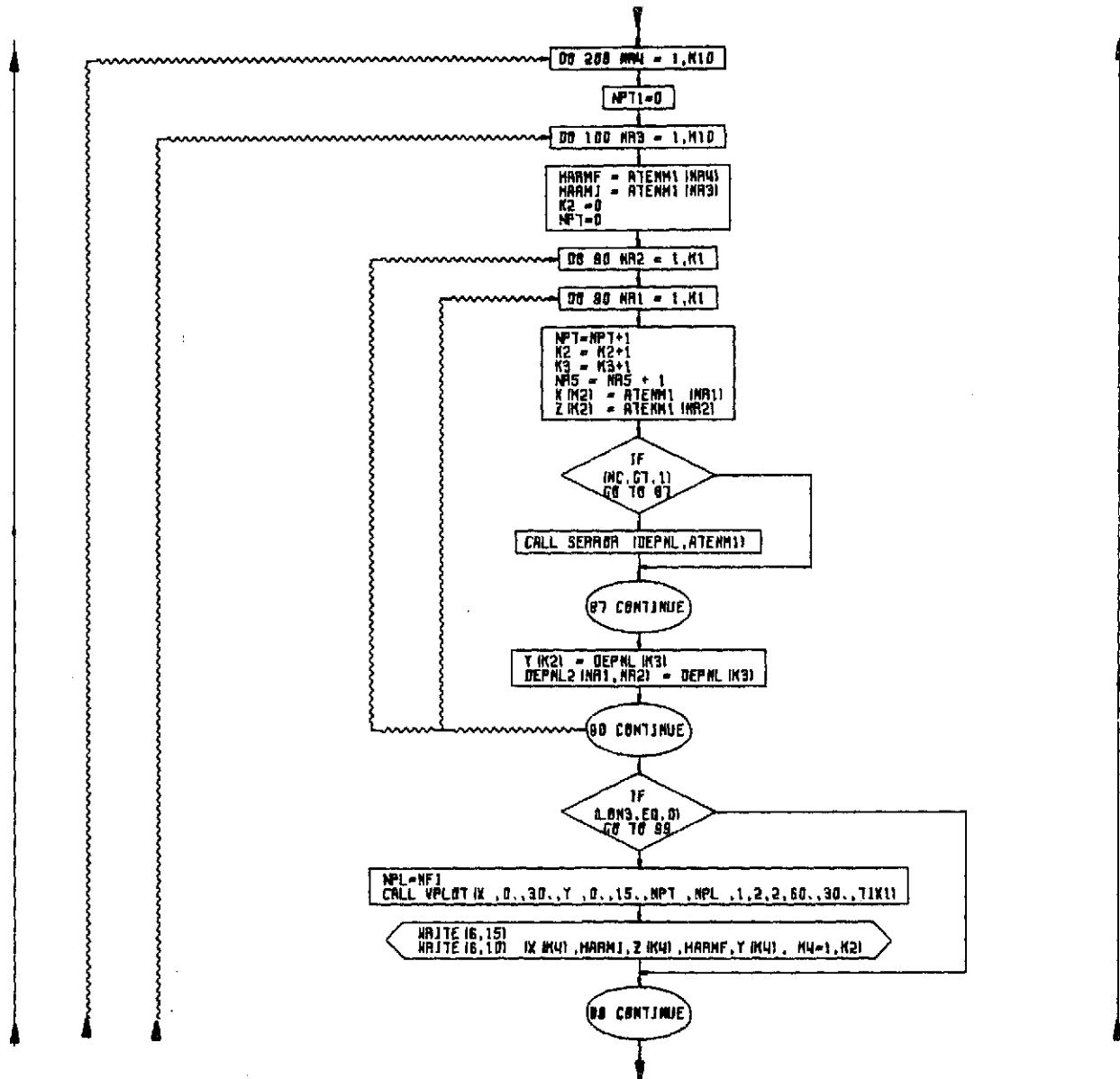
The difference between the first EPNL calculated and the reference EPNL value.

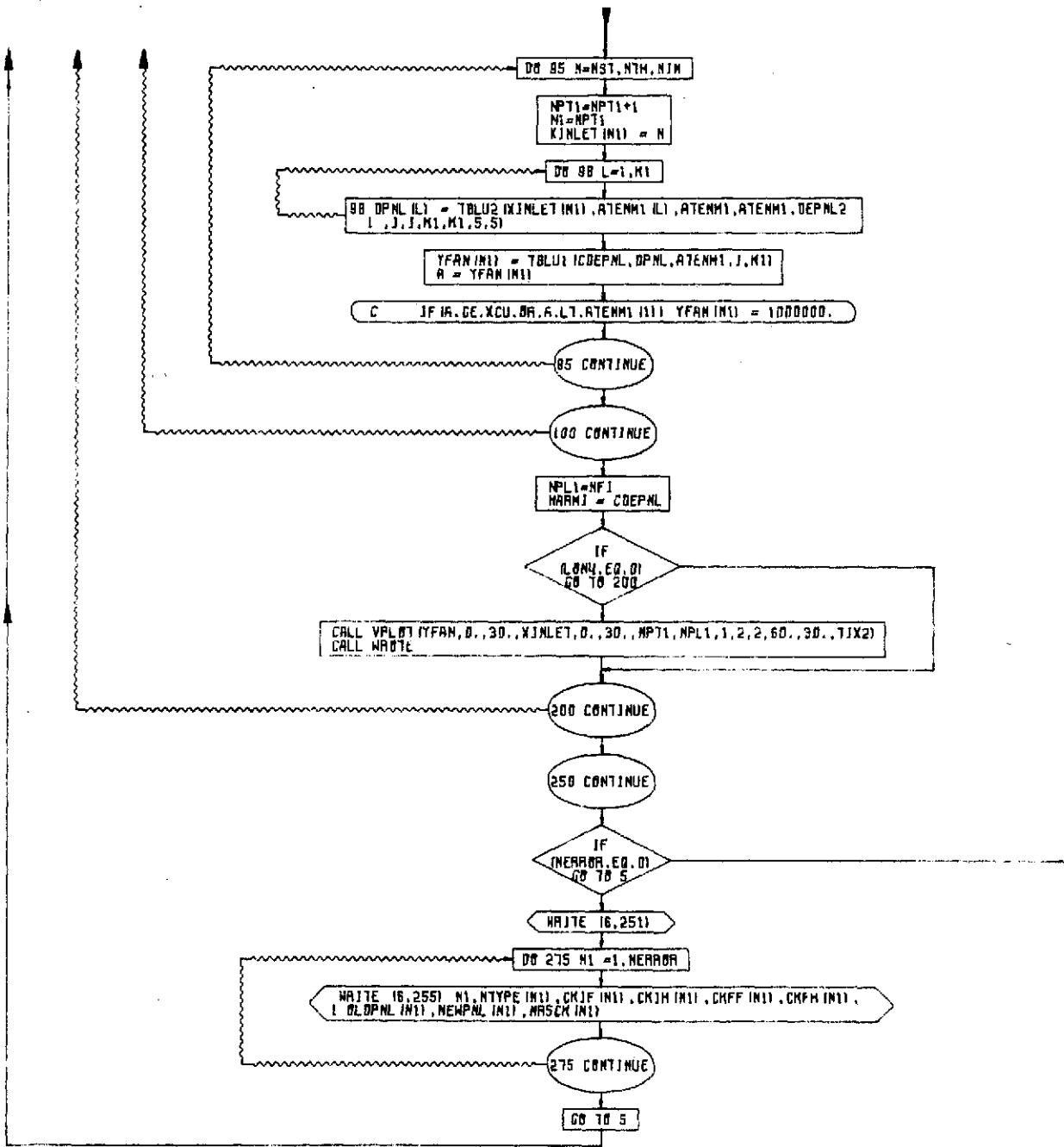
The program is set up to accept inputs for attenuations of 0, 5, 10, 20, and 30 dB (ATENM2). The revised input (ATENM1) can be changed to allow a detailed description of the attenuation, for instance 0., 2.5, 5, 7.5, 10 dB.

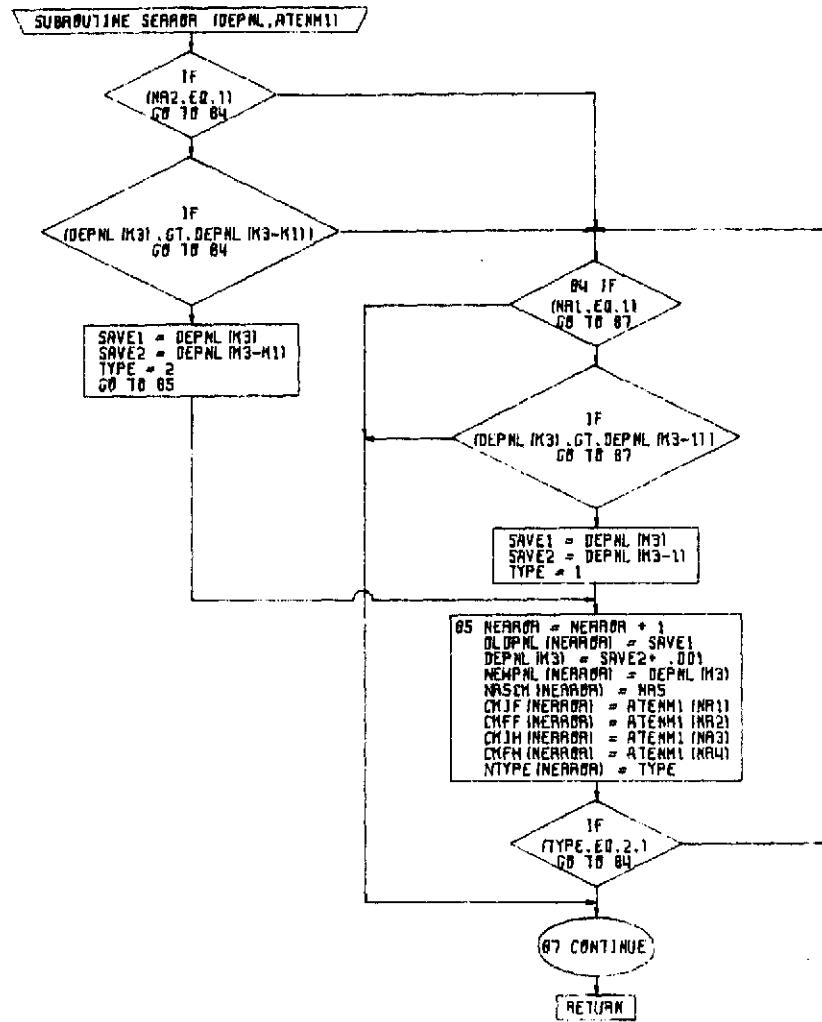
**APPENDIX 5**  
**PHI PROGRAM FLOW CHART**

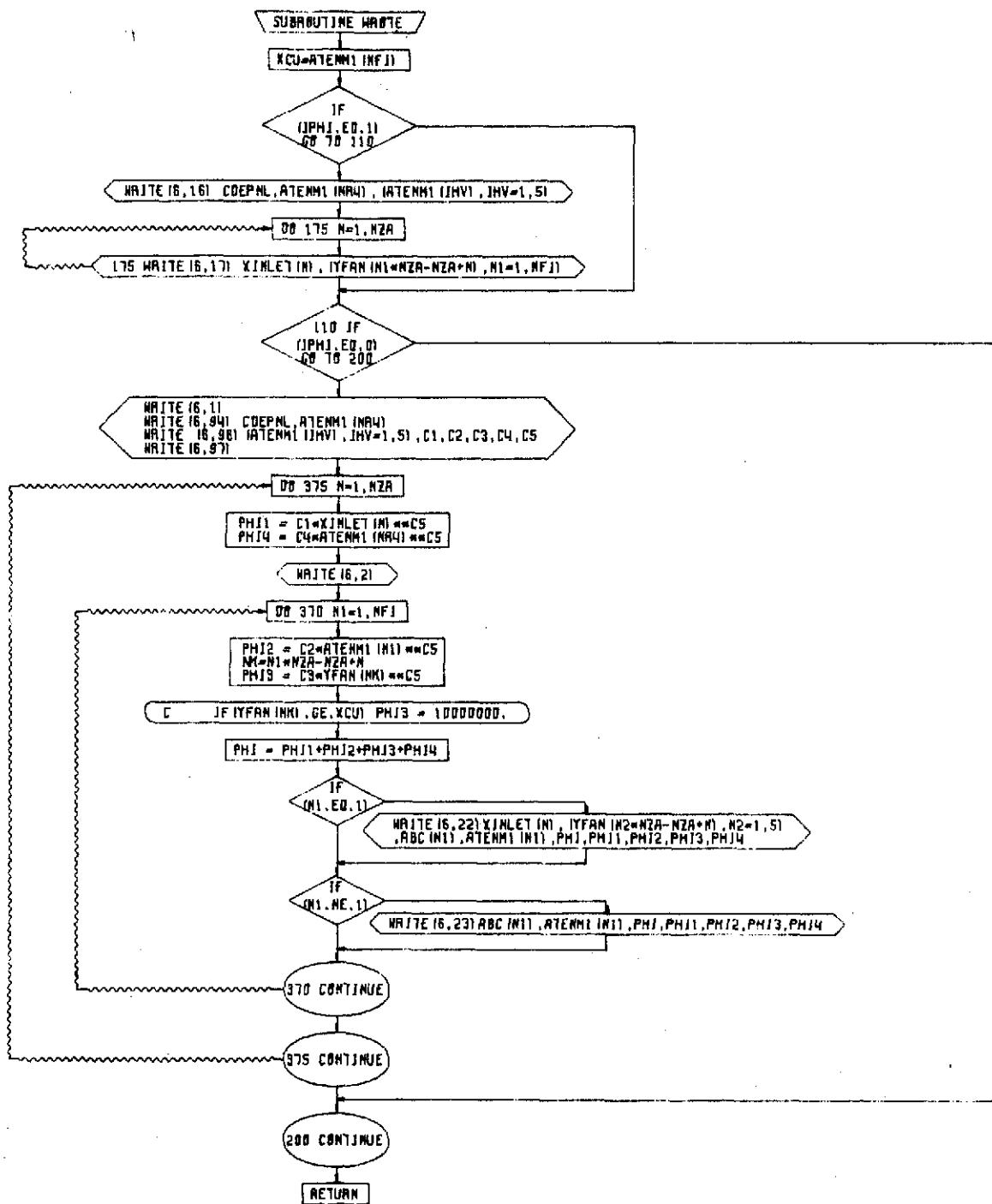


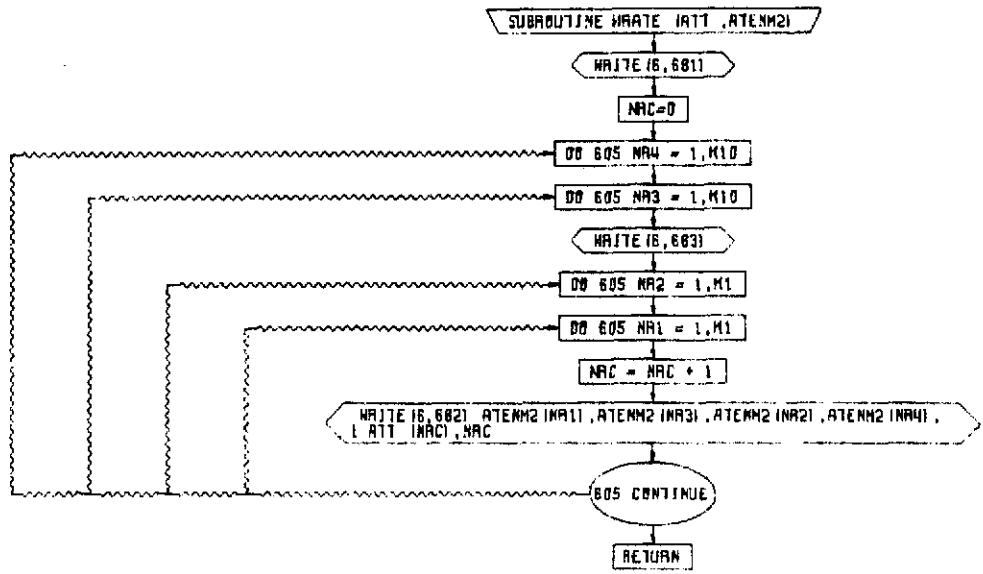
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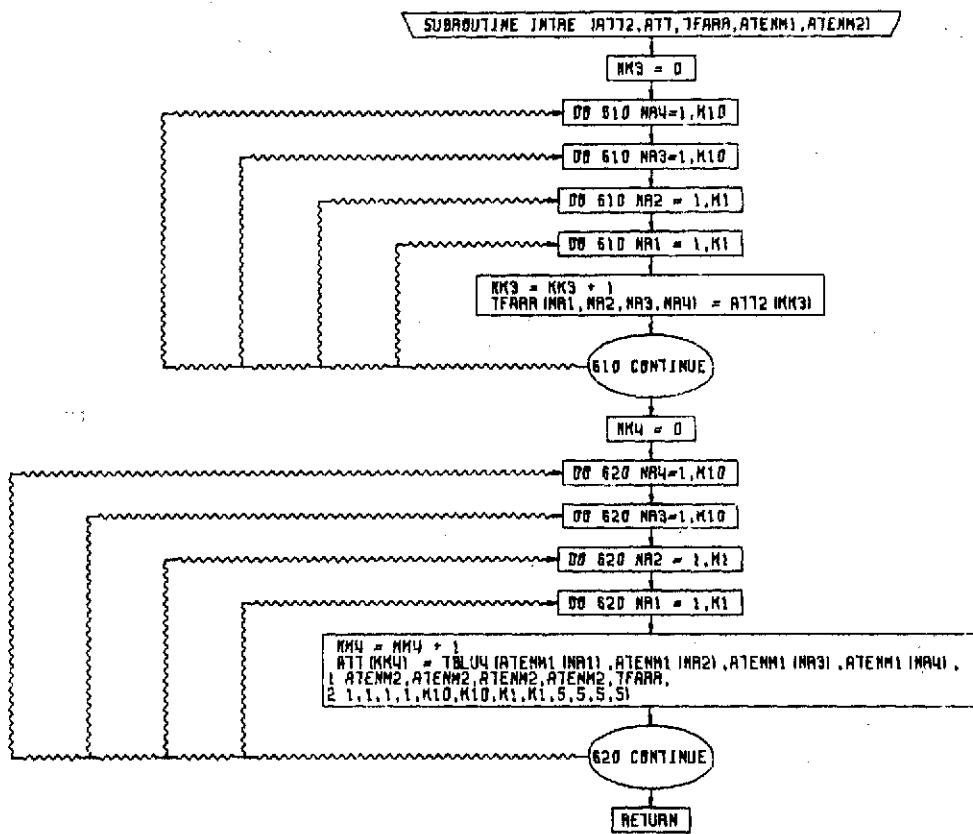












FUNCTION TBLU1 IX,X,Y,MDX,MDY,MLX

```
NO X = MDX-1
CALL SEARCH IX,XX,NO X,MLX,I,DELTA 1,A 11
TBLU1 = Y(I) + (Y(I+1)-Y(I)) * DELTA 1/A 1
RETURN
```

FUNCTION TBLU2 IX,YY,X,Y,F2,MDX,MDY,MLY,MFX,MFY

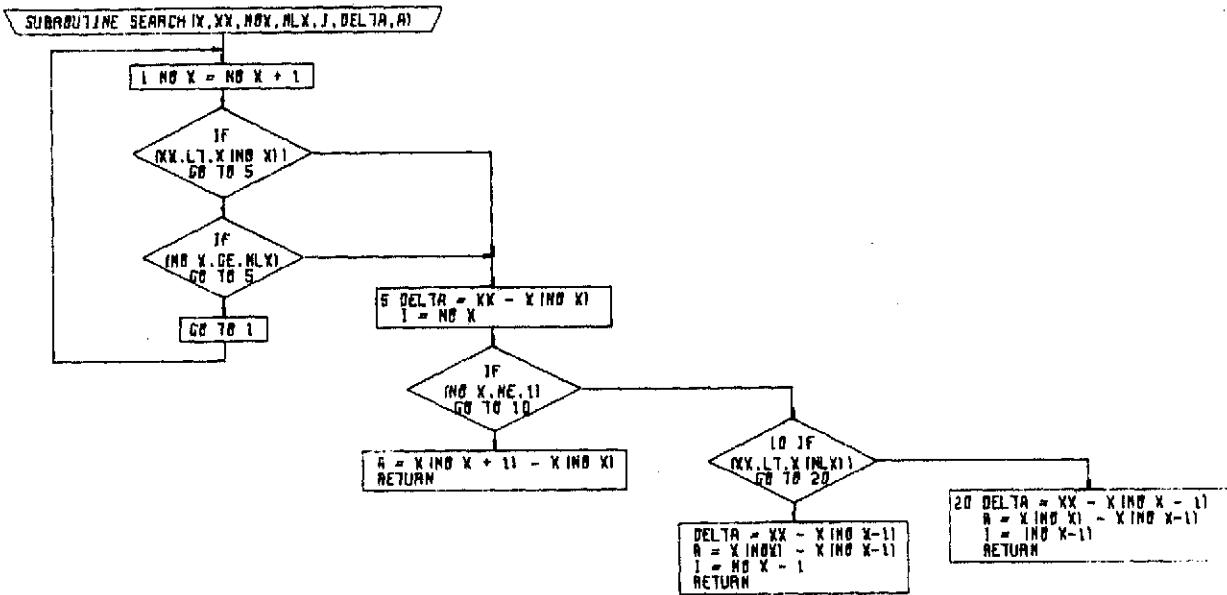
```
NO X = MDX-1
NO Y = MDY-1
CALL SEARCH IX,XX,NO X,MLX,J,DELTA 1,A 11
CALL SEARCH YY,YY,NO Y,MLY,J,DELTA 2,A 21
DFX1 = (F2(J+1)-F2(J),J) * DELTA 1/A 1
DFX2 = (F2(J+1,J+1)-F2(J,J+1)) * DELTA 1/A 1
TBLU2 = F2(J,J) + DFX1 + (F2(J,J+1)-F2(J,J)) * DFX2 - DFX1 * DELTA 2/A 2
RETURN
```

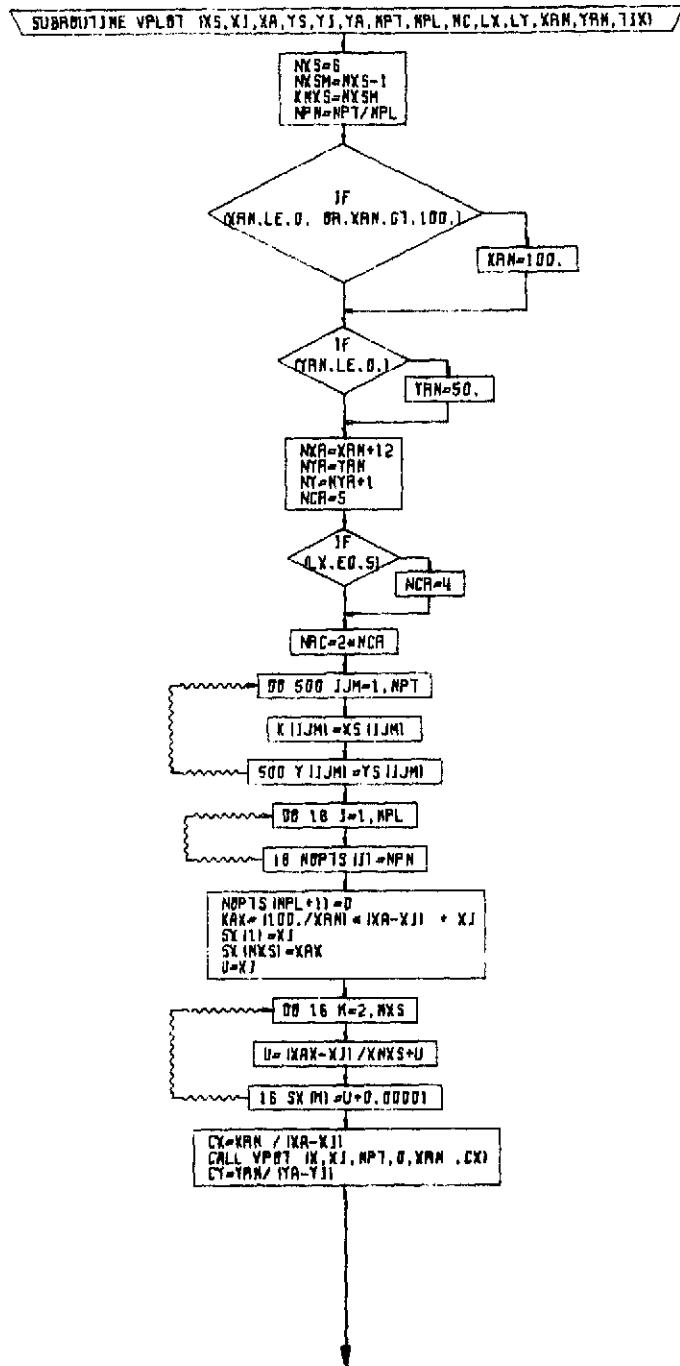
FUNCTION TBLU4 IX,X,Y,Z,W,V,MDX,MDY,MDZ,MDW,  
MLX,MLY,MLZ,MLW,MFX,MFY,MFZ,MFW

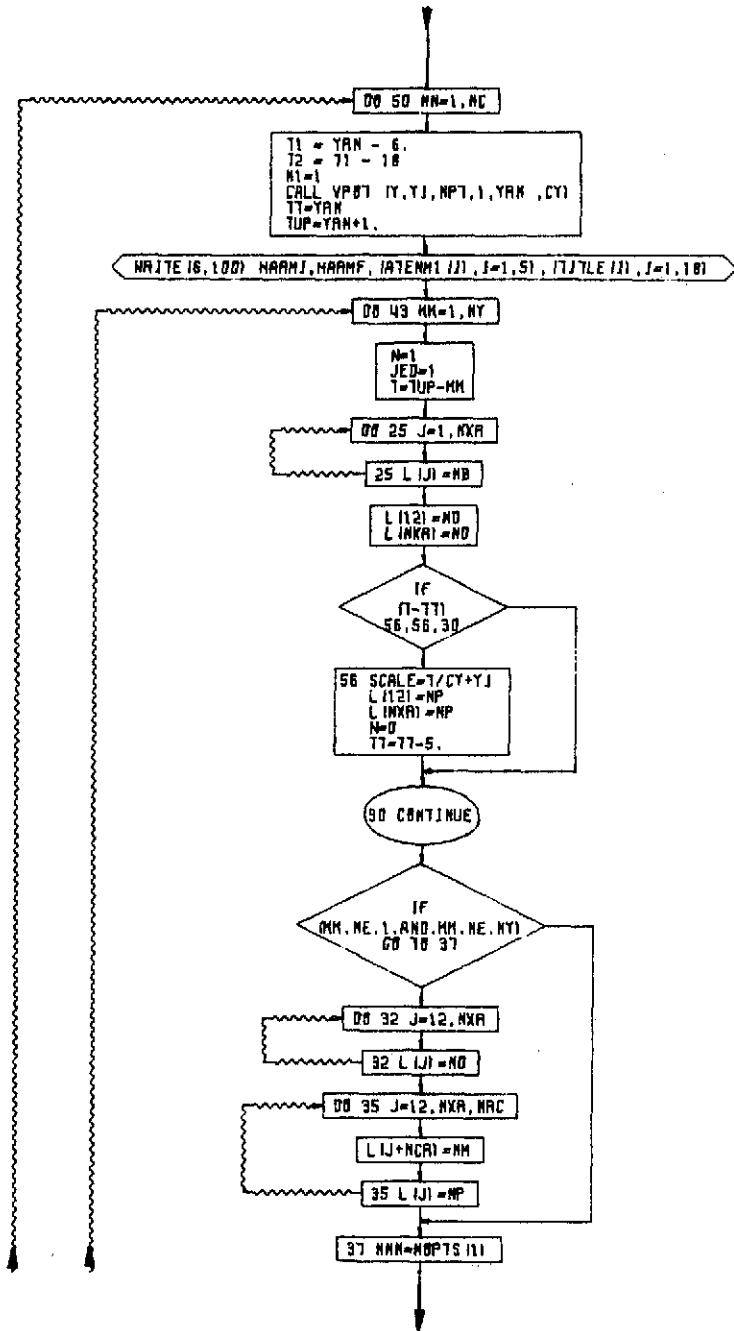
```
NO Y = MDY - 1
NO Z = MDZ - 1
NO W = MDW - 1
NO X = MDX - 1
CALL SEARCH IX,XX,NO X,MLX,J,DELTA 1,A 11
CALL SEARCH IY,YY,NO Y,MLY,J,DELTA 2,A 21
CALL SEARCH IZ,ZZ,NO Z,MLZ,K,DELTA 3,A 31
CALL SEARCH IW,WW,NO W,MLW,L,DELTA 4,A 41
DR1 = DELTA 1/A 1
DR2 = DELTA 2/A 2
DR3 = DELTA 3/A 3
DR4 = DELTA 4/A 4
VPX1 = VPX IDA1,V(I),J,M,L,I,V(I+1),J,M,L,I
VPX2 = VPX IDA1,V(I),J+1,M,L,I,V(I+1),J+1,M,L,I
VPX3 = VPX IDA1,V(I),J,M+1,L,I,V(I+1),J,M+1,L,I
VPX4 = VPX IDA1,V(I),J+1,M+1,L,I,V(I+1),J+1,M+1,L,I
VPX5 = VPX IDA1,V(I),J,M,L+1,V(I+1),J,M,L+1
VPX6 = VPX IDA1,V(I),J+1,M,L+1,V(I+1),J+1,M,L+1
VPX7 = VPX IDA1,V(I),J,M+1,L+1,V(I+1),J,M+1,L+1
VPX8 = VPX IDA1,V(I),J+1,M+1,L+1,V(I+1),J+1,M+1,L+1
VPX9 = VPX IDA2,V(I),J,M,L,I,V(I+1),J,M,L,I
VPX10 = VPX IDA2,VPX1,VPX2
VPX11 = VPX IDA2,VPX1,VPX3
VPX12 = VPX IDA2,VPX5,VPX6
VPX13 = VPX IDA2,VPX5,VPX6
VPX14 = VPX IDA2,VPX7,VPX8
VPXY1 = VPX IDA3,VPXY1,VPXY2
VPXY2 = VPX IDA3,VPXY3,VPXY4
TBLU4 = VPX IDA4,VPXY1,VPXY2
RETURN
```

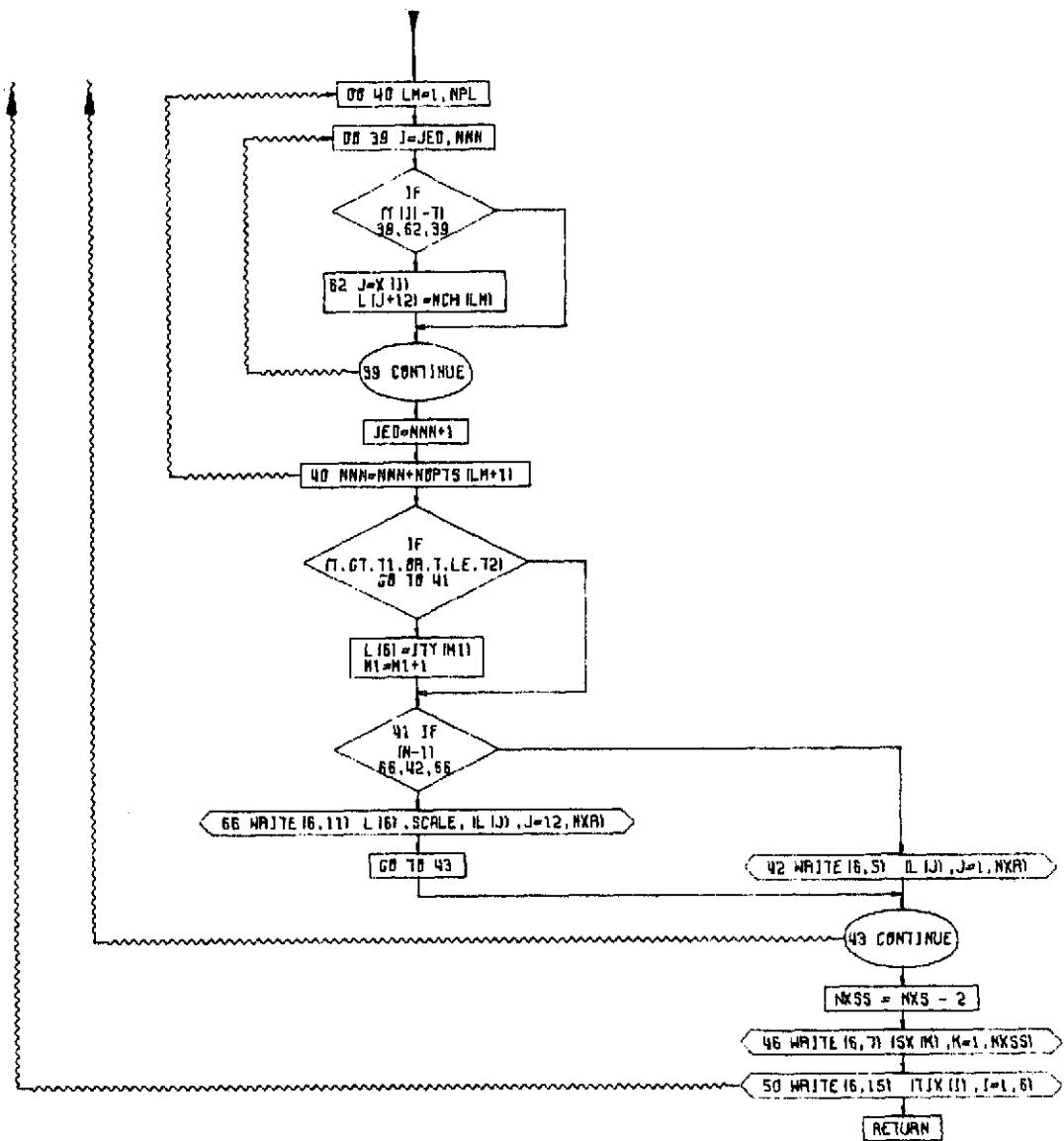
FUNCTION VPX (A,B,C)

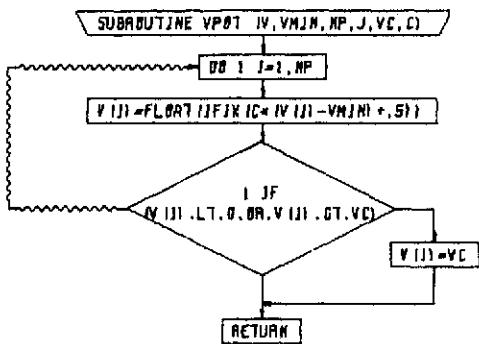
```
VPX = B + A*(C-B)
RETURN
```











**APPENDIX 6**  
**PHI PROGRAM LISTING**

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```

C
C
C PROGRAM TO EVALUATE THE RELATIVE EFFECTIVENESS OF INLET AND
C FDUCT LINING MATERIALS
C
C REFERENCE DOCUMENT D3-8686
C
C LON1=1      WRITE INPUT
C LON2=1      WRITE REVISED INPUT
C LON3=1      PLOT ATTENUATIONS
C LON4=1      PLOT PHI STUDY
C IPHI = C SHORT PRINT OUT FOR DOCUMENTATION
C IPHI = 1 LONG PRINT OUT TO GET DESIGN FACTORS
C
C
REAL          NEWPNL
DIMENSION      X(25),      Y(25),      Z(25)
DIMENSION      ATT(625),    DEPNL(625)
DIMENSION      DEPNL2(5,5)
DIMENSION      TIX1(6),    TIX2(6)
DIMENSION      ATENM2(5),   ATT2(625),   TFARR(5,5,5,5)
COMMON / COMM1 / OLDPNL(625), NEWPNL(625), NR5CK(625), NTYPE(625)
1             ,CKIF(625), CKFF(625), CKIH(625), CMFH(625)
COMMON / COMM2 / K1,K2,K3,K4,NR1,NR2,NR3,NR4,NR5,NERROR,K10,K14
COMMON / COMM3 / IFLAG,IPHI,NTH,NFI,NST,NEN,NZA,NZR
COMMON/COMM4/DPNL(5),XINLET(150),YFAN(150)
COMMON/COMM5/ C1,C2,C3,C4,C5,CDEPNL
COMMON / COMM6 / ATENM1(5),ABC(5)
COMMON / COMM7 / LON1,LON2,LON3,LON4,LON5,LON6,LON7,LON8
DIMENSION      CDEPN(6)
DATA CDEPN     /10.,5*0./
DATA ATENM2    /0.,5.,10.,20.,30./
DATA ATENM1    /0.,5.,10.,20.,30./
EQUIVALENCE    (ATT(1),DEPNL(1))
DATA           ABC/'A','B','C','D','E'
DATA TIX1      /*INLE', 'T AT', 'TENU', 'ATIO', 'N   ', '  /
DATA TIX2      /*FAN ', 'DUCT', ' ATT', 'ENUA', 'TION', '  /
COMMON         / RFOPL0 / HARM1,HARMF,TITLE(36)
1 FORMAT('1')
10 FORMAT(4F10.1,F10.2)
15 FORMAT(/10X,'INLET'15X*'FDUCT'6X*D-EPNL'/5X*'FUND.'5X*'HARM.'
1           5X*'FUND.'5X*'HARM.'/)
20 FORMAT(10I5)
251 FORMAT('1      DATA CORRECTIONS'//)
1* TYPE = 1 INCREASED INLET FUND. ATT. GAVE DECREASED DELTA EPNL*/
2* TYPE = 2 INCREASED FAN FUND. ATT. GAVE DECREASED DELTA EPNL*/
3//3X* NO. '3X* TYPE '3X*INLET'3X*INLET'3X* FAN '3X* FAN '3X* OLD *
43X* NEW '3X* DATA'/19X*'FUND.'3X*'HARM.'3X*'FUND.'3X*'HARM.'3X*DEPNL*
53X*DEPNL'3X* CARD'/
255 FORMAT(3X,I3,8X,I1,1X,4(3X,F5.2),2(2X,F6.3),5X,I3)
300 FORMAT(4CX,F6.0)

```

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```

301 FORMAT(1PA4)
302 FORMAT(F10.0)
XCU = ATENM1(5)
5 CONTINUE
DO 125 I=1,150
125 YFAN(I)=0.
NERROR = 0
NR5 = C
READ (5,301,END=500) TITLE
READ (5,2C) K1,K10
READ (5,20) LON1,LON2,LON3,LON4,NTH,NST,NIN,IPHI
NZA=(NTH-NST)/NIN+1
IF (K1.EQ.5) GO TO 120
ATENM1(4)=ATENM1(5)
ATENM2(4)=ATENM2(5)
120 IF (K10.EQ.0) K10=K1
NFI=K1
K14=K1*K1*K1*K1
READ(5,10) C1,C2,C3,C4,C5
RFAD (5,300) (ATT2(I),I=1,K14)
READ (5,202) CORREC
DO 30 N=1,K14
30 ATT2(N) = ATT2(N) + CORREC
IF (LON1.EQ.1) CALL WRATE (ATT2,ATENM2)
CALL INTRE (ATT2,ATT,TFARR,ATENM1,ATENM2)
IF (LON2.EQ.1) CALL WRATE (ATT ,ATENM1)
NC=1
C DO 250 NC = 1,1
I = 1
CDEPNL = CDEPN(NC)
K3 =0
DO 200 NR4 = 1,K10
NPT1=0
DO 100 NR3 = 1,K10
HARMF = ATENM1(NR4)
HARMI = ATENM1(NR3)
K2 =C
NPT=C
DO 90 NR2 = 1,K1
DO 90 NR1 = 1,K1
NPT=NPT+1
K2 = K2+1
K3 = K3+1
NR5 = NR5 + I
X(K2) = ATENM1 (NR1)
Z(K2) = ATENM1(NR2)
IF(NC.GT.1) GO TO 87
CALL SERROR (DEPNL,ATENM1)
87 CONTINUE
Y(K2) = DEPNL(K2)
DEPNL2(NR1,NR2) = DEPNL(K3)

```

```

90 CONTINUE
IF (LON3.EQ.0) GO TO 99
NPL=NFI
CALL VPLOT(X ,0.,30.,Y ,0.,15.,NPT ,NPL ,1,2,2,60.,30.,TIX1)
WRITE(6,15)
WRITE(6,101) (X(K4),HARMI,Z(K4),HARMF,Y(K4), K4=1,K2)
99 CONTINUE
DO 95 N=NST,NTH,NIN
NPT1=NPT1+1
N1=NPT1
XINLET(N1) = N
DO 98 L=1,K1
98 DPNL(L) = TBLU2(XINLET(N1),ATENM1(L),ATENM1,ATENM1,DEPNL2
1 ,I,I,K1,K1,5,5)
YFAN(N1) = TBLUI(CDEPNL,DPNL,ATENM1,I,K1)
A = YFAN(N1)
C IF(A.GE.XCU.OR.A.LT.ATENM1(1)) YFAN(N1) = 1000000.
95 CONTINUE
100 CONTINUE
NPL1=NFI
HARMI = CDEPNL
IF (LON4.EQ.0) GO TO 200
CALL VPLOT(YFAN,0.,30.,XINLET,0.,30.,NPT1,NPL1,1,2,2,60.,30.,TIX2)
CALL WROTE
200 CONTINUE
250 CONTINUE
IF(NERROR.EQ.0) GO TO 5
WRITE (6,251)
DO 275 N1 =1,NERROR
WRITE (6,2551 N1,NTYPE(N1),CKIF(N1),CKIH(N1),CKFF(N1),CKFH(N1),
1 OLDPNL(N1),NEWPNL(N1),NR5CK(N1))
275 CONTINUE
GO TO 5
500 CONTINUE
WRITE(6,1)
STOP
END

```

```

SUBROUTINE SERROR (DEPNL,ATENM1)
REAL           NEWPNL
COMMON / COMM1 / OLDPNL(625), NEWPNL(625), NR5CK(625), NTYPE(625)
1           ,CKIF(625), CKFF(625), CKIH(625), CKFH(625)
COMMON / COMM2 / K1,K2,K3,K4,NR1,NR2,NR3,NR4,NR5,NERROR,K10,K14
COMMON / COMM7 / LON1,LON2,LON3,LON4,LON5,LON6,LON7,LON8
DIMENSION      DEPNL(625),ATENM1(5)
IF(NR2.EQ.1) GO TO 84
IF(DEPNL(K3).GT.DEPNL(K3-K1)) GO TO 84
SAVE1 = DEPNL(K3)
SAVE2 = DEPNL(K3-K1)
TYPE = 2
GO TO 85
84 IF(NR1.EQ.1) GO TO 87
IF(DEPNL(K3).GT.DEPNL(K3-1)) GO TO 87
SAVE1 = DEPNL(K3)
SAVE2 = DEPNL(K3-1)
TYPE = 1
85 NERROR = NERROR + 1
OLDPNL(NERROR) = SAVE1
DEPNL(K3) = SAVE2+.001
NEWPNL(NERROR) = DEPNL(K3)
NR5CK(NERROR) = NR5
CKIF(NERROR) = ATENM1(NR1)
CKFF(NERROR) = ATENM1(NR2)
CKIH(NERROR) = ATENM1(NR3)
CKFH(NERROR) = ATENM1(NR4)
NTYPE(NERROR) = TYPE
IF(TYPE.EQ.2.) GO TO 84
87 CONTINUE
RETURN
END

```

```

SUBROUTINE WROTE
COMMON / COMM2 / K1,K2,K3,K4,NR1,NR2,NR3,NR4,NR5,NERROR,K10,K14
COMMON / COMM3 / IFLAG,IPHI,NTH,NFI,NST,NIN,NZA,NZB
COMMON/COMM4/DPNL(5),XINLET(150),YFAN(150)
COMMON/COMM5/ C1,C2,C3,C4,C5,CDEPNL
COMMON / COMM6 / ATENM1(5),ABC(5)
COMMON / COMM7 / LON1,LON2,LON3,LON4,LON5,LON6,LON7,LON8
1 FORMAT('1')
2 FORMAT (/)
16 FORMAT(//10X*DELTA EPNL = 'F5.1,10X*FAN HARMONIC = 'F5.1
1 . . . . . / . . . . .
1 4X*INLET HARMONIC = '5X,5(F4.1,7X)/
1 . . . . . /10X*INLET'2X,5(8X*FAN*)/10X*FUND.*3X,
2 . . . . . E(6X*FUND.*)/)

17 FORMAT(9X,F6.2,3X,5(5X,F6.2))
22 FORMAT(5X,F5.2,EX,5(3X,F5.2),12X,A1,1X,F5.2,5(2X,F7.2))
23 FORMAT(68X,A1,1X,F5.2,5(2X,F7.2))
94 FORMAT(5X*DELTA EPNL = 'F4.1,10X*FAN HARMONIC = 'F4.1/)
96 FORMAT(21X*A*7X*B*7X*C*7X*D*7X*E*16X*FUNCTION PHI//3X*INLET HARMONIC
1 1C = '2X,5(F4.1,4X),12X, 'C1 = 'F5.3,3X*C2 = 'F5.3,3X
3 'C3 = 'F5.3,3X*C4 = 'F5.3/70X*EXPONENT C5 = 'F5.3/5X*INLET*5X,
4 5(5X*FAN*),15X*INLET'2X* PHI '2X*PHI(IF)*2X*PHI(TH)*2X*PHI(FF)*
5 2X*PHI(FH)*)
97 FORMAT(5X*FUND.,'6X,5(3X,*FUND.*),14X*HARM.* )
XCU=ATENM1(NFI)
IF( IPHI.EQ.1) GO TO 110
WRITE(6,16) CDEPNL,ATENM1(NR4),(ATENM1(IHV),IHV=1,5)
DO 175 N=1,NZA
175 WRITE(6,17) XINLET(N),(YFAN(N1*NZA-NZA+N),N1=1,NFI)
110 IF( IPHI.EQ.0) GO TO 200
WRITE(6,1)
WRITE(6,54) CDEPNL,ATENM1(NR4)
WRITE(6,96)(ATENM1(IHV),IHV=1,5),C1,C2,C3,C4,C5
WRITE(6,97)
DO 375 N=1,NZA
PHI1 = C1*XINLET(N)**C5
PHI4 = C4*ATENM1(NR4)**C5
WRITE(6,2)
DO 370 N1=1,NFI
PHI2 = C2*ATENM1(N1)**C5
NK=N1*NZA-NZA+N
PHI3 = C3*YFAN(NK)**C5
IF(YFAN(NK).GE.XCU) PHI3 = 10000000.
PHI = PHI1+PHI2+PHI3+PHI4
IF(N1.EQ.1) WRITE(6,22)XINLET(N),(YFAN(N2*NZA-NZA+N),N2=1,5),
,ABC(N1),ATENM1(N1),PHI,PHI1,PHI2,PHI3,PHI4
1 IF(N1.NE.1) WRITE(6,23)ABC(N1),ATENM1(N1),PHI,PHI1,PHI2,PHI3,PHI4
370 CONTINUE
375 CONTINUE
200 CONTINUE
RETURN

```

```
SUBROUTINE WRATE (ATT ,ATENM2)
DIMENSION ATT(1),ATENM2(1)
COMMON / COMM2 / K1,K2,K3,K4,NR1,NR2,NR3,NR4,NR5,NERROR,K10,K14
681 FORMAT('1'20X,'INLET'15X'FANDUCT'6X'D-EPNL'14X'DATA'/
1           17X'FUND.'5X'HARM.'
2           5X'FUND.'5X'HARM.'22X'CARD')
682 FORMAT(11X,4F10.1,F10.2,12X,15)
683 FORMAT(/)
      WRITE(6,681)
      NRC=0
      DO 605 NR4 = 1,K10
      DO 605 NR3 = 1,K10
      WRITE(6,683)
      DO 605 NR2 = 1,K1
      DO 605 NR1 = 1,K1
      NRC = NRC + 1
      WRITE(6,682) ATENM2(NR1),ATENM2(NR3),ATENM2(NR2),ATENM2(NR4),
1                  ATT (NRC),NRC
605 CONTINUE
      RETURN
      END
```

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```

SUBROUTINE INTRE (ATT2,ATT,TFARR,ATENM1,ATENM2)
COMMON / COMM2 / K1,K2,K3,K4,NR1,NR2,NR3,NR4,NR5,NERROR,K10,K14
DIMENSION           TFARR(5,5,5,5)
DIMENSION           ATT2(1),ATT (1),ATENM1(1),ATENM2(1)
KK3 = 0
DO 610 NR4=1,K10
DO 610 NR3=1,K10
DO 610 NR2 = 1,K1
DO 610 NR1 = 1,K1
KK3 = KK2 + 1
TFARR(NR1,NR2,NR3,NR4) = ATT2(KK3)
510 CONTINUE
KK4 = 0
DO 620 NR4=1,K10
DO 620 NR3=1,K10
DO 620 NR2 = 1,K1
DO 620 NR1 = 1,K1
KK4 = KK4 + 1
ATT(KK4) = TBLU4(ATENM1(NR1),ATENM1(NR2),ATENM1(NR3),ATENM1(NR4),
1                         ATENM2,ATENM2,ATENM2,ATENM2,TFARR,
2                         1,1,1,1,K10,K10,K1,K1,5,5,5,5)
520 CONTINUE
RETURN
END

```

```
FUNCTION TBLUI(XX,X,Y,MDX,NLX)
DIMENSION X(1),Y(1)
NO X = MDX-1
CALL SEARCH(X,XX,NO X,NLX,I,DELTA 1,A 1)
TBLUI = Y(I) + (Y(I+1)-Y(I))*DELTA 1/A 1
RETURN
END
```

```
FUNCTION TBLU2(XX,YY,X,Y,F2,MDX,MDY,NLX,NLY,NFX,NFY)
DIMENSION F2(NFX,NFY),X(1),Y(1)
NO X = MDX-1
NO Y = MDY-1
CALL SEARCH(X,XX,NO X,NLX,I,DELTA 1,A 1)
CALL SEARCH(Y,YY,NO Y,NLY,J,DELTA 2,A 2)
DFX1 = (F2(I+1,J)-F2(I,J))*DELTA 1/A1
DFX2 = (F2(I+1,J+1)-F2(I,J+1))*DELTA 1/A1
TBLU2 = F2(I,J) + DFX1 + (F2(I,J+1)-F2(I,J)+DFX2-DFX1)*DELTA 2/A 2
RETURN
END
```

```

FUNCTION TBLU4(XX,YY,ZZ,WW,X,Y,Z,K,V,MDX,MDY,MDZ,MDW,
1           NLX,NLY,NLZ,NLW,NFX,NFY,NFZ,NFW)
DIMENSION V(NFX,NFY,NFZ,NFW),X(1),Y(1),Z(1),W(1)
NO X = MDX - 1
NO Y = MDY - 1
NO Z = MDZ - 1
NO W = MDW - 1
CALL SEARCH(X,XX,NO X,NLX,I,DELTA 1,A 1)
CALL SEARCH(Y,YY,NO Y,NLY,J,DELTA 2,A 2)
CALL SEARCH(Z,ZZ,NO Z,NLZ,K,DELTA 3,A 3)
CALL SEARCH(W,WW,NO W,NLW,L,DELTA 4,A 4)
DA1 = DELTA 1/A 1
DA2 = DELTA 2/A 2
DA3 = DELTA 3/A 3
DA4 = DELTA 4/A 4
VPX1 = VPX(DA1,V(I,J ,K ,L ),V(I+1,J ,K ,L ))
VPX2 = VPX(DA1,V(I,J+1,K ,L ),V(I+1,J+1,K ,L ))
VPX3 = VPX(DA1,V(I,J ,K+1,L ),V(I+1,J ,K+1,L ))
VPX4 = VPX(DA1,V(I,J+1,K+1,L ),V(I+1,J+1,K+1,L ))
VPX5 = VPX(DA1,V(I,J ,K ,L+1),V(I+1,J ,K ,L+1))
VPX6 = VPX(DA1,V(I,J+1,K ,L+1),V(I+1,J+1,K ,L+1))
VPX7 = VPX(DA1,V(I,J ,K+1,L+1),V(I+1,J ,K+1,L+1))
VPX8 = VPX(DA1,V(I,J+1,K+1,L+1),V(I+1,J+1,K+1,L+1))
VPXY1 = VPX(DA2,VPX1,VPX2)
VPXY2 = VPX(DA2,VPX3,VPX4)
VPXY3 = VPX(DA2,VPX5,VPX6)
VPXY4 = VPX(DA2,VPX7,VPX8)
VPXYW1 = VPX(DA3,VPXY1,VPXY2)
VPXYW2 = VPX(DA3,VPXY3,VPXY4)
TBLU4 = VPX(DA4,VPXYW1,VPXYW2)
RETURN
END

```

```
FUNCTION VPX(A,B,C)
VPX = B + A*(C-B)
RETURN
END
```

```
SUBROUTINE SEARCH(XX,XX,NOX,NLX,I,DELTA,A)
DIMENSION X(1)
1 NO X = NO X + 1
IF(XX.LT.X(NO X)) GO TO 5
IF(NO X.GE.NLX)GO TO 5
GO TO 1
5 DELTA = XX - X(NO X)
I = NO X
IF(NO X.NE.1) GO TO 10
A = X(NO X + 1) - X(NO X)
RETURN
10 IF(XX.LT.X(NLX)) GO TO 20
DELTA = XX - X(NO X-1)
A = X(NO X) - X(NO X-1)
I = NO X - 1
RETURN
20 DELTA = XX - X(NO X - 1)
A = X(NO X) - X(NO X-1)
I = (NO X-1)
RETURN
END
```

```

SUBROUTINE VPLOT (XS,XI,XA,YS,YI,YA,NPT,NPL,NC,LX,LY,XRN,YRN,TIX)
DIMENSION XS(1),YS(1),NOPTS(30),SX(200),X(200),Y(200)
DIMENSION ITY(18),TIX(6),L(120),NCH(27)
COMMON / RFOPLO / HARM1,HARMF,TITLE(36)
COMMON / COMM6 / ATENM1(5),ABC(5)
DATA ND/'*'/,NP/*+*/,NM/*-/ ,NB/* */
DATA NCH/'A','B','C','D','E','F','G','H','I','J',
1      'K','L','M','N','O','P','Q','R','S','T','U',
2      'V','W','X','Y','Z',' '
DATA ITY    /*I*,*N*,*L*,*E*,*T*,* ,*A*,*T*,*T*,*E*,*N*,*U*,
1      *A*,*T*,*I*,*O*,*N*,* /
3 FORMAT('1'18A4/18A4)
100 FORMAT('1      DELTA EPNL = 'F5.1,10X'FAN HARMONIC = 'F5.1/7X
1      'INLET HARMONIC (A = 'F4.1 'B = 'F4.1 ' C = 'F4.1
1      ' D = 'F4.1 ' E = 'F4.1
1      /7X,18A4/)
5 FORMAT(120A1)
7 FORMAT(/F13.0,10F20.0)
11 FORMAT(5X,A1,1X,F3.0,1X,109A1)
15 FORMAT(/30X,6A4)
NXS=6
NXSM=NXS-1
XNXS=NXSM
NPN=NPT/NPL
IF (XRN.LE.0. OR.XRN.GT.100.) XRN=100.
IF (YRN.LE.0.) YRN=50.
NXR=XRN+12
NYR=YRN
NY=NYR+1
NCR=5
IF (LX.EQ.5)NCR=4
NRC=2*NCR
DO 500 IJM=1,NPT
X(IJM)=XS(IJM)
500 Y(IJM)=YS(IJM)
DO 18 I=1,NPL
18 NOPTS(I)=NPN
NOPTS(NPL+1)=0
XAX=(1CC./XRN)*(XA-XI) + XI
SX(1)=XI
SX(NXS)=XAX
U=XI
DO 16 K=2,NXS
U=(XAX-XI)/XNXS+U
16 SX(K)=U+C.C0001
CX=XRN /(XA-XII)
CALL VPOT (X,XI,NPT,0,XRN + CX)
CY=YRN/(YA-YII)
DO 50 NN=1,NC
T1 = YRN - 6.
T2 = T1 - 18

```

```

M1=1
CALL VPOT (Y,YI,NPT,1,YRN ,CY)
TT=YRN
TUP=YRN+1.
WRITE(6,100) HARM1,HARMF,(ATENM1(I),I=1,5),(TITLE(I),I=1,18)
DO 43 KK=1,NY
N=1
JED=1
T=TUP-KK
DO 25 J=1,NXR
25 L(J)=NB
L(12)=NC
L(NXR)=NC
IF(T-TT) 56,56,30
56 SCALE=T/CY+YI
L(12)=NP
L(NXR)=NP
N=0
TT=TT-5.
30 CONTINUE
IF (KK.NE.1.AND.KK.NE.NY) GO TO 37
DO 32 J=12,NXR
32 L(J)=ND
DO 35 J=12,NXR,NRC
L(J+NCR)=NM
35 L(J)=NP
37 NNN=NOPTS(1)
DO 40 LM=1,NPL
DO 39 I=JED,NNN
IF(Y(I)-T) 39,62,39
62 J=X(I)
L(J+12)=NCH(LM)
39 CONTINUE
JED=NNN+1
40 NNN=NNN+NPTS(LM+1)
IF (T.GT.T1.OR.T.LE.T2) GO TO 41
L(6)=ITY(M1)
M1=M1+1
41 IF(N-1) 66,42,66
66 WRITE(6,11) L(6),SCALE,(L(J),J=12,NXR)
GO TO 43
42 WRITE(6,5) (L(J),J=1,NXR)
43 CONTINUE
NXSS = NXS - 2
46 WRITE(6,7)(SX(K),K=1,NXSS)
50 WRITE(6,15) (TIX(I),I=1,6)
RETURN
END

```

```
SUBROUTINE VPOT (V,VMIN,NP,J,VC,C)
DIMENSION      V(1)
DO 1 I=1,NP
V(I)=FLCAT(IFIX(C*(V(I)-VMIN)+.5))
1 IF (V(I).LT.0.OR.V(I).GT.VC) V(I)=VC
RETURN
END
```

**APPENDIX 7**  
**EPNL ACOUSTIC TRADE STUDY RESULTS**

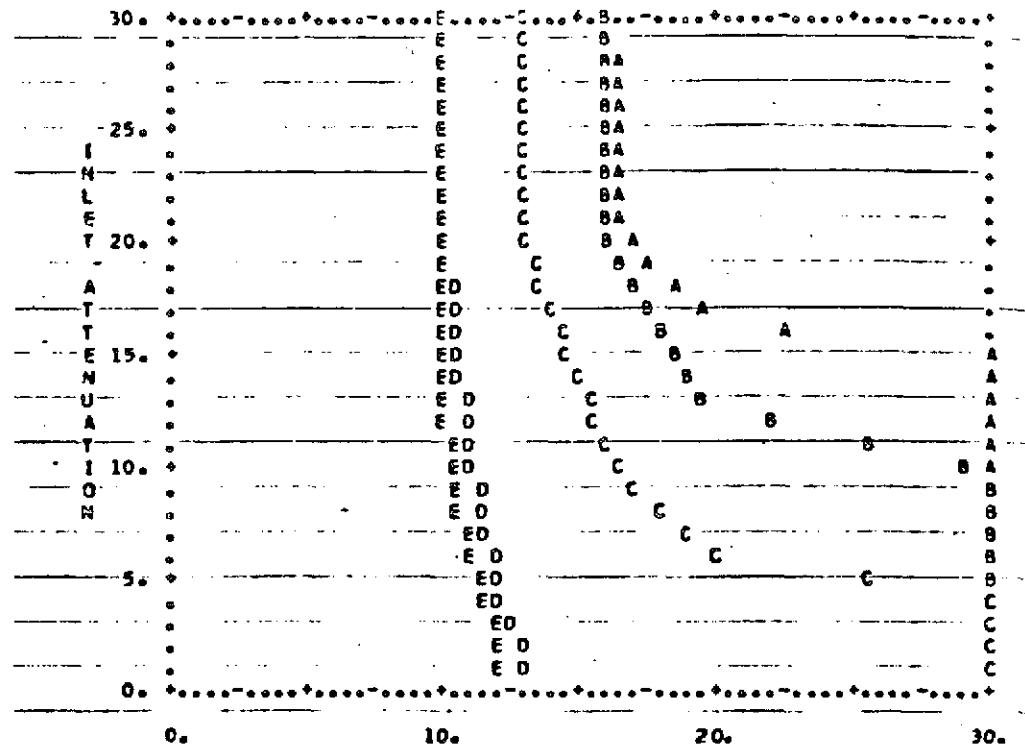
DELTA EPNL = 5.0      FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TONE, 0. JET ATT. WITH TURB. EPNL = 101.80, RUN 1 A

	30.	ED	C	-BA		
I		ED	C	B		
N		ED	C	B		
L		ED	C	B		
E		ED	C	B		
T	20.	ED	C	BA		
A		E	C	BA		
T		E	C	B	A	
E	15.	E	C	B		A
N		E	C	B		A
U		ED	C	B		A
A		ED	C	B	A	
T		ED	C	B	A	
I	10.	ED	C		BA	
O		E	C		B	
N		ED	C		B	
S		ED	C		B	
0		ED	C		C	
		ED			C	
		ED			C	
		ED			C	
		ED			C	
0.	10.	20.	30.			

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 0.0	DELTA EPNL = 5.0			
	A	B	C	D	E
INLET HARMONIC =	0	5	10	20	30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	15.57	15.99
2.00	*****	*****	*****	15.32	15.79
3.00	*****	*****	*****	16.07	15.59
4.00	*****	*****	*****	15.84	15.39
5.00	*****	*****	28.44	15.61	15.19
6.00	*****	*****	23.98	15.47	15.08
7.00	*****	*****	19.98	15.34	14.97
8.00	*****	*****	19.43	15.20	14.87
9.00	*****	*****	18.92	15.06	14.77
10.00	*****	29.75	18.43	14.92	14.67
11.00	*****	27.89	18.20	14.88	14.62
12.00	*****	25.90	17.98	14.84	14.57
13.00	*****	23.79	17.77	14.79	14.53
14.00	*****	21.53	17.56	14.75	14.48
15.00	*****	19.88	17.35	14.71	14.44
16.00	27.35	19.55	17.15	14.67	14.39
17.00	22.69	19.22	16.96	14.62	14.35
18.00	19.75	18.90	16.77	14.58	14.31
19.00	19.21	18.59	16.59	14.54	14.27
20.00	18.70	18.29	16.41	14.50	14.22
21.00	18.66	18.28	16.40	14.50	14.22
22.00	18.62	18.28	16.40	14.50	14.22
23.00	18.58	18.28	16.40	14.50	14.22
24.00	18.54	18.27	16.40	14.50	14.22
25.00	18.50	18.27	16.40	14.50	14.22
26.00	18.46	18.26	16.40	14.49	14.22
27.00	18.42	18.26	16.40	14.49	14.22
28.00	18.38	18.25	16.40	14.49	14.22
29.00	18.34	18.25	16.40	14.49	14.22
30.00	18.30	18.24	16.40	14.49	14.22

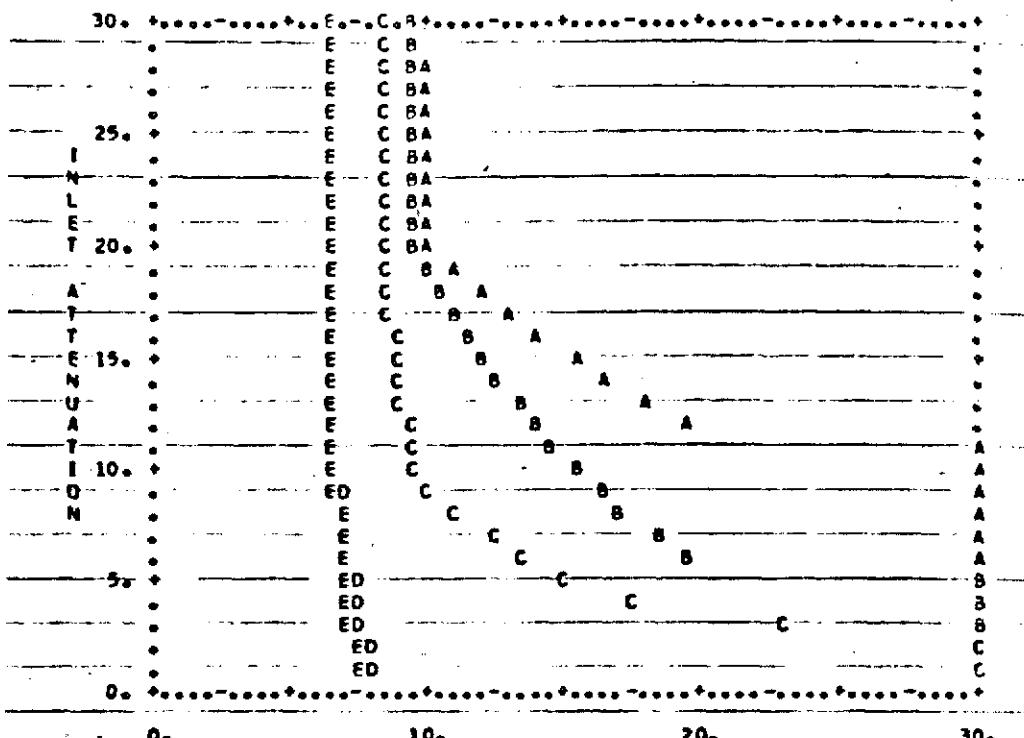
DELTA EPNL = 5.0      FAN HARMONIC = 5.0  
 INLET HARMONIC LA = 0, B = 5, C = 10, D = 20, E = 30  
 APP. WITH TOYE, O. JET ATT, WITH TURB. EPNL = 101.80, RUN 1 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 5.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	13.22	12.20
2.00	*****	*****	*****	*****	12.97	12.00
3.00	*****	*****	*****	*****	12.53	11.81
4.00	*****	*****	*****	*****	12.21	11.63
5.00	*****	*****	*****	25.71	11.91	11.46
6.00	*****	*****	*****	19.83	11.76	11.21
7.00	*****	*****	*****	18.86	11.51	10.98
8.00	*****	*****	*****	17.96	11.46	10.74
9.00	*****	*****	*****	17.11	11.32	10.51
10.00	*****	29.09	16.32	11.18	10.28	
11.00	*****	25.53	15.96	11.07	10.25	
12.00	*****	21.81	15.60	10.96	10.23	
13.00	*****	19.71	15.27	10.85	10.21	
14.00	****	19.17	14.94	10.74	10.19	
15.00	30.59	18.65	14.62	10.63	10.17	
16.00	22.32	18.14	14.31	10.52	10.15	
17.00	19.33	17.64	14.02	10.41	10.13	
18.00	18.44	17.15	13.73	10.31	10.12	
19.00	17.59	16.68	13.45	10.20	10.10	
20.00	16.77	16.21	13.18	10.09	10.03	
21.00	16.71	16.20	13.18	10.09	10.03	
22.00	16.64	16.19	13.18	10.08	10.07	
23.00	16.58	16.17	13.18	10.08	10.07	
24.00	16.52	16.16	13.18	10.07	10.06	
25.00	16.45	16.15	13.17	10.07	10.05	
26.00	16.39	16.14	13.17	10.06	10.06	
27.00	16.33	16.12	13.17	10.06	10.05	
28.00	16.26	16.11	13.17	10.05	10.05	
29.00	16.20	16.10	13.17	10.05	10.04	
30.00	16.13	16.09	13.17	10.05	10.04	

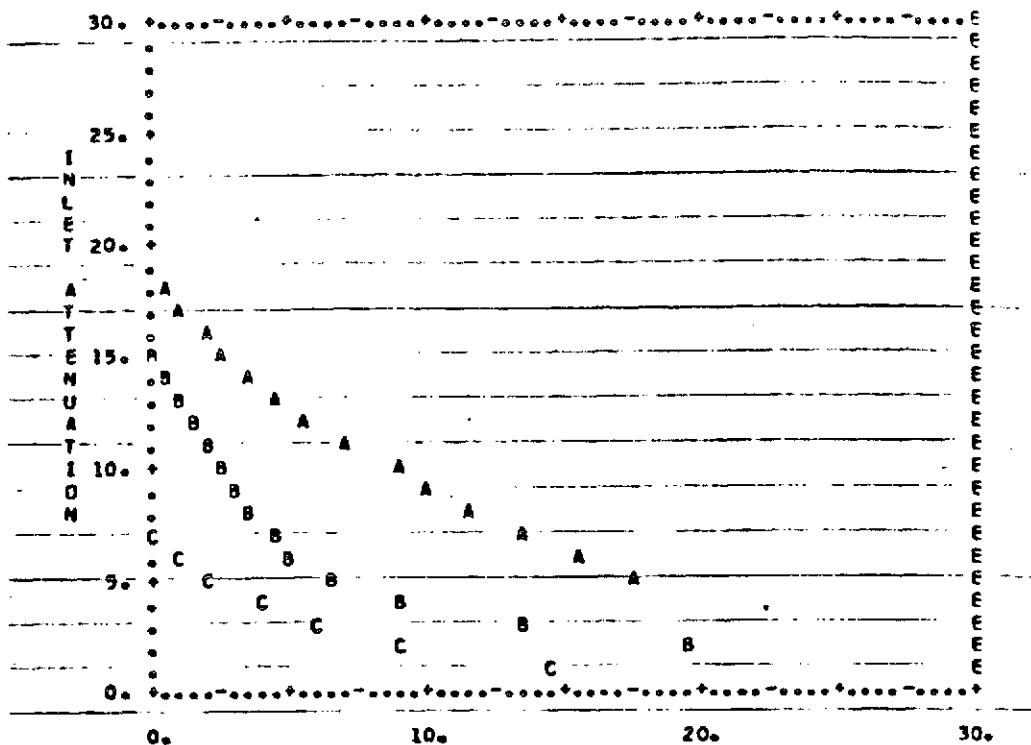
DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC IA = 0, R = 5, C = 10, D = 20, E = 30  
 APP, WITH TONE, 0° JET ATT, WITH TURB, EPNL = 101.80, RUN 1/A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA-EPNL = 5.0			
INLET HARMONIC		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	7.92	7.32
2.00	*****	*****	*****	7.76	7.25
3.00	*****	*****	23.08	7.60	7.19
4.00	*****	*****	17.49	7.45	7.13
5.00	*****	*****	14.95	7.30	7.07
6.00	*****	19.33	13.58	7.19	6.97
7.00	*****	18.26	12.31	7.07	6.85
8.00	*****	17.24	11.14	6.95	6.63
9.00	*****	16.26	10.05	6.83	6.66
10.00	*****	15.32	9.59	6.71	6.56
11.00	*****	14.63	9.46	6.69	6.54
12.00	19.29	13.96	9.33	6.57	6.51
13.00	17.92	13.30	9.20	6.65	6.49
14.00	16.61	12.66	9.08	6.63	6.47
15.00	15.36	12.04	8.95	6.61	6.45
16.00	14.17	11.43	8.83	6.59	6.43
17.00	13.04	10.84	8.72	6.57	6.41
18.00	11.96	10.26	8.60	6.55	6.39
19.00	10.92	9.86	8.49	6.53	6.38
20.00	9.97	9.61	8.38	6.51	6.36
21.00	9.94	9.61	8.37	6.51	6.36
22.00	9.92	9.61	8.36	6.51	6.36
23.00	9.89	9.61	8.34	6.50	6.35
24.00	9.87	9.61	8.33	6.50	6.35
25.00	9.84	9.61	8.32	6.50	6.35
26.00	9.81	9.61	8.31	6.50	6.35
27.00	9.79	9.61	8.30	6.50	6.35
28.00	9.76	9.61	8.28	6.50	6.35
29.00	9.74	9.61	8.27	6.49	6.35
30.00	9.71	9.61	8.26	6.49	6.35

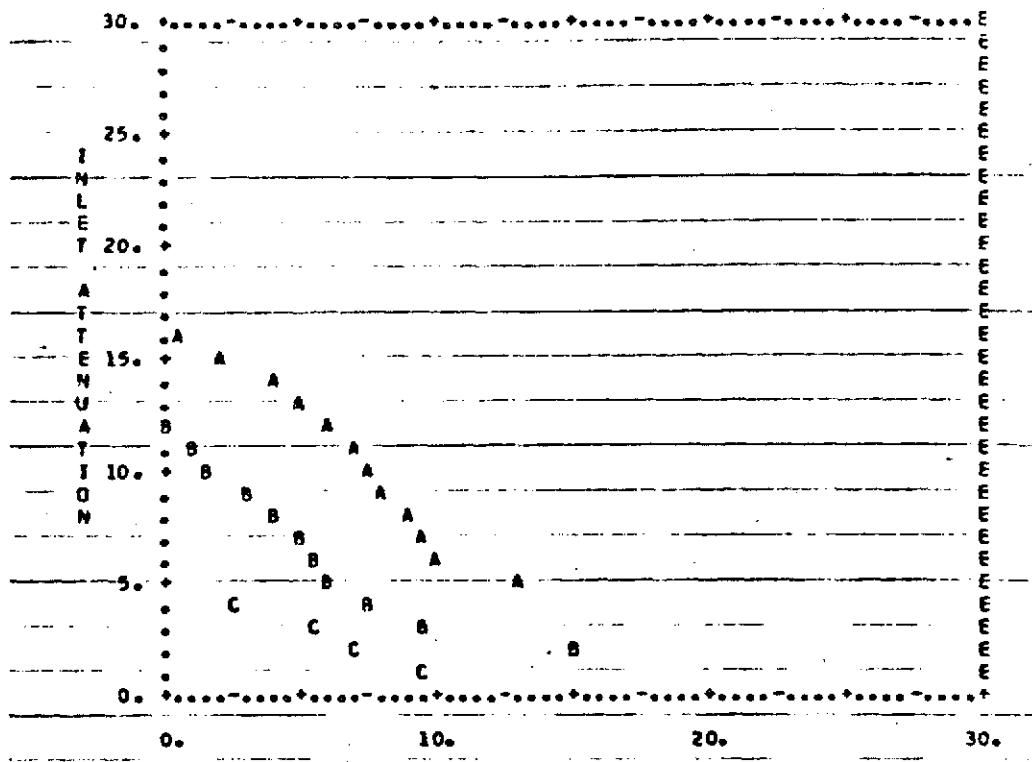
DELTA EPNL = 5.0      FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TONE, 0. JET ATT. WITH TURB, EPNL = 101.80, RUN 1 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 5.0			
INLET FUND.	FAN FUND.	INLET FUND.	FAN FUND.	INLET FUND.	FAN FUND.
0	5	10	20	30	
1.00	*****	*****	14.53	*****	*****
2.00	*****	19.33	9.04	*****	*****
3.00	*****	13.40	6.00	*****	*****
4.00	*****	8.97	3.79	*****	*****
5.00	17.65	6.37	1.96	*****	*****
6.00	15.55	5.21	0.96	*****	*****
7.00	13.58	4.38	0.05	*****	*****
8.00	11.72	3.66	*****	*****	*****
9.00	9.98	2.98	*****	*****	*****
10.00	8.94	2.32	*****	*****	*****
11.00	7.20	1.88	*****	*****	*****
12.00	5.67	1.45	*****	*****	*****
13.00	4.50	1.02	*****	*****	*****
14.00	3.57	0.61	*****	*****	*****
15.00	2.69	0.20	*****	*****	*****
16.00	1.84	*****	*****	*****	*****
17.00	1.04	*****	*****	*****	*****
18.00	0.28	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

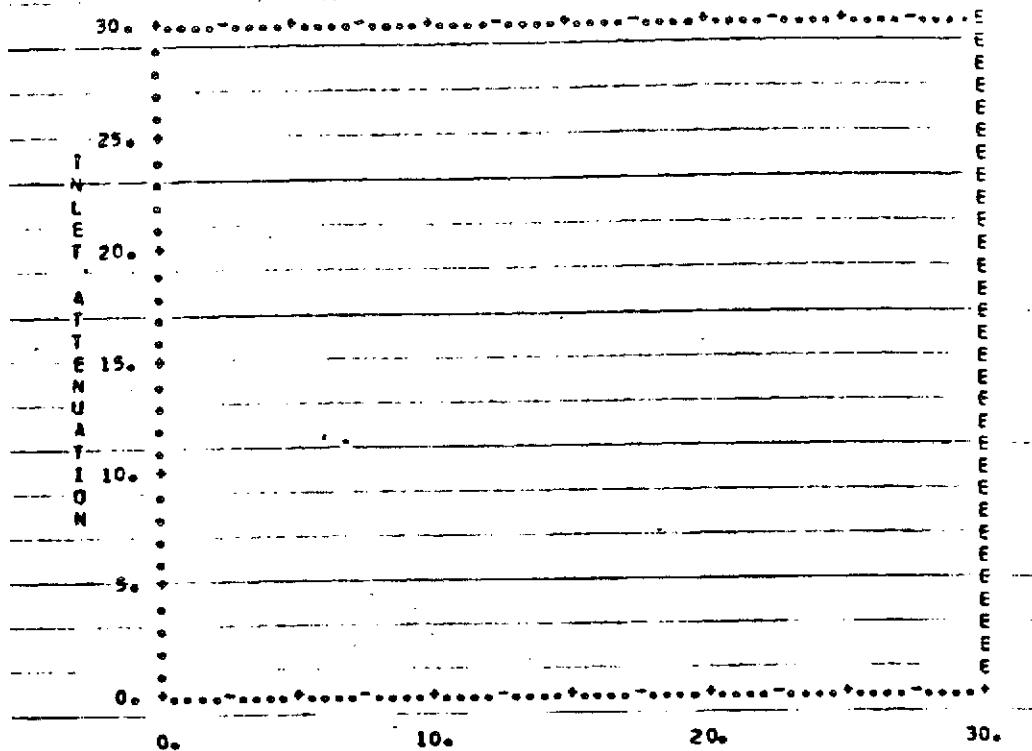
DELTA EPNL = 5.0      FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TUBE, O. JET ATT. WITH TURB. EPNL = 101.80, RUN 1 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EPNL = 5.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	9.55	*****	*****	*****
2.00	*****	15.20	7.21	*****	*****	*****
3.00	*****	9.26	5.44	*****	*****	*****
4.00	*****	7.58	2.71	*****	*****	*****
5.00	13.24	6.21	*****	*****	*****	*****
6.00	10.11	5.59	*****	*****	*****	*****
7.00	9.38	5.02	*****	*****	*****	*****
8.00	8.78	3.90	*****	*****	*****	*****
9.00	8.22	2.80	*****	*****	*****	*****
10.00	7.69	1.73	*****	*****	*****	*****
11.00	6.75	0.86	*****	*****	*****	*****
12.00	5.91	0.00	*****	*****	*****	*****
13.00	5.15	*****	*****	*****	*****	*****
14.00	3.77	*****	*****	*****	*****	*****
15.00	2.22	*****	*****	*****	*****	*****
16.00	0.69	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

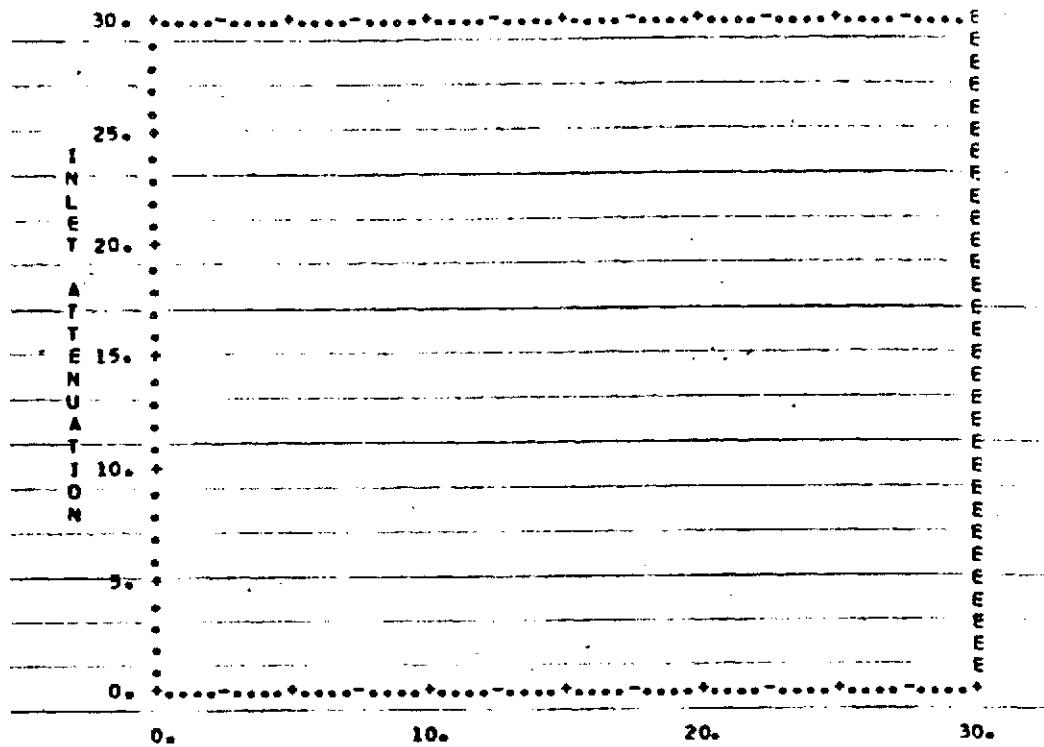
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TOVE, 0. JET ATT. WITH TURB. EPNL = 101.80, RUM 1 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 10.0				
INLET FUND.	FAN FUND.	A	B	C	D	E
INLET FUND.	FAN FUND.	0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

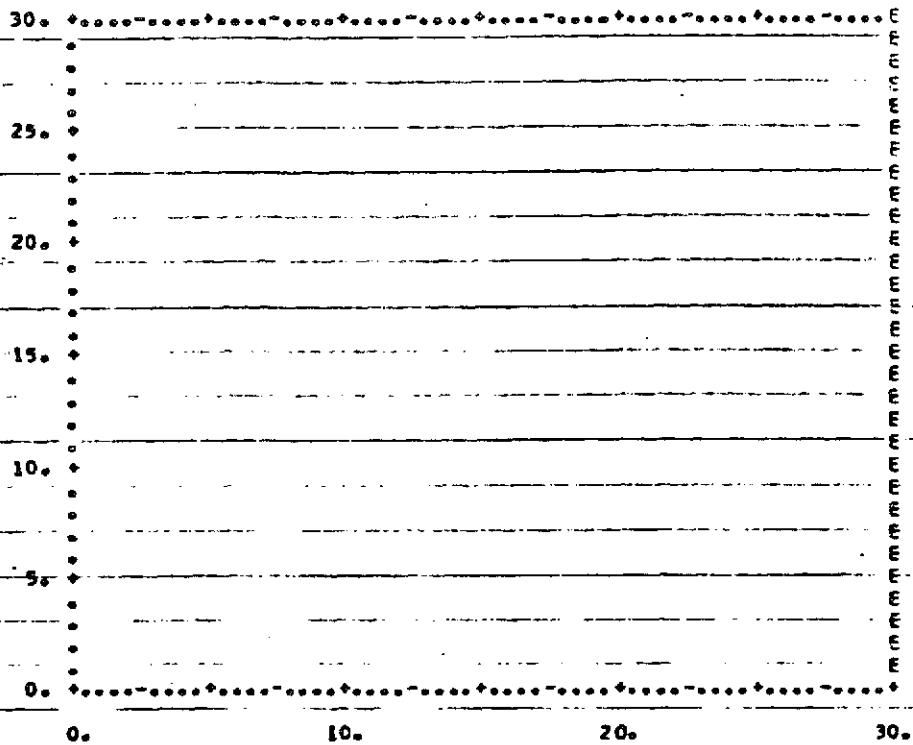
DELTA EPNL = 10.0      FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TONE, O. JET ATT. WITH TURA, EPNL = 101.80, RUN 1 A



#### FAN DUCT ATTENUATION

	FAN HARMONIC = 5.0		DELTA EPNL = 10.0		
	A	B	C	D	E
INLET HARMONIC	0	5	10	20	30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TONE, 0° JET ATT, WITH TURB, EPNL = 101.50, RUM 1 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0			
INLET HARMONIC *	A	B	C	D	E
	0	5	10	20	30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

DELTA EPNL = 10.0      FAN HARMONIC = 20.0  
 INLET HARMONIC IA = 0, B = 5, C = 10, D = 20, E = 30  
 APP, WITH TONE, O. JET ATT, WITH TURB, EPNL = 101.80, RUN 1 A

	30.	20.	10.	0.
	E	D		C
	E	D		C
	E	D		C
	E	D		C
25.	E	D		C
I	E	D		C
N	E	D		C
L	E	D		C
E	E	D		C
T 20.	E	D		C
	E	D		C
A	E	D		C
T	E	D		C
T	E	D		C
E 15.	E	D		C
N	E	D		C
U	E	D		C
A	E	D		C
T	E	D		C
E 10.	E	D		C
D	E	D		C
N	E	D		C
	E	D		C
5.	E			D
			E	E
				E
				E
0.				

0.      10.      20.      30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET	FUND.	A	B	C	D	E
HARMONIC		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	20.02
5.00	*****	*****	*****	*****	*****	18.63
6.00	*****	*****	*****	*****	18.84	16.99
7.00	*****	*****	*****	*****	17.64	15.57
8.00	*****	*****	*****	*****	16.55	14.35
9.00	*****	*****	*****	*****	15.54	13.28
10.00	*****	*****	*****	*****	14.60	12.34
11.00	*****	*****	*****	*****	14.30	12.12
12.00	*****	*****	*****	*****	14.00	11.91
13.00	*****	*****	*****	*****	13.71	11.70
14.00	*****	*****	*****	*****	13.43	11.59
15.00	*****	*****	*****	*****	13.16	11.30
16.00	*****	*****	*****	*****	12.89	11.11
17.00	*****	*****	*****	*****	12.64	10.93
18.00	*****	*****	*****	*****	12.39	10.75
19.00	*****	*****	*****	*****	12.15	10.57
20.00	*****	*****	*****	11.91	10.40	
21.00	*****	*****	*****	11.91	10.40	
22.00	*****	*****	*****	11.91	10.40	
23.00	*****	*****	*****	11.91	10.40	
24.00	*****	*****	*****	11.91	10.40	
25.00	*****	*****	*****	11.91	10.40	
26.00	*****	*****	*****	11.91	10.40	
27.00	*****	*****	*****	11.91	10.40	
28.00	*****	*****	*****	11.91	10.40	
29.00	*****	*****	*****	11.91	10.40	
30.00	*****	*****	*****	11.91	10.40	

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC IA = 0, B = 5, C = 10, D = 20, E = 30  
 APP, WITH TONE, O, JET ATT, WITH TURB, EPNL = 101.80, RUN 1 A

	30.	25.	20.	15.	10.	5.	0.	
I	E D							C
N	E D							C
L	E D							C
E	E D							C
T	E D							C
A	E D							C
T	E D							C
E	E D							C
N	E D							C
U	E D							C
A	E D							C
T	E D							C
I	E D							C
O	E D							C
N	E D							C
	E D							C
	E D							C
	E D							C
	E D							C
	E D							C

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 30.0	DELTA EPNL = 10.0			
	A	B	C	D	E
INLET HARMONIC =	0	5	10	20	30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	18.84
3.00	*****	*****	*****	19.71	16.74
4.00	*****	*****	*****	17.69	15.14
5.00	*****	*****	*****	15.49	13.77
6.00	*****	*****	*****	13.95	12.
7.00	*****	*****	*****	12.38	10.00
8.00	*****	*****	*****	11.06	9.31
9.00	*****	*****	*****	9.95	9.42
10.00	*****	*****	*****	9.61	9.06
11.00	*****	*****	*****	9.53	8.96
12.00	*****	*****	*****	9.45	8.86
13.00	*****	*****	*****	9.37	8.77
14.00	*****	*****	*****	9.29	8.67
15.00	*****	*****	*****	9.22	8.58
16.00	*****	*****	*****	9.14	8.48
17.00	*****	*****	*****	9.07	8.39
18.00	*****	*****	*****	8.99	8.30
19.00	*****	*****	*****	8.92	8.21
20.00	*****	*****	*****	8.84	8.11
21.00	*****	*****	*****	8.84	8.11
22.00	*****	*****	*****	8.84	8.11
23.00	*****	*****	*****	8.84	8.11
24.00	*****	*****	*****	8.84	8.11
25.00	*****	*****	*****	8.84	8.11
26.00	*****	*****	*****	8.84	8.11
27.00	*****	*****	*****	8.84	8.11
28.00	*****	*****	*****	8.84	8.11
29.00	*****	*****	*****	8.84	8.11
30.00	*****	*****	*****	8.84	8.11

DELTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC IA = 0, B = 5, C = 10, D = 20, E = 30  
 T/D. WITH TONE, 0, JET ATT, NO TURB, EPNL = 103.02, RUN 2 A

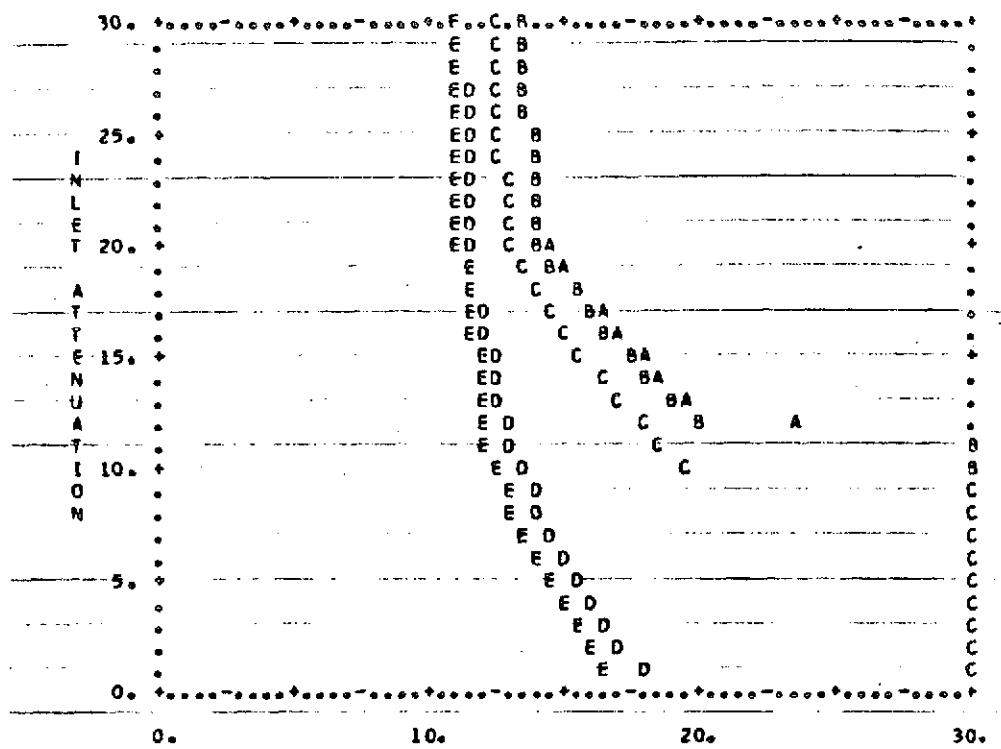
	30.	FC	B			
	.	EC	B			
	.	EC	B			
	.	EC	B			
	.	EC	BA			
	25.		EC	BA		
I	.		EC	B		
N	.		EC	B		
L	.		EC	BA		
E	.		EC	BA		
T	20.		EC	BA		
A	.		EC	BA		
	.		ED	C	BA	
T	.		E	C	BA	
E	15.		E	C	BA	
N	.		E	C	BA	
U	.		ED	C	BA	
A	.		ED	C	B	A
	.		E	C	B	A
T	.		E	C		B
I	10.		ED			C
O	.		ED			C
N	.		ED			C
	.		ED			C
	5.		ED			C
	.		ED			C
	.		ED			C
	.		ED			C
	0.		ED			C

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 0.0	DELTA EPNL = 5.0				
	A	B	C	D	E	
	INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
	1.00	*****	*****	*****	17.38	16.88
	2.00	*****	*****	*****	16.95	16.45
	3.00	*****	*****	*****	16.54	16.05
	4.00	*****	*****	*****	16.15	15.68
	5.00	*****	*****	*****	15.77	15.32
	6.00	*****	*****	*****	15.43	15.02
	7.00	*****	*****	*****	15.10	14.73
	8.00	*****	*****	*****	14.77	14.44
	9.00	*****	*****	*****	14.44	14.15
	10.00	*****	*****	18.98	14.12	13.88
	11.00	28.57	22.91	18.34	14.01	13.78
	12.00	21.81	19.54	17.73	13.91	13.69
	13.00	19.49	18.88	17.14	13.80	13.60
	14.00	18.82	18.25	16.57	13.70	13.51
	15.00	18.19	17.65	16.03	13.60	13.43
	16.00	17.58	17.07	15.50	13.50	13.34
	17.00	17.00	16.51	15.00	13.41	13.25
	18.00	16.45	15.98	14.51	13.31	13.17
	19.00	15.92	15.47	14.04	13.22	13.09
	20.00	15.42	14.98	13.58	13.12	13.00
	21.00	15.31	14.92	13.57	13.12	13.00
	22.00	15.21	14.87	13.55	13.11	13.00
	23.00	15.10	14.81	13.53	13.11	13.00
	24.00	15.00	14.76	13.51	13.10	12.99
	25.00	14.90	14.71	13.49	13.10	12.99
	26.00	14.80	14.65	13.47	13.09	12.99
	27.00	14.70	14.60	13.45	13.08	12.98
	28.00	14.60	14.55	13.43	13.08	12.98
	29.00	14.50	14.49	13.41	13.07	12.98
	30.00	14.40	14.44	13.39	13.07	12.97

DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0% JET ATT, NO TURB, EPNL = 103.02, RUN 2 A



#### FAN DUCT ATTENUATION

	FAN HARMONIC = 5.0	DELTA EPNL = 5.0	A	B	C	D	E
	INLET HARMONIC = 0		0	5	10	20	30
1.00	*****	*****	*****	*****	17.87	16.59	
2.00	*****	*****	*****	*****	17.20	16.04	
3.00	*****	*****	*****	*****	16.59	15.52	
4.00	*****	*****	*****	*****	15.99	15.03	
5.00	*****	*****	*****	*****	15.44	14.57	
6.00	*****	*****	*****	*****	14.99	14.	
7.00	*****	*****	*****	*****	14.56	13.65	
8.00	*****	*****	*****	*****	14.15	13.21	
9.00	*****	*****	*****	*****	13.76	12.79	
10.00	*****	*****	19.57	13.39	12.38		
11.00	30.72	*****	18.66	13.17	12.25		
12.00	23.74	19.89	17.82	12.95	12.12		
13.00	19.59	19.02	17.04	12.74	11.99		
14.00	18.73	18.21	16.31	12.53	11.87		
15.00	17.91	17.44	15.64	12.32	11.75		
16.00	17.13	16.71	15.01	12.12	11.64		
17.00	16.38	16.02	14.41	11.91	11.52		
18.00	15.66	15.37	13.86	11.72	11.41		
19.00	14.97	14.75	13.34	11.52	11.32		
20.00	14.30	14.16	12.84	11.33	11.20		
21.00	14.19	14.08	12.81	11.32	11.19		
22.00	14.09	14.00	12.79	11.31	11.19		
23.00	13.98	13.93	12.76	11.29	11.18		
24.00	13.88	13.85	12.73	11.28	11.18		
25.00	13.78	13.78	12.70	11.27	11.17		
26.00	13.68	13.70	12.67	11.26	11.17		
27.00	13.59	13.63	12.65	11.25	11.16		
28.00	13.50	13.55	12.62	11.24	11.16		
29.00	13.40	13.48	12.59	11.23	11.15		
30.00	13.32	13.41	12.56	11.22	11.15		

DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0. JET ATT, NO TURB, EPNL = 103.02, RUN 2 A

	30.	EDCB	
		ECDR	
		ECB	
		ECBA	
25.		EC B	
I		EC B	
N		EC B	
L		EC B	
E		E CB	
T	20.	E CB	
		E CBA	
A		ED C B	
F		ED C BA	
T		ED C B	
E	15.	ED C BA	
N		ED C BA	
U		ED C BA	
A		ED C BA	
T		ED C BA	
I	10.	ED C B A	
O		ED C	8
N		ED	C
		ED	C
S.		ED	C
		ED	C
		ED	C
		ED	C
0.			

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 10.0		DEELTA EPNL = 5.0		
	A	B	C	D	E
INLET HARMONIC =	0	5	10	20	30
1.00	*****	*****	*****	14.73	13.23
2.00	*****	*****	*****	14.13	12.77
3.00	*****	*****	*****	13.58	12.37
4.00	*****	*****	*****	13.08	11.99
5.00	*****	*****	*****	12.62	11.66
6.00	*****	*****	*****	12.19	11.19
7.00	*****	*****	*****	11.78	10.73
8.00	*****	*****	*****	11.39	10.29
9.00	*****	*****	19.62	11.02	9.94
10.00	21.33	19.59	17.35	10.67	9.77
11.00	18.96	18.43	16.44	10.51	9.72
12.00	17.84	17.38	15.58	10.35	9.67
13.00	16.81	16.37	14.78	10.20	9.62
14.00	15.84	15.45	14.03	10.06	9.57
15.00	14.95	14.59	13.32	9.96	9.52
16.00	14.11	13.80	12.66	9.89	9.47
17.00	13.33	13.06	12.02	9.83	9.43
18.00	12.59	12.36	11.43	9.77	9.38
19.00	11.90	11.71	10.86	9.70	9.33
20.00	11.25	11.09	10.33	9.64	9.29
21.00	11.17	11.04	10.28	9.64	9.28
22.00	11.09	10.98	10.24	9.64	9.28
23.00	11.01	10.92	10.20	9.63	9.27
24.00	10.93	10.86	10.15	9.63	9.27
25.00	10.85	10.81	10.11	9.63	9.26
26.00	10.77	10.75	10.07	9.63	9.26
27.00	10.69	10.69	10.02	9.62	9.25
28.00	10.61	10.64	9.99	9.62	9.25
29.00	10.54	10.58	9.97	9.62	9.25
30.00	10.46	10.52	9.95	9.62	9.24

DELTA EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0, JET ATT, NO TURB, EPNL = 103.02, RUN 2 A

30.		ED-C-B			
	*	ED	C B		
	*	ED	C B		
	*	ED	C B		
	*	ED	C B		
	25.	+	ED	C B	
I	.	ED	C B		
N	.	ED	C B		
L	.	ED	C B		
E	.	ED	C BA		
T	20.	+	ED	C BA	
A	.	ED	C B		
T	.	ED	C BA		
T	.	ED	C BA		
E	15.	+	ED	C BA	
N	.	ED	C BA		
U	.	ED	C BA		
A	.	ED	C BA		
T	.	ED	C BA		
I	10.	+	ED	C BA	
O	.	ED	C BA		
N	.	ED	C	B	A
	.	ED		C	B
	.	ED		C	C
S	.	ED		C	C
	.	ED		C	C
	.	ED		C	C
D.	.	ED		C	C

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 5.0				
INLET HARMONIC	FUND.	A	B	C	D	E
0		0	5	10	20	30
1.00	*****	*****	*****	8.04	7.41	
2.00	*****	*****	*****	7.49	6.89	
3.00	*****	*****	*****	6.93	6.38	
4.00	*****	*****	*****	6.37	5.87	
5.00	*****	*****	*****	5.80	5.37	
6.00	*****	*****	*****	5.42	4.97	
7.00	*****	*****	19.84	5.04	4.47	
8.00	30.56	26.67	15.96	4.65	3.96	
9.00	17.85	16.94	12.84	4.25	3.42	
10.00	14.80	13.59	10.26	3.85	2.86	
11.00	13.35	12.30	9.39	3.67	2.70	
12.00	12.04	11.14	8.61	3.49	2.55	
13.00	10.86	10.08	7.88	3.32	2.41	
14.00	9.81	9.24	7.20	3.15	2.28	
15.00	8.95	8.46	6.56	2.97	2.16	
16.00	8.14	7.72	5.96	2.80	2.04	
17.00	7.39	7.03	5.40	2.63	1.93	
18.00	6.67	6.38	4.86	2.46	1.83	
19.00	6.00	5.76	4.30	2.29	1.74	
20.00	5.36	5.17	3.77	2.12	1.65	
21.00	5.26	5.10	3.73	2.11	1.64	
22.00	5.15	5.03	3.70	2.10	1.64	
23.00	5.05	4.96	3.66	2.09	1.63	
24.00	4.95	4.89	3.63	2.08	1.62	
25.00	4.84	4.81	3.59	2.06	1.61	
26.00	4.74	4.74	3.56	2.05	1.61	
27.00	4.63	4.66	3.53	2.04	1.60	
28.00	4.53	4.59	3.49	2.03	1.59	
29.00	4.43	4.52	3.46	2.01	1.58	
30.00	4.33	4.44	3.43	2.00	1.58	

DELTA EPNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0. JET ATT, NO TURB, EPNL = 103.02, RUN 2 A

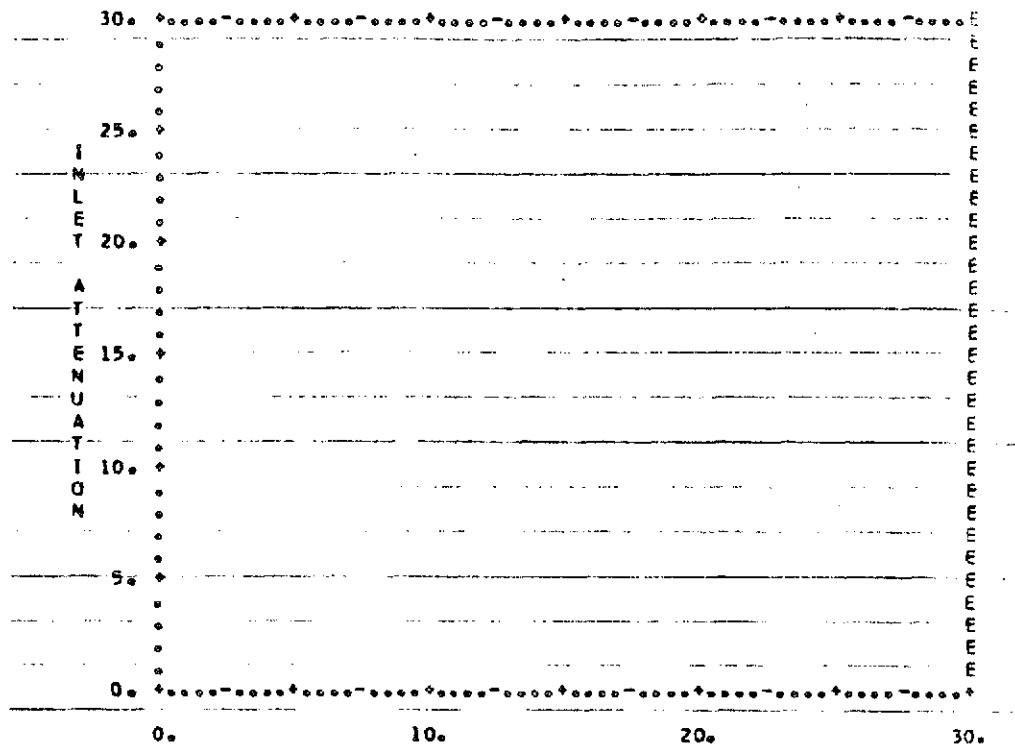
		ED	CB						
30.	+	ED	CB						
	-	ED	CB						
	-	ED	CB						
	-	ED	CB						
25.	+	E	CB						
I	-	E	CBA						
N	-	E	CBA						
L	-	E	C B						
E	-	E	C B						
T	20.	E	C B						
	-	E	CBA						
A	-	ED	C B						
T	-	ED	C A						
T	-	ED	C B						
E	15.	ED	C BA						
N	-	ED	C BA						
U	-	ED	C BA						
A	-	ED	C BA						
T	-	ED	C C B A						
I	10.	ED	C B A						
D	-	ED	C C B A						
N	-	ED	C C B A						
	-	E							
	-	ED							
5.	-	E							
	-	ED							
	-	E							
	-	ED							
	-	E							
0.	-	ED							
	-	E							

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 30.0	DELTA EPNL = 5.0			
	A	B	C	D	E
INLET HARMONIC =	0	5	10	20	30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	7.56	7.47
2.00	*****	*****	*****	7.27	7.22
3.00	*****	*****	*****	7.01	5.96
4.00	*****	*****	*****	6.76	6.70
5.00	*****	*****	*****	6.54	6.43
6.00	*****	*****	*****	6.28	6.10
7.00	*****	*****	18.77	6.02	5.76
8.00	25.45	20.48	14.44	5.76	5.42
9.00	16.75	15.72	10.98	5.50	5.03
10.00	13.33	12.04	9.04	5.25	4.56
11.00	11.71	10.67	8.52	5.15	4.48
12.00	10.25	9.72	8.05	5.08	4.41
13.00	9.50	9.15	7.63	4.99	4.33
14.00	8.95	8.64	7.25	4.85	4.25
15.00	8.46	8.18	6.91	4.70	4.12
16.00	8.01	7.75	6.60	4.56	4.01
17.00	7.61	7.36	6.32	4.43	4.04
18.00	7.24	7.00	6.06	4.29	3.95
19.00	6.91	6.66	5.82	4.16	3.89
20.00	6.60	6.35	5.59	4.02	3.81
21.00	6.32	6.30	5.57	4.01	3.80
22.00	6.45	6.26	5.55	3.99	3.71
23.00	6.37	6.21	5.53	3.97	3.77
24.00	6.29	6.16	5.51	3.96	3.76
25.00	6.21	6.11	5.49	3.94	3.75
26.00	6.13	6.06	5.47	3.93	3.74
27.00	6.05	6.01	5.45	3.91	3.73
28.00	5.97	5.96	5.43	3.90	3.72
29.00	5.89	5.91	5.40	3.88	3.70
30.00	5.81	5.86	5.38	3.86	3.69

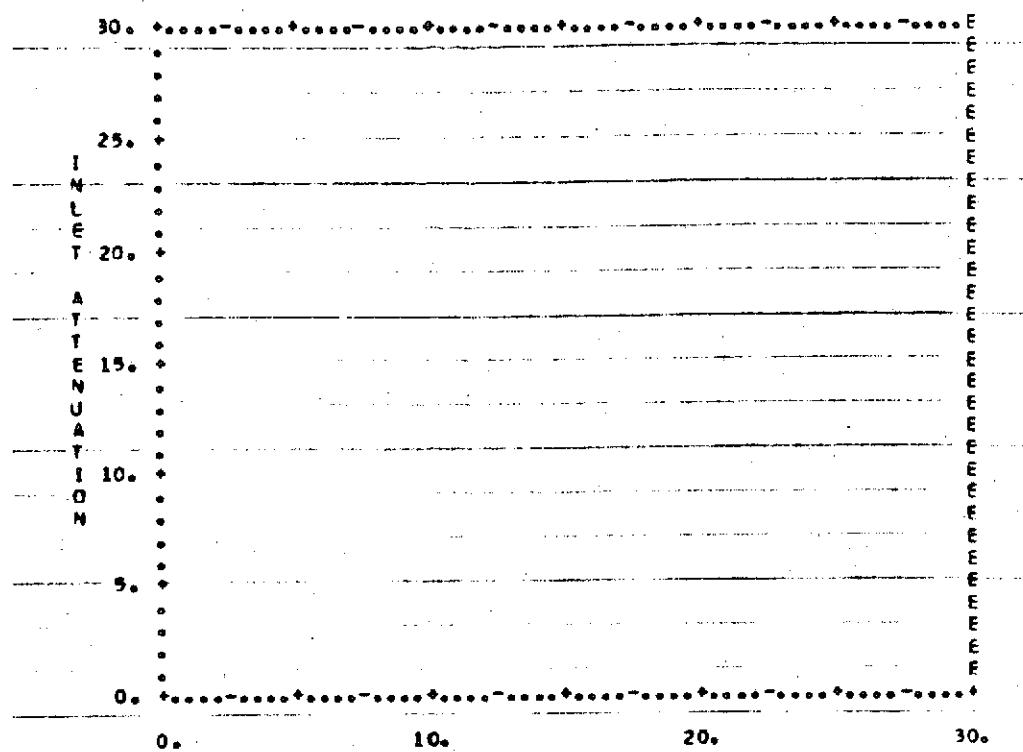
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/Q, WITH TONE, Q, JET ATT, NO TURB, EPNL = 103.02, RUN 2 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

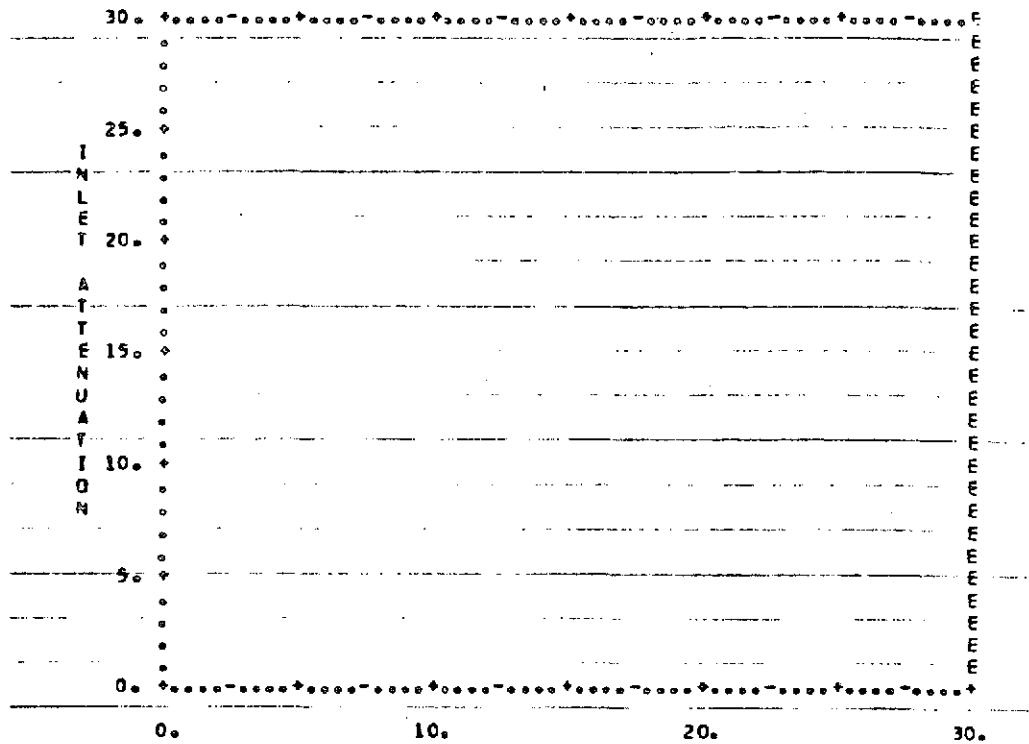
DELTA EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0, JET ATT, NO TURB, EPNL = 103.02, RUN 2 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 10.0				
INLET HARMONIC =	A	B	C	D	E	
	0	5	10	20	30	
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0° JET ATT, NO TURB, EPNL = 103.02, RUN 2 A

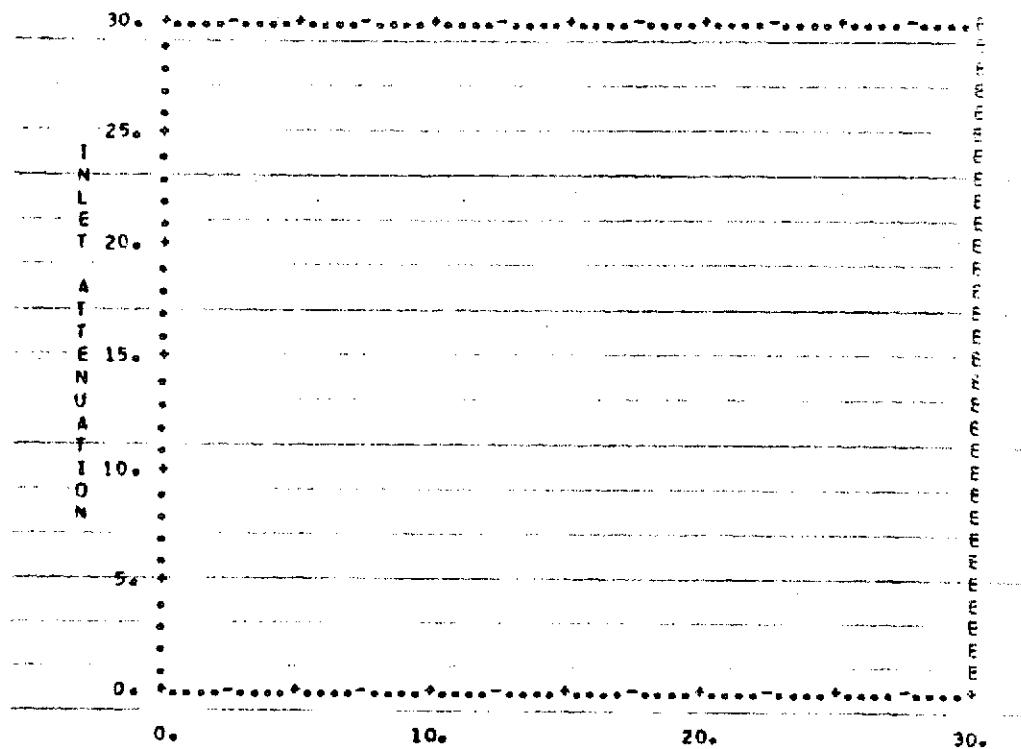


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0			
INLET HARMONIC	A	B	C	D	E
	0	5	10	20	30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

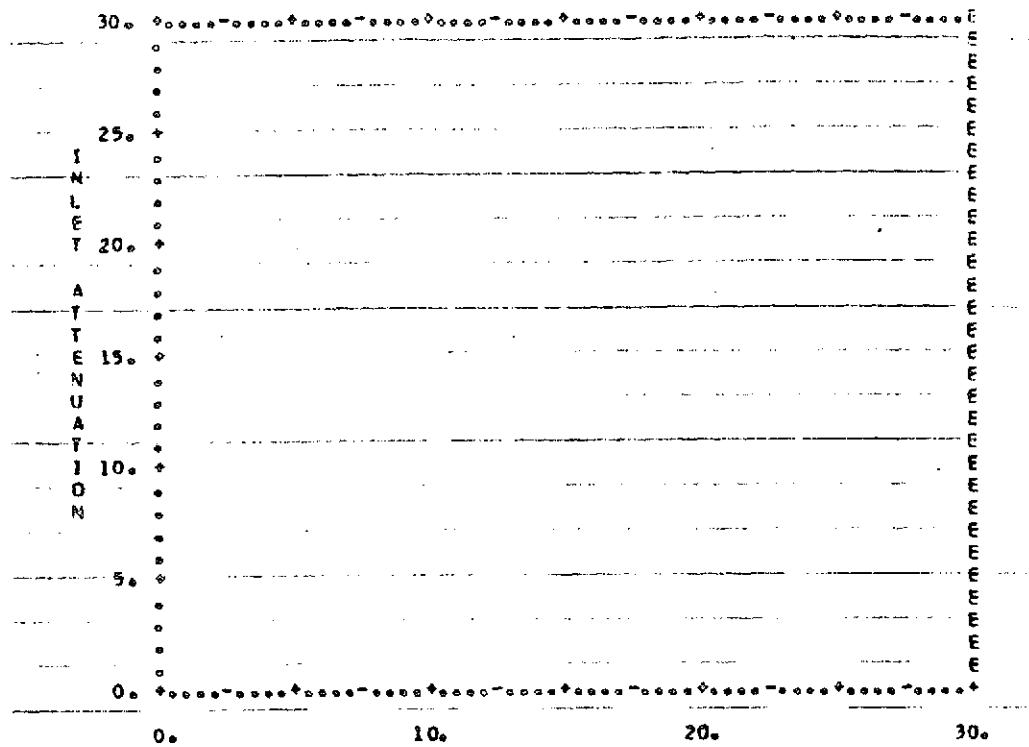
DELTA EPNL = 10.0      FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O. WITH TONE, 0, JET ATT. NO TURB, EPNL = 103.02, RUN 2 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET HARMONIC *		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

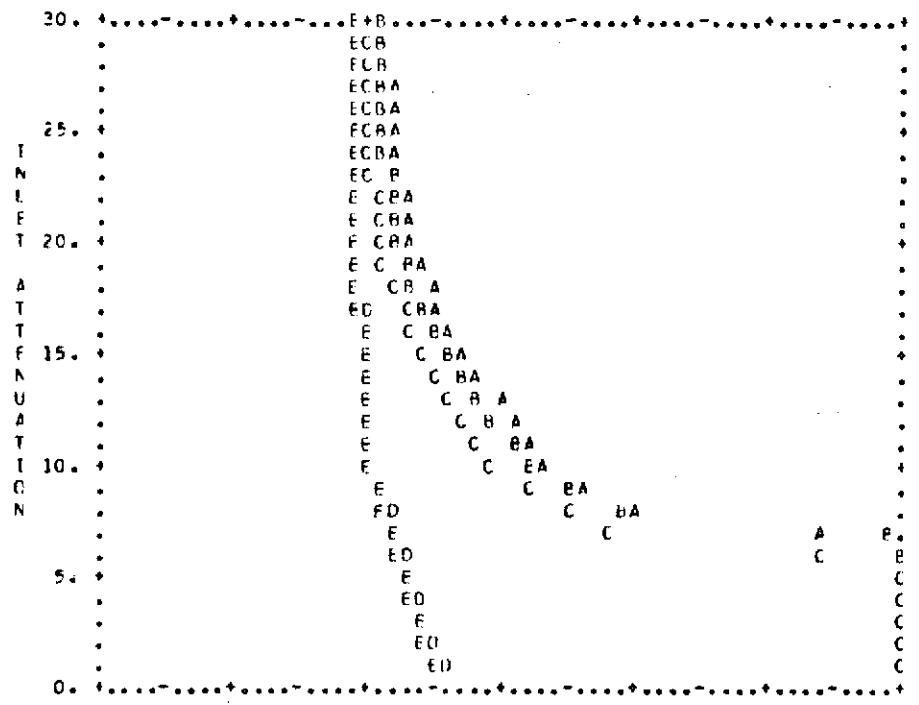
DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0, JET ATT, NO TURB, EPNL = 103.02, RUN 2 A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EPNL = 10.0				
INLET FUND.	FAN FUND.	A	B	C	D	E
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

DELTA EFLN = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE, 10. JET ATT, NO TURB, EFLN = 101.83, RUN 5A



0. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC =	FAN HARMONIC = 0.0		DELTA EFLN = 5.0			
	A C	B S	C 10	D 20	E ?C	
1.00	*****	*****	*****	12.85	12.42	
2.00	*****	*****	*****	12.53	12.14	
3.00	*****	*****	*****	12.23	11.86	
4.00	*****	*****	*****	11.95	11.65	
5.00	*****	*****	*****	11.68	11.43	
6.00	*****	*****	26.59	11.37	11.13	
7.00	27.21	25.54	19.03	11.08	10.84	
8.00	19.90	19.54	17.29	10.79	10.58	
9.00	18.14	17.63	15.77	10.51	10.32	
10.00	16.65	15.03	14.43	10.24	10.08	
11.00	15.59	15.35	13.87	10.17	10.04	
12.00	15.36	14.72	13.36	10.11	9.99	
13.00	14.77	14.13	12.89	10.05	9.94	
14.00	14.22	13.59	12.46	9.99	9.89	
15.00	13.69	13.08	12.05	9.93	9.84	
16.00	13.20	12.60	11.68	9.87	9.79	
17.00	12.72	12.15	11.33	9.81	9.74	
18.00	12.29	11.73	11.00	9.75	9.69	
19.00	11.86	11.33	10.69	9.69	9.65	
20.00	11.46	10.96	10.41	9.63	9.60	
21.00	11.35	10.91	10.34	9.63	9.59	
22.00	11.25	10.85	10.27	9.62	9.59	
23.00	11.15	10.80	10.20	9.61	9.58	
24.00	11.06	10.75	10.13	9.61	9.58	
25.00	10.56	10.70	10.05	9.60	9.57	
26.00	10.47	10.65	9.98	9.60	9.57	
27.00	10.78	10.60	9.90	9.59	9.56	
28.00	10.70	10.55	9.83	9.59	9.56	
29.00	10.62	10.50	9.76	9.58	9.55	
30.00	10.54	10.45	9.69	9.57	9.55	

DELTA EFNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE, 10. JET ATT, NO TURB, EPNL = 101.83, RUN 5A

30.	+	....	-	E.C. +	....	-	....	+	....	-	....	+
.	.			E C	.			.				.
.	.			ECC	.			.				.
.	.			ECCA	.			.				.
.	.			ECB	.			.				.
25.	+			EDC	.			.				.
I	.			EDCR	.			.				.
N	.			ECBR	.			.				.
L	.			ECB	.			.				.
E	.			ECB	.			.				.
T	20.	+		ECB	.			.				.
A	.			E CB	.			.				.
T	.			E CPA	.			.				.
T	.			E CBA	.			.				.
E	15.	+		E C BA	.			.				.
N	.			ED C BA	.			.				.
U	.			ED C B	.			.				.
A	.			ED C BA	.			.				.
T	.			ED C BA	.			.				.
I	10.	+		ED C BA	.			.				.
C	.			ED C RA	.			.				.
N	.			ED C BA	.			.				.
.	.			ED C	.			.		A	B	
S.	0			E	.			.		C	B	
.	.			ED	.			.		C		
.	.			ED	.			.		C		
.	.			ED	.			.		C		
0.	0			ED	.			.				

C.

10.

20.

30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EFNL = 5.0		
	A	B	C	D	E
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	9.9999	9.9999	9.9999	11.23	10.28
2.00	9.9999	9.9999	9.9999	10.85	10.02
3.00	9.9999	9.9999	9.9999	10.49	9.87
4.00	9.9999	9.9999	9.9999	10.16	9.73
5.00	9.9999	9.9999	9.9999	9.90	9.59
6.00	9.9999	9.9999	27.68	9.75	9.38
7.00	27.18	30.00	18.84	9.59	9.15
8.00	19.26	19.23	16.72	9.43	9.00
9.00	17.26	16.95	14.84	9.28	8.82
10.00	15.48	15.00	13.18	9.13	8.65
11.00	14.68	14.19	12.53	9.03	8.61
12.00	13.93	13.46	11.96	8.94	8.57
13.00	13.24	12.80	11.44	8.85	8.52
14.00	12.59	12.19	10.97	8.76	8.49
15.00	11.95	11.64	10.54	8.68	8.45
16.00	11.43	11.13	10.15	8.60	8.41
17.00	10.90	10.66	9.84	8.52	8.37
18.00	10.40	10.22	9.56	8.44	8.33
19.00	9.95	9.87	9.32	8.36	8.29
20.00	9.66	9.58	9.07	8.29	8.26
21.00	9.61	9.53	9.05	8.29	8.25
22.00	9.56	9.48	9.02	8.28	8.25
23.00	9.51	9.43	9.00	8.28	8.25
24.00	9.46	9.38	8.98	8.27	8.24
25.00	9.41	9.33	8.96	8.27	8.24
26.00	9.35	9.28	8.94	8.26	8.24
27.00	9.30	9.23	8.92	8.26	8.22
28.00	9.25	9.18	8.89	8.25	8.23
29.00	9.20	9.13	8.87	8.25	8.23
30.00	9.14	9.08	8.85	8.24	8.22

DELTA EPNL = E.C FAN HARMONIC = 10.0  
 INLET HARMONIC (A = C, B = 5, C = 10, E = 20, E = 30)  
 T/O, WITH TCNE, 10. JET ATT, NO TURB, EPNL = 101.83, RUN 5A

30.	.....	ECA	.....	.....	.....	.....	.....
.	.	ECA	.	.	.	.	.
.	.	ECA	.	.	.	.	.
.	.	ECA	.	.	.	.	.
25.	+	ECA					
I	.	ECA					
N	.	ECA					
L	.	ECA					
E	.	EDEA					
T	20.	EDCA					
A	.	EDCR					
T	.	EC CA					
E	.	ED CB					
E	15.	E CPA					
N	.	E CB					
U	.	E CRA					
A	.	E CPA					
T	.	ED CRA					
I	10.	ED CB					
O	.	ED C R					
N	.	E C B					
.	.	ED C A B					
.	.	ED C					
5.	+	ED					
.	.	ED					
.	.	ED					
.	.	ED					
0.	.....	.....	.....	.....	.....	.....	.....

C. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 10.0		DELTA EPNL = 5.0		
	A	B	C	D	E
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	5.37	8.91
2.00	*****	*****	*****	5.23	8.74
3.00	*****	*****	*****	5.10	8.56
4.00	*****	*****	*****	5.07	8.38
5.00	*****	*****	*****	5.05	8.20
6.00	*****	*****	22.47	5.05	8.07
7.00	19.93	22.17	17.63	5.45	7.95
8.00	17.40	17.44	15.26	5.25	7.84
9.00	15.19	14.96	13.31	5.05	7.73
10.00	13.22	12.84	11.67	5.86	7.63
11.00	12.34	11.95	10.93	5.80	7.57
12.00	11.55	11.24	10.29	7.73	7.51
12.00	10.84	10.57	9.83	7.68	7.45
14.00	10.20	9.98	9.45	7.62	7.40
15.00	9.77	9.63	9.17	7.56	7.34
16.00	9.44	9.29	8.87	7.50	7.29
17.00	9.14	8.96	8.57	7.45	7.24
18.00	8.46	8.64	8.29	7.40	7.19
19.00	8.59	8.34	8.02	7.34	7.14
20.00	8.34	8.04	7.75	7.29	7.09
21.00	8.28	8.01	7.73	7.29	7.09
22.00	8.23	7.98	7.70	7.29	7.09
23.00	8.17	7.95	7.66	7.29	7.08
24.00	8.11	7.91	7.65	7.29	7.08
25.00	8.05	7.88	7.63	7.29	7.08
26.00	8.00	7.85	7.61	7.29	7.08
27.00	7.94	7.82	7.58	7.29	7.08
28.00	7.88	7.79	7.56	7.29	7.08
29.00	7.82	7.76	7.53	7.29	7.08
30.00	7.76	7.73	7.51	7.29	7.08

DELTA EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 1/D, WITH TCNE, 10° JET ATT, NO TURB, EPNL = 101.83, RUN 5A

20.	C.B	E
	C.B	E
	C.B	E
	.CA	E
25.	+CHA	E
I	.CHA	E
N	.CHA	E
L	.CHA	E
E	.C.B	E
T	2C.	E
A	+C EA	E
T	.C B	E
T	.C PA	E
E	.C BA	E
K	15. +C BA	E
U	C EA	E
A	C BA	E
T	C EA	E
D	C BA	E
T	10. +C BA	E
O	EC	E
N	.ED	E
	.ED	E
	.ED	E
5.	+E C	E
	.ED	E
C.	+.....	E

C. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0	DELTA EPNL = 5.0				
	A	B	C	D	E
INLET HARMONIC = 0	5	10	20	30	
1.00	*****	*****	*****	3.54	2.74
2.00	*****	*****	*****	3.21	2.31
3.00	*****	*****	*****	2.90	1.92
4.00	*****	*****	*****	2.60	1.56
5.00	*****	*****	28.46	2.32	1.24
6.00	24.41	*****	15.57	1.87	0.93
7.00	10.39	16.60	12.10	1.45	0.63
8.00	12.57	12.64	9.29	1.05	0.33
9.00	10.11	5.51	7.15	C.66	0.03
10.00	7.67	7.12	5.24	C.29	*****
11.00	6.48	6.24	4.56	0.22	*****
12.00	6.01	5.46	3.97	C.15	*****
13.00	5.25	4.80	3.42	C.08	*****
14.00	4.63	4.23	2.91	C.01	*****
15.00	4.09	3.69	2.44	*****	*****
16.00	3.57	3.17	1.99	*****	*****
17.00	3.08	2.68	1.56	*****	*****
18.00	2.62	2.21	1.13	*****	*****
19.00	2.18	1.76	C.82	*****	*****
20.00	1.76	1.33	C.47	*****	*****
21.00	1.66	1.27	C.43	*****	*****
22.00	1.56	1.22	C.40	*****	*****
23.00	1.46	1.16	C.36	*****	*****
24.00	1.37	1.11	C.33	*****	*****
25.00	1.27	1.06	C.29	*****	*****
26.00	1.18	1.00	C.25	*****	*****
27.00	1.09	0.95	C.22	*****	*****
28.00	0.99	0.90	C.18	*****	*****
29.00	0.90	0.85	C.14	*****	*****
30.00	0.81	0.80	C.11	*****	*****

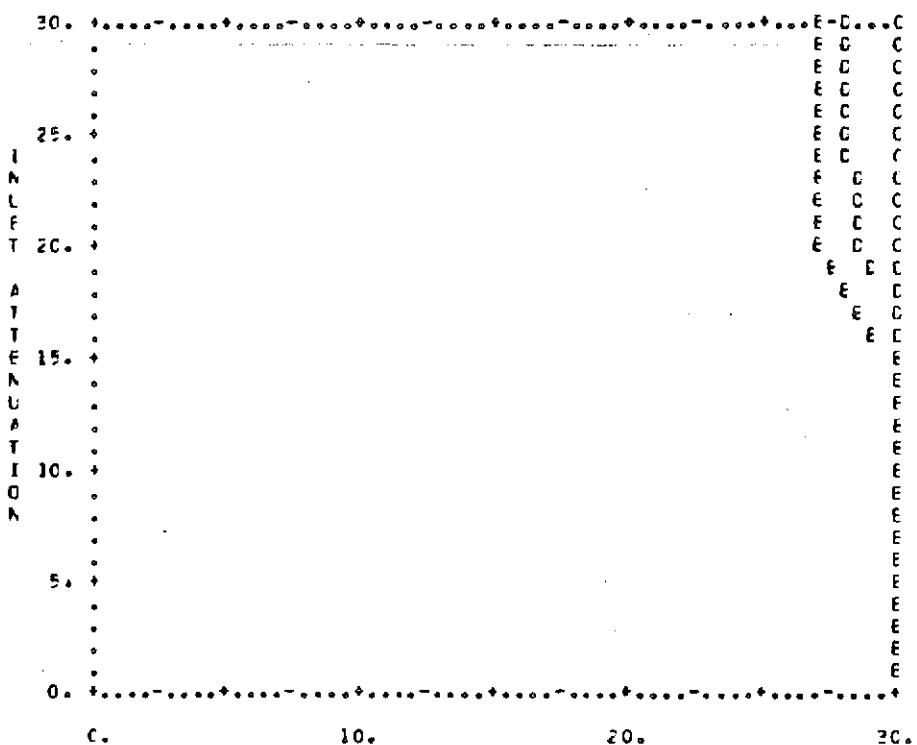
DELTAEFNL = 5.0 . FANHARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 1/C, WITH TCNE, 10. JET ATT, NO TURE, EPNL = 101.82, RUN 5A

C. 10. 20. 30.

## FAN DUCT ATTENATION

FAN HARMONIC = 30.0		DELTA EPNL = 5.0			
INLET HARMONIC =	A	B	C	C	E
	0	5	10	20	30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	4.52	3.79
2.00	*****	*****	*****	4.00	3.09
3.00	*****	*****	*****	2.51	2.45
4.00	*****	*****	*****	3.04	1.85
5.00	*****	*****	18.57	2.61	1.20
6.00	19.22	20.30	13.72	1.95	0.55
7.00	14.67	14.76	10.00	1.33	*****
8.00	10.92	10.59	8.63	0.73	*****
9.00	9.01	8.70	7.43	0.10	*****
10.00	7.77	7.34	6.35	*****	*****
11.00	7.21	6.81	5.86	*****	*****
12.00	6.71	6.32	5.42	*****	*****
13.00	6.27	5.89	5.03	*****	*****
14.00	5.87	5.49	4.13	*****	*****
15.00	5.51	5.13	3.24	*****	*****
16.00	5.18	4.46	2.42	*****	*****
17.00	4.69	3.64	1.66	*****	*****
18.00	3.92	2.85	0.96	*****	*****
19.00	3.16	2.09	0.30	*****	*****
20.00	2.43	1.35	*****	*****	*****
21.00	2.18	1.18	*****	*****	*****
22.00	1.93	1.02	*****	*****	*****
23.00	1.68	0.86	*****	*****	*****
24.00	1.43	0.70	*****	*****	*****
25.00	1.19	0.54	*****	*****	*****
26.00	0.94	0.38	*****	*****	*****
27.00	0.70	0.22	*****	*****	*****
28.00	0.47	0.06	*****	*****	*****
29.00	0.23	*****	*****	*****	*****
30.00	0.00	*****	*****	*****	*****

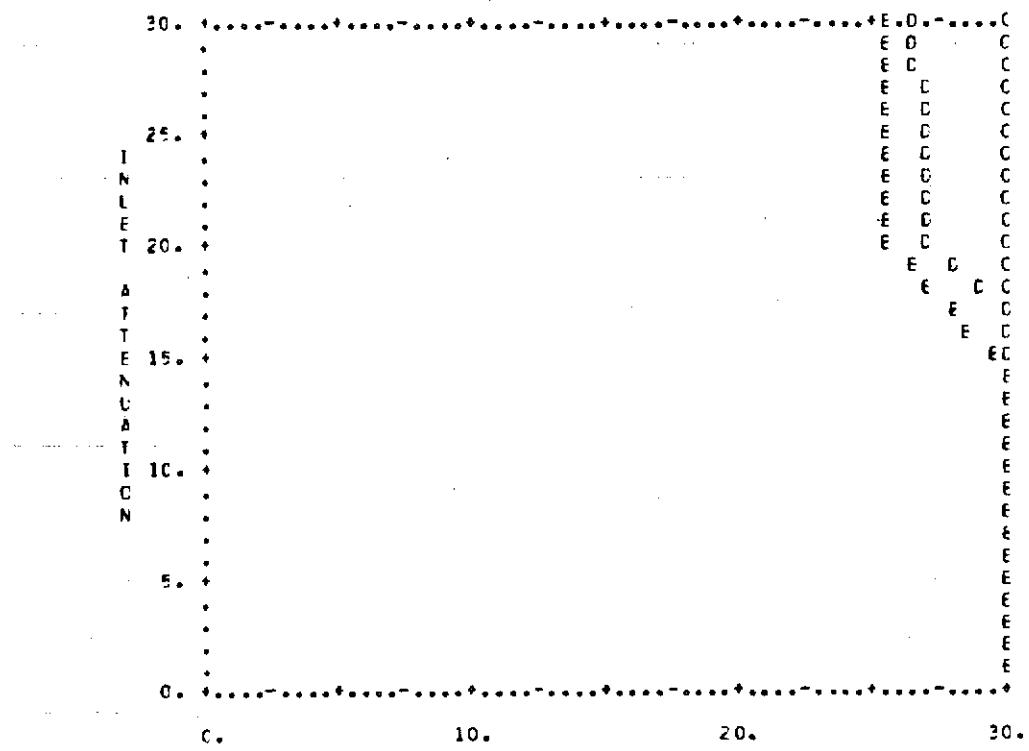
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 1/0, WITH TCNE, 10. JET ATT, NO TURE, EPNL = 101.83, RUN 5A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0	DELTA EPNL = 10.0				
	A C	B S	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	30.45
15.00	*****	*****	*****	*****	29.84
16.00	*****	*****	*****	*****	29.23
17.00	*****	*****	*****	30.61	28.65
18.00	*****	*****	*****	29.80	28.10
19.00	*****	*****	*****	29.03	27.56
20.00	*****	*****	*****	28.31	27.06
21.00	*****	*****	*****	28.29	27.06
22.00	*****	*****	*****	28.28	27.06
23.00	*****	*****	*****	28.26	27.06
24.00	*****	*****	*****	28.24	27.06
25.00	*****	*****	*****	28.23	27.06
26.00	*****	*****	*****	28.21	27.06
27.00	*****	*****	*****	28.20	27.06
28.00	*****	*****	*****	28.18	27.06
29.00	*****	*****	*****	28.16	27.06
30.00	*****	*****	*****	28.15	27.06

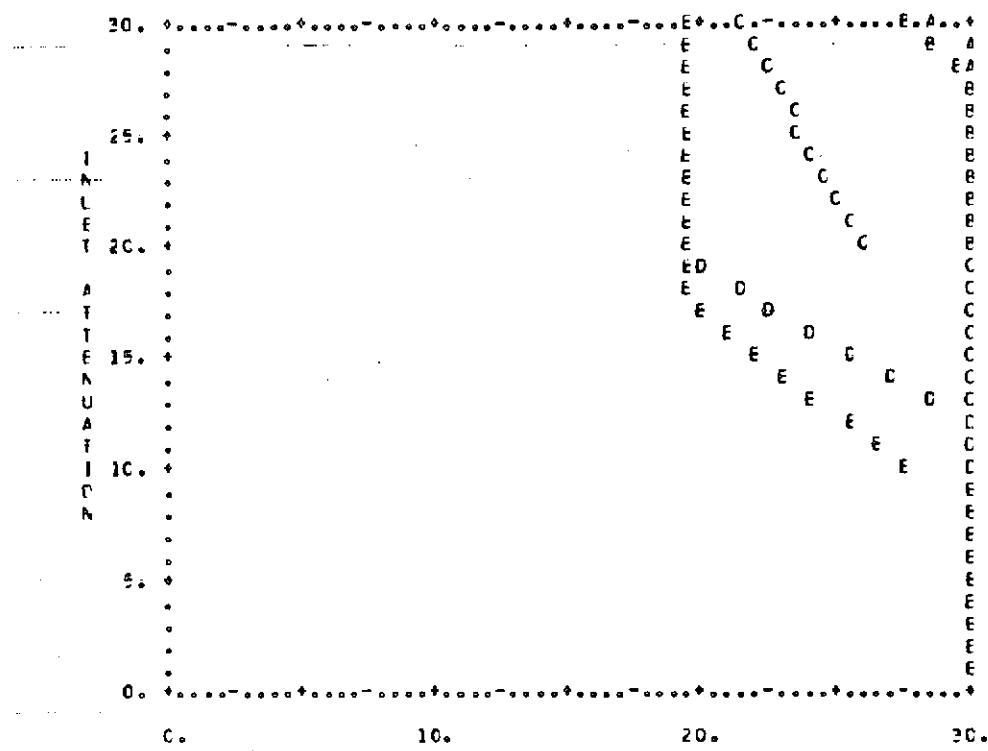
DELTA EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = C, B = S, C = 1C, D = 20, E = 30)  
 T/D, WITH TCNE, 10. JET ATT, NO TLRE, EPNL = 101.83, RUN 5A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EPNL = 10.0		
	A C	B S	C 1C	D 20	E 30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
2.0C	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.0C	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.0C	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.0C	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	30.4E
15.00	*****	*****	*****	*****	29.56
16.0C	*****	*****	*****	*****	28.68
17.00	*****	*****	*****	30.19	27.84
18.0C	*****	*****	*****	29.10	27.04
19.00	*****	*****	*****	28.07	26.27
20.0C	*****	*****	*****	27.09	25.53
21.0C	*****	*****	*****	27.05	25.53
22.00	*****	*****	*****	27.00	25.52
23.0C	*****	*****	*****	26.95	25.52
24.00	*****	*****	*****	26.91	25.52
25.0C	*****	*****	*****	26.86	25.52
26.00	*****	*****	*****	26.82	25.52
27.0C	*****	*****	30.90	26.77	25.52
28.00	*****	*****	30.56	26.73	25.52
29.00	*****	*****	30.23	26.68	25.52
30.0G	*****	*****	29.90	26.64	25.52

DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 1C, D = 20, E = 30)  
 T/O, WITH TCNE, 10. JET ATT, NC TLRB, EPNL = 101.83, RUN 5A

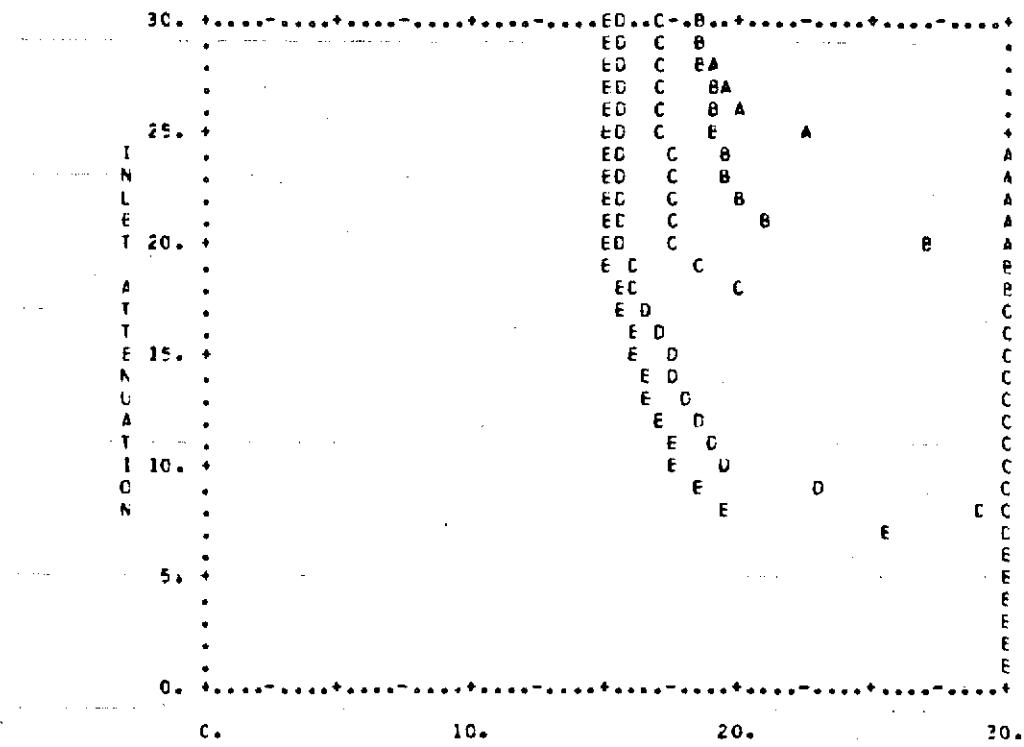


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0			
INLET HARMONIC =		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	***	*****	*****	*****	*****
7.00	***	*****	*****	*****	*****
8.00	***	*****	*****	*****	*****
9.00	***	*****	*****	*****	30.65
10.00	***	*****	*****	*****	27.71
11.00	***	*****	*****	*****	26.52
12.00	***	*****	*****	30.22	25.36
13.00	***	*****	*****	28.58	24.23
14.00	***	*****	*****	27.00	23.13
15.00	***	*****	*****	25.48	22.05
16.00	***	*****	*****	24.03	20.99
17.00	***	*****	*****	22.62	19.59
18.00	***	*****	*****	21.27	19.74
19.00	***	*****	30.48	19.59	19.49
20.00	***	*****	26.06	19.71	19.26
21.00	***	*****	25.57	19.69	19.26
22.00	***	*****	25.10	19.68	19.25
23.00	***	*****	24.63	19.66	19.25
24.00	***	*****	24.18	19.64	19.25
25.00	***	*****	23.73	19.62	19.25
26.00	***	*****	23.30	19.61	19.25
27.00	***	30.17	22.87	19.59	19.25
28.00	***	29.26	22.45	19.57	19.25
29.00	29.77	28.38	22.05	19.55	19.25
30.00	28.26	27.53	21.65	19.54	19.25

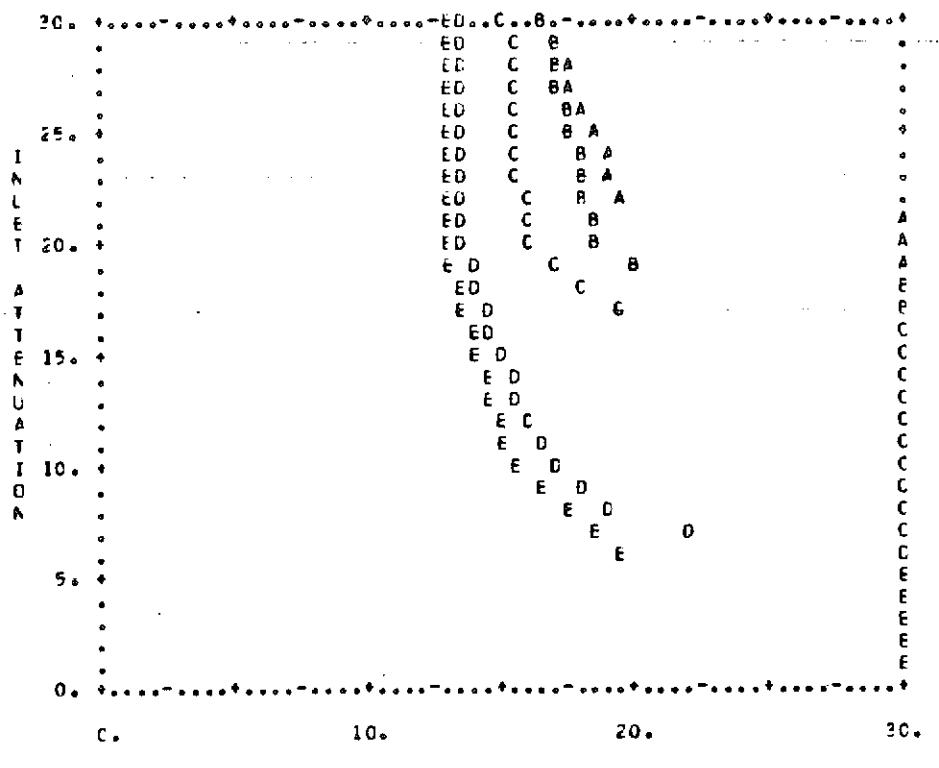
DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TCNE, 10% JET ATT, NO TURB, EPNL = 101.83, RUN 5A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 20.0		DELTA EPNL = 10.0		
	A C	B 5	C 10	D 20	E 30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	25.45
8.00	*****	*****	*****	29.23	19.54
9.00	*****	*****	*****	22.48	18.51
10.00	*****	*****	*****	14.50	17.58
11.00	*****	*****	*****	19.01	17.26
12.00	*****	*****	*****	18.55	16.96
13.00	*****	*****	*****	18.11	16.66
14.00	*****	*****	*****	17.69	16.38
15.00	*****	*****	*****	17.29	16.11
16.00	*****	*****	*****	16.91	15.85
17.00	*****	*****	*****	16.55	15.59
18.00	*****	*****	19.89	16.20	15.35
19.00	*****	*****	18.72	15.87	15.11
20.00	*****	27.00	17.66	15.55	14.88
21.00	*****	21.10	17.57	15.54	14.88
22.00	*****	19.84	17.47	15.52	14.88
23.00	*****	19.63	17.38	15.51	14.88
24.00	*****	19.43	17.29	15.50	14.88
25.00	22.65	19.23	17.20	15.48	14.87
26.00	19.75	19.04	17.11	15.47	14.87
27.00	19.39	18.85	17.02	15.46	14.87
28.00	19.05	18.67	16.93	15.44	14.87
29.00	18.72	18.49	16.84	15.43	14.87
30.00	18.46	18.32	16.75	15.42	14.87

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE, 10% JET ATT, NO TURB, EPNL = 101.83, RUN 5A



C. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC =	FAN HARMONIC = 30.0		DELTA EPNL = 10.0		
	A C	B 5	C 10	D 20	E 30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	15.54
7.00	*****	*****	*****	22.03	18.36
8.00	*****	*****	*****	18.98	17.26
9.00	*****	*****	*****	17.87	16.34
10.00	*****	*****	*****	16.87	15.51
11.00	*****	*****	*****	16.44	15.20
12.00	*****	*****	*****	16.05	14.90
13.00	*****	*****	*****	15.68	14.62
14.00	*****	*****	*****	15.33	14.35
15.00	*****	*****	*****	15.01	14.08
16.00	*****	*****	*****	14.71	13.83
17.00	*****	*****	15.50	14.42	13.59
18.00	*****	*****	18.19	14.15	13.26
19.00	*****	19.79	16.99	13.90	13.13
20.00	*****	18.56	15.91	13.66	12.92
21.00	*****	18.35	15.83	13.64	12.91
22.00	19.65	18.15	15.76	13.61	12.91
23.00	19.21	17.96	15.68	13.58	12.91
24.00	18.75	17.77	15.61	13.56	12.90
25.00	18.39	17.59	15.54	13.53	12.90
26.00	18.02	17.41	15.47	13.50	12.89
27.00	17.67	17.24	15.40	13.48	12.89
28.00	17.34	17.06	15.33	13.45	12.89
29.00	17.03	16.90	15.26	13.42	12.88
30.00	16.73	16.73	15.20	13.40	12.88

OFLTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, O, JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1

30.		FCB						30.	
I	25.	.	ECH	.	.	.	.	.	.
N	.	.	EBC	.	.	.	.	.	.
L	.	.	ECH	.	.	.	.	.	.
F	.	.	ECBA	.	.	.	.	.	.
T	20.	+	E CA	.	.	.	.	.	.
E	.	.	E CA	.	.	.	.	.	.
N	.	.	E CB	.	.	.	.	.	.
U	.	.	E CRA	.	.	.	.	.	.
A	.	.	E CBA	.	.	.	.	.	.
T	.	.	ED CRA	.	.	.	.	.	.
E	15.	+	FD CRA	.	.	.	.	.	.
N	.	.	ED C BA	.	.	.	.	.	.
U	.	.	E C BA	.	.	.	.	.	.
A	.	.	ED C BA	.	.	.	.	.	.
T	.	.	ED C RA	.	.	.	.	.	.
I	10.	+	ED C RA	.	.	.	.	.	.
O	.	.	ED C BA	.	.	.	.	.	.
N	.	.	ED C BA	.	.	.	.	.	.
-5.		.	ED C BA	E	B	A	.	.	.
.	.	.	FD	C	C	C	.	.	.
.	.	.	ED	C	C	C	.	.	.
.	.	.	ED	C	C	C	.	.	.
0.		.	FD	C	C	C	.	.	.

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 0.0			DELT A EPNL = 5.0			F
	A	B	C	D	E	F	
	0	5	10	20	30		
1.00	*****	*****	*****	12.31	11.66		
2.00	*****	*****	*****	12.02	11.41		
3.00	*****	*****	*****	11.74	11.19		
4.00	*****	*****	*****	11.47	10.99		
5.00	*****	*****	*****	11.21	10.81		
6.00	*****	*****	20.73	10.98	10.54		
7.00	24.94	22.83	18.73	10.76	10.28		
8.00	19.27	18.63	16.57	10.57	10.03		
9.00	17.59	16.91	15.12	10.38	9.78		
10.00	16.10	15.42	13.85	10.21	9.53		
11.00	15.38	14.71	13.27	10.10	9.49		
12.00	14.72	14.07	12.75	9.99	9.44		
13.00	14.11	13.48	12.28	9.87	9.39		
14.00	13.54	12.93	11.85	9.74	9.34		
15.00	13.01	12.43	11.46	9.62	9.29		
16.00	12.52	11.97	11.09	9.51	9.24		
17.00	12.07	11.53	10.76	9.40	9.20		
18.00	11.64	11.13	10.45	9.29	9.15		
19.00	11.24	10.76	10.17	9.19	9.11		
20.00	10.86	10.41	9.87	9.09	9.06		
21.00	10.55	10.35	9.85	9.08	9.06		
22.00	10.25	10.29	9.82	9.08	9.05		
23.00	10.56	10.24	9.80	9.07	9.05		
24.00	10.46	10.19	9.78	9.07	9.04		
25.00	10.37	10.13	9.75	9.06	9.04		
26.00	10.28	10.08	9.73	9.06	9.03		
27.00	10.19	10.03	9.70	9.05	9.03		
28.00	10.11	9.97	9.68	9.05	9.02		
29.00	10.03	9.90	9.65	9.04	9.02		
30.00	9.94	9.84	9.63	9.04	9.01		

DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, O. JET ATT, NO TURA, EPNL = 101.70, RUN 10A NO JET1

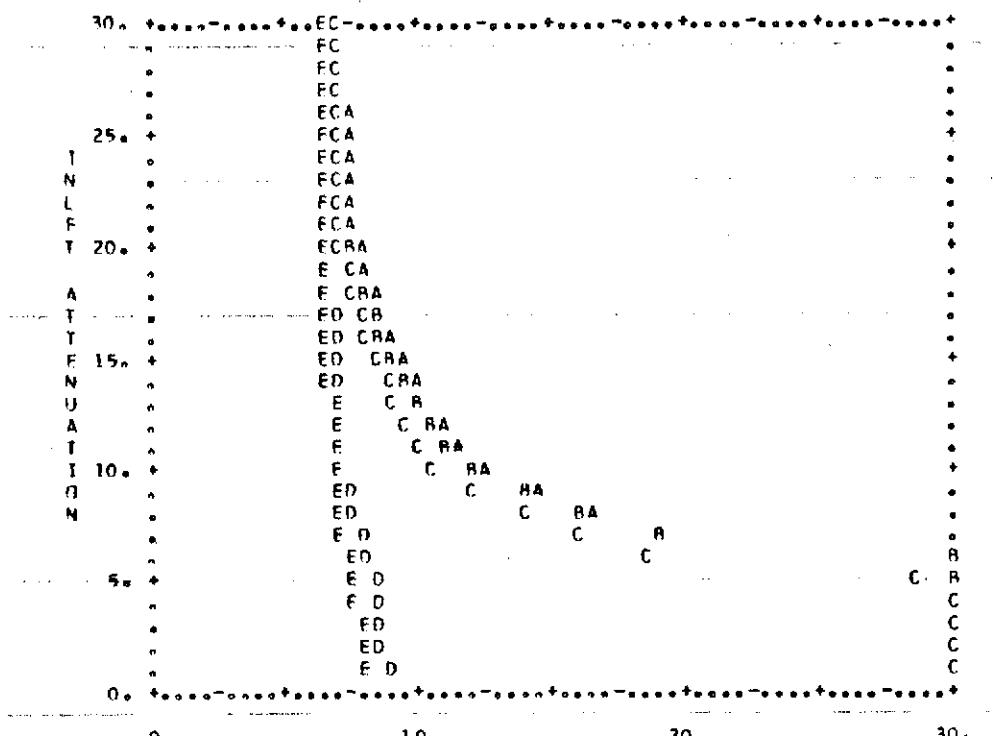
	10.	20.	30.
I	ECB		
N	ECB		
L	ECB		
E	ECB		
T	ECB		
A	ED CBA		
T	E C B		
E	E CBA		
N	E C BA		
U	ED C BA		
A	ED C BA		
T	ED C BA		
I	ED C BA		
O	ED C BA		
N	FD C BA		
	FD C B A		
	ED C		
5.	ED		
	ED		
0.	+	+	+

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EPNL = 5.0		F
	A	B	C	D	
1.00	*****	*****	*****	10.46	9.69
2.00	*****	*****	*****	10.12	9.52
3.00	*****	*****	*****	9.86	9.36
4.00	*****	*****	*****	9.65	9.20
5.00	*****	*****	*****	9.45	9.05
6.00	*****	*****	10.56	9.27	8.93
7.00	23.71	21.05	17.30	9.09	8.63
8.00	18.52	17.83	15.41	8.91	8.43
9.00	16.53	15.82	13.79	8.72	8.24
10.00	14.78	14.09	12.40	8.54	8.05
11.00	13.93	13.28	11.78	8.44	8.01
12.00	13.16	12.55	11.22	8.34	7.97
13.00	12.47	11.89	10.72	8.25	7.93
14.00	11.83	11.28	10.26	8.16	7.89
15.00	11.24	10.73	9.87	8.08	7.85
16.00	10.70	10.23	9.53	7.99	7.81
17.00	10.21	9.81	9.21	7.91	7.77
18.00	9.80	9.46	8.89	7.84	7.73
19.00	9.47	9.13	8.59	7.76	7.70
20.00	9.15	8.81	8.30	7.69	7.66
21.00	9.06	8.75	8.28	7.69	7.66
22.00	8.97	8.70	8.25	7.69	7.65
23.00	8.89	8.64	8.23	7.69	7.65
24.00	8.80	8.58	8.21	7.68	7.64
25.00	8.71	8.52	8.18	7.68	7.64
26.00	8.63	8.47	8.16	7.68	7.63
27.00	8.54	8.41	8.14	7.68	7.63
28.00	8.45	8.35	8.12	7.68	7.63
29.00	8.37	8.29	8.09	7.67	7.62
30.00	8.28	8.24	8.07	7.67	7.62

DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, D, JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1

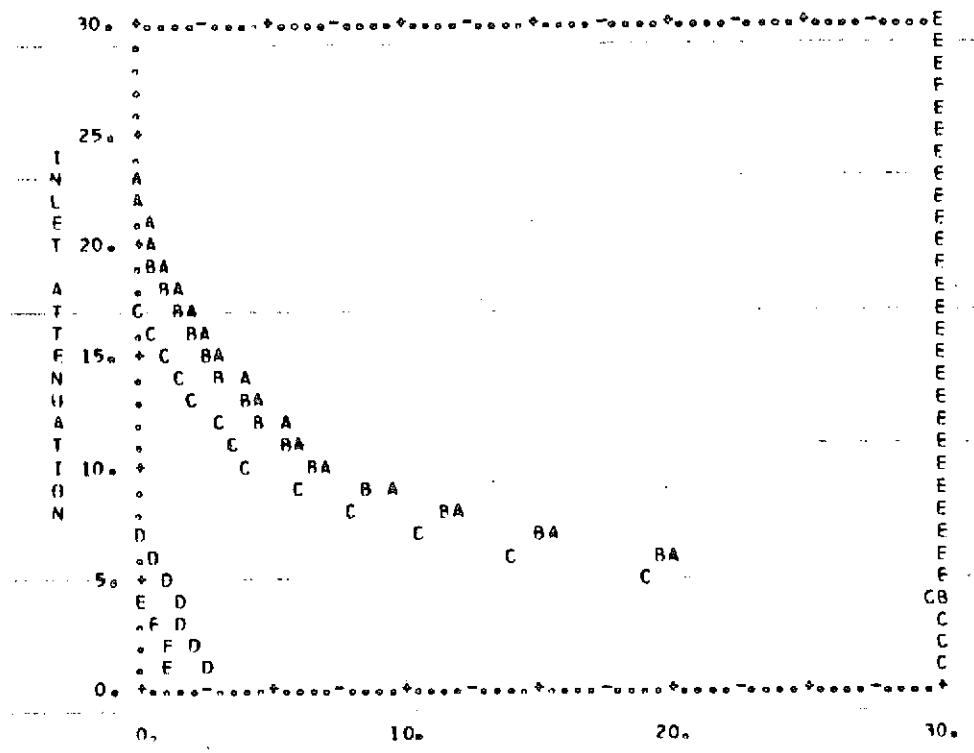


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 5.0			
INLET HARMONIC	FUND.	A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	8.89	8.22
2.00	*****	*****	*****	9.73	8.03
3.00	*****	*****	*****	8.57	7.84
4.00	*****	*****	*****	8.42	7.64
5.00	*****	*****	28.75	8.27	7.45
6.00	*****	*****	18.68	8.04	7.34
7.00	19.22	18.76	15.98	7.81	7.23
8.00	16.68	16.12	13.82	7.58	7.13
9.00	14.50	13.92	12.05	7.36	7.03
10.00	12.59	12.07	10.58	7.15	6.94
11.00	11.70	11.72	9.94	7.09	6.89
12.00	10.92	10.47	9.54	7.03	6.83
13.00	10.24	9.87	9.17	6.97	6.78
14.00	9.75	9.44	8.91	6.91	6.73
15.00	9.37	9.03	8.47	6.86	6.68
16.00	9.02	8.65	8.14	6.81	6.64
17.00	8.68	8.27	7.83	6.75	6.59
18.00	8.37	7.92	7.54	6.70	6.55
19.00	8.08	7.58	7.26	6.65	6.50
20.00	7.80	7.25	6.99	6.61	6.46
21.00	7.52	7.22	6.96	6.60	6.46
22.00	7.23	7.19	6.94	6.60	6.46
23.00	7.05	7.16	6.91	6.60	6.45
24.00	7.47	7.13	6.89	6.59	6.45
25.00	7.38	7.10	6.87	6.59	6.45
26.00	7.30	7.07	6.84	6.59	6.44
27.00	7.22	7.04	6.82	6.58	6.44
28.00	7.13	7.01	6.80	6.58	6.44
29.00	7.05	6.98	6.77	6.57	6.44
30.00	6.97	6.95	6.75	6.57	6.43

DELTA EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0° JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JETI

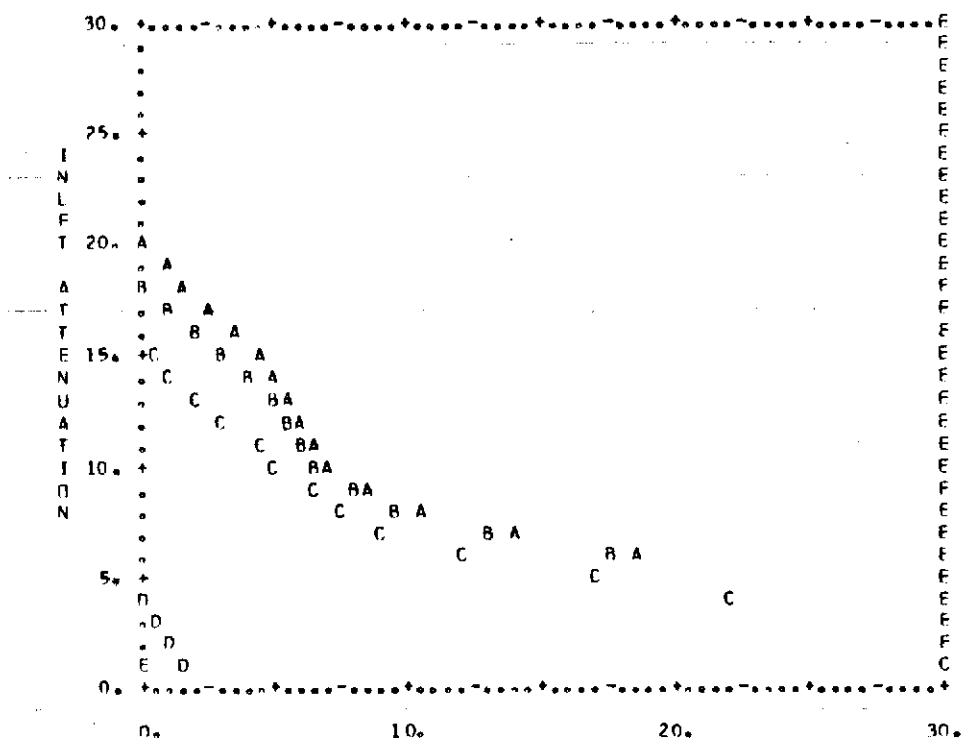


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 5.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	2.30	1.15
2.00	*****	*****	*****	*****	1.99	0.76
3.00	*****	*****	*****	*****	1.69	0.39
4.00	*****	*****	29.65	1.37	0.05	
5.00	*****	*****	19.06	1.06	*****	
6.00	19.78	1.9.29	14.02	0.61	*****	
7.00	15.58	14.79	10.42	0.19	*****	
8.00	12.14	11.25	7.97	*****	*****	
9.00	9.35	8.57	5.87	*****	*****	
10.00	7.12	6.39	4.12	*****	*****	
11.00	6.13	5.44	3.44	*****	*****	
12.00	5.28	4.63	2.81	*****	*****	
13.00	4.55	3.90	2.21	*****	*****	
14.00	3.88	3.21	1.65	*****	*****	
15.00	3.24	2.56	1.12	*****	*****	
16.00	2.63	1.94	0.63	*****	*****	
17.00	2.05	1.35	0.15	*****	*****	
18.00	1.50	0.79	*****	*****	*****	
19.00	0.97	0.26	*****	*****	*****	
20.00	0.47	*****	*****	*****	*****	
21.00	0.34	*****	*****	*****	*****	
22.00	0.21	*****	*****	*****	*****	
23.00	0.08	*****	*****	*****	*****	
24.00	*****	*****	*****	*****	*****	
25.00	*****	*****	*****	*****	*****	
26.00	*****	*****	*****	*****	*****	
27.00	*****	*****	*****	*****	*****	
28.00	*****	*****	*****	*****	*****	
29.00	*****	*****	*****	*****	*****	
30.00	*****	*****	*****	*****	*****	

DELTA EPNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0° JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1

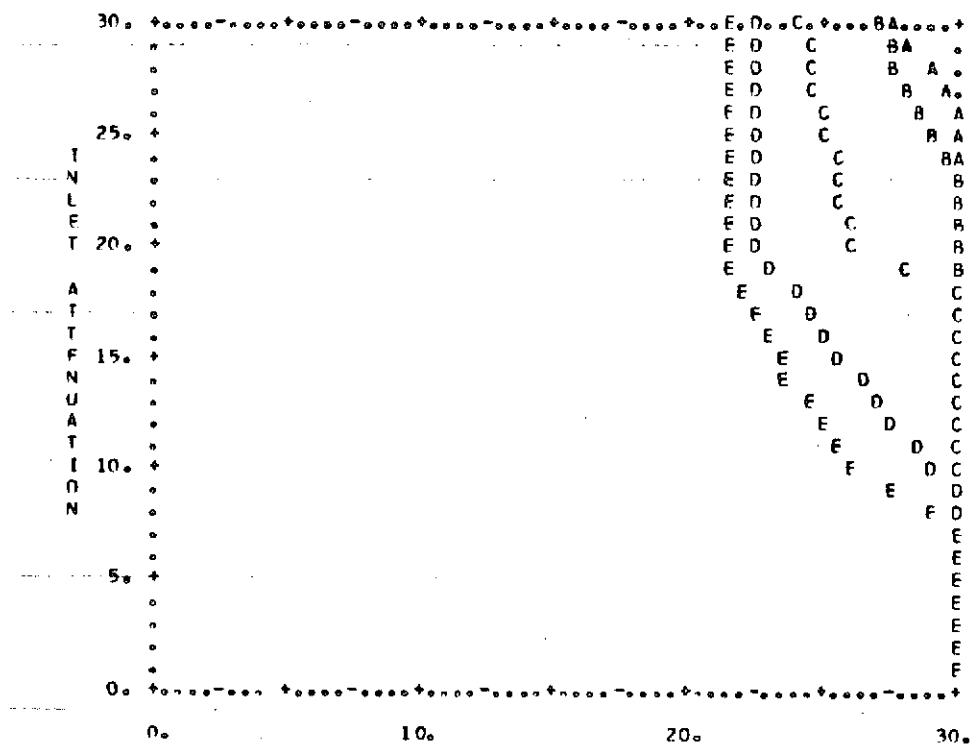


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 30.0		DELTA EPNL = 5.0		
INLET HARMONIC =	A 0	B 5	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	1.60	0.10
2.00	*****	*****	*****	1.10	*****
3.00	*****	*****	*****	0.64	*****
4.00	*****	*****	22.09	0.20	*****
5.00	*****	*****	17.18	*****	*****
6.00	18.42	17.66	11.88	*****	*****
7.00	14.01	13.05	9.09	*****	*****
8.00	10.34	9.66	7.59	*****	*****
9.00	8.51	8.08	6.30	*****	*****
10.00	7.25	6.72	5.18	*****	*****
11.00	6.63	6.10	4.25	*****	*****
12.00	6.08	5.55	3.16	*****	*****
13.00	5.59	5.06	2.16	*****	*****
14.00	5.14	4.03	1.24	*****	*****
15.00	4.32	2.98	0.40	*****	*****
16.00	3.35	1.99	*****	*****	*****
17.00	2.44	1.07	*****	*****	*****
18.00	1.58	0.20	*****	*****	*****
19.00	0.77	*****	*****	*****	*****
20.00	0.00	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

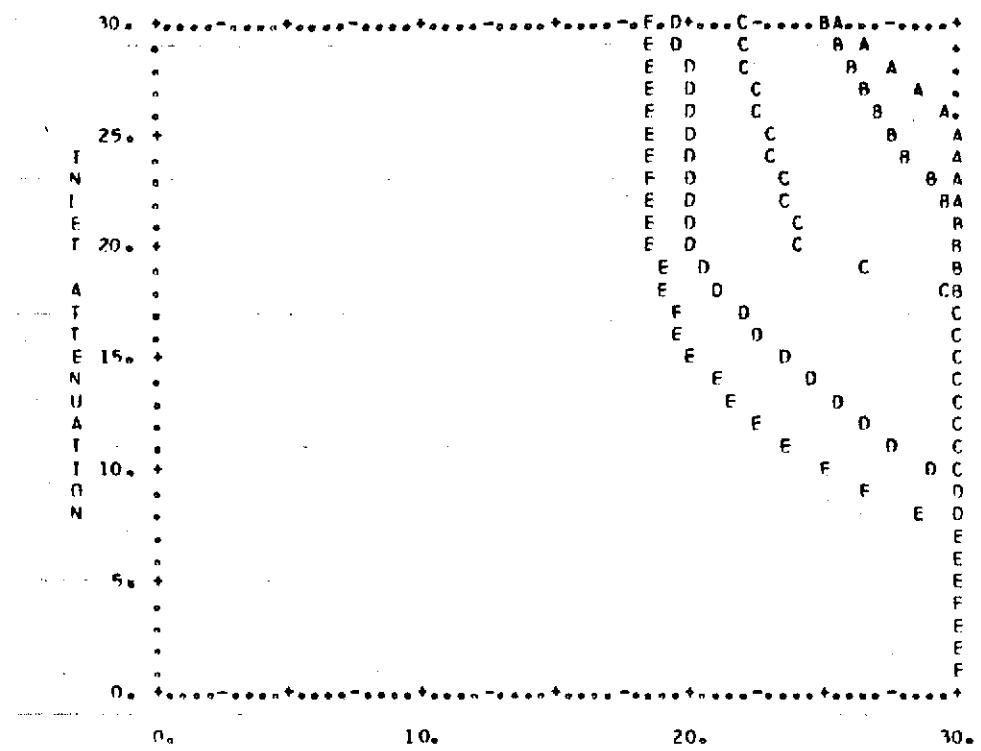
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, D, JET ATT, NO TURB, EPNL = 101.70, RUN 10A ND JETT



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A 0	B 5	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	30.94
8.00	*****	*****	*****	*****	*****	29.18
9.00	*****	*****	*****	*****	*****	27.59
10.00	*****	*****	*****	29.25	26.14	
11.00	*****	*****	23.44	25.45		
12.00	*****	*****	27.67	24.82		
13.00	*****	*****	26.94	24.25		
14.00	*****	*****	26.25	23.74		
15.00	*****	*****	25.59	23.26		
16.00	*****	*****	24.96	22.83		
17.00	*****	*****	24.36	22.43		
18.00	*****	30.54	23.78	22.06		
19.00	*****	28.22	23.23	21.72		
20.00	*****	26.11	22.70	21.40		
21.00	*****	25.90	22.67	21.40		
22.00	*****	25.68	22.64	21.40		
23.00	*****	25.47	22.61	21.40		
24.00	*****	25.27	22.58	21.40		
25.00	*****	25.07	22.55	21.40		
26.00	30.55	25.87	22.52	21.40		
27.00	29.73	28.17	24.67	22.49	21.40	
28.00	28.94	27.73	24.48	22.47	21.40	
29.00	28.18	27.30	24.29	22.46	21.40	
30.00	27.45	26.89	24.10	22.41	21.40	

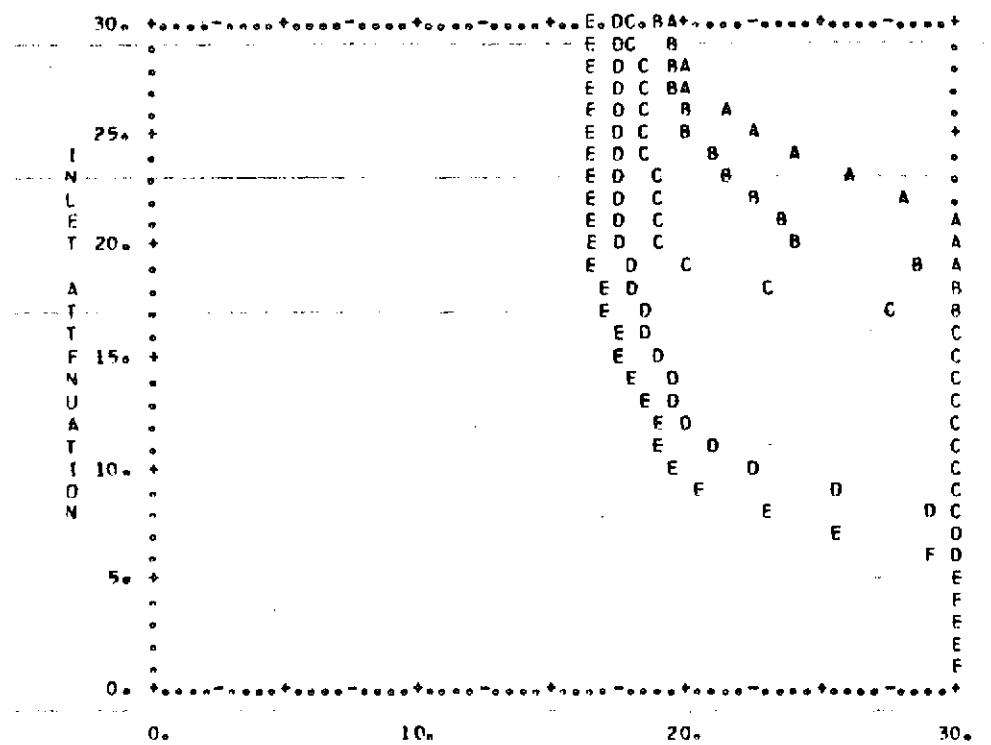
DELTA EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, O, JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1



#### FAN DUCT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 10.0			
INLET HARMONIC =		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	30.97
8.00	*****	*****	*****	*****	28.73
9.00	*****	*****	*****	*****	26.71
10.00	*****	*****	*****	28.80	24.88
11.00	*****	*****	*****	27.58	23.72
12.00	*****	*****	*****	26.46	22.67
13.00	*****	*****	*****	25.41	21.71
14.00	*****	*****	*****	24.44	20.82
15.00	*****	*****	*****	23.54	20.00
16.00	*****	*****	*****	22.69	19.71
17.00	*****	*****	*****	21.90	19.43
18.00	*****	*****	29.44	21.16	19.16
19.00	*****	*****	26.66	20.46	18.91
20.00	*****	30.88	24.13	19.92	18.67
21.00	*****	30.16	23.87	19.40	18.67
22.00	*****	29.47	23.61	19.88	18.67
23.00	*****	28.82	23.36	19.86	18.66
24.00	*****	28.18	23.12	19.84	18.66
25.00	30.64	27.58	22.88	19.82	18.66
26.00	29.47	26.99	22.65	19.80	18.66
27.00	28.37	26.44	22.42	19.79	18.66
28.00	27.35	25.90	22.20	19.77	18.66
29.00	26.40	25.38	21.98	19.75	18.66
30.00	25.50	24.88	21.76	19.73	18.66

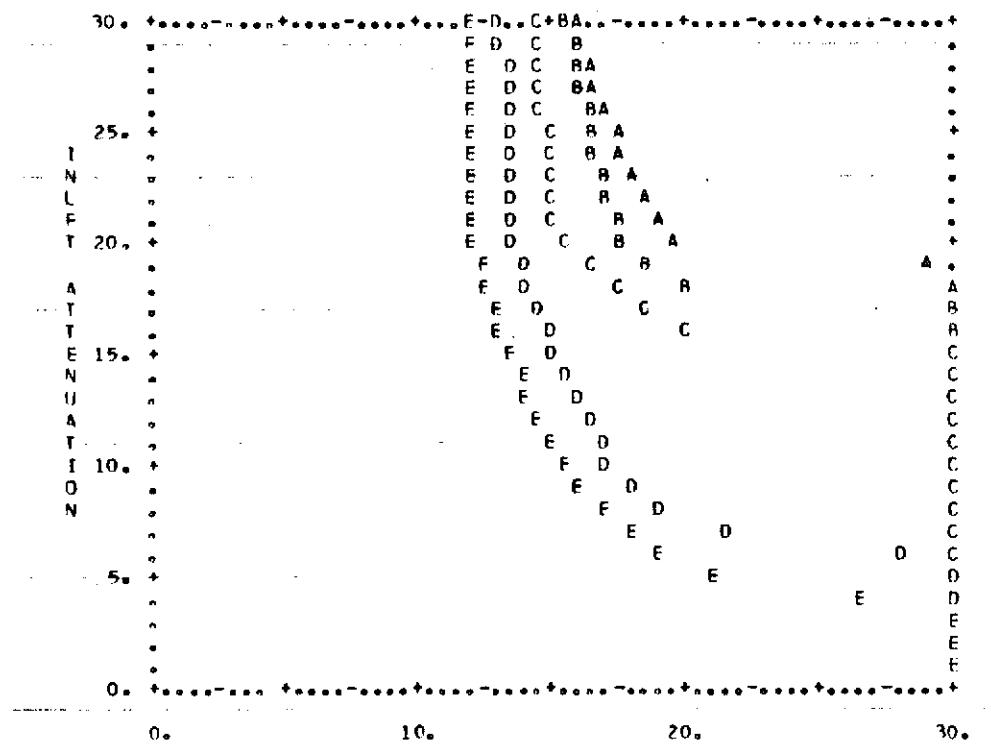
DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0% JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1



#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0			
INLET HARMONIC =		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	28.89
7.00	*****	*****	*****	*****	25.66
8.00	*****	*****	*****	28.92	22.92
9.00	*****	*****	*****	25.55	20.56
10.00	*****	*****	*****	22.60	19.55
11.00	*****	*****	*****	21.22	19.14
12.00	*****	*****	*****	19.97	18.76
13.00	*****	*****	*****	19.63	18.40
14.00	*****	*****	*****	19.30	18.06
15.00	*****	*****	*****	18.98	17.73
16.00	*****	*****	*****	18.68	17.43
17.00	*****	*****	27.25	18.40	17.14
18.00	*****	*****	23.07	18.12	16.86
19.00	*****	28.66	19.84	17.96	16.60
20.00	*****	24.21	19.03	17.60	16.35
21.00	30.21	23.29	18.94	17.58	16.35
22.00	27.93	22.42	18.85	17.56	16.35
23.00	25.95	21.59	18.76	17.53	16.35
24.00	24.22	20.81	18.67	17.51	16.35
25.00	22.68	20.06	18.58	17.49	16.35
26.00	21.32	19.85	18.50	17.47	16.35
27.00	20.09	19.68	18.41	17.44	16.35
28.00	19.78	19.52	18.33	17.42	16.35
29.00	19.55	19.36	18.24	17.40	16.35
30.00	19.33	19.21	18.16	17.38	16.34

DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 0. JET ATT, NO TURB, EPNL = 101.70, RUN 10A NO JET1



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 20.0		DELTA EPNL = 10.0		E
	A	B	C	D	
0.00	*****	*****	*****	*****	*****
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	26.42
5.00	*****	*****	*****	*****	20.85
6.00	*****	*****	*****	24.13	18.96
7.00	*****	*****	*****	21.70	17.91
8.00	*****	*****	*****	19.20	16.98
9.00	*****	*****	*****	18.17	16.16
10.00	*****	*****	*****	17.24	15.42
11.00	*****	*****	*****	16.76	14.96
12.00	*****	*****	*****	16.31	14.53
13.00	*****	*****	*****	15.89	14.13
14.00	*****	*****	*****	15.49	13.76
15.00	*****	*****	*****	15.12	13.42
16.00	*****	*****	19.83	14.76	13.10
17.00	*****	*****	18.51	14.43	12.80
18.00	*****	19.92	17.33	14.12	12.52
19.00	29.16	18.65	16.27	13.82	12.26
20.00	19.31	17.50	15.32	13.54	12.01
21.00	18.88	17.29	15.21	13.50	12.01
22.00	18.47	17.08	15.10	13.47	12.01
23.00	18.08	16.89	14.99	13.43	12.01
24.00	17.71	16.69	14.89	13.40	12.01
25.00	17.36	16.51	14.78	13.36	12.01
26.00	17.02	16.33	14.68	13.33	12.01
27.00	16.70	16.15	14.58	13.29	12.01
28.00	16.39	15.98	14.47	13.26	12.01
29.00	16.09	15.82	14.38	13.23	12.01
30.00	15.80	15.66	14.28	13.19	12.01

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 0° JET ATT, NO TURB, EPNL = 101.70, RUN 104 NO JET

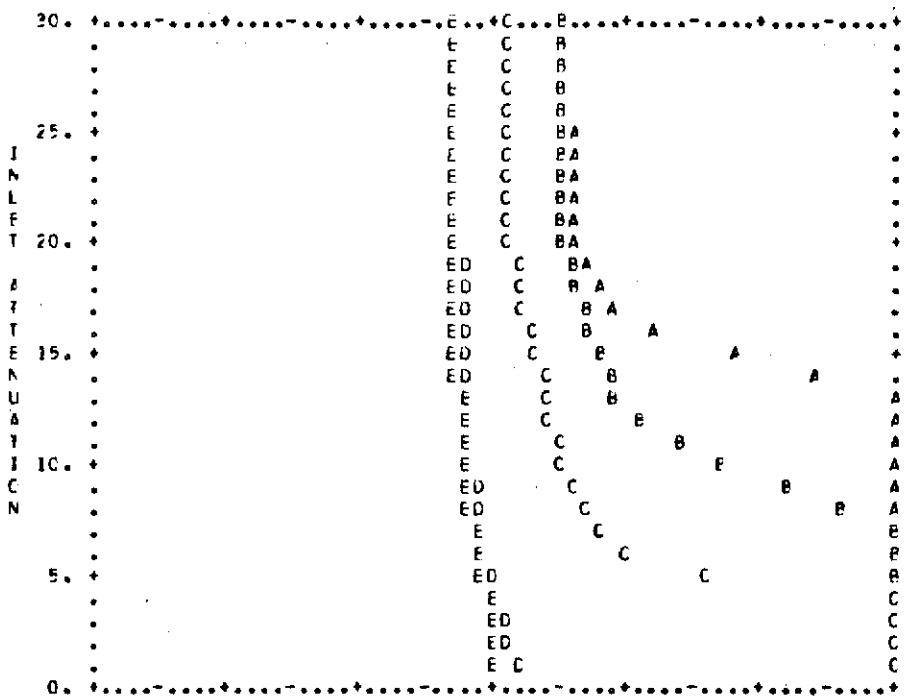
30.0					
	F	D	C	B	
	E	D	C	BA	
	E	D	C	BA	
	E	D	C	BA	
75.0		E	D	C	BA
I		E	D	C	BA
N		E	D	C	BA
L		F	D	C	BA
E		E	D	C	BA
T	20.0	F	D	C	BA
A		F	D	C	BA
T		E	D	G	H
T		E	D	C	H
E	15.0	E	D		C
N		E	D		C
U		E	D		C
A		E	D		C
T		E	D		C
I	10.0	E	D		C
O		E	D		C
N		E	D		C
		E	D		C
5.0		E			D
		E			D
		E			E
0.0					E

0.0 10.0 20.0 30.0

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 30.0	DELTA EPNL = 10.0			
INLET HARMONIC =	A 0	B 5	C 10	D 20	E 30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	24.65
4.00	*****	*****	*****	*****	19.49
5.00	*****	*****	*****	*****	18.46
6.00	*****	*****	*****	19.72	17.35
7.00	*****	*****	*****	19.35	16.38
8.00	*****	*****	*****	17.14	15.52
9.00	*****	*****	*****	16.06	14.76
10.00	*****	*****	*****	15.11	14.04
11.00	*****	*****	*****	14.71	13.61
12.00	*****	*****	*****	14.35	13.19
13.00	*****	*****	*****	14.01	12.80
14.00	*****	*****	*****	13.71	12.45
15.00	*****	*****	21.84	13.43	12.13
16.00	*****	*****	18.61	13.17	11.83
17.00	*****	25.76	17.28	12.93	11.56
18.00	*****	18.85	16.11	12.70	11.31
19.00	19.18	17.53	15.08	12.49	11.07
20.00	17.95	16.36	14.15	12.30	10.85
21.00	17.56	16.15	14.05	12.27	10.85
22.00	17.19	15.95	13.94	12.23	10.85
23.00	16.84	15.76	13.84	12.20	10.85
24.00	16.50	15.54	13.74	12.17	10.85
25.00	16.18	15.40	13.64	12.14	10.85
26.00	15.88	15.23	13.55	12.11	10.85
27.00	15.59	15.07	13.45	12.07	10.85
28.00	15.31	14.92	13.36	12.04	10.85
29.00	15.04	14.77	13.27	12.01	10.85
30.00	14.79	14.62	13.19	11.98	10.85

DELTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = C, B = S, C = 10, D = 20, E = 30)  
 APP, WITH TCNE, 3. JET ATT., NO TURE, EPNL = 101.66, RUN 11A NC JET1

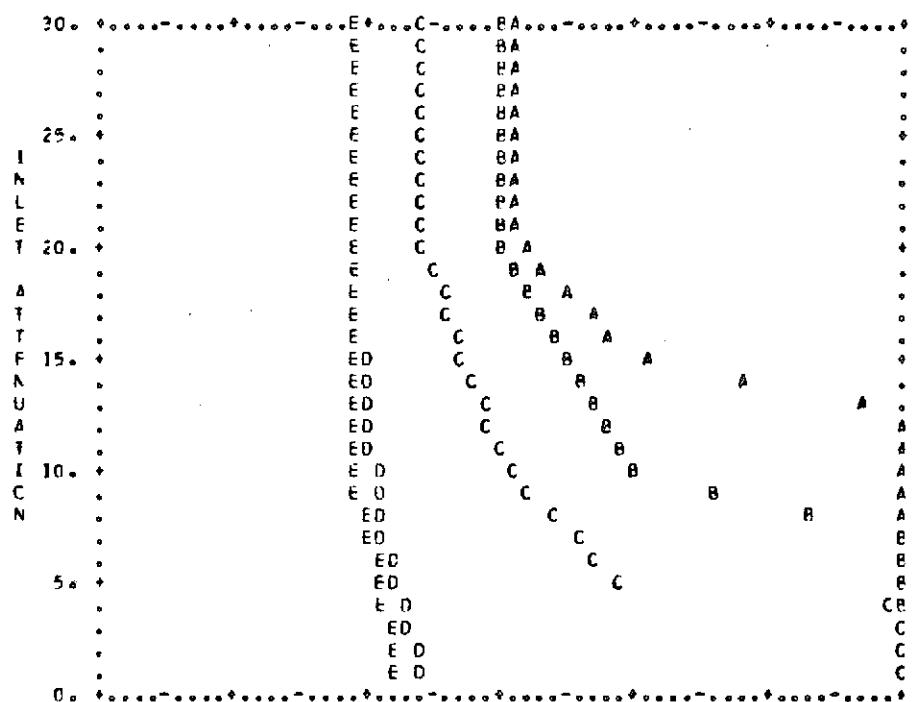


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 5.0				
INLET HARMONIC *		A C	B S	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	15.81	15.23	
2.00	*****	*****	*****	15.56	15.07	
3.00	*****	*****	*****	15.33	14.91	
4.00	*****	*****	30.76	15.10	14.75	
5.00	*****	*****	22.97	14.88	14.60	
6.00	*****	*****	15.63	14.73	14.45	
7.00	*****	30.83	15.24	14.59	14.31	
8.00	*****	28.20	18.69	14.44	14.17	
9.00	*****	25.77	18.17	14.29	14.03	
10.00	*****	23.52	17.68	14.15	13.89	
11.00	*****	21.93	17.45	14.11	13.85	
12.00	*****	20.30	17.23	14.05	13.81	
13.00	30.00	19.72	17.02	14.02	13.77	
14.00	26.89	19.38	16.81	13.98	13.73	
15.00	23.81	19.06	16.61	13.94	13.70	
16.00	20.78	18.74	16.42	13.90	13.66	
17.00	19.58	18.42	16.22	13.86	13.62	
18.00	19.04	18.12	16.04	13.83	13.59	
19.00	18.52	17.82	15.86	13.79	13.55	
20.00	18.02	17.53	15.66	13.75	13.52	
21.00	17.57	17.52	15.68	13.75	13.51	
22.00	17.53	17.52	15.68	13.75	13.51	
23.00	17.88	17.51	15.68	13.74	13.51	
24.00	17.83	17.51	15.68	13.74	13.51	
25.00	17.78	17.50	15.68	13.74	13.51	
26.00	17.73	17.49	15.68	13.74	13.51	
27.00	17.68	17.49	15.68	13.74	13.50	
28.00	17.63	17.48	15.68	13.74	13.50	
29.00	17.59	17.48	15.68	13.73	13.50	
30.00	17.54	17.47	15.68	13.73	13.50	

DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = C, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TENE, 3% JET ATT., NO TURB, EPNL = 101.66, RUN 11A NC JET1



0. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EPNL = 5.0		
	A C	B 5	C 10	D 20	E 30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	12.22	11.19
2.00	*****	*****	*****	11.89	11.02
3.00	*****	*****	*****	11.58	10.87
4.00	*****	*****	29.67	11.29	10.71
5.00	*****	*****	19.73	11.01	10.57
6.00	*****	*****	18.71	10.86	10.32
7.00	*****	30.90	17.76	10.71	10.08
8.00	*****	26.58	16.88	10.56	9.88
9.00	*****	22.79	16.06	10.42	9.71
10.00	*****	19.87	15.29	10.29	9.54
11.00	*****	19.37	14.93	10.18	9.53
12.00	*****	19.87	14.58	10.07	9.52
13.00	28.32	18.38	14.25	9.97	9.50
14.00	24.22	17.90	13.93	9.89	9.49
15.00	20.26	17.43	13.62	9.80	9.48
16.00	19.14	16.97	13.32	9.72	9.47
17.00	18.25	16.51	13.03	9.64	9.46
18.00	17.39	16.06	12.75	9.57	9.44
19.00	16.56	15.61	12.48	9.49	9.43
20.00	15.76	15.18	12.22	9.42	9.42
21.00	15.12	15.18	12.22	9.42	9.42
22.00	15.68	15.18	12.22	9.42	9.41
23.00	15.64	15.17	12.22	9.42	9.41
24.00	15.60	15.17	12.22	9.42	9.41
25.00	15.56	15.17	12.22	9.42	9.41
26.00	15.52	15.17	12.22	9.42	9.40
27.00	15.48	15.17	12.22	9.42	9.40
28.00	15.44	15.17	12.22	9.42	9.40
29.00	15.40	15.17	12.22	9.42	9.39
30.00	15.36	15.17	12.22	9.42	9.39

DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TONE, 3. JET ATT., NO TURB, EPNL = 101.66, RUN 11A AC JET1

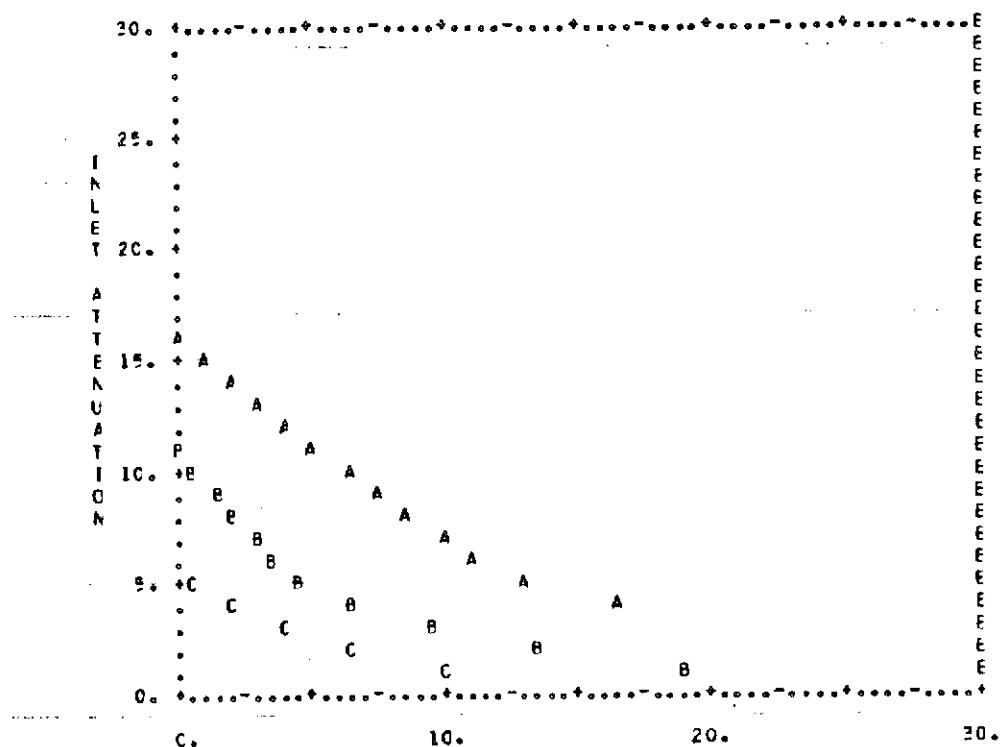
30.	.....+ED..C..R.+.....+	.....+.....+.....+.....+.....+
.	EC C B	.
.	EC C R	.
.	EC C B	.
.	EC C R	.
25.	+ED C B	.
I	EC C B	.
N	EC C B	.
L	EC C R	.
E	EC C R	.
T 20.	+ED C B	.
A	E C B A	.
T	E C B A	.
T	E C B A	.
E 15.	+E C B A	.
N	E C B A	.
U	E C B A	.
A	E C B A	.
T	ED C B A	.
E 10.	+ED C B A	.
C	ED C B A	.
N	ED C B A	A
.	ED C B A	A
-5.	+E C B B	A
.	ED C B B	A
.	ED C B B	A
0.	+ED C B B	A

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 10.0		DELTA EPNL = 5.0		E
	A	B	C	D	
FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	7.31	6.70
2.00	*****	*****	26.45	7.14	6.62
3.00	*****	*****	18.06	6.98	6.55
4.00	*****	25.45	15.50	6.83	6.48
5.00	*****	18.55	13.20	6.68	6.42
6.00	*****	17.45	11.91	6.60	6.31
7.00	*****	16.41	10.72	6.52	6.21
8.00	*****	15.43	9.82	6.45	6.10
9.00	28.65	14.51	9.36	6.37	5.99
10.00	21.67	13.63	8.93	6.30	5.89
11.00	18.72	12.97	8.79	6.25	5.87
12.00	17.28	12.33	8.66	6.21	5.86
13.00	15.93	11.69	8.53	6.16	5.84
14.00	14.66	11.07	8.40	6.12	5.83
15.00	13.47	10.46	8.27	6.07	5.81
16.00	12.34	9.93	8.15	6.02	5.80
17.00	11.28	9.66	8.03	5.98	5.78
18.00	10.27	9.39	7.91	5.93	5.77
19.00	9.67	9.14	7.80	5.89	5.76
20.00	9.24	8.90	7.69	5.84	5.74
21.00	9.22	8.90	7.67	5.84	5.74
22.00	9.19	8.90	7.66	5.84	5.74
22.00	9.16	8.90	7.65	5.84	5.74
24.00	9.13	8.90	7.63	5.83	5.74
25.00	9.11	8.90	7.62	5.83	5.73
26.00	9.08	8.90	7.61	5.83	5.73
27.00	9.05	8.90	7.59	5.83	5.73
28.00	9.02	8.90	7.58	5.83	5.73
29.00	9.00	8.90	7.57	5.82	5.73
30.00	8.97	8.90	7.55	5.82	5.73

DELTA EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TONE, 3. JET ATT., NO TUBE, EPNL = 101.66, RUN 11A NC JET1

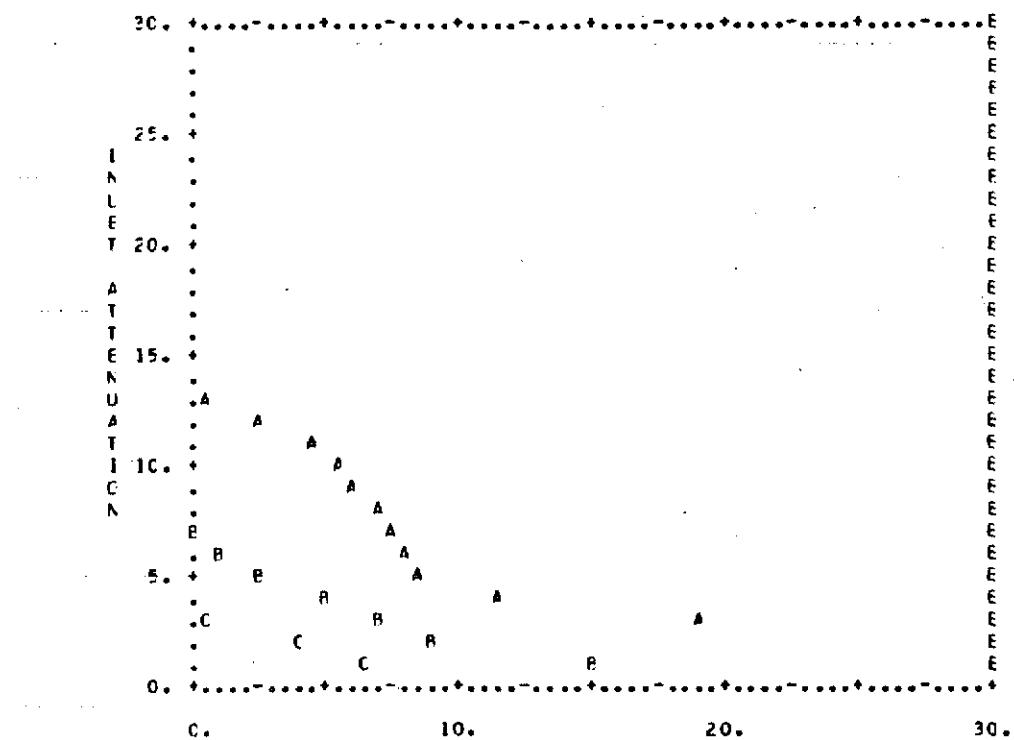


C. 10. 20.

#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 20.0		DELTA EPNL = 5.0		E 30
	A 5	B 10	C 20	D 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	18.80	5.85	*****	*****
2.00	*****	13.69	6.57	*****	*****
3.00	*****	9.44	4.08	*****	*****
4.00	15.68	6.74	2.08	*****	*****
5.00	12.84	4.50	0.26	*****	*****
6.00	11.21	3.63	*****	*****	*****
7.00	9.77	2.80	*****	*****	*****
8.00	8.69	2.01	*****	*****	*****
9.00	7.68	1.27	*****	*****	*****
10.00	6.73	0.56	*****	*****	*****
11.00	5.16	0.10	*****	*****	*****
12.00	4.00	*****	*****	*****	*****
13.00	2.95	*****	*****	*****	*****
14.00	1.98	*****	*****	*****	*****
15.00	1.06	*****	*****	*****	*****
16.00	0.21	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

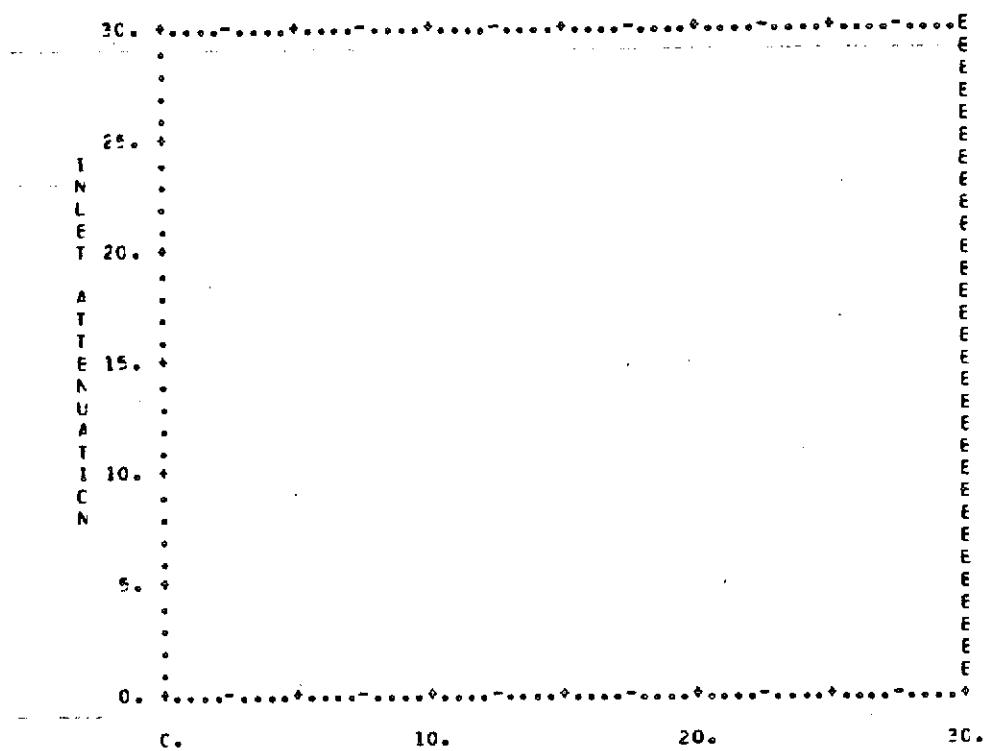
DELTA EPNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TCNE, 3. JET ATT., NO TURB, EPNL = 101.66, RUN 11A NC JET1



#### FAN DUCT ATTENUATION

INLET FUND.	FAN HARMONIC = 30.0		DELTA EPNL = 5.0		E
	A	B	C	D	
	0.00	5.00	10.00	20.00	30.00
1.00	*****	15.08	6.65	*****	*****
2.00	*****	9.05	3.78	*****	*****
3.00	19.02	6.91	0.30	*****	*****
4.00	11.39	5.18	*****	*****	*****
5.00	8.75	2.50	*****	*****	*****
6.00	8.03	1.20	*****	*****	*****
7.00	7.37	0.02	*****	*****	*****
8.00	6.75	*****	*****	*****	*****
9.00	6.18	*****	*****	*****	*****
10.00	5.64	*****	*****	*****	*****
11.00	4.30	*****	*****	*****	*****
12.00	2.46	*****	*****	*****	*****
13.00	0.68	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

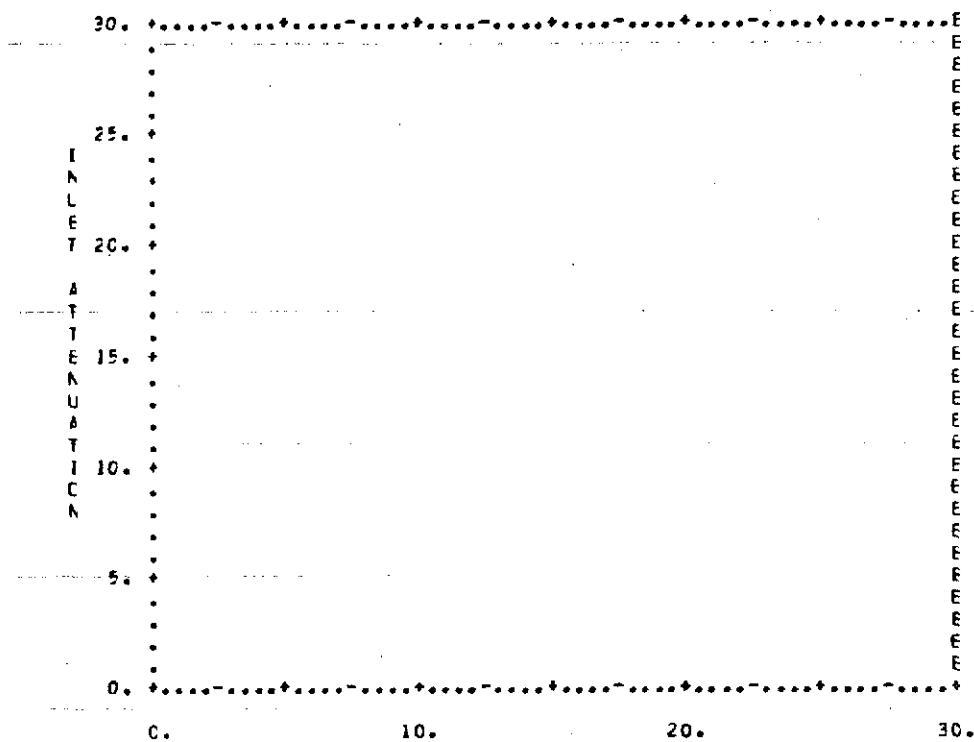
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP, WITH TCNE, 3. JET ATT., NO TURB, EPNL = 101.66, RUN 11A NC JET1



#### FAN DUCT ATTENUATION

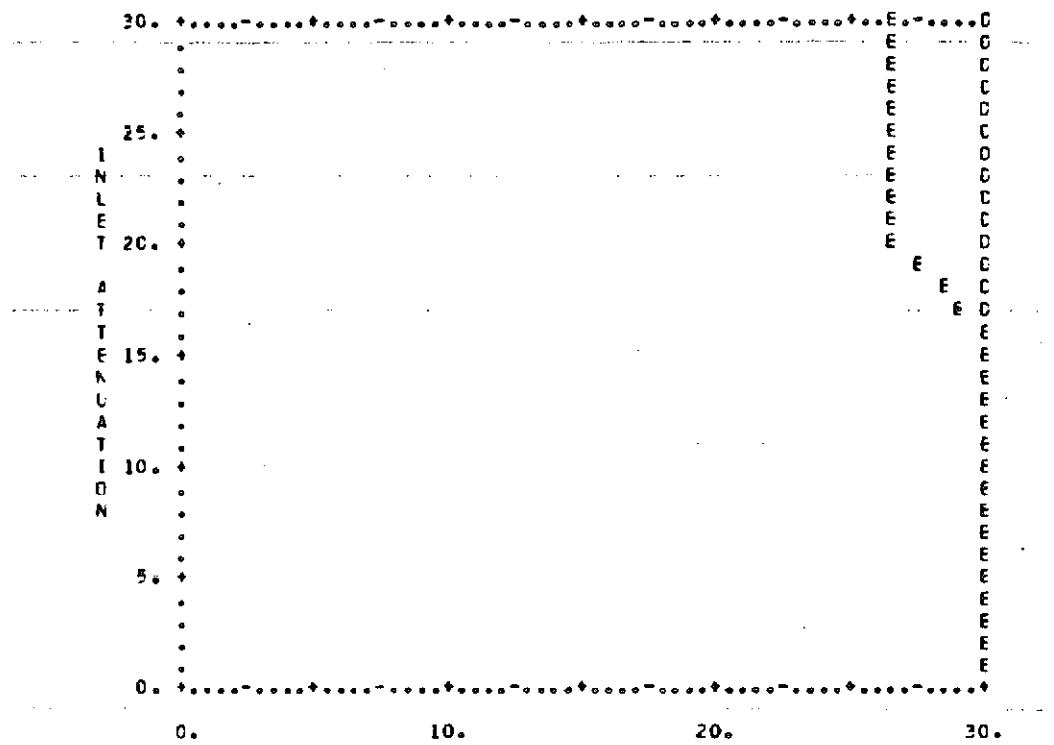
FAN HARMONIC = 0.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

DELTA EPNL = 10.0      FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TCNE, 3% JET ATT., NO TURB, EPAL = 101.66, RUN 11A AC JETI



FAN HARMONIC = 5.0		DELTA EPNL = 10.0				
INLET HARMONIC	FUND.	A	B	C	D	E
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

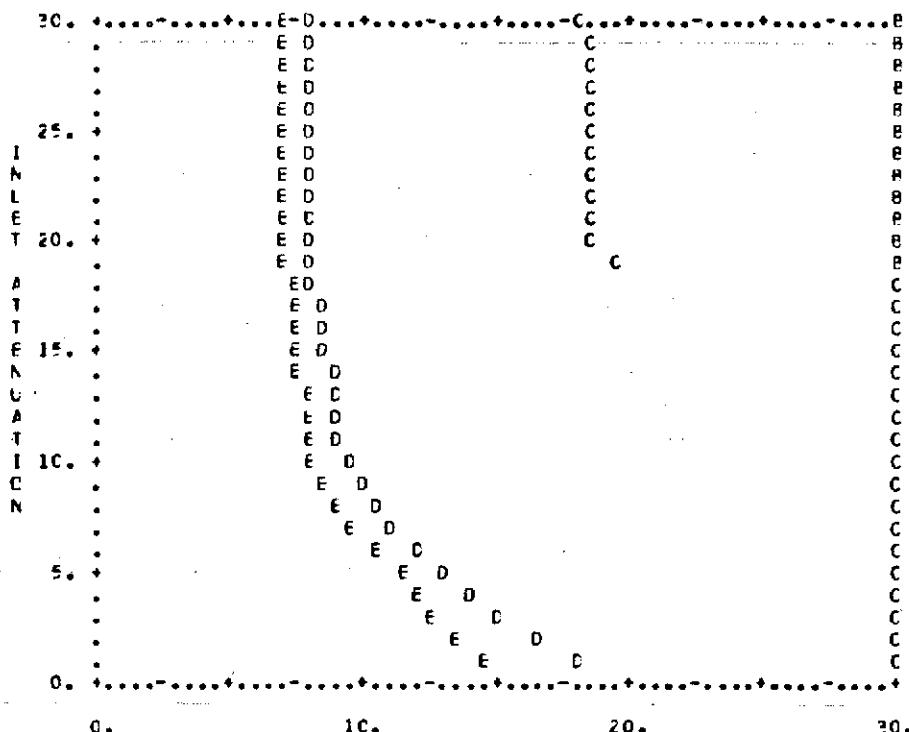
DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC IA = C, B = 5, C = 10, D = 20, E = 30  
 APP, WITH TCNE, 3. JET ATT., NO TURB, EPNL = 101.66, RUN 11A NC JET1



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 10.0		DELTA EPNL = 10.0		
	A C	B 5	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	30.07
17.00	*****	*****	*****	*****	29.16
18.00	*****	*****	*****	*****	28.29
19.00	*****	*****	*****	*****	27.46
20.00	*****	*****	*****	*****	26.67
21.00	*****	*****	*****	*****	26.67
22.00	*****	*****	*****	*****	26.66
23.00	*****	*****	*****	*****	26.66
24.00	*****	*****	*****	*****	26.66
25.00	*****	*****	*****	*****	26.66
26.00	*****	*****	*****	*****	26.66
27.00	*****	*****	*****	*****	26.66
28.00	*****	*****	*****	*****	26.65
29.00	*****	*****	*****	*****	26.65
30.00	*****	*****	*****	*****	26.65

DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TONE, 3. JET ATT., NO TURE, EPNL = 101.66, RUN 11A NC JET1



#### FAN DUCT ATTENUATION

INLET HARMONIC *	FAN HARMONIC = 20.0		DELTA EPNL = 10.0			
	A C	B 5	C 10	D 20	E 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
1.00	*****	*****	*****	17.84	14.46	
2.00	*****	*****	*****	16.36	13.53	
3.00	*****	*****	*****	15.05	12.71	
4.00	*****	*****	*****	13.89	11.57	
5.00	*****	*****	*****	12.84	11.31	
6.00	*****	*****	*****	11.97	10.43	
7.00	*****	*****	*****	11.19	9.74	
8.00	*****	*****	*****	10.49	9.18	
9.00	*****	*****	*****	9.88	8.67	
10.00	*****	*****	*****	9.39	8.19	
11.00	*****	*****	*****	9.23	8.08	
12.00	*****	*****	*****	9.07	7.96	
13.00	*****	*****	*****	8.91	7.85	
14.00	*****	*****	*****	8.76	7.74	
15.00	*****	*****	*****	8.61	7.64	
16.00	*****	*****	*****	8.47	7.53	
17.00	*****	*****	*****	8.33	7.43	
18.00	*****	*****	*****	8.20	7.33	
19.00	*****	*****	19.45	8.06	7.24	
20.00	*****	*****	18.68	7.93	7.14	
21.00	*****	*****	18.63	7.93	7.14	
22.00	*****	*****	18.59	7.93	7.14	
23.00	*****	*****	18.54	7.93	7.14	
24.00	*****	*****	18.49	7.93	7.14	
25.00	*****	*****	18.44	7.93	7.14	
26.00	*****	*****	18.39	7.93	7.14	
27.00	*****	*****	18.35	7.93	7.14	
28.00	*****	*****	18.30	7.93	7.14	
29.00	*****	*****	18.25	7.93	7.14	
30.00	*****	*****	18.21	7.93	7.14	

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 APP. WITH TCNE, 3. JET ATT., NO TUBE, EPNL = 101.66, RUN 114 AC JET1

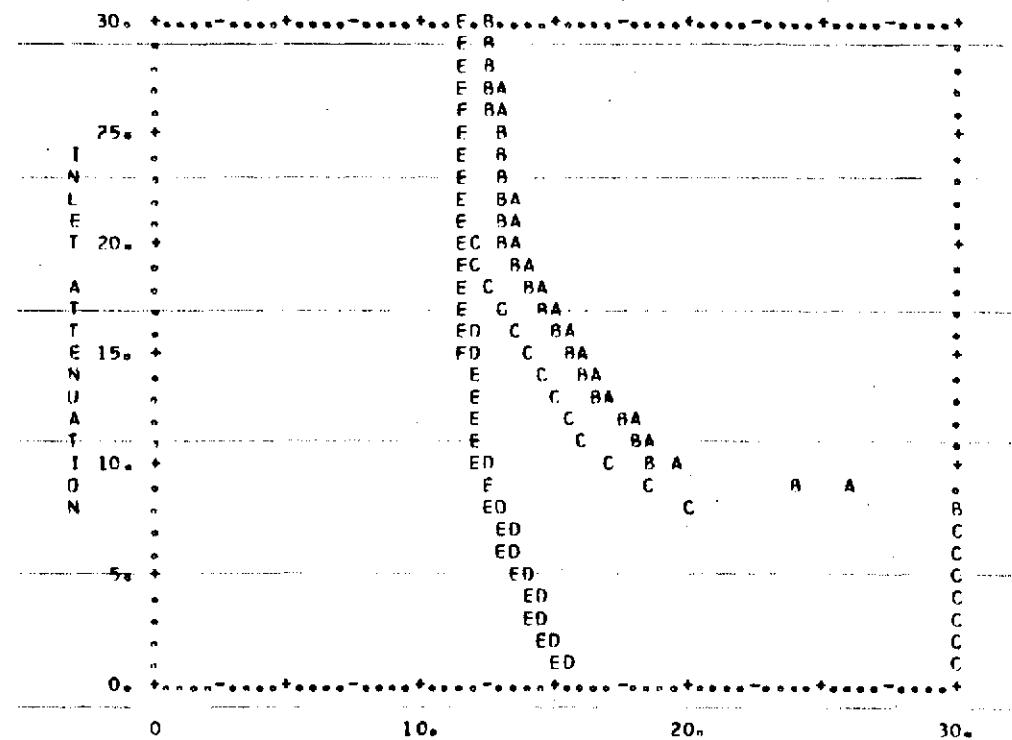
20.	*****	E	*****	C	*****	E
.	E	D	C			A
.	E	D	C			B
.	E	D	C			B
.	E	D	C			B
25.	+	E	C	C		B
I	.	E	C	C		B
N	.	E	D	C		B
L	.	E	C	C		B
E	.	E	C	C		B
T	20.	+	E	D	C	B
A	.	E	C	C		A
T	.	E	D	G		B
T	.	E	D	C		B
E	15.	+	E	D	C	B
R	.	E	D	C		B
U	.	E	D	C		C
A	.	E	D			C
T	.	E	D			C
I	10.	+	E	D		C
O	.	E	D			C
N	.	E	D			C
.	.	E	D			C
5.	+	E	D			C
.	.	E	D			C
.	.	E	D			C
0.	.	E	D			C

C. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	12.79	9.55
2.00	*****	*****	*****	*****	11.10	9.15
3.00	*****	*****	*****	*****	9.84	8.78
4.00	*****	*****	*****	*****	9.25	8.42
5.00	*****	*****	*****	*****	8.71	8.09
6.00	*****	*****	*****	*****	8.28	7.58
7.00	*****	*****	*****	*****	7.88	7.12
8.00	*****	*****	*****	*****	7.51	6.70
9.00	*****	*****	*****	*****	7.17	6.32
10.00	*****	*****	*****	*****	6.85	5.57
11.00	*****	*****	*****	*****	6.73	5.88
12.00	*****	*****	*****	*****	6.63	5.80
13.00	*****	*****	*****	*****	6.52	5.72
14.00	*****	*****	*****	15.46	6.42	5.64
15.00	*****	*****	*****	18.46	6.32	5.56
16.00	*****	*****	*****	17.50	6.22	5.49
17.00	*****	*****	*****	16.59	6.13	5.42
18.00	*****	*****	*****	15.71	6.04	5.35
19.00	*****	*****	*****	14.86	5.95	5.28
20.00	*****	*****	*****	14.06	5.86	5.21
21.00	*****	*****	*****	14.05	5.86	5.21
22.00	*****	*****	*****	14.03	5.86	5.21
23.00	*****	*****	*****	14.00	5.86	5.21
24.00	*****	*****	*****	13.97	5.86	5.21
25.00	*****	*****	*****	13.95	5.86	5.21
26.00	*****	*****	*****	13.92	5.86	5.21
27.00	*****	*****	*****	13.90	5.86	5.21
28.00	*****	*****	*****	13.87	5.86	5.21
29.00	*****	*****	*****	13.85	5.86	5.21
30.00	*****	*****	*****	13.82	5.86	5.21

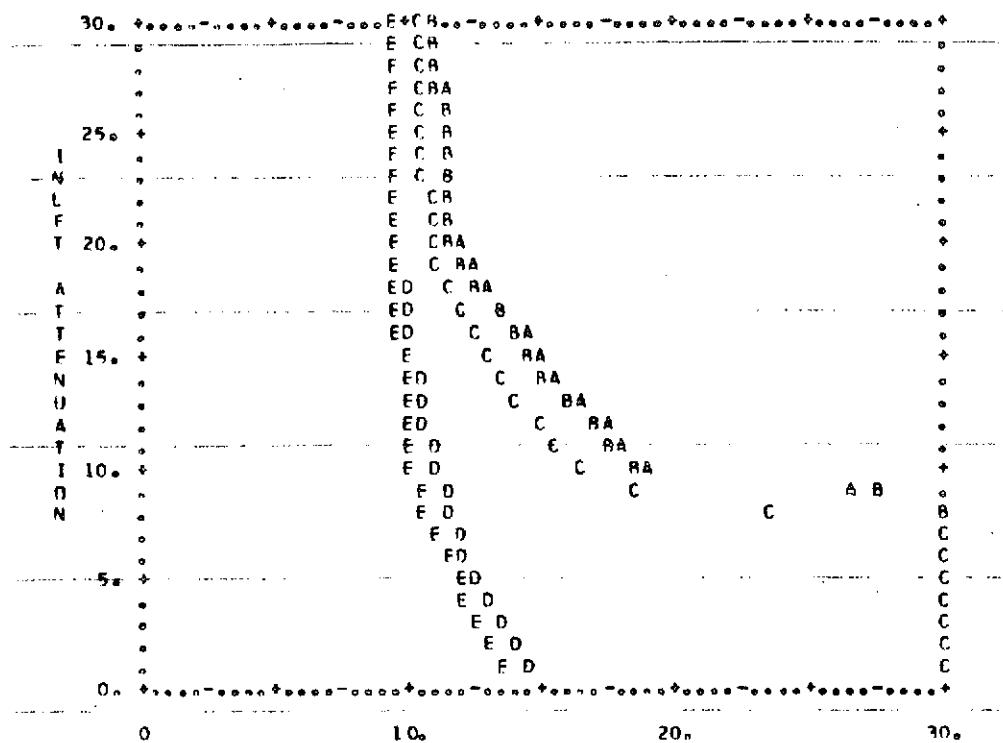
DELTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 2.5 JET ATT, NO FURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 5.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	15.37	14.90	
2.00	*****	*****	*****	14.98	14.52	
3.00	*****	*****	*****	14.61	14.18	
4.00	*****	*****	*****	14.26	13.85	
5.00	*****	*****	*****	13.93	13.55	
6.00	*****	*****	*****	13.59	13.23	
7.00	*****	*****	*****	13.26	12.91	
8.00	*****	*****	19.92	12.94	12.61	
9.00	26.03	23.78	18.28	12.62	12.32	
10.00	19.37	18.74	16.82	12.31	12.04	
11.00	18.62	18.02	16.20	12.21	11.97	
12.00	17.91	17.34	15.61	12.12	11.90	
13.00	17.25	16.69	15.05	12.02	11.82	
14.00	16.62	16.09	14.51	11.93	11.75	
15.00	16.03	15.51	14.00	11.84	11.68	
16.00	15.47	14.97	13.51	11.75	11.62	
17.00	14.93	14.45	13.05	11.66	11.55	
18.00	14.43	13.96	12.60	11.58	11.48	
19.00	13.95	13.50	12.18	11.50	11.42	
20.00	13.49	13.05	11.77	11.42	11.36	
21.00	13.39	12.99	11.75	11.41	11.35	
22.00	13.29	12.94	11.73	11.41	11.34	
23.00	13.19	12.88	11.70	11.40	11.34	
24.00	13.10	12.83	11.68	11.39	11.33	
25.00	13.00	12.78	11.66	11.39	11.33	
26.00	12.91	12.72	11.64	11.38	11.32	
27.00	12.81	12.67	11.62	11.38	11.31	
28.00	12.72	12.61	11.59	11.37	11.31	
29.00	12.63	12.56	11.57	11.36	11.30	
30.00	12.54	12.51	11.55	11.36	11.30	

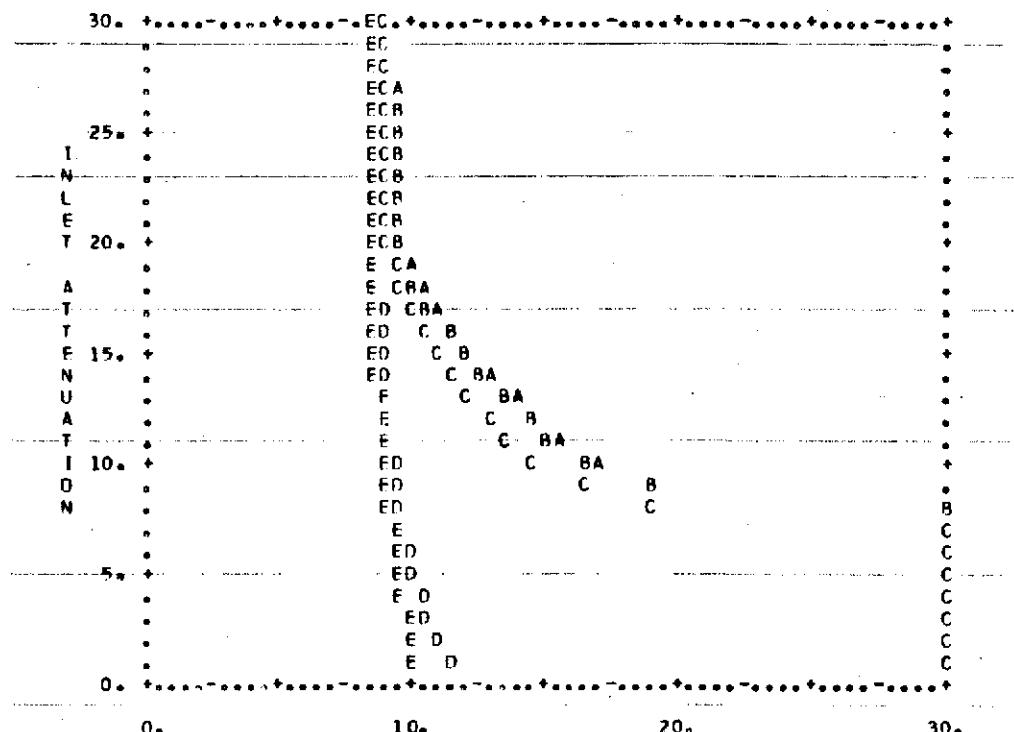
DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

	FAN HARMONIC = 5.0		DELTA EPNL = 5.0			
INLET HARMONIC =	A 0	B 5	C 10	D 20	E 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
1.00	*****	*****	*****	14.48	13.37	
2.00	*****	*****	*****	13.93	12.92	
3.00	*****	*****	*****	13.42	12.50	
4.00	*****	*****	*****	12.95	12.12	
5.00	*****	*****	*****	12.51	11.76	
6.00	*****	*****	*****	12.17	11.38	
7.00	*****	*****	*****	11.85	11.02	
8.0	*****	*****	23.37	11.55	10.67	
9.0	26.51	27.56	18.28	11.26	10.33	
10.00	19.09	18.54	16.37	10.99	10.00	
11.00	18.16	17.61	15.58	10.80	9.95	
12.00	17.28	16.76	14.87	10.62	9.91	
13.00	16.46	15.97	14.27	10.44	9.86	
14.00	15.68	15.24	13.62	10.26	9.82	
15.00	14.94	14.55	13.06	10.09	9.77	
16.00	14.25	13.91	12.55	9.96	9.73	
17.00	13.59	13.31	12.08	9.96	9.69	
18.00	12.97	12.75	11.64	9.78	9.64	
19.00	12.38	12.22	11.22	9.69	9.60	
20.00	11.81	11.72	10.84	9.60	9.56	
21.00	11.73	11.64	10.81	9.60	9.56	
22.00	11.64	11.57	10.77	9.60	9.56	
23.00	11.56	11.50	10.74	9.59	9.55	
24.00	11.48	11.42	10.70	9.59	9.55	
25.00	11.40	11.35	10.67	9.58	9.55	
26.00	11.33	11.28	10.63	9.58	9.55	
27.00	11.25	11.21	10.60	9.57	9.54	
28.00	11.18	11.14	10.56	9.57	9.54	
29.00	11.11	11.07	10.53	9.56	9.54	
30.00	11.04	11.00	10.49	9.56	9.54	

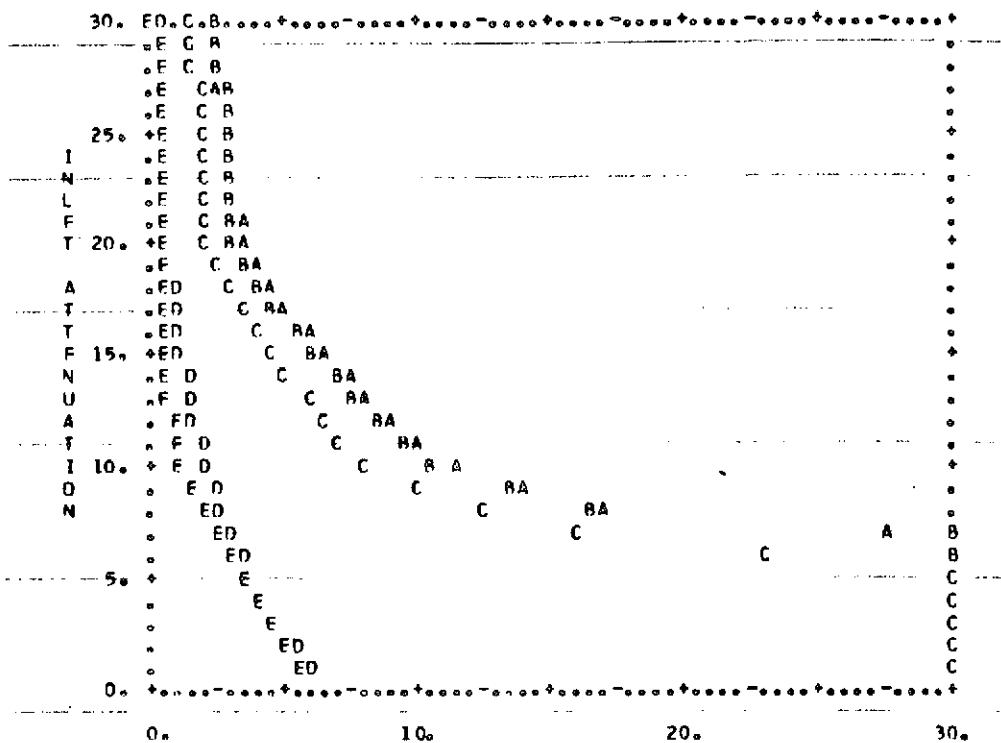
DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 10.0		DELTA EPNL = 5.0		
	A 0	B 5	C 10	D 20	E 30
1.00	*****	*****	*****	11.57	10.19
2.00	*****	*****	*****	11.13	9.96
3.00	*****	*****	*****	10.74	9.85
4.00	*****	*****	*****	10.38	9.74
5.00	*****	*****	*****	10.06	9.62
6.00	*****	*****	*****	9.86	9.49
7.00	*****	*****	*****	9.70	9.37
8.00	9.81	9.65	9.17	9.55	9.24
9.00	9.07	8.82	8.73	9.40	9.13
10.00	8.79	8.28	8.64	9.26	9.01
11.00	8.68	8.25	8.72	9.19	8.94
12.00	8.68	8.31	8.90	9.12	8.86
13.00	8.77	8.45	8.16	9.05	8.79
14.00	8.95	8.66	8.50	8.99	8.72
15.00	8.19	8.94	8.89	8.92	8.65
16.00	8.50	8.27	8.33	8.86	8.58
17.00	8.86	8.65	9.90	8.80	8.52
18.00	8.27	8.07	9.63	8.73	8.45
19.00	9.86	9.74	9.37	9.67	8.39
20.00	9.59	9.45	9.12	8.62	8.33
21.00	9.55	9.42	9.09	8.61	8.32
22.00	9.51	9.39	9.07	8.61	8.32
23.00	9.46	9.36	9.05	8.61	8.32
24.00	9.42	9.33	9.03	8.61	8.32
25.00	9.38	9.30	9.00	8.60	8.32
26.00	9.33	9.27	8.98	8.60	8.31
27.00	9.29	9.24	8.96	8.60	8.31
28.00	9.24	9.21	8.93	8.60	8.31
29.00	9.20	9.18	8.91	8.59	8.31
30.00	9.15	9.15	8.89	8.59	8.30

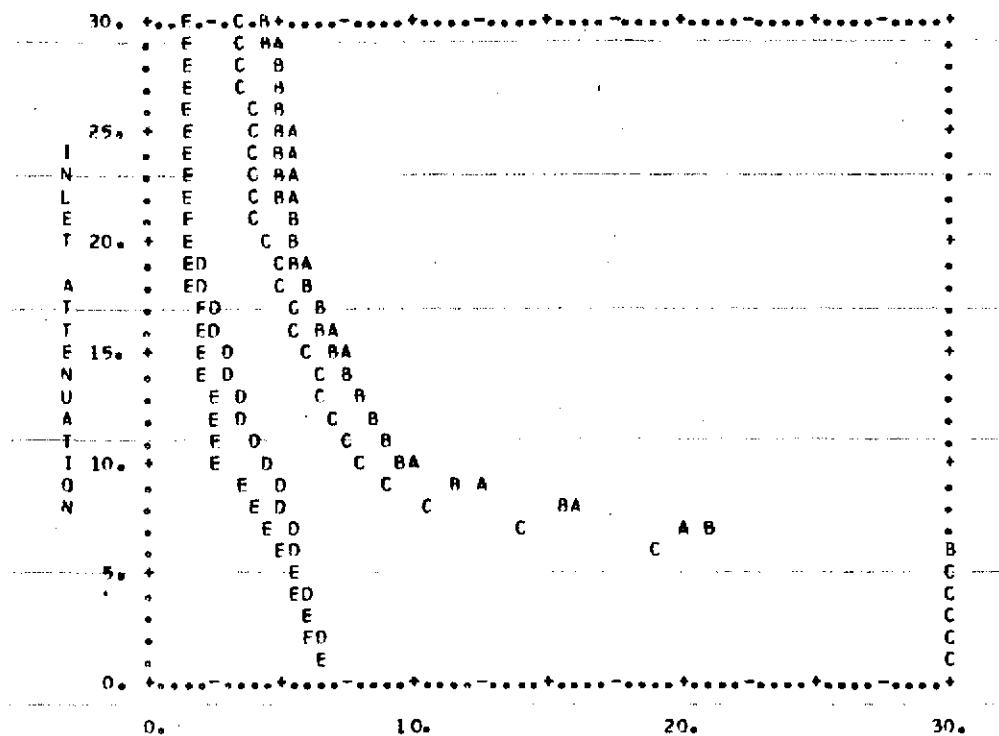
DELTA EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 20.0		DELTA EPNL = 5.0		
	A 0	B 5	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	5.86	5.40
2.00	*****	*****	*****	5.27	4.90
3.00	*****	*****	*****	4.70	4.34
4.00	*****	*****	*****	4.17	3.83
5.00	*****	*****	*****	3.69	3.35
6.00	*****	*****	22.80	3.35	2.87
7.00	27.44	*****	15.85	3.01	2.38
8.00	17.09	16.69	12.44	2.66	1.90
9.00	14.02	13.25	9.75	2.31	1.41
10.00	11.37	10.40	7.93	1.95	0.91
11.00	10.14	9.36	7.17	1.80	0.84
12.00	9.11	8.43	6.46	1.65	0.77
13.00	8.18	7.57	5.78	1.50	0.70
14.00	7.31	6.77	5.13	1.36	0.64
15.00	6.51	6.03	4.51	1.21	0.58
16.00	5.76	5.34	3.94	1.07	0.52
17.00	5.07	4.72	3.40	0.93	0.47
18.00	4.49	4.15	2.91	0.79	0.42
19.00	3.96	3.62	2.45	0.65	0.37
20.00	3.46	3.11	2.03	0.52	0.32
21.00	3.35	3.06	1.99	0.51	0.32
22.00	3.24	3.01	1.95	0.49	0.31
23.00	3.13	2.97	1.92	0.48	0.30
24.00	3.02	2.92	1.88	0.46	0.29
25.00	2.91	2.87	1.85	0.45	0.28
26.00	2.80	2.83	1.81	0.44	0.28
27.00	2.69	2.78	1.77	0.42	0.27
28.00	2.58	2.74	1.74	0.41	0.26
29.00	2.47	2.69	1.70	0.39	0.25
30.00	2.36	2.64	1.67	0.38	0.25

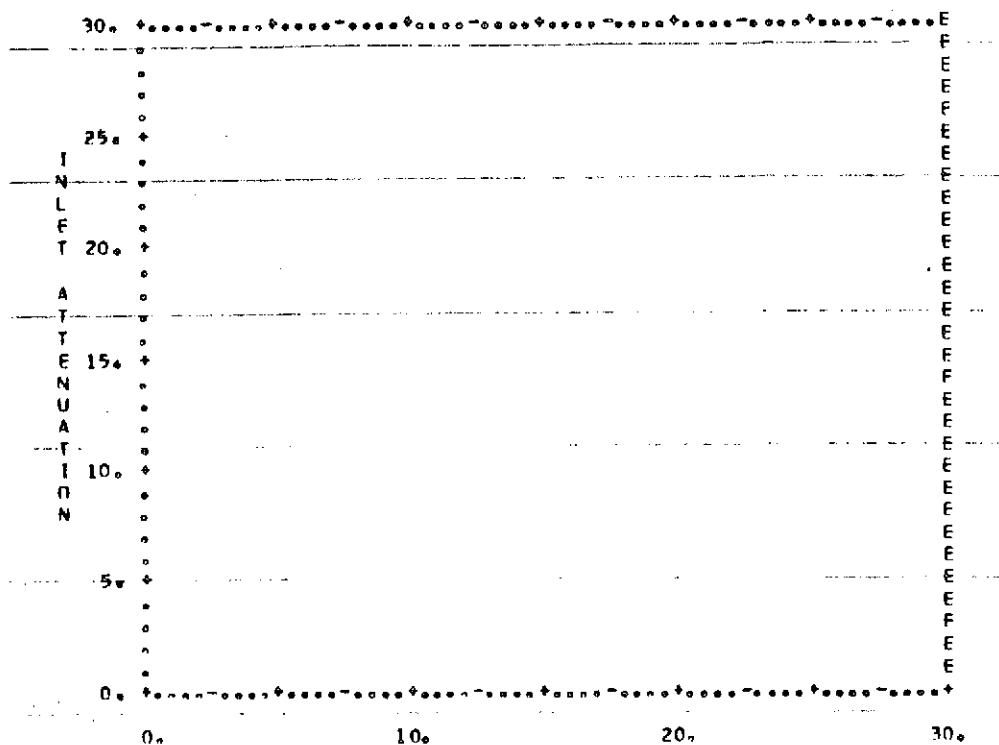
DELTA EPNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

INLET HARMONIC =	FAN HARMONIC = 30.0		DELTA EPNL = 5.0			
	A 0	B 5	C 10	D 20	E 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
1.00	*****	*****	*****	6.64	6.51	
2.00	*****	*****	*****	6.38	6.23	
3.00	*****	*****	*****	6.14	5.95	
4.00	*****	*****	*****	5.91	5.68	
5.00	*****	*****	*****	5.69	5.42	
6.00	*****	*****	19.15	5.51	5.16	
7.00	19.95	21.20	14.20	5.33	4.72	
8.00	15.87	15.43	10.32	5.15	4.04	
9.00	12.40	11.62	8.89	4.95	3.39	
10.00	9.76	9.33	7.79	4.36	2.75	
11.00	9.16	8.79	7.36	4.00	2.60	
12.00	8.63	8.29	6.98	3.67	2.47	
13.00	8.14	7.83	6.62	3.36	2.33	
14.00	7.69	7.40	6.29	3.07	2.19	
15.00	7.28	6.99	5.99	2.80	2.06	
16.00	6.91	6.62	5.70	2.55	1.93	
17.00	6.56	6.27	5.44	2.32	1.80	
18.00	6.24	5.94	5.20	2.09	1.68	
19.00	5.94	5.63	4.91	1.89	1.55	
20.00	5.66	5.34	4.31	1.69	1.43	
21.00	5.38	5.29	4.23	1.48	1.41	
22.00	5.50	5.24	4.14	1.67	1.40	
23.00	5.42	5.19	4.06	1.66	1.38	
24.00	5.34	5.13	3.97	1.65	1.37	
25.00	5.25	5.08	3.88	1.65	1.35	
26.00	5.17	5.03	3.78	1.64	1.34	
27.00	5.09	4.96	3.67	1.63	1.33	
28.00	5.01	4.83	3.57	1.62	1.31	
29.00	4.80	4.70	3.45	1.61	1.30	
30.00	4.59	4.58	3.33	1.60	1.28	

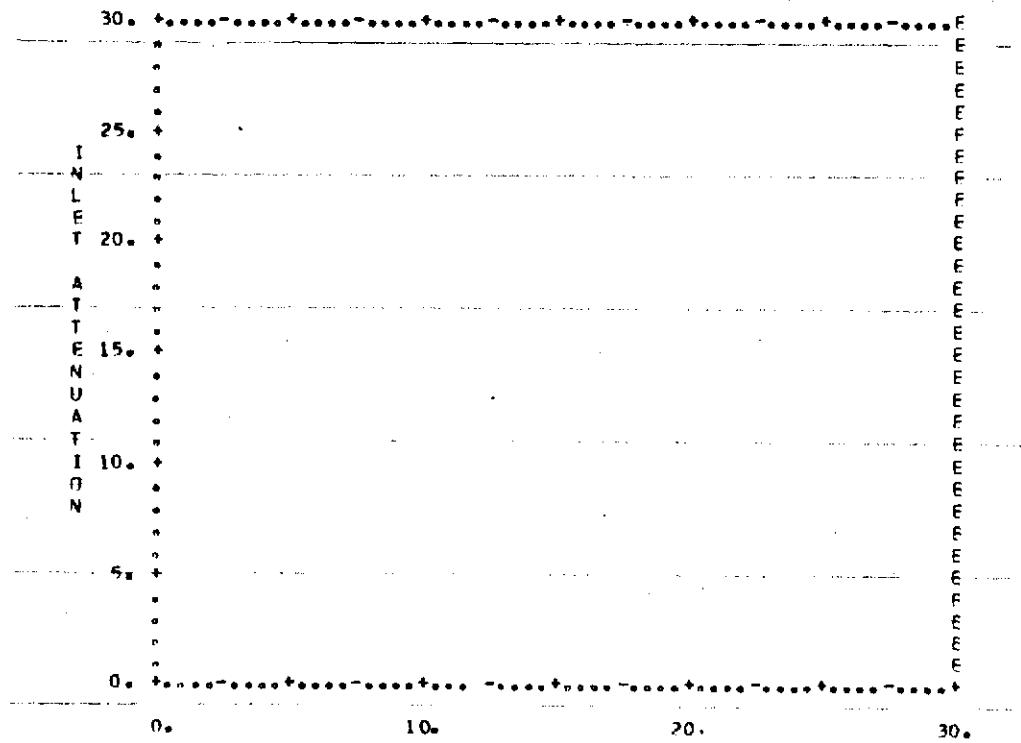
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 0.0					DELTA EPNL = 10.0
	A 0	B 5	C 10	D 20	E 30	
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

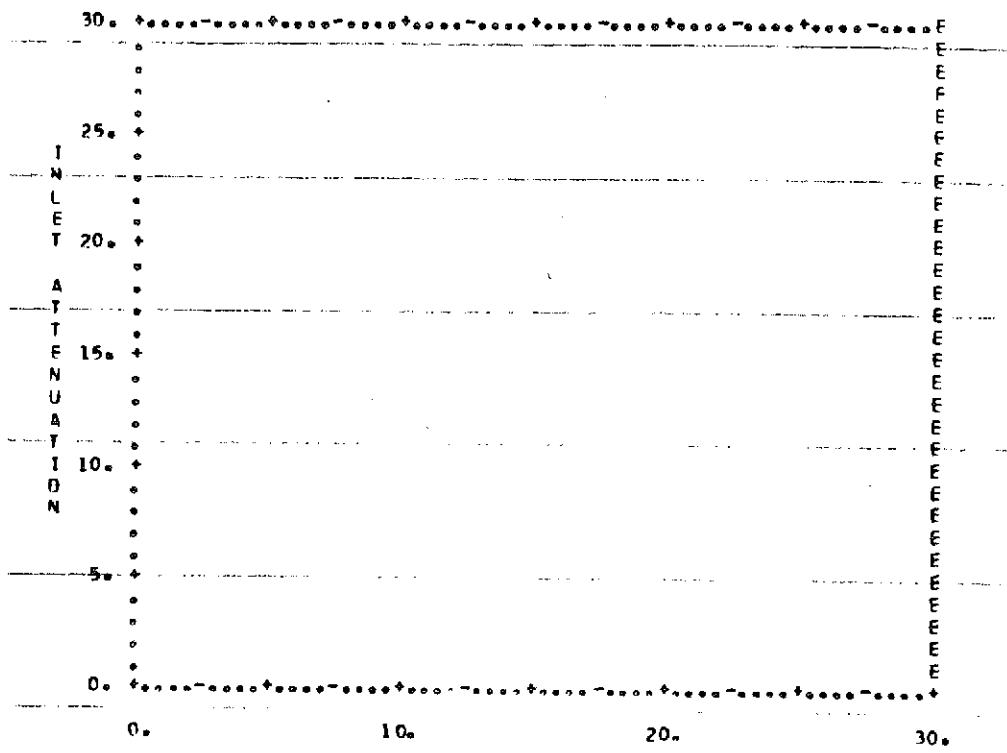
DELTA EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DICT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 10.0				
INLET HARMONIC =	A 0	B 5	C 10	D 20	E 30	
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

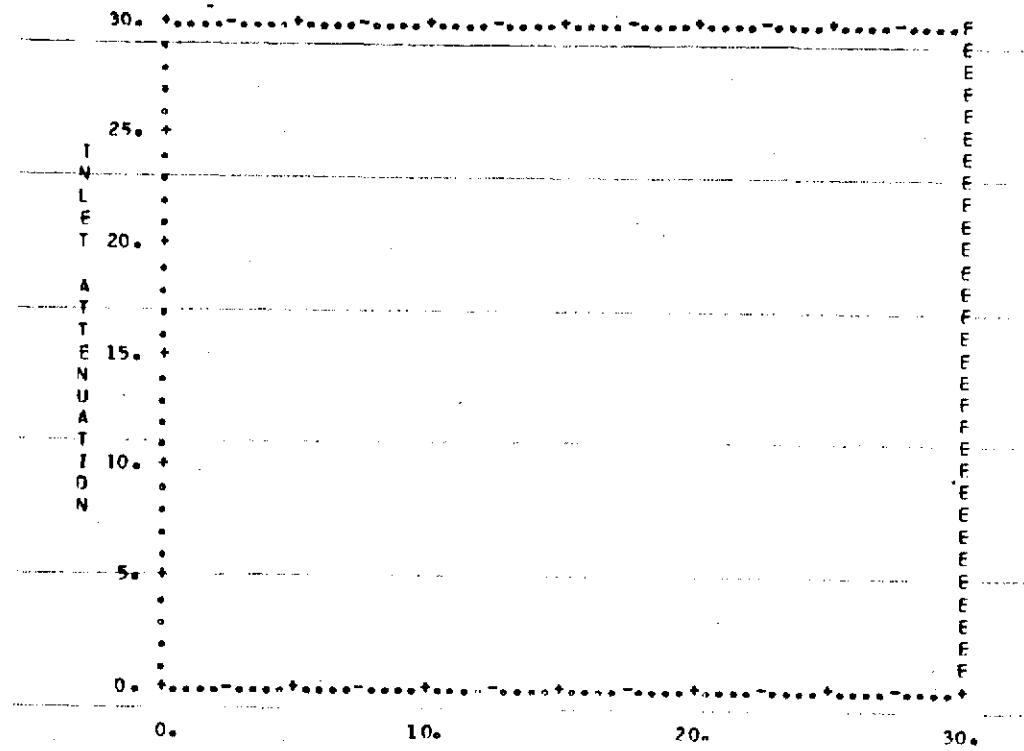
DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

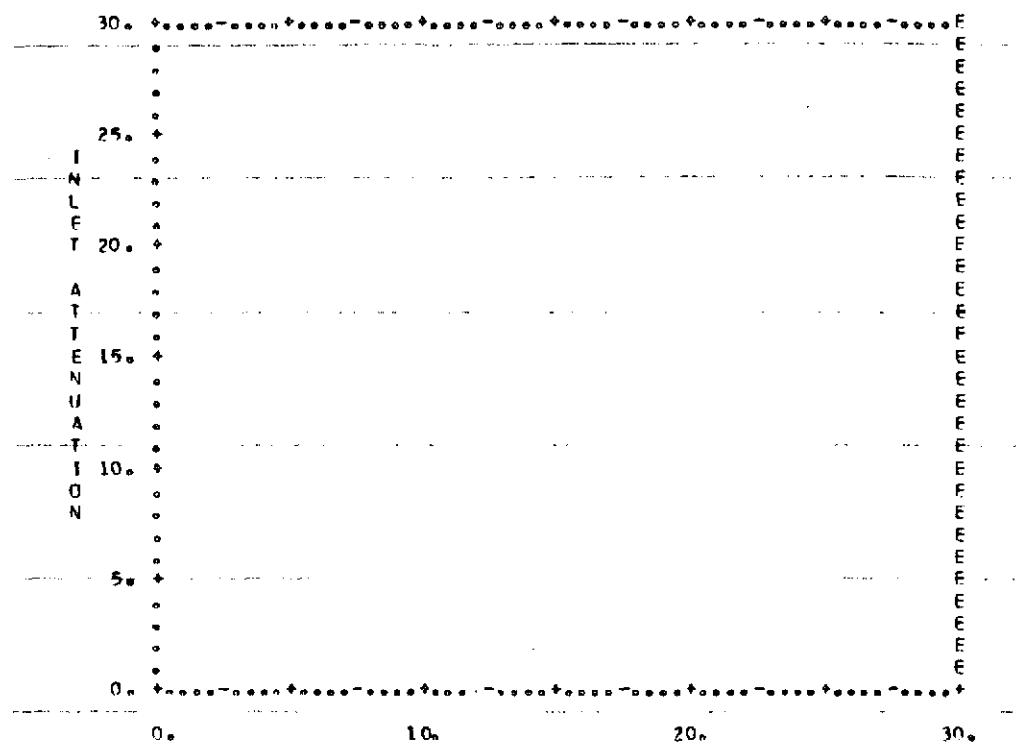
FAN HARMONIC = 10.0		DELTA EPNL = 10.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
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23.00	*****	*****	*****	*****	*****	*****
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26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
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12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
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26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
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30.00	*****	*****	*****	*****	*****	*****

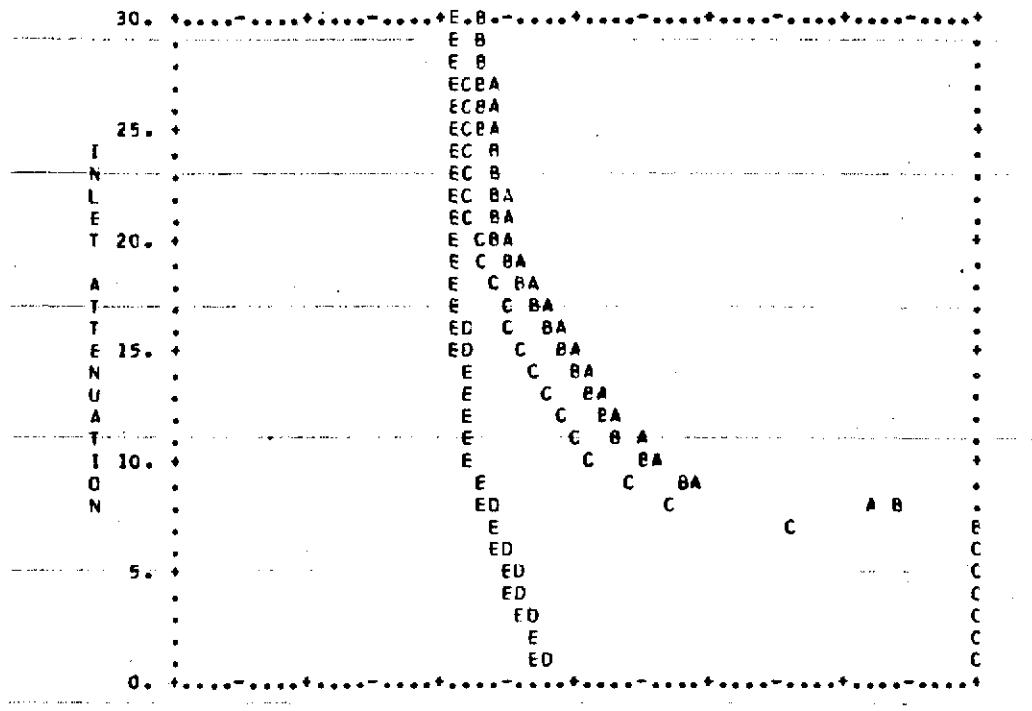
DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 2.5 JET ATT, NO TURB, EPNL = 102.56, RUN 12A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EPNL = 10.0			
INLET HARMONIC	A	B	C	D	E
	0	5	10	20	30
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
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16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

DELTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE.5. JET ATT, NO TURB, EPNL = 102.22, RUN-13A

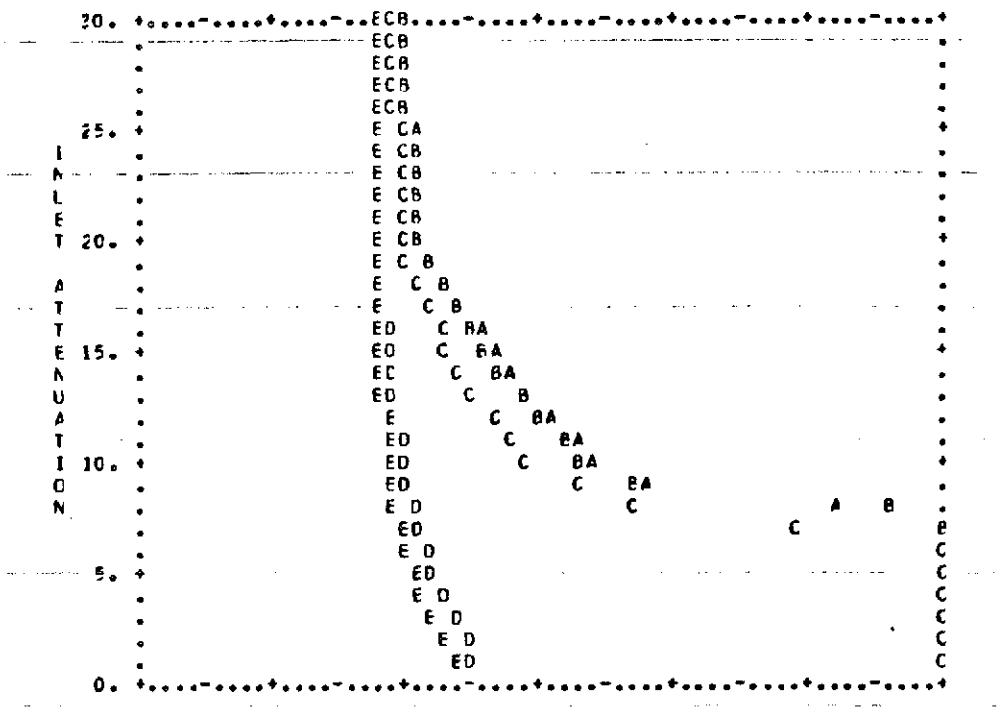


0. 10. 20. 30.

#### FAN DUCT ATTENUATION

	FAN HARMONIC = 0.0		DELTA EPNL = 5.0		
INLET HARMONIC =	A	B	C	D	E
	0	5	10	20	30
1.00	*****	*****	*****	14.06	13.64
2.00	*****	*****	*****	13.71	13.28
3.00	*****	*****	*****	13.38	12.95
4.00	*****	*****	*****	13.07	12.65
5.00	*****	*****	*****	12.77	12.37
6.00	*****	*****	*****	12.45	12.08
7.00	*****	*****	22.86	12.14	11.80
8.00	26.22	27.18	18.48	11.83	11.53
9.00	19.55	19.03	16.90	11.53	11.28
10.00	18.02	17.33	15.51	11.24	11.04
11.00	17.31	16.63	14.95	11.16	10.97
12.00	16.64	15.98	14.44	11.08	10.91
13.00	16.01	15.37	13.95	11.01	10.85
14.00	15.41	14.80	13.50	10.93	10.79
15.00	14.85	14.26	13.18	10.86	10.73
16.00	14.33	13.76	12.68	10.79	10.68
17.00	13.82	13.28	12.30	10.72	10.62
18.00	13.35	12.82	11.95	10.66	10.56
19.00	12.90	12.39	11.62	10.59	10.51
20.00	12.47	11.99	11.30	10.52	10.45
21.00	12.37	11.93	11.23	10.52	10.45
22.00	12.26	11.87	11.16	10.51	10.45
23.00	12.16	11.82	11.09	10.51	10.44
24.00	12.06	11.76	11.02	10.50	10.44
25.00	11.96	11.71	10.94	10.49	10.44
26.00	11.87	11.65	10.87	10.49	10.44
27.00	11.77	11.60	10.80	10.48	10.43
28.00	11.68	11.54	10.72	10.48	10.43
29.00	11.59	11.49	10.65	10.47	10.43
30.00	11.50	11.44	10.58	10.46	10.42

DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TCNE, S. JET ATT, NO TURB, EPNL = 102.22, RUN-13A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 5.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	12.73	11.78	
2.00	*****	*****	*****	12.27	11.39	
3.00	*****	*****	*****	11.85	11.04	
4.00	*****	*****	*****	11.47	10.72	
5.00	*****	*****	*****	11.11	10.42	
6.00	*****	*****	*****	10.83	10.11	
7.00	*****	*****	24.39	10.57	9.88	
8.00	26.15	27.86	18.30	10.31	9.70	
9.00	19.03	18.73	16.32	10.07	9.53	
10.00	17.16	16.65	14.56	9.90	9.37	
11.00	16.30	15.75	13.50	9.80	9.33	
12.00	15.45	15.01	13.29	9.70	9.29	
13.00	14.74	14.29	12.73	9.60	9.25	
14.00	14.04	13.63	12.21	9.51	9.21	
15.00	13.38	12.02	11.74	9.42	9.16	
16.00	12.76	12.45	11.30	9.33	9.12	
17.00	12.18	11.92	10.89	9.25	9.08	
18.00	11.63	11.43	10.51	9.16	9.04	
19.00	11.11	10.97	10.16	9.08	9.00	
20.00	10.62	10.53	9.87	9.00	8.96	
21.00	10.54	10.47	9.85	9.00	8.96	
22.00	10.47	10.40	9.82	8.99	8.96	
23.00	10.39	10.33	9.80	8.99	8.95	
24.00	10.32	10.27	9.78	8.98	8.95	
25.00	10.25	10.20	9.76	8.97	8.94	
26.00	10.18	10.14	9.73	8.97	8.94	
27.00	10.12	10.08	9.71	8.96	8.93	
28.00	10.05	10.01	9.69	8.96	8.93	
29.00	9.99	9.97	9.67	8.95	8.93	
30.00	9.95	9.92	9.65	8.94	8.92	

DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TCNE,5. JET ATT, NO TURE, EPNL = 102.22, RUN-13A

30.	+	.....	ECB	.....	.....	.....
.	-	EDC	.	.	.	.
.	-	EDC	.	.	.	.
.	-	EDC	.	.	.	.
.	-	EDC	.	.	.	.
25.	+	ECC	.	.	.	.
I	.	EOCA	.	.	.	.
N	.	EDCA	.	.	.	.
L	.	ECCA	.	.	.	.
E	.	ECCA	.	.	.	.
T	20.	ECCB	.	.	.	.
A	.	ECBA	.	.	.	.
T	.	E CA	.	.	.	.
T	.	E CB	.	.	.	.
E	15.	EO CBA	.	.	.	.
N	.	EC CEA	.	.	.	.
U	.	ED C B	.	.	.	.
A	.	ED C BA	.	.	.	.
T	.	E C BA	.	.	.	.
I	10.	E C BA	.	.	.	.
C	.	ED C BA	.	.	.	.
N	.	ED C B	.	.	.	.
.	.	ED C	.	.	.	.
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0.	+	.....	.....	.....	.....	.....

0. 10. 20. 30.

#### FAN DLCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 5.0			
INLET HARMONIC		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	9.99	9.64
2.00	*****	*****	*****	9.84	9.46
3.00	*****	*****	*****	9.69	9.28
4.00	*****	*****	*****	9.54	9.10
5.00	*****	*****	*****	9.40	8.92
6.00	*****	*****	*****	9.24	8.79
7.00	*****	*****	19.82	9.08	8.68
8.00	19.25	19.39	17.19	8.92	8.56
9.00	16.55	16.69	14.59	8.75	8.45
10.00	14.84	14.37	13.14	8.59	8.35
11.00	13.87	13.45	12.30	8.52	8.28
12.00	12.95	12.62	11.56	8.46	8.21
13.00	12.20	11.87	10.89	8.39	8.15
14.00	11.48	11.19	10.30	8.33	8.08
15.00	10.82	10.56	9.86	8.27	8.02
16.00	10.22	9.99	9.56	8.21	7.96
17.00	9.82	9.68	9.27	8.15	7.90
18.00	9.53	9.37	9.00	8.10	7.85
19.00	9.25	9.07	8.73	8.04	7.79
20.00	9.00	8.78	8.47	7.99	7.73
21.00	8.85	8.75	8.44	7.98	7.73
22.00	8.60	8.71	8.42	7.98	7.73
23.00	8.85	8.68	8.39	7.98	7.73
24.00	8.80	8.65	8.37	7.97	7.72
25.00	8.74	8.62	8.34	7.97	7.72
26.00	8.65	8.58	8.32	7.97	7.72
27.00	8.64	8.55	8.30	7.96	7.72
28.00	8.55	8.52	8.27	7.96	7.72
29.00	8.54	8.49	8.25	7.96	7.72
30.00	8.48	8.46	8.23	7.96	7.72

CELT A EPNL = 5.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE, 5. JET ATT, NO TURB, EPNL = 102.22, RUA-13A

C. 10. 20. 30.

## FAN CUCT ATTENUATION

FAN HARMONIC =	20.0	DELTA EPAL =			5.0	
INLET HARMONIC =	A C	B S	C 10	D 20	E 30	
INLET FLNG.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	4.64	3.93	
2.00	*****	*****	*****	4.15	3.40	
3.00	*****	*****	*****	3.67	2.91	
4.00	*****	*****	*****	3.22	2.47	
5.00	*****	*****	*****	2.79	2.06	
6.00	*****	*****	18.13	2.37	1.58	
7.00	18.27	18.64	13.83	1.97	1.10	
8.00	14.85	14.60	10.38	1.57	0.62	
9.00	11.98	11.31	8.26	1.19	0.14	
10.00	9.57	8.72	6.54	0.82	*****	
11.00	8.48	7.75	5.83	0.67	*****	
12.00	7.49	6.87	5.15	0.52	*****	
13.00	6.60	6.06	4.48	0.38	*****	
14.00	5.80	5.33	3.84	0.23	*****	
15.00	5.06	4.67	3.26	0.09	*****	
16.00	4.44	4.06	2.72	*****	*****	
17.00	3.87	3.49	2.23	*****	*****	
18.00	3.34	2.96	1.77	*****	*****	
19.00	2.84	2.45	1.35	*****	*****	
20.00	2.37	1.96	0.96	*****	*****	
21.00	2.29	1.92	0.93	*****	*****	
22.00	2.20	1.87	0.90	*****	*****	
23.00	2.12	1.82	0.87	*****	*****	
24.00	2.03	1.78	0.83	*****	*****	
25.00	1.94	1.73	0.80	*****	*****	
26.00	1.85	1.68	0.77	*****	*****	
27.00	1.76	1.63	0.73	*****	*****	
28.00	1.67	1.58	0.70	*****	*****	
29.00	1.58	1.53	0.67	*****	*****	
30.00	1.48	1.48	0.63	*****	*****	

DELTA EFNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TCNE,5. JET ATT, NO TURE, EPAL = 102.22, RUN=138.

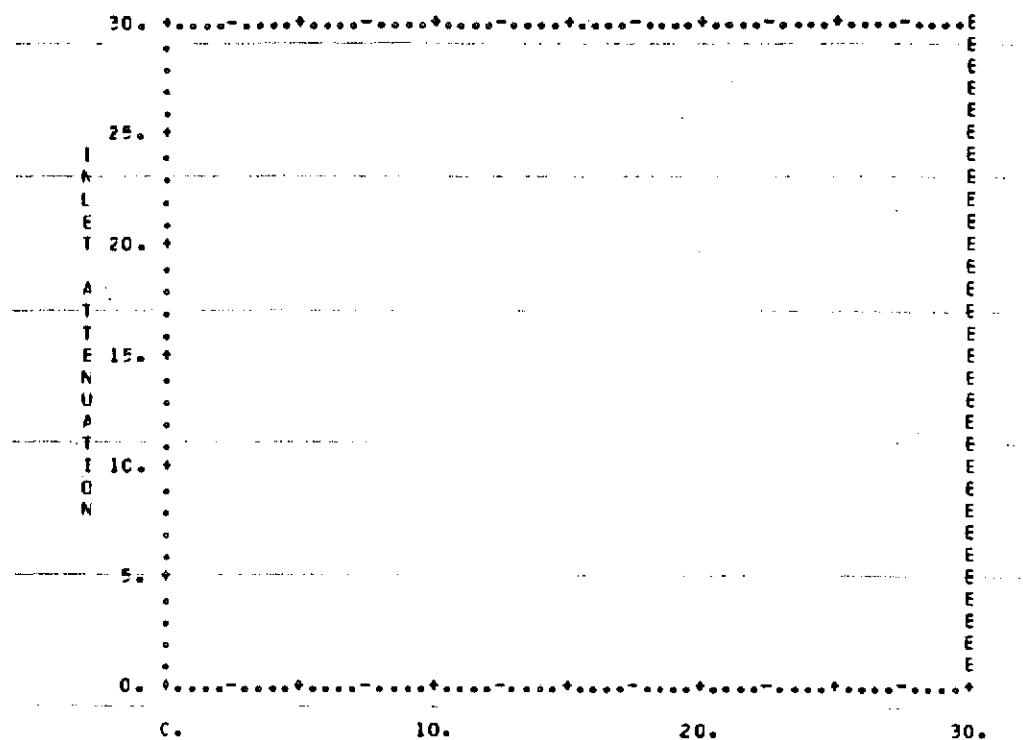
20.	+C...B....+	.....+	.....+	.....+	.....+	.....+
	. C BA					E
	. C BA					E
	. C B					E
	. C BA					E
25.	+ C BA					E
I	. C BA					E
N	. C BA					E
L	. C BA					E
E	. C EA					E
T	20. + C BA					E
A	. C EA					E
T	. C C BA					E
T	. C C BA					E
E	15. + D C BA					E
R	E D C BA					.
U	E D C B					.
A	E D C B					.
T	.E D C B					.
I	10. +E C C BA					.
O	. E D C BA					.
N	. E D C BA					.
	. E D C AB					.
	. E D C					B
5.	+ ED					C
	. E					C
	. EC					C
	. E					C
	. E					C
0.	+.....+	.....+	.....+	.....+	.....+	.....+

0. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EFNL = 5.0			
INLET HARMONIC *		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	5.74	5.66
2.00	*****	*****	*****	5.51	5.41
3.00	*****	*****	*****	5.28	5.17
4.00	*****	*****	*****	5.07	4.81
5.00	*****	*****	*****	4.63	4.15
6.00	*****	*****	16.25	4.13	3.31
7.00	17.14	17.28	11.89	3.65	2.54
8.00	13.25	12.86	9.36	3.18	1.81
9.00	9.99	9.69	8.11	2.73	1.14
10.00	8.78	8.35	6.97	2.30	0.51
11.00	8.20	7.82	6.52	2.05	0.39
12.00	7.68	7.33	6.11	1.80	0.27
13.00	7.22	6.89	5.75	1.54	0.14
14.00	6.80	6.47	5.43	1.27	0.03
15.00	6.42	6.09	5.14	1.00	*****
16.00	6.07	5.74	4.67	0.72	*****
17.00	5.75	5.41	4.03	0.44	*****
18.00	5.45	5.10	3.40	0.15	*****
19.00	5.18	4.52	2.80	*****	*****
20.00	4.80	3.82	2.22	*****	*****
21.00	4.56	3.67	2.08	*****	*****
22.00	4.33	3.53	1.94	*****	*****
23.00	4.10	3.38	1.80	*****	*****
24.00	3.87	3.24	1.65	*****	*****
25.00	3.64	3.09	1.49	*****	*****
26.00	3.42	2.95	1.32	*****	*****
27.00	3.19	2.80	1.15	*****	*****
28.00	2.97	2.66	0.96	*****	*****
29.00	2.75	2.51	0.77	*****	*****
30.00	2.53	2.37	0.56	*****	*****

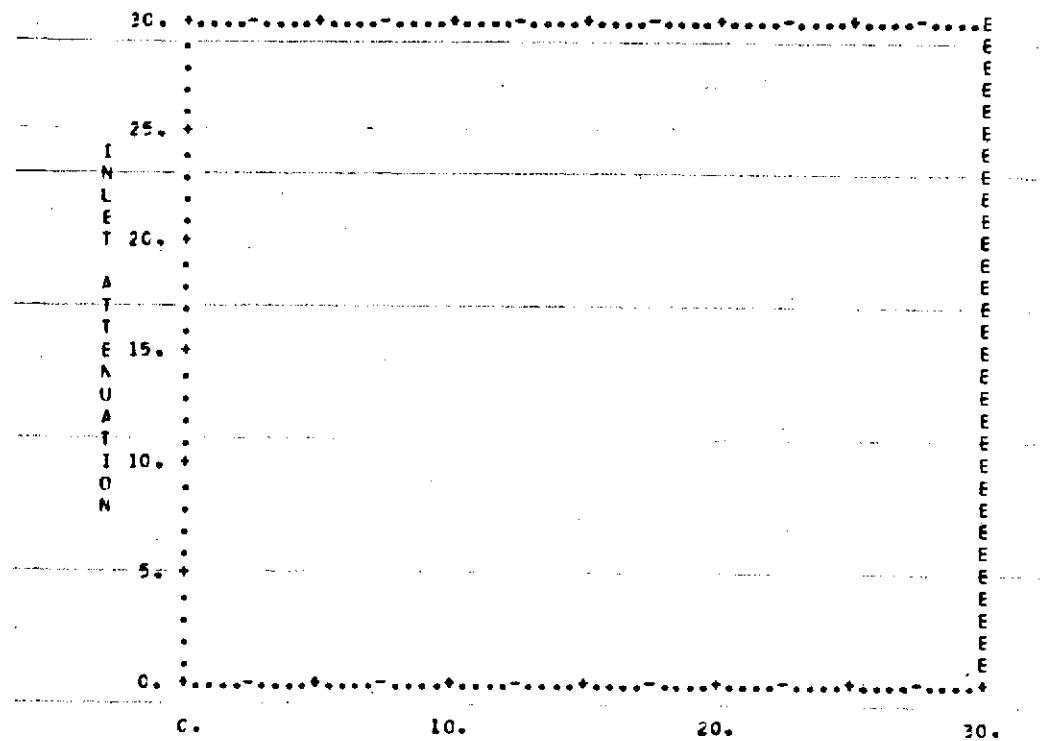
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TCNE=5, JET ATT, NO TURB, EPNL = 102.22, RUN-13A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 0.0		DELTA EPAL = 10.0		
	A C	B 5	C 10	D 20	E 30
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****

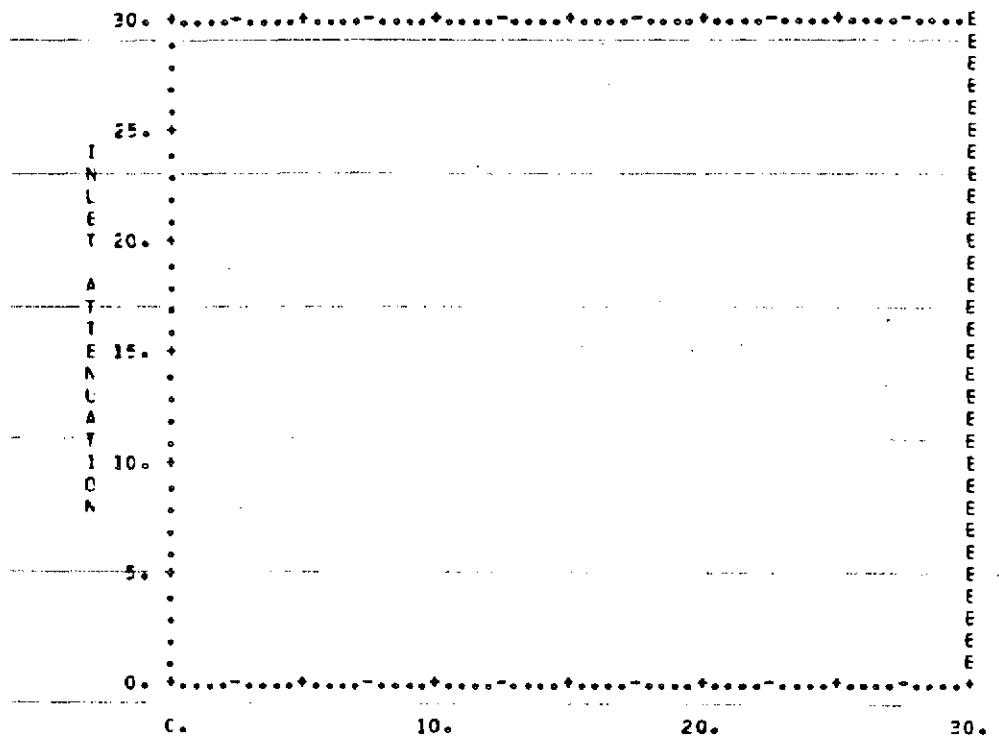
CELT A EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C, WITH TONE, JET ATT, NO TURE, EPAL = 102.22, RUN-13A



#### FAN DUCT ATTENLATION

INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EPNL = 10.0			
	A C	B 5	C 10	D 20	E 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
1.00	*****	*****	*****	*****	*****	
2.00	*****	*****	*****	*****	*****	
3.00	*****	*****	*****	*****	*****	
4.00	*****	*****	*****	*****	*****	
5.00	*****	*****	*****	*****	*****	
6.00	*****	*****	*****	*****	*****	
7.00	*****	*****	*****	*****	*****	
8.00	*****	*****	*****	*****	*****	
9.00	*****	*****	*****	*****	*****	
10.00	*****	*****	*****	*****	*****	
11.00	*****	*****	*****	*****	*****	
12.00	*****	*****	*****	*****	*****	
13.00	*****	*****	*****	*****	*****	
14.00	*****	*****	*****	*****	*****	
15.00	*****	*****	*****	*****	*****	
16.00	*****	*****	*****	*****	*****	
17.00	*****	*****	*****	*****	*****	
18.00	*****	*****	*****	*****	*****	
19.00	*****	*****	*****	*****	*****	
20.00	*****	*****	*****	*****	*****	
21.00	*****	*****	*****	*****	*****	
22.00	*****	*****	*****	*****	*****	
23.00	*****	*****	*****	*****	*****	
24.00	*****	*****	*****	*****	*****	
25.00	*****	*****	*****	*****	*****	
26.00	*****	*****	*****	*****	*****	
27.00	*****	*****	*****	*****	*****	
28.00	*****	*****	*****	*****	*****	
29.00	*****	*****	*****	*****	*****	
30.00	*****	*****	*****	*****	*****	

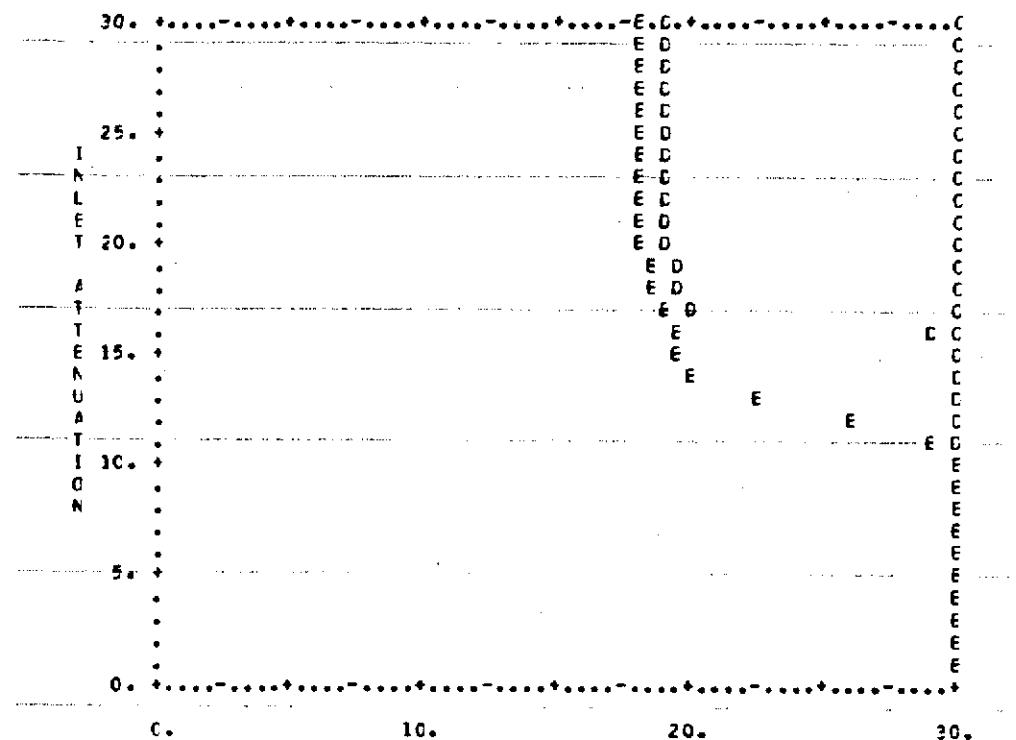
DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/C; WITH TCNE,5. JET ATT, NO TURB. EPNL = 102.22, RUN-13A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A C	B 5	C 10	D 20	E 30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

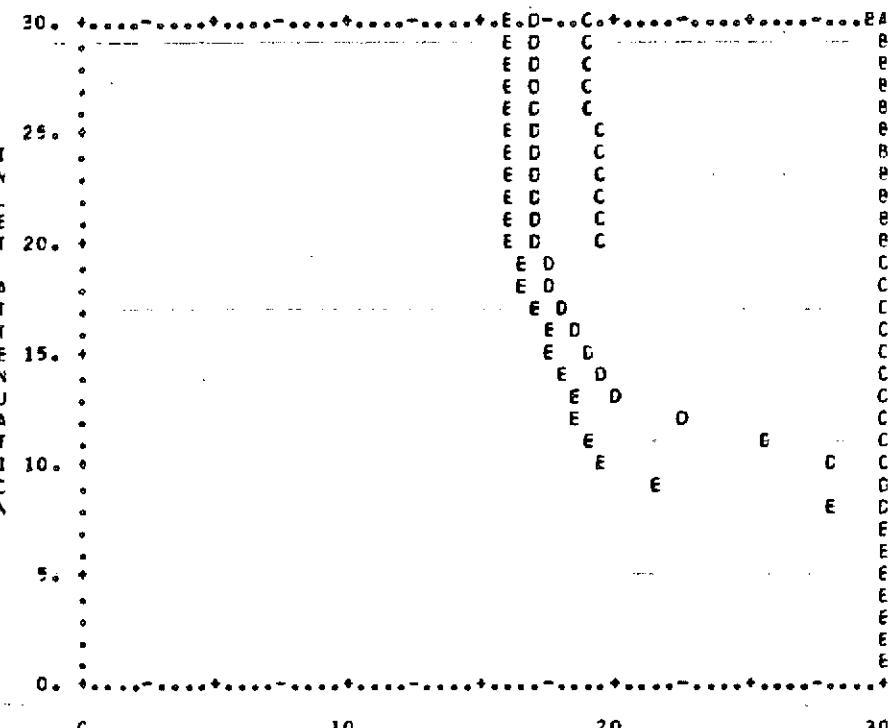
DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = C, B = S, C = 10, D = 20, E = 30)  
 T/D, WITH TCNE,5. JET ATT, NO TURB, EPNL = 102.22, RUN-134



#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET HARMONIC *	C	A	B	C	D	E
			5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	29.18
12.00	*****	*****	*****	*****	*****	26.16
13.00	*****	*****	*****	*****	*****	22.43
14.00	*****	*****	*****	*****	*****	19.86
15.00	*****	*****	*****	*****	*****	19.57
16.00	*****	*****	*****	*****	28.93	19.28
17.00	*****	*****	*****	*****	20.00	19.00
18.00	*****	*****	*****	*****	19.62	18.72
19.00	*****	*****	*****	*****	19.25	18.46
20.00	*****	*****	*****	*****	18.90	18.20
21.00	*****	*****	*****	*****	18.89	18.20
22.00	*****	*****	*****	*****	18.88	18.20
23.00	*****	*****	*****	*****	18.88	18.20
24.00	*****	*****	*****	*****	18.87	18.20
25.00	*****	*****	*****	*****	18.86	18.20
26.00	*****	*****	*****	*****	18.85	18.20
27.00	*****	*****	*****	*****	18.85	18.20
28.00	*****	*****	*****	*****	18.84	18.20
29.00	*****	*****	*****	*****	18.83	18.19
30.00	*****	*****	*****	*****	18.82	18.19

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TENE,5. JET ATT, NO TURB, EPNL = 102.22, RUN-13A

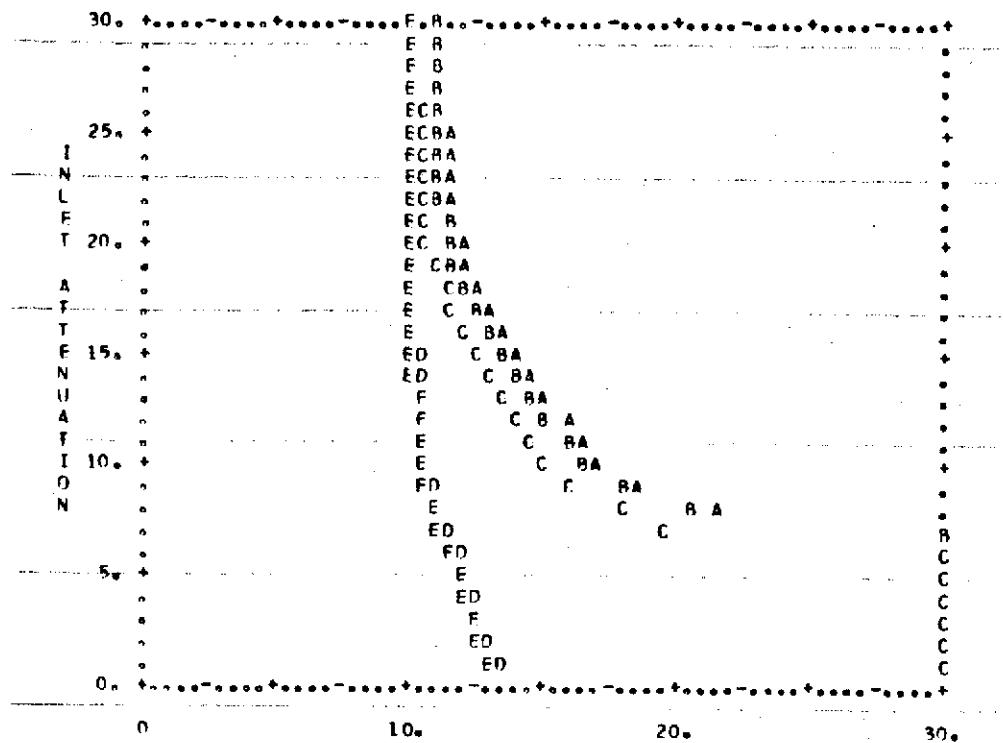


C. 10. 20. 30.

#### FAN DUCT ATTENUATION

FAN HARMONIC = 30.0		DELTA EPNL = 10.0			
INLET HARMONIC =		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	27.77
9.00	*****	*****	*****	*****	21.47
10.00	*****	*****	*****	27.95	19.26
11.00	*****	*****	*****	25.42	18.92
12.00	*****	*****	*****	22.53	18.59
13.00	*****	*****	*****	19.89	18.26
14.00	*****	*****	*****	19.40	17.94
15.00	*****	*****	*****	18.94	17.62
16.00	*****	*****	*****	18.50	17.31
17.00	*****	*****	*****	18.07	17.01
18.00	*****	*****	*****	17.66	16.71
19.00	*****	*****	*****	17.27	16.42
20.00	*****	*****	19.63	16.88	16.13
21.00	*****	*****	19.56	16.88	16.13
22.00	*****	*****	19.50	16.88	16.13
23.00	*****	*****	19.43	16.88	16.13
24.00	*****	*****	19.37	16.88	16.13
25.00	*****	*****	19.30	16.88	16.12
26.00	*****	*****	19.24	16.88	16.12
27.00	*****	*****	19.18	16.88	16.12
28.00	*****	*****	19.11	16.88	16.13
29.00	*****	*****	19.05	16.88	16.13
30.00	*****	29.52	18.99	16.88	16.13

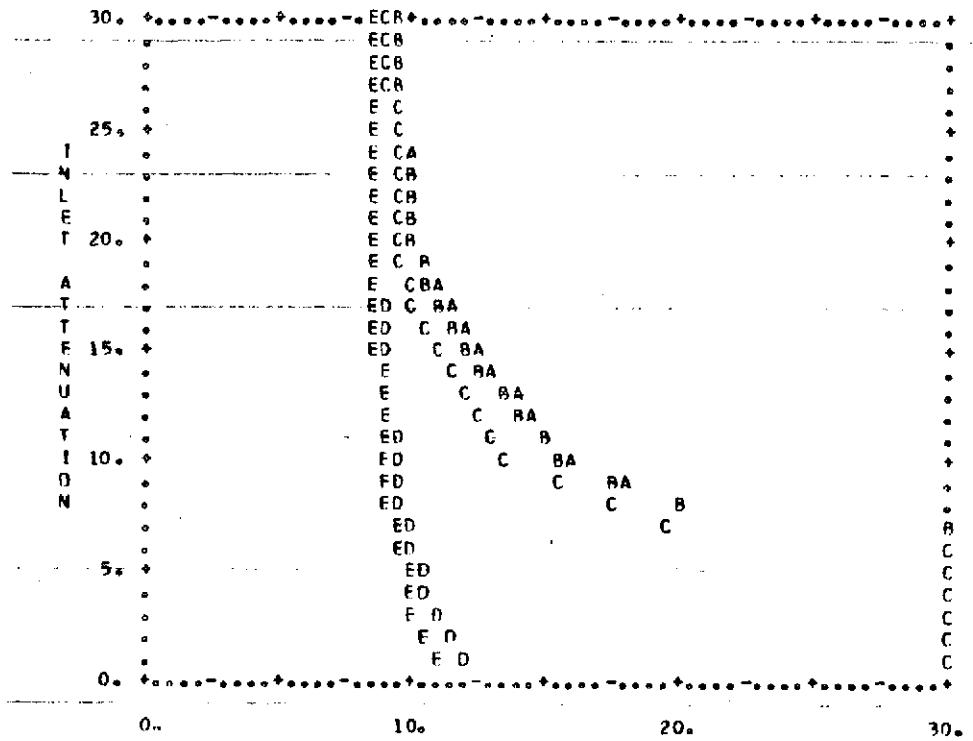
DELTA EPNL = 5.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

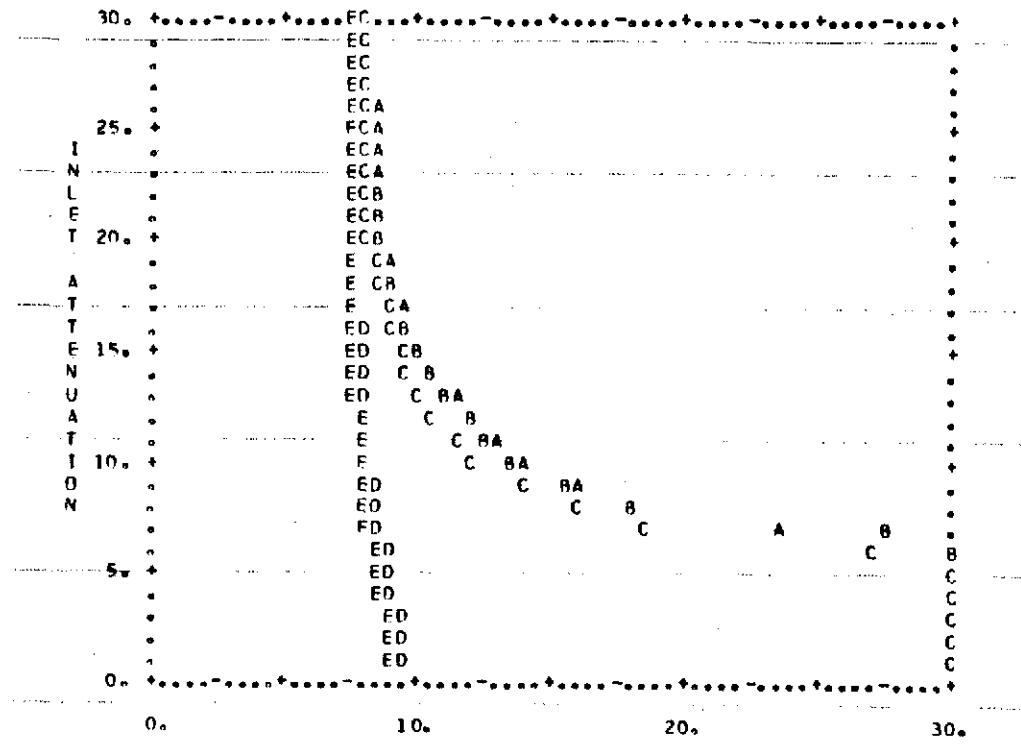
FAN HARMONIC = 0.0		DELTA EPNL = 5.0				
INLET HARMONIC =		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	13.31	12.88	
2.00	*****	*****	*****	12.98	12.57	
3.00	*****	*****	*****	12.66	12.28	
4.00	*****	*****	*****	12.37	12.01	
5.00	*****	*****	*****	12.09	11.76	
6.00	*****	*****	*****	11.79	11.47	
7.00	10.30	*****	10.55	11.48	11.19	
8.00	21.59	20.43	17.79	11.19	10.92	
9.00	18.68	18.13	16.24	10.91	10.68	
10.00	17.17	16.49	14.88	10.63	10.44	
11.00	16.46	15.81	14.31	10.56	10.39	
12.00	15.80	15.18	13.79	10.49	10.34	
13.00	15.19	14.59	13.31	10.43	10.29	
14.00	14.61	14.04	12.86	10.36	10.24	
15.00	14.07	13.52	12.45	10.30	10.18	
16.00	13.57	13.04	12.06	10.23	10.14	
17.00	13.09	12.58	11.70	10.17	10.09	
18.00	12.64	12.15	11.36	10.11	10.04	
19.00	12.21	11.74	11.04	10.06	9.99	
20.00	11.81	11.36	10.74	10.00	9.94	
21.00	11.71	11.30	10.67	9.99	9.94	
22.00	11.61	11.24	10.60	9.98	9.93	
23.00	11.51	11.18	10.53	9.97	9.92	
24.00	11.42	11.13	10.46	9.97	9.92	
25.00	11.32	11.07	10.39	9.96	9.91	
26.00	11.24	11.02	10.32	9.95	9.91	
27.00	11.15	10.96	10.25	9.94	9.90	
28.00	11.06	10.91	10.17	9.93	9.90	
29.00	10.98	10.86	10.10	9.92	9.89	
30.00	10.90	10.81	10.03	9.91	9.88	

DELTA EPNL = 5.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TUNE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



INLET HARMONIC	FAN HARMONIC = 5.0		DELTA EPNL = 5.0		E
	A 0	B 5	C 10	D 20	
1.00	*****	*****	*****	11.77	10.81
2.00	*****	*****	*****	11.37	10.49
3.00	*****	*****	*****	10.99	10.19
4.00	*****	*****	*****	10.64	9.95
5.00	*****	*****	*****	10.32	9.79
6.00	*****	*****	*****	10.07	9.60
7.00	*****	*****	19.48	9.49	9.41
8.00	19.93	19.89	17.33	9.74	9.24
9.00	17.90	17.58	15.43	9.58	9.08
10.00	16.08	15.60	13.74	9.43	8.93
11.00	15.25	14.78	13.08	9.33	8.89
12.00	14.48	14.03	12.48	9.24	8.85
13.00	13.76	13.36	11.94	9.15	8.81
14.00	13.10	12.73	11.45	9.06	8.77
15.00	12.48	12.15	11.01	8.97	8.73
16.00	11.90	11.62	10.60	8.89	8.69
17.00	11.36	11.13	10.22	8.80	8.65
18.00	10.85	10.60	9.90	8.73	8.61
19.00	10.37	10.26	9.65	8.65	8.58
20.00	9.95	9.90	9.40	8.57	8.54
21.00	9.90	9.85	9.37	8.57	8.53
22.00	9.86	9.80	9.35	8.56	8.53
23.00	9.81	9.76	9.33	8.56	8.53
24.00	9.76	9.71	9.30	8.55	8.52
25.00	9.71	9.66	9.28	8.55	8.52
26.00	9.66	9.61	9.26	8.54	8.51
27.00	9.61	9.56	9.23	8.53	8.51
28.00	9.56	9.51	9.21	8.53	8.50
29.00	9.51	9.46	9.19	8.52	8.50
30.00	9.46	9.41	9.17	8.52	8.50

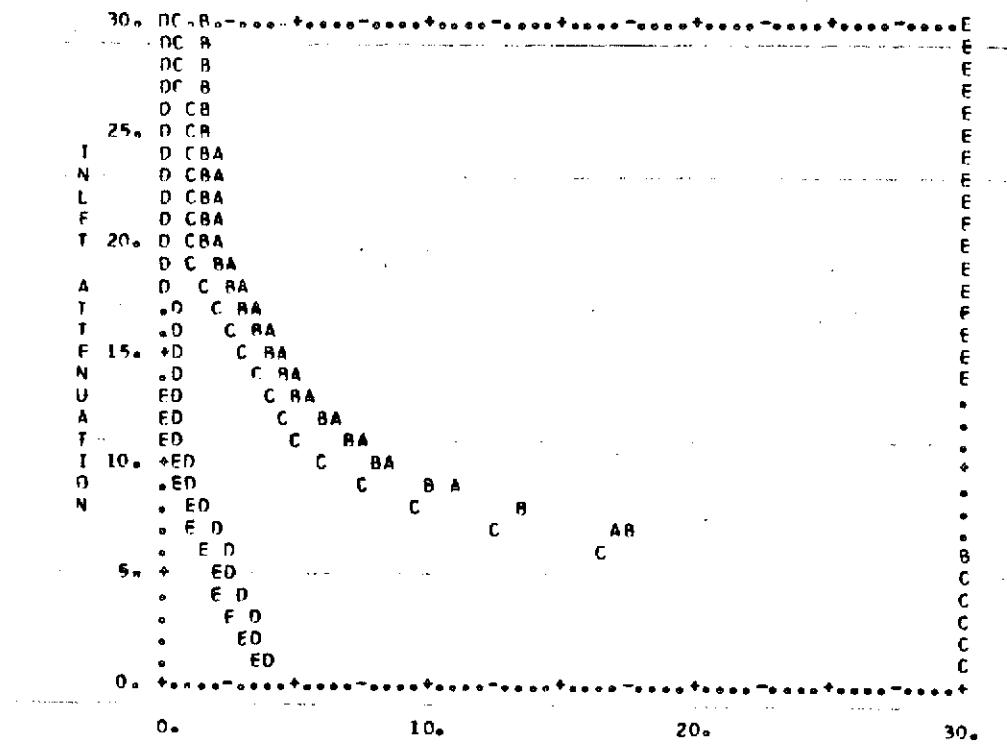
DELTA EPNL = 5.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TUNE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 10.0		DELTA EPNL = 5.0			
	A 0	B 5	C 10	D 20	E 30	
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	
1.00	*****	*****	*****	9.59	9.19	
2.00	*****	*****	*****	9.43	9.01	
3.00	*****	*****	*****	9.28	8.82	
4.00	*****	*****	*****	9.14	8.64	
5.00	*****	*****	*****	9.00	8.45	
6.00	*****	*****	26.91	8.83	8.33	
7.00	23.50	27.36	18.40	8.66	8.22	
8.00	18.10	18.12	15.94	8.48	8.12	
9.00	15.84	15.58	13.90	8.31	8.01	
10.00	13.83	13.41	12.19	8.14	7.91	
11.00	12.91	12.54	11.41	8.08	7.85	
12.00	12.09	11.76	10.73	8.01	7.79	
13.00	11.35	11.06	10.13	7.96	7.73	
14.00	10.69	10.43	9.75	7.90	7.67	
15.00	10.09	9.91	9.43	7.84	7.61	
16.00	9.72	9.57	9.13	7.79	7.55	
17.00	9.42	9.25	8.84	7.73	7.50	
18.00	9.13	8.93	8.56	7.68	7.45	
19.00	8.86	8.63	8.29	7.63	7.39	
20.00	8.60	8.33	8.03	7.57	7.34	
21.00	8.35	8.30	8.00	7.57	7.34	
22.00	8.49	8.27	7.98	7.57	7.34	
23.00	8.43	8.24	7.96	7.56	7.34	
24.00	8.38	8.20	7.94	7.56	7.34	
25.00	8.32	8.17	7.92	7.55	7.34	
26.00	8.27	8.14	7.89	7.55	7.34	
27.00	8.21	8.11	7.87	7.54	7.33	
28.00	8.15	8.08	7.85	7.54	7.33	
29.00	8.09	8.05	7.83	7.54	7.33	
30.00	8.04	8.02	7.81	7.53	7.33	

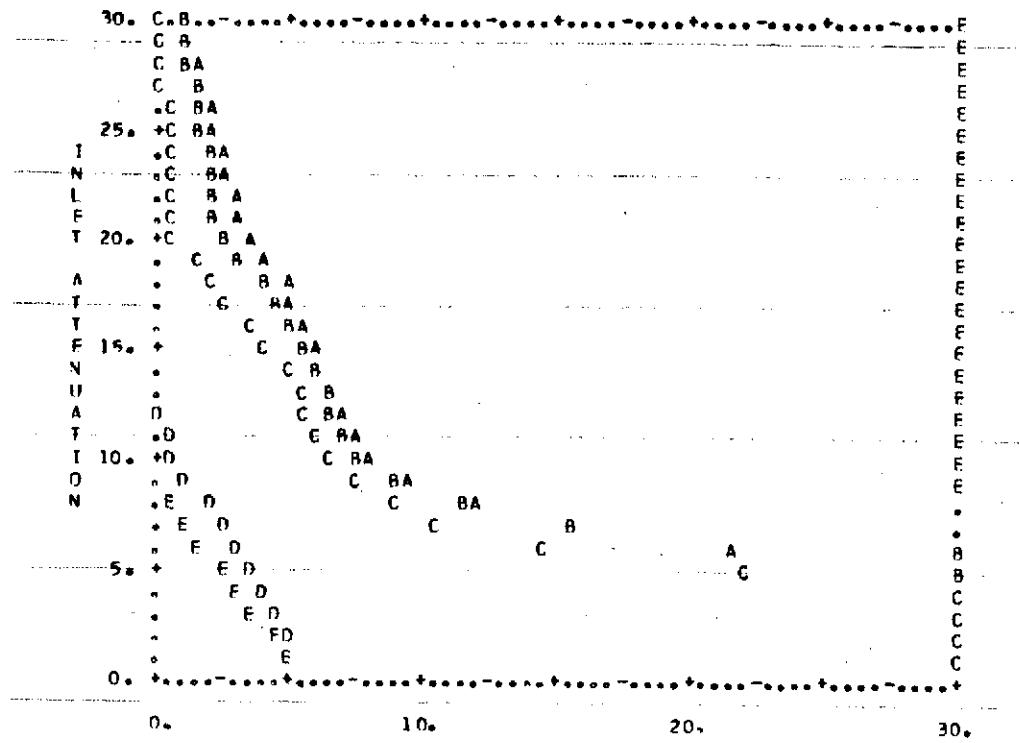
DELTA FPNL = 5.0      FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 20.0		DELTA FPNL = 5.0		
	A	B	C	D	F
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	4.00	3.29
2.00	*****	*****	*****	3.65	2.86
3.00	*****	*****	*****	3.12	2.47
4.00	*****	*****	*****	2.99	2.11
5.00	*****	*****	*****	2.69	1.77
6.00	*****	*****	16.72	2.27	1.47
7.00	17.02	17.51	12.60	1.88	1.18
8.00	13.63	13.46	9.53	1.51	0.89
9.00	10.79	10.17	7.51	1.15	0.60
10.00	8.49	7.76	5.77	0.80	0.31
11.00	7.45	6.81	5.03	0.73	0.27
12.00	6.51	5.96	4.41	0.66	0.14
13.00	5.67	5.19	3.84	0.58	0.06
14.00	4.92	4.57	3.32	0.51	*****
15.00	4.38	4.03	2.84	0.44	*****
16.00	3.86	3.52	2.39	0.37	*****
17.00	3.38	3.03	1.98	0.30	*****
18.00	2.93	2.57	1.60	0.24	*****
19.00	2.51	2.14	1.25	0.17	*****
20.00	2.11	1.73	0.92	0.10	*****
21.00	2.03	1.69	0.89	0.10	*****
22.00	1.96	1.65	0.87	0.10	*****
23.00	1.88	1.61	0.84	0.10	*****
24.00	1.81	1.57	0.82	0.10	*****
25.00	1.73	1.53	0.79	0.10	*****
26.00	1.66	1.49	0.76	0.10	*****
27.00	1.58	1.45	0.74	0.10	*****
28.00	1.50	1.41	0.71	0.10	*****
29.00	1.42	1.37	0.68	0.10	*****
30.00	1.34	1.33	0.65	0.10	*****

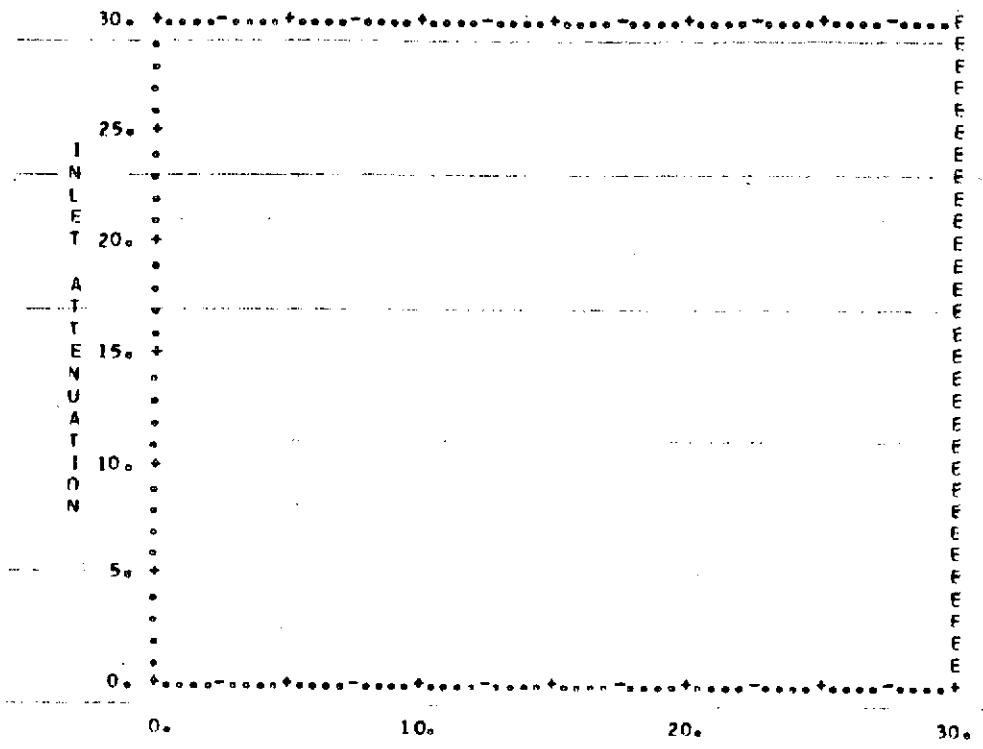
DELTA EPNL = 5.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

INLET HARMONIC	FAN HARMONIC = 30.0		DELTA EPNL = 5.0			
	A 0	B 5	C 10	D 20	E 30	
1.00	*****	*****	*****	5.21	5.02	
2.00	*****	*****	*****	5.01	4.32	
3.00	*****	*****	*****	4.45	3.64	
4.00	*****	*****	*****	3.92	3.01	
5.00	*****	*****	22.00	3.43	2.43	
6.00	21.46	*****	14.63	2.85	1.75	
7.00	15.60	15.65	10.66	2.29	1.06	
8.00	11.84	11.43	8.89	1.75	0.37	
9.00	9.41	9.09	7.64	1.23	*****	
10.00	8.17	7.73	6.51	0.72	*****	
11.00	7.60	7.20	6.04	0.45	*****	
12.00	7.09	6.72	5.63	0.17	*****	
13.00	6.64	6.28	5.26	*****	*****	
14.00	6.23	5.87	4.83	*****	*****	
15.00	5.86	5.50	4.06	*****	*****	
16.00	5.53	5.16	3.32	*****	*****	
17.00	5.22	4.60	2.63	*****	*****	
18.00	4.85	3.83	1.97	*****	*****	
19.00	4.13	3.09	1.34	*****	*****	
20.00	3.42	2.37	0.74	*****	*****	
21.00	3.17	2.22	0.67	*****	*****	
22.00	2.92	2.06	0.59	*****	*****	
23.00	2.68	1.91	0.52	*****	*****	
24.00	2.44	1.76	0.44	*****	*****	
25.00	2.19	1.60	0.37	*****	*****	
26.00	1.95	1.45	0.29	*****	*****	
27.00	1.72	1.29	0.22	*****	*****	
28.00	1.48	1.14	0.15	*****	*****	
29.00	1.25	0.99	0.07	*****	*****	
30.00	1.01	0.83	0.00	*****	*****	

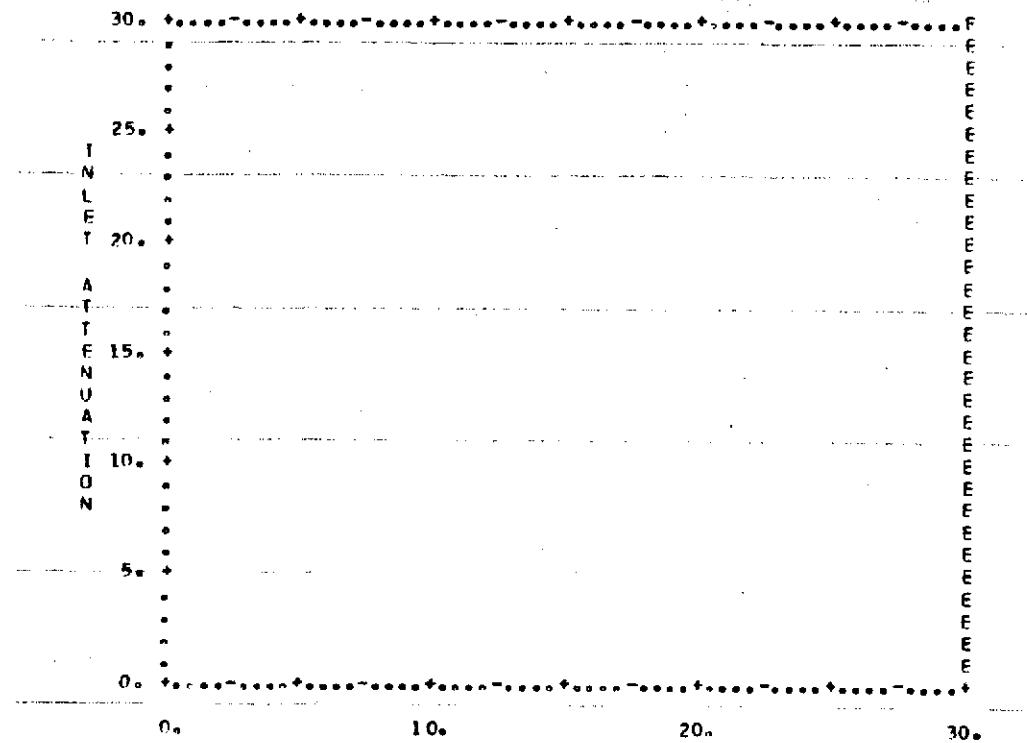
DELTA EPNL = 10.0 FAN HARMONIC = 0.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 0.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A	B	C	D	F
INLET FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.	FAN FUND.
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

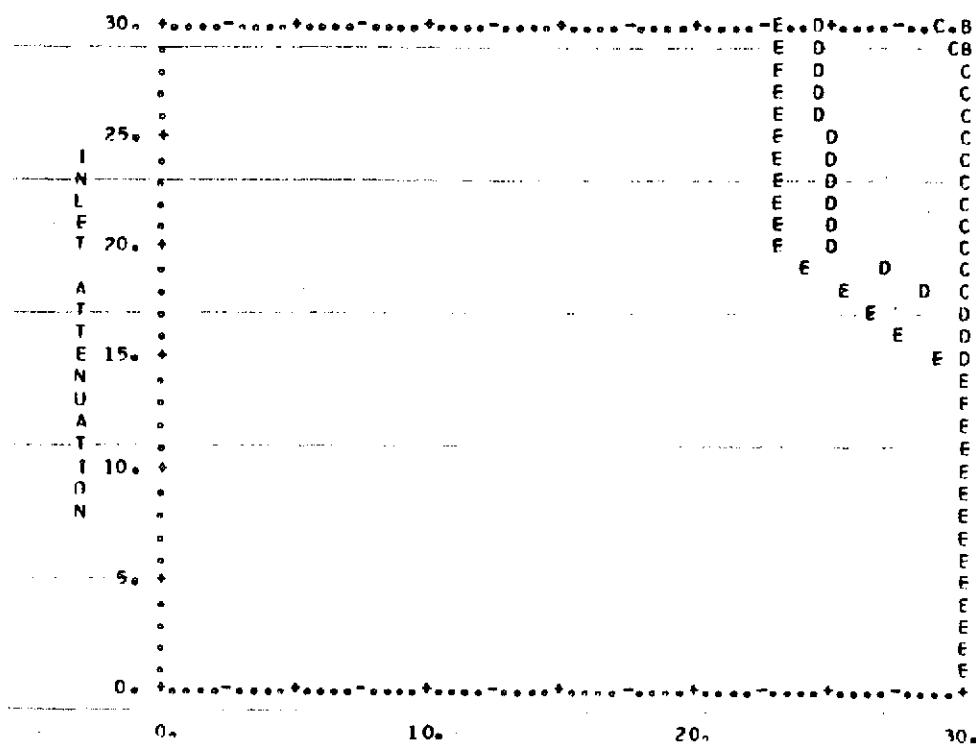
DELTA EPNL = 10.0 FAN HARMONIC = 5.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 5.0		DELTA EPNL = 10.0				
INLET HARMONIC =		A 0	B 5	C 10	D 20	E 30
INLET FUND.	FUND.	FUND.	FUND.	FUND.	FUND.	FUND.
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	*****	*****
15.00	*****	*****	*****	*****	*****	*****
16.00	*****	*****	*****	*****	*****	*****
17.00	*****	*****	*****	*****	*****	*****
18.00	*****	*****	*****	*****	*****	*****
19.00	*****	*****	*****	*****	*****	*****
20.00	*****	*****	*****	*****	*****	*****
21.00	*****	*****	*****	*****	*****	*****
22.00	*****	*****	*****	*****	*****	*****
23.00	*****	*****	*****	*****	*****	*****
24.00	*****	*****	*****	*****	*****	*****
25.00	*****	*****	*****	*****	*****	*****
26.00	*****	*****	*****	*****	*****	*****
27.00	*****	*****	*****	*****	*****	*****
28.00	*****	*****	*****	*****	*****	*****
29.00	*****	*****	*****	*****	*****	*****
30.00	*****	*****	*****	*****	*****	*****

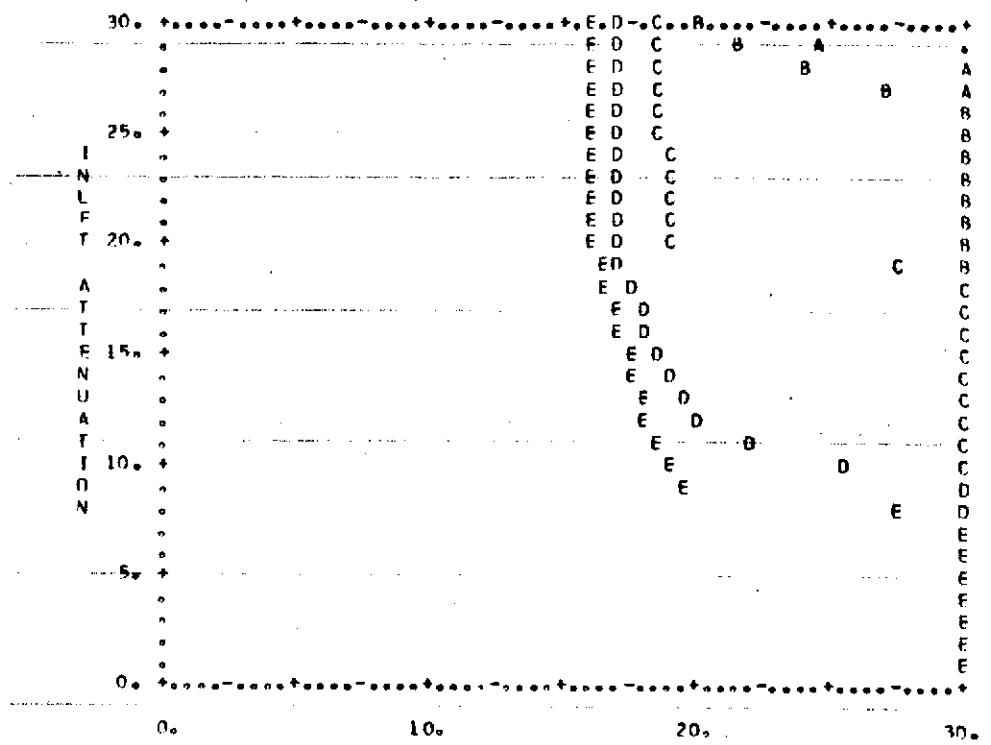
DELTA EPNL = 10.0 FAN HARMONIC = 10.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TUNE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN=14A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 10.0		DELTA EPNL = 10.0			
INLET HARMONIC		A	B	C	D
		0	5	10	20
1.00	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****
9.00	*****	*****	*****	*****	*****
10.00	*****	*****	*****	*****	*****
11.00	*****	*****	*****	*****	*****
12.00	*****	*****	*****	*****	*****
13.00	*****	*****	*****	*****	*****
14.00	*****	*****	*****	*****	30.20
15.00	*****	*****	*****	*****	29.94
16.00	*****	*****	*****	*****	27.69
17.00	*****	*****	*****	30.02	26.46
18.00	*****	*****	*****	28.37	25.25
19.00	*****	*****	*****	26.78	24.06
20.00	*****	*****	*****	25.22	22.88
21.00	*****	*****	*****	25.14	22.88
22.00	*****	*****	*****	25.06	22.87
23.00	*****	*****	*****	24.98	22.87
24.00	*****	*****	*****	24.90	22.87
25.00	*****	*****	*****	24.81	22.87
26.00	*****	*****	30.86	24.73	22.87
27.00	*****	*****	30.36	24.65	22.87
28.00	*****	*****	29.88	24.57	22.87
29.00	*****	*****	29.40	24.49	22.86
30.00	*****	*****	28.92	24.41	22.86

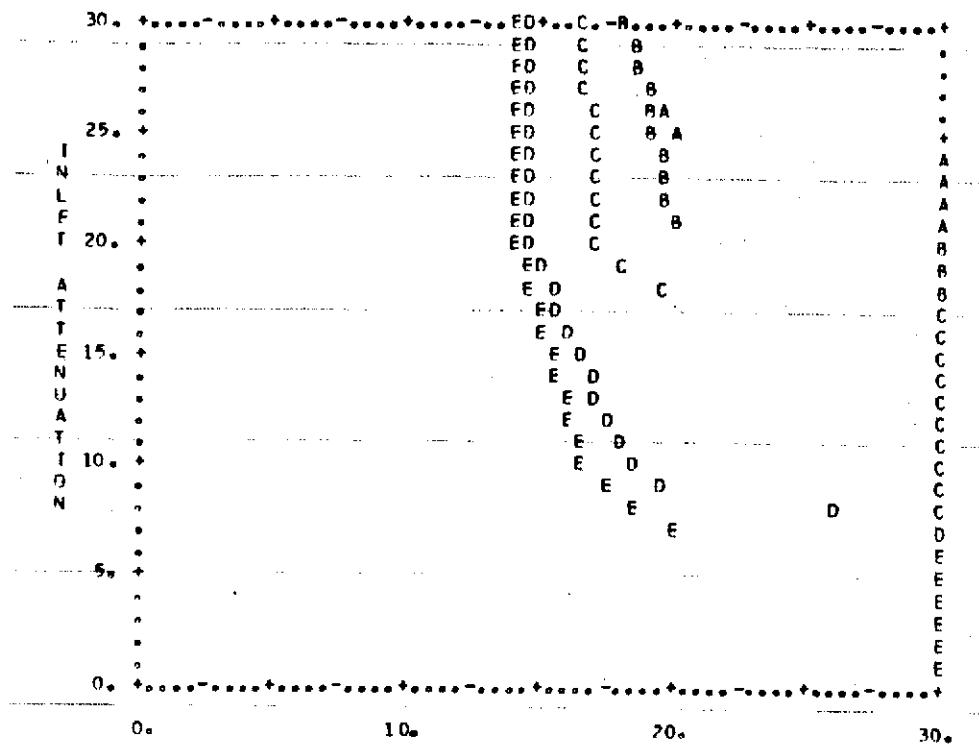
DELTA EPNL = 10.0 FAN HARMONIC = 20.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/D, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



#### FAN DUCT ATTENUATION

FAN HARMONIC = 20.0		DELTA EPNL = 10.0				
INLET HARMONIC		A	B	C	D	E
		0	5	10	20	30
1.00	*****	*****	*****	*****	*****	*****
2.00	*****	*****	*****	*****	*****	*****
3.00	*****	*****	*****	*****	*****	*****
4.00	*****	*****	*****	*****	*****	*****
5.00	*****	*****	*****	*****	*****	*****
6.00	*****	*****	*****	*****	*****	*****
7.00	*****	*****	*****	*****	*****	*****
8.00	*****	*****	*****	*****	*****	27.32
9.00	*****	*****	*****	*****	*****	19.66
10.00	*****	*****	*****	25.71	18.76	
11.00	*****	*****	*****	21.92	18.45	
12.00	*****	*****	*****	19.78	18.15	
13.00	*****	*****	*****	19.37	17.86	
14.00	*****	*****	*****	18.98	17.58	
15.00	*****	*****	*****	18.60	17.31	
16.00	*****	*****	*****	18.23	17.04	
17.00	*****	*****	*****	17.88	16.79	
18.00	*****	*****	*****	17.53	16.54	
19.00	*****	*****	27.44	17.20	16.30	
20.00	*****	*****	19.12	16.88	16.06	
21.00	*****	*****	19.04	16.88	16.06	
22.00	*****	*****	19.95	16.87	16.06	
23.00	*****	*****	18.87	16.87	16.06	
24.00	*****	*****	18.78	16.87	16.06	
25.00	*****	*****	18.70	16.87	16.06	
26.00	*****	*****	18.62	16.87	16.05	
27.00	*****	27.16	18.54	16.86	16.05	
28.00	30.40	23.98	18.45	16.86	16.05	
29.00	24.32	21.33	18.37	16.86	16.05	
30.00	19.96	19.93	18.29	16.86	16.05	

DELTA EPNL = 10.0 FAN HARMONIC = 30.0  
 INLET HARMONIC (A = 0, B = 5, C = 10, D = 20, E = 30)  
 T/O, WITH TONE, 7.5 JET ATT, NO TURB, EPNL = 101.98, RUN-14A



INLET HARMONIC	FAN DUCT ATTENUATION				
	A	B	C	D	E
0	30.0	29.0	28.0	27.0	26.0
5	29.0	28.0	27.0	26.0	25.0
10	28.0	27.0	26.0	25.0	24.0
20	26.0	25.0	24.0	23.0	22.0
30	25.0	24.0	23.0	22.0	21.0
1.00	24.0	23.0	22.0	21.0	20.0
2.00	23.0	22.0	21.0	20.0	19.0
3.00	22.0	21.0	20.0	19.0	18.0
4.00	21.0	20.0	19.0	18.0	17.0
5.00	20.0	19.0	18.0	17.0	16.0
6.00	19.0	18.0	17.0	16.0	15.0
7.00	18.0	17.0	16.0	15.0	14.0
8.00	17.0	16.0	15.0	14.0	13.0
9.00	16.0	15.0	14.0	13.0	12.0
10.00	15.0	14.0	13.0	12.0	11.0
11.00	14.0	13.0	12.0	11.0	10.0
12.00	13.0	12.0	11.0	10.0	9.0
13.00	12.0	11.0	10.0	9.0	8.0
14.00	11.0	10.0	9.0	8.0	7.0
15.00	10.0	9.0	8.0	7.0	6.0
16.00	9.0	8.0	7.0	6.0	5.0
17.00	8.0	7.0	6.0	5.0	4.0
18.00	7.0	6.0	5.0	4.0	3.0
19.00	6.0	5.0	4.0	3.0	2.0
20.00	5.0	4.0	3.0	2.0	1.0
21.00	4.0	3.0	2.0	1.0	0.0
22.00	3.0	2.0	1.0	0.0	0.0
23.00	2.0	1.0	0.0	0.0	0.0
24.00	1.0	0.0	0.0	0.0	0.0
25.00	19.80	19.14	18.82	18.70	18.13
26.00	19.43	18.95	18.76	18.69	18.13
27.00	19.08	18.76	18.71	18.68	18.13
28.00	18.75	18.58	18.66	18.66	18.13
29.00	18.43	18.40	18.62	18.65	18.13
30.00	18.13	18.23	18.57	18.64	18.12

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